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**SUBSONIC AND TRANSONIC DYNAMIC STABILITY
DERIVATIVES OF A MODIFIED 089B SHUTTLE ORBITER**

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AND
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- (A) CNQ, CNA, CLMQ, CLMA VS ALPHA
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- (C) CBLP, CBLBS, CYNP, CYNBS VS ALPHA

SUBSONIC-AND TRANSONIC DYNAMIC STABILITY
DERIVATIVES OF A MODIFIED O89B SHUTTLE ORBITER
(LA20)

by

Richmond P. Boyden and Delma C. Freeman, Jr.

SUMMARY

An experimental test program has been conducted to measure the subsonic and transonic dynamic stability derivatives of a modified O89B shuttle orbiter configuration. The test were conducted in the Langley 8-Foot Transonic Pressure Tunnel utilizing forced oscillation equipment. Rotary derivatives were measured about all three axis with both the in-phase and out-of-phase derivatives reported herein. The test were conducted at Mach numbers of 0.8, 0.9, 0.98 and 1.2 for angles of attack up to 22° . The data were measured at the model resonant frequency with pitch and yaw amplitude of 1° and roll amplitude of 2.5° .

INTRODUCTION

One of the current major national goals is the development of the Space Shuttle. As part of this effort a program has been initiated at the Langley Research Center to experimentally measure the dynamic damping derivatives of the shuttle orbiter for all flight phases from entry to landing. These experimentally measured values will be utilized in computations to assess the importance of the dynamic damping derivatives on overall vehicle dynamics. Supersonic results are presented in reference 1 and hypersonic results in reference 2.

As part of this study subsonic and transonic forced oscillation tests of a 0.0165 scale model of a modified 089B shuttle orbiter have been made in the Langley 2.4384-meter (8-Foot) Transonic Pressure Tunnel. These tests were made for several configurations over a Mach number range from 0.3 to 1.2 measuring the pitch, roll, and yaw damping. The normal force due to pitch rate, cross-derivatives yawing moment due to roll rate, and rolling moment due to yaw rate were also measured. Data plots and tables were prepared by Chrysler Corporation under IASA contract. For this investigation, designated 8TPT 653 (LA20), the tunnel occupancy time was 160 hours.

SYMBOLS

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
b	BREF	reference span, meters
C_{ℓ}	CBL	rolling-moment coefficient, $\frac{\text{rolling moment}}{q_{\infty} S b}$
$C_{\ell p}$	-	$\frac{\partial C_{\ell}}{\partial \left(\frac{pb}{2V}\right)}$, per radian
$C_{\ell \dot{p}}$	-	$\frac{\partial C_{\ell}}{\partial \left(\frac{pb^2}{4V^2}\right)}$, per radian
$C_{\ell p} + C_{\ell \dot{\beta}} \sin \alpha$	CBLP	damping-in-roll parameter, per radian
$C_{\ell r}$	-	$\frac{\partial C_{\ell}}{\partial \left(\frac{rb}{2V}\right)}$, per radian
$C_{\ell \dot{r}}$	-	$\frac{\partial C_{\ell}}{\partial \left(\frac{rb^2}{4V^2}\right)}$, per radian
$C_{\ell r} - C_{\ell \dot{\beta}} \cos \alpha$	CBLR	rolling moment due to yaw rate parameter, per radian

SYMBOLS (Continued)

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
$C_{\dot{\alpha}} \beta$	-	$\frac{\partial C_{\dot{\alpha}}}{\partial \beta}$, per radian
$C_{\dot{\alpha}} \dot{\beta}$	-	$\frac{\partial C_{\dot{\alpha}}}{\partial \left(\frac{\dot{\beta} b}{2V} \right)}$, per radian
$C_{\dot{\alpha}} \cos \alpha - k^2 C_{\dot{\alpha}} \dot{\beta}$	CBLBS	rolling moment due to roll displacement parameter, per radian
$C_{\dot{\alpha}} \cos \alpha + k^2 C_{\dot{\alpha}} \dot{\beta}$	CBLBC	effective dihedral parameter, per radian
C_m	CLM	pitching-moment coefficient, $\frac{\text{pitching moment}}{q_{\infty} S \bar{c}}$
$C_{m \dot{\alpha}}$	-	$\frac{\partial C_m}{\partial \left(\frac{q \bar{c}}{2V} \right)}$, per radian
$C_{m \dot{q}}$	-	$\frac{\partial C_m}{\partial \left(\frac{q \bar{c}^2}{4V^2} \right)}$, per radian
$C_{m \dot{q}} + C_{m \dot{\alpha}}$	CLMQ	damping-in-pitch parameter, per radian
$C_{m \alpha}$	-	$\frac{\partial C_m}{\partial \alpha}$, per radian
$C_{m \dot{\alpha}}$	-	$\frac{\partial C_m}{\partial \left(\frac{\dot{\alpha} \bar{c}}{2V} \right)}$, per radian
$C_{m \alpha} - k^2 C_{m \dot{q}}$	CLMA	oscillatory longitudinal stability parameter, per radian
C_N	CN	normal force coefficient, $\frac{\text{normal force}}{q_{\infty} S}$

SYMBOLS (Continued)

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
C_{N_q}	-	$\frac{\partial C_N}{\partial \left(\frac{q\dot{c}}{2V}\right)}$, per radian
$C_{N_{\dot{q}}}$	-	$\frac{\partial C_N}{\partial \left(\frac{q\dot{c}^2}{4V^2}\right)}$, per radian
$C_{N_q} + C_{N_{\dot{q}}}$	CNQ	normal force due to pitch rate parameter, per radian
C_{N_α}	-	$\frac{\partial C_N}{\partial \alpha}$, per radian
$C_{N_{\dot{\alpha}}}$	-	$\frac{\partial C_N}{\partial \left(\frac{\dot{\alpha}c}{2V}\right)}$, per radian
$C_{N_\alpha} - k^2 C_{N_{\dot{q}}}$	CNA	normal force due to pitch displacement parameter, per radian
C_n	CYN	yawing-moment coefficient, $\frac{\text{yawing moment}}{q_\infty S b}$
C_{n_p}	-	$\frac{\partial C_n}{\partial \left(\frac{pb}{2V}\right)}$, per radian
$C_{n_{\dot{p}}}$	-	$\frac{\partial C_n}{\partial \left(\frac{\dot{p}b^2}{4V^2}\right)}$, per radian
$C_{n_p} + C_{n_{\dot{\beta}}} \sin \alpha$	CYNP	yawing moment due to roll rate parameter, per radian
C_{n_r}	-	$\frac{\partial C_n}{\partial \left(\frac{rb}{2V}\right)}$, per radian

SYMBOLS (Continued)

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
C_{n_r}	-	$\frac{\partial C_n}{\partial \left(\frac{rb^2}{4V^2}\right)}$, per radian
$C_{n_r} - C_{n_\beta} \cos \alpha$	CYNR	damping-in-yaw parameter, per radian
C_{n_β}	-	$\frac{\partial C_n}{\partial \beta}$, per radian
$C_{n_{\dot{\beta}}}$	-	$\frac{\partial C_n}{\partial \left(\frac{\dot{\beta} b}{2V}\right)}$, per radian
$C_{n_\beta} \cos \alpha + k^2 C_{n_r}$	CYNBC	oscillatory directional-stability parameter, per radian
$C_{n_\beta} \sin \alpha - k^2 C_{n_p}$	CYNBS	yawing moment due to roll displacement parameter, per radian
\bar{c}	LREF	reference chord, meters
c.g.	CG-LOC	reference center of gravity location position
f	-	frequency of oscillation, Hertz
k	-	reduced frequency parameter, $\frac{\omega \bar{c}}{2V}$ in pitch; $\frac{\omega b}{2V}$ in roll and yaw, radians
M	MACH	free-stream Mach number
p	-	angular velocity of model about X axis, radians/second
q	-	angular velocity of model about Y axis, radians/second
q_∞	-	free-stream dynamic pressure, k Pa, psf
R	RN/L	Keynolds number, millions/meter, millions/foot, millions based on body length

SYMBOLS (Concluded)

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
r	-	angular velocity of model about Z axis, radians/second
S	SREF	reference area, meters ²
V	-	free-stream velocity, meters/second
α	ALPHA	angle of attack, radians or degrees
β	BETA	angle of sideslip, radians or degrees
ω	-	angular velocity, $2\pi f$, radians/second
δ_a	AILRON	total aileron control surface deflection angle, degrees
δ_{bf}	BDFLAP	body flap surface deflection angle, positive trailing edge down, degrees
δ_e	ELEVTR	elevator surface deflection angle, positive trailing edge down, degrees
δ_r	RUDDER	rudder surface deflection angle, positive deflection trailing edge to the left, degrees
δ_{rf}	RUDFLR	rudder flare deflection angle, included angle between split rudder used to decrease speed, degrees

Note: A dot over a quantity indicates a first derivative with respect to time.

CONFIGURATION INVESTIGATED

A 0.0165-scale model of a blend of Rockwell International shuttle configurations was tested. The model consisted of a 089B orbiter configuration with a 139B configuration nose forward of fuselage station 500. A sketch and photograph of the model are shown in figures 2 and 3, respectively. Data were measured at two moment reference point locations, 65% and 67% of the body length, (positions 1.0 and 2.0 respectively). Variations of rudder flare, body flap and OMS installation were also investigated. The data tabulation sheets are presented in Table II and the model component dimensional data are presented in Table III.

TEST CONDITIONS

Tunnel conditions during the tests are summarized in table I. The Langley forced oscillation apparatus (see reference 3) was used with the model oscillating at resonant frequency. Amplitudes of 1° were used during the pitch and yaw tests and 2.5° for the roll tests. Both the in-phase and the out-of-phase derivatives are presented for the primary derivatives as well as the cross-derivatives of yawing moment due to rolling velocity and rolling moment due to yaw velocity. All data presented are average values of three data samples at each angle of attack.

TABLE I. TEST CONDITIONS

MACH NUMBER	REYNOLDS NUMBER X 10^6 (BASED ON MODEL LENGTH)	DYNAMIC PRESSURE	
		k Pa	PSF
0.30	3.2	5.985	125
0.80	6.7	29.781	622
0.90	7.0	33.947	709
0.98	3.6	18.434	385
1.20	3.7	21.115	441

TEST FACILITY DESCRIPTION

The NASA/Langley Research Center 2.4384-Meter (8-Foot) Transonic Pressure Tunnel is an air-medium, facility capable of attaining continuously variable Mach numbers from 0.20 to 1.30. It is a single-return, closed-circuit tunnel, having controlled stagnation temperature, total pressure, and dew-point temperature. The test section is 2.16 meters square (7.1 feet square). Reynolds numbers are variable from 0.985 to 23×10^6 /meter (0.3 to 7×10^6 /foot), depending on Mach number and tunnel total-pressure limitations. Models are generally supported in the test section by a sting-sector system, but wall-mounting is possible. Schlieren photography is available for flow and shock-wave studies.

DATA REDUCTION

Static aerodynamic forces and moments were reduced to coefficient form based on the following model reference values:

$b = BREF = \text{reference wing span} = 0.39256 \text{ meter} \quad (15.455 \text{ in.})$
 $\bar{c} = LREF = \text{reference MAC} = 0.1990 \text{ meter} \quad (7.834 \text{ in.})$
 $S = SREF = \text{reference wing area} = 0.0680 \text{ m}^2 \quad (0.7323 \text{ ft}^2)$

Moment coefficients are referenced to two theoretical center of gravity locations, which are; 1) 65% and 2) 67% of the body length. These dimensions are:

x_{cg1}	longitudinal length, nose to forward moment reference center	0.3513 meter (13.830 in.)
x_{cg2}	longitudinal length, nose to aft moment reference center	0.3621 meter (14.256 in.)
z_{cg}	vertical distance, water plane 0 to moment reference center	0.0106 meter (0.419 in.)

PRESENTATION OF RESULTS

The forced oscillation test results are presented in figures 4 through 13. The effect of c.g. position and OMS installation on the pitch damping and the normal force due to pitch rate are presented in figures 4 and 5 respectively. The model exhibited positive damping in pitch throughout the test angle-of-attack range for Mach numbers up to 0.9. At Mach numbers of 0.98 and 1.2 there were regions of negative damping between 10 and 12 degrees angle-of-attack. The effect of OMS installation on the yaw and roll damping are presented in figures 6 and 7 respectively. These results show that the model exhibited positive damping in yaw throughout the test angle of attack range and that the model exhibited positive roll damping over the test Mach number range except for the highest angles-of-attack (excess of 20°) for Mach numbers of 0.98 and 1.2. The vertical tail contributions to yaw and roll damping can be seen in figures 8 and 9 and the effect of configuration variables such as body flap deflection in combination with rudder flare and body flap removal are presented in figures 10 through 13.

REFERENCES

1. Freeman, Delma C., Jr.; Boyden, Richmond P., and Davenport, Edwin, E., Supersonic Dynamic Stability Derivatives of a Modified O89B Shuttle Orbiter, TMX-72630, January 1975.
2. Usselton, Bob L., Hypersonic Dynamic Stability Derivatives of a Modified O89B Shuttle Orbiter, DMS-DR-2132, December 1974.
3. Bielat, Ralph P. and Wiley, Harleth G., Dynamic Longitudinal and Directional Stability Derivatives for a 45° Sweptback-Wing Airplane Model at Transonic Speeds, TM X-39, August 1959.

TABLE II.

TEST : LaRC 8 FT. TPT 653,692										DATE : 10 September 73									
DATA SET / RUN NUMBER COLLATION SUMMARY																			
DATA SET IDENTIFIER	CONFIGURATION	SCHD.		PARAMETERS/VALUES			NO. OF C.G. RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)					TEST RUN NUMBERS						
		α	β	δe	$\delta r f$	$\delta b f$		0.3	0.8	0.9	0.98	1.2							
RPKY02	BWV F (Test 692)	A	0	0	0	10	0	1	5										
03	BW MF	A	0	0	0	-	0	1	5										
04	BWVMF	A	0	0	0	10	0	1	5										
05	BWVMF	A	0	0	0	85	13	1	5										
RPKF00	BWVM (Test 653)	A	0	0	0	10	-	1	5										
02	BWV F	A	0	0	0	10	0	1	5										
04	BWVMF	A	0	0	0	10	0	1	5										
05	BWVMF	A	0	0	0	85	13	1	5										
06	BWVMF	A	0	0	0	10	0	2	5										
RPKR01	BW F (Test 653)	A	0	0	0	-	0	1	2										
02	BWV F	A	0	0	0	10	0	1	5										
03	BW MF	A	0	0	0	-	0	1	5										
04	BWVMF	A	0	0	0	10	0	1	5										
05	BWVMF	A	0	0	0	85	13	1	5										
"Y" DATA	CYNR	CBLR	CBLR	CYBRC	CBLBC												MACH	ALPHA	4
"P" DATA	CLMQ	CLMA	CNQ	CNA													MACH	ALPHA	4
"R" DATA	CBLP	CBLBS	CYNP	CYNBS													MACH	ALPHA	4
TYPE OF DATA		α OR β		A) = $-4^\circ \rightarrow 24^\circ$, $\Delta\alpha = 2^\circ$		COEFFICIENT SCHEDULES		C.G.		1- α FORWARD=65% Body Length		2- α FORWARD=67% Body Length		1- α FORWARD=65% Body Length		2- α FORWARD=67% Body Length		NDV	
SCHEDULES																			

TABLE III. - COMPONENT DIMENSIONAL DATA

COMPONENT- BODY - B

GENERAL DESCRIPTION- 0898-1398(MODIFIED NOSE). NOSE SECTION FROM FULL-SCALE STATION 238.0 TO STATION 500 FROM NAR DRAWING VL70-0001398. REMAINING BODY AFT OF STATION 500 FROM NAR VL70-000093.

MODEL SCALE- 0.0165

DRAWING NUMBER- VL70-000093, VL70-0001398.

TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
LENGTH	3277.4	CM.	1290.3	IN.	54.077	CM.	21.290	IN.
MAX. WIDTH	673.1	CM.	265.0	IN.	11.106	CM.	4.372	IN.
MAX. DEPTH	629.9	CM.	248.0	IN.	10.394	CM.	4.092	IN.
FINENESS RATIO	4.869		4.869		4.869		4.869	
MAX CROSS-SECTIONAL AREA	42.4011	SQ.M.	456.4000	SQ.FT.	115.4370	SQ.CM.	17.8927	SQ.IN.

FOOTNOTE- GENERAL-MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES
UNLESS NOTED OTHERWISE.

COMPONENT- ELEVON - E

GENERAL DESCRIPTION- CONFIGURATION PER LINES VL70-000093, DATA FOR
(1) OF (2) SIDES.

MODEL SCALE- 0.0165

DRAWING NUMBER- VL70-000093

TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
AREA	19.0932	SQ.M.	205.5170	SQ.FT.	51.9813	SQ.CM.	8.0571	SQ.IN.
EQUIVALENT SPAN	897.5	CM.	353.34	IN.	14.81	CM.	5.83	IN.
INBOARD EQUIVALENT CHORD	291.5	CM.	114.78	IN.	4.81	CM.	1.89	IN.
OUTBOARD EQUIVALENT CHORD	139.7	CM.	55.00	IN.	2.31	CM.	.91	IN.
RATIO MOVABLE SURFACE CHORD/ TOTAL SURFACE CHORD								
AT INBOARD EQUIVALENT CHORD	.208		.208		.208		.208	
AT OUTBOARD EQUIVALENT CHORD	.400		.400		.400		.400	
SWEEP-BACK ANGLES								
LEADING EDGE	.00	DEG.	.00	DEG.	.00	DEG.	.00	DEG.
TAILING EDGE	-10.02	DEG.	-10.02	DEG.	-10.02	DEG.	-10.02	DEG.
HINGELINE	.00	DEG.	.00	DEG.	.00	DEG.	.00	DEG.
AREA MOMENT								
NORMAL TO HINGELINE	43.8367	CU.M.	1548.0700	CU.FT.	196.9200	CU.CM.	12.0167	CU.IN.

FOOTNOTE- GENERAL-MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES
UNLESS NOTED OTHERWISE.

TABLE III. - COMPONENT DIMENSIONAL DATA

COMPONENT- WING - W

GENERAL DESCRIPTION- ORBITER CONFIGURATION PER LINES VL70-000093, (DIHEDRAL IS DEFINED AT THE LOWER SURFACE OF THE WING AT THE 75.33 PERCENT ELEMENT LINE PROJECTED INTO A PLANE PERPENDICULAR TO THE FUSELAGE REFERENCE LINE).

MODEL SCALE- 0.0165

DRAWING NUMBER- VL70-000093

TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
TOTAL DATA								
PLANFORM AREA	249.9102	SQ.M.	2690.0000	SQ.FT.	680.3804	SQ.CM.	105.4588	SQ.IN.
EQUIVALENT SPAN	2379.172	CM.	936.680	IN.	39.256	CM.	15.455	IN.
ASPECT RATIO	2.265		2.265		2.265		2.265	
RATE OF TAPER	1.177		1.177		1.177		1.177	
TAPER RATIO	.200		.200		.200		.200	
DIHEDRAL ANGLE	3.500	DEG.	3.500	DEG.	3.500	DEG.	3.500	DEG.
INCIDENCE ANGLE	3.000	DEG.	3.000	DEG.	3.000	DEG.	3.000	DEG.
AERODYNAMIC TWIST	3.000	DEG.	3.000	DEG.	3.000	DEG.	3.000	DEG.
SWEEP-BACK ANGLES								
LEADING EDGE	45.000	DEG.	45.000	DEG.	45.000	DEG.	45.000	DEG.
TRAILING EDGE	-10.240	DEG.	-10.240	DEG.	-10.240	DEG.	-10.240	DEG.
0.25 ELEMENT LINE	35.209	DEG.	35.209	DEG.	35.209	DEG.	35.209	DEG.
CHORDS								
ROOT (WING STATION 0.0)	1750.67	CM.	689.24	IN.	28.89	CM.	11.37	IN.
TIP (EQUIVALENT)	350.14	CM.	137.85	IN.	5.78	CM.	2.27	IN.
MAC	1206.02	CM.	474.01	IN.	19.90	CM.	7.83	IN.
FUS. STA. OF 0.25 MAC	2887.71	CM.	1136.89	IN.	47.85	CM.	18.76	IN.
V.P. OF 0.25 MAC	759.97	CM.	299.20	IN.	12.54	CM.	4.94	IN.
B.L. OF 0.25 MAC	462.61	CM.	182.13	IN.	7.63	CM.	3.01	IN.
EXPOSED DATA								
AREA	162.7937	SQ.M.	1752.2900	SQ.FT.	443.2059	SQ.CM.	68.6968	SQ.IN.
EQUIVALENT SPAN	1830.53	CM.	720.68	IN.	30.20	CM.	11.89	IN.
ASPECT RATIO	2.058		2.058		2.058		2.058	
TAPER RATIO	.2451		.2451		.2451		.2451	
CHORDS								
ROOT	1428.50	CM.	562.40	IN.	23.57	CM.	9.28	IN.
TIP	350.14	CM.	137.85	IN.	5.78	CM.	2.27	IN.
MAC	998.30	CM.	393.03	IN.	16.47	CM.	6.48	IN.
FUS. STA. OF 0.25 MAC	3010.69	CM.	1185.31	IN.	49.68	CM.	19.56	IN.
V.P. OF 0.25 MAC	762.51	CM.	300.20	IN.	12.58	CM.	4.95	IN.
B.L. OF 0.25 MAC	365.15	CM.	143.76	IN.	6.02	CM.	2.37	IN.

FOOTNOTE- GENERAL- MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES UNLESS NOTED OTHERWISE.

TABLE III. - COMPONENT DIMENSIONAL DATA

COMPONENT- VERTICAL TAIL - V
 GENERAL DESCRIPTION- CENTERLINE VERTICAL TAIL DOUBLE VEDGE AIRFOIL
 WITH ROUNDED LEADING EDGE.
 MODEL SCALE- 0.0165
 DRAWING NUMBER- VL70-000095
 TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
THEORETICAL AREA	38.3923	SQ.M.	413.2500	SQ.FT.	104.5231	SQ.CM.	16.2011	SQ.IN.
EQUIVALENT SPAN	801.93	CM.	315.72	IN.	13.23	CM.	5.21	IN.
INBOARD EQUIVALENT CHORD	681.99	CM.	268.50	IN.	11.25	CM.	4.43	IN.
OUTBOARD EQUIVALENT CHORD	275.51	CM.	108.47	IN.	4.55	CM.	1.79	IN.
SWEEP-BACK ANGLES								
LEADING EDGE	45.00	DEG.	45.00	DEG.	45.00	DEG.	45.00	DEG.
TAILING EDGE	26.25	DEG.	26.249	DEG.	26.25	DEG.	26.25	DEG.

FOOTNOTE-GENERAL-MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES
 UNLESS NOTED OTHERWISE.

COMPONENT- RUDDER - R
 GENERAL DESCRIPTION- CONFIGURATION PER LINES VL70-000095.
 MODEL SCALE- 0.0165
 DRAWING NUMBER- VL70-000095
 TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
AREA	9.883	SQ.M.	106.380	SQ.FT.	.0027	SQ.M.	.0293	SQ.FT.
EQUIVALENT SPAN	510.54	CM.	201.00	IN.	8.42	CM.	3.32	IN.
INBOARD EQUIVALENT CHORD	232.63	CM.	91.585	IN.	3.84	CM.	1.51	IN.
OUTBOARD EQUIVALENT CHORD	129.12	CM.	50.833	IN.	2.13	CM.	.84	IN.
RATIO MOVABLE SURFACE CHORD/ TOTAL SURFACE CHORD								
AT INBOARD EQUIVALENT CHORD	.400		.400		.400		.400	
AT OUTBOARD EQUIVALENT CHORD	.400		.400		.400		.400	
SWEEP-BACK ANGLES								
LEADING EDGE	34.83	DEG.	34.83	DEG.	34.83	DEG.	34.83	DEG.
TRAILING EDGE	26.25	DEG.	26.25	DEG.	26.25	DEG.	26.25	DEG.
HINGELINE	34.83	DEG.	34.83	DEG.	34.83	DEG.	34.83	DEG.
AREA MOMENT								
NORMAL TO HINGELINE	14.8983	CU.M.	526.1259	CU.FT.	66.9250	CU.CM.	4.0840	CU.IN.

FOOTNOTE-GENERAL-MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES
 UNLESS NOTED OTHERWISE.

ORIGINAL PAGE IS
 OF POOR QUALITY

TABLE III. - COMPONENT DIMENSIONAL DATA

COMPONENT- OMS PODS- M
GENERAL DESCRIPTION- 2A LIGHTWEIGHT CONFIGURATION PER NC120074
MODEL SCALE- 0.0165
DRAWING NUMBER- VL70-000094
TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
LENGTH	878.842	CM.	346.000	IN.	14.501	CM.	5.709	IN.
MAX WIDTH (AT XO=1450)	274.321	CM.	108.000	IN.	4.526	CM.	1.782	IN.
MAX DEPTH (AT XO=1500)	289.053	CM.	113.800	IN.	4.769	CM.	1.878	IN.
OMS POD CENTERLINE								
Z AXIS ORBITER	1178.308	CM.	463.900	IN.	19.442	CM.	7.654	IN.
Y AXIS ORBITER	203.200	CM.	80.000	IN.	3.353	CM.	1.320	IN.

FOOTNOTE- GENERAL- MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES
UNLESS NOTED OTHERWISE.

COMPONENT- BODY FLAP- F
GENERAL DESCRIPTION- BODY FLAP 2A CONFIGURATION PER LINES VL70-000094.
MODEL SCALE- 0.0165
DRAWING NUMBER- VL70-000094 A
TEST IDENTIFICATION- LA20

	FULL SCALE METRIC		FULL SCALE ENGLISH		MODEL SCALE METRIC		MODEL SCALE ENGLISH	
LENGTH	215.138	CM.	84.700	IN.	3.550	CM.	1.398	IN.
MAXIMUM WIDTH	673.101	CM.	265.000	IN.	11.106	CM.	4.372	IN.
MAXIMUM DEPTH	53.340	CM.	21.000	IN.	.880	CM.	.346	IN.
AREA PLANFORM	13.2517	SQ.M.	142.6400	SQ.FT.	36.8779	SQ.CM.	5.5921	SQ.IN.
AREA BASE	3.5903	SQ.M.	38.6460	SQ.FT.	9.7747	SQ.CM.	1.5151	SQ.IN.

FOOTNOTE- GENERAL- MODEL SCALE VALUES ARE DERIVED FROM FULL SCALE VALUES
UNLESS NOTED OTHERWISE.

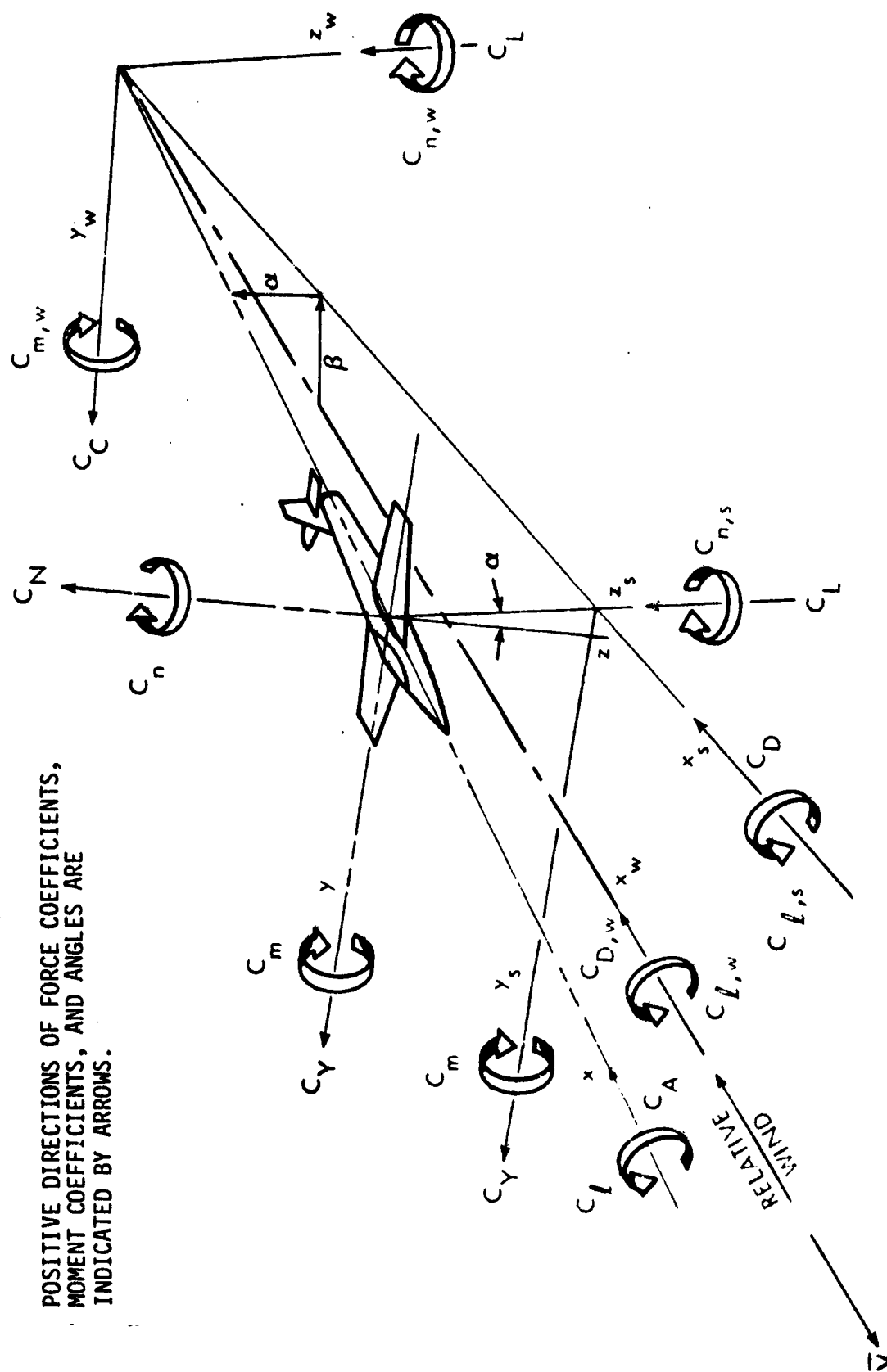


FIGURE 1. AXIS SYSTEMS

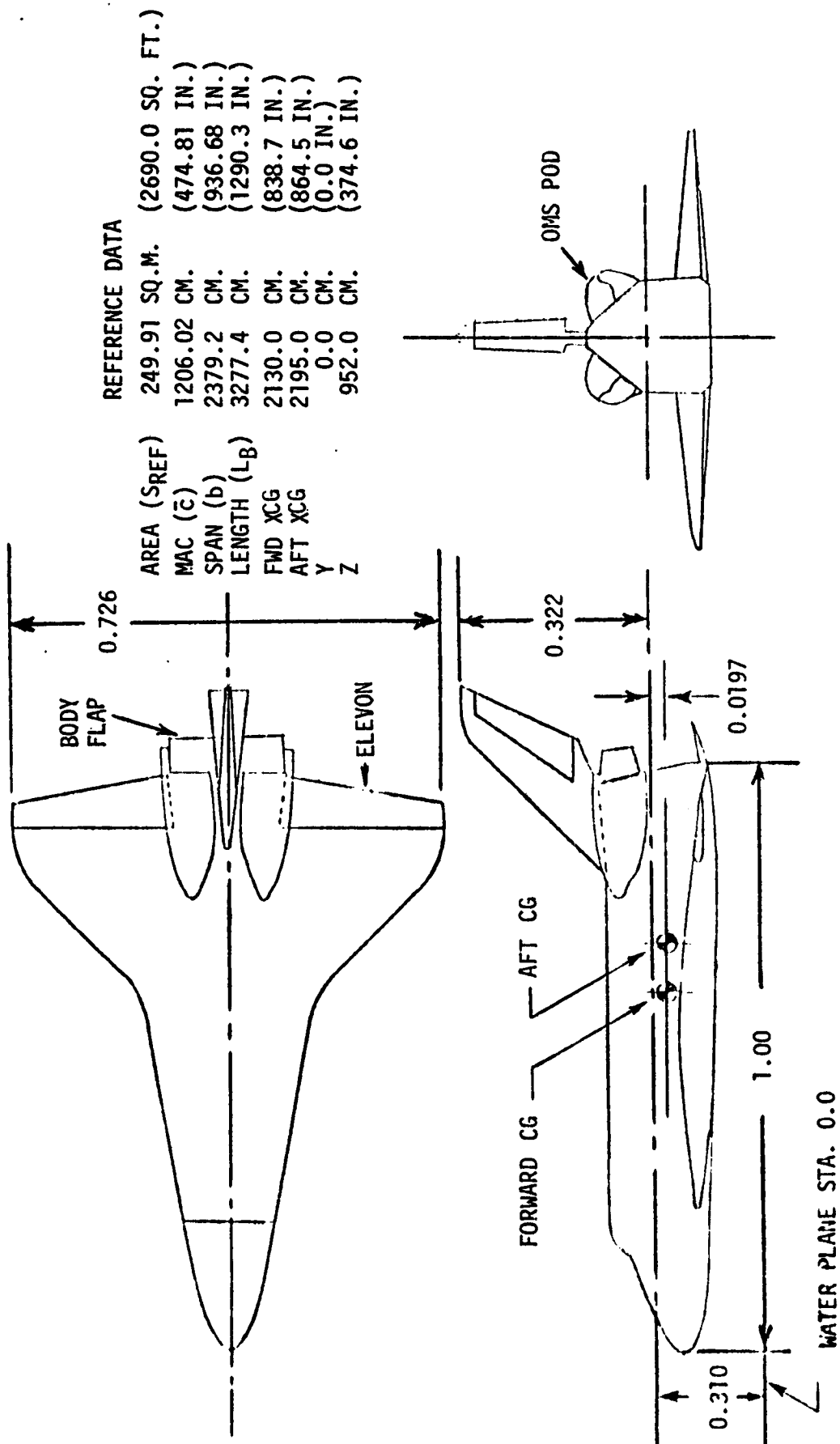
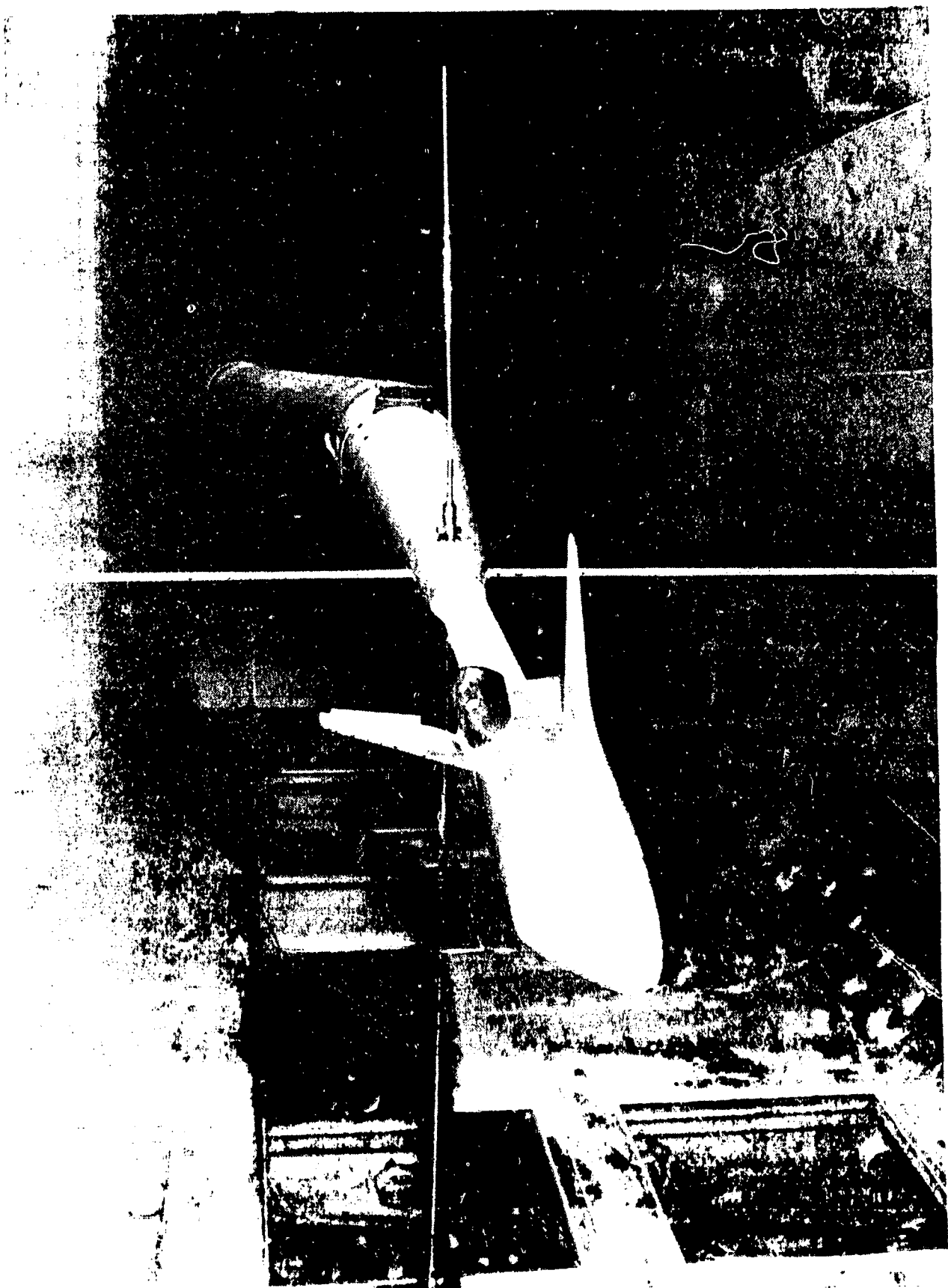


FIGURE 2. SSV ORBITER WITH DIMENSIONS NORMALIZED WITH RESPECT TO BODY LENGTH OF 3277.4 CM. (1290.3 IN.)



PHOTOGRAPH OF MOBILE UNIT TO BE USED FOR EDD - IN VACUUM PERFORMING UNIT

DATA FIGURES

CG-LOC ELEVTR BOFLAP RUDELIR
1.000 .000 10.000
2.000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
(RPM:PG4) LA-20: ROCKWELL 0898 QRB V/MOD NOSE (BNVME)
(RPM:PG5) LA-20: ROCKWELL 0898 QRB V/MOD NOSE (BNVME)

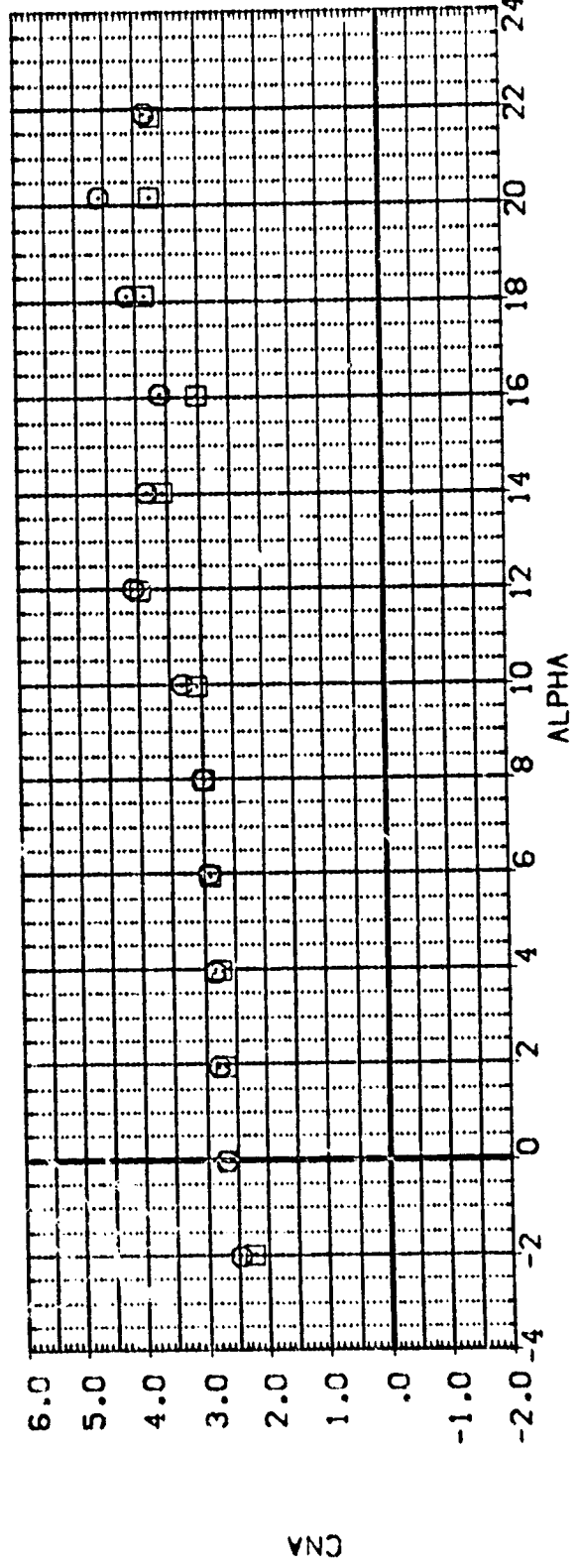
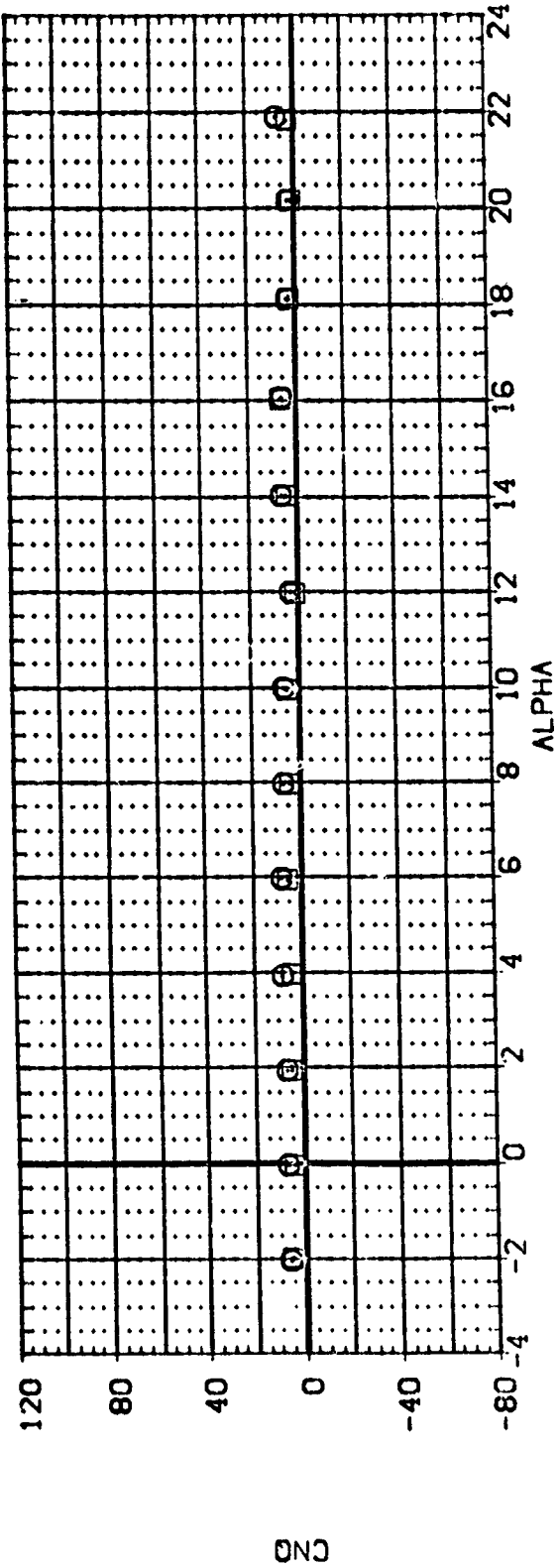


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

(RPMPO4) LA-20, ROCKWELL 0898 OR8 V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

(RPMPC6) LA-20, ROCKWELL 0898 OR8 V/MOD NOSE (BVMF) 2.000 .000 .000 10.000

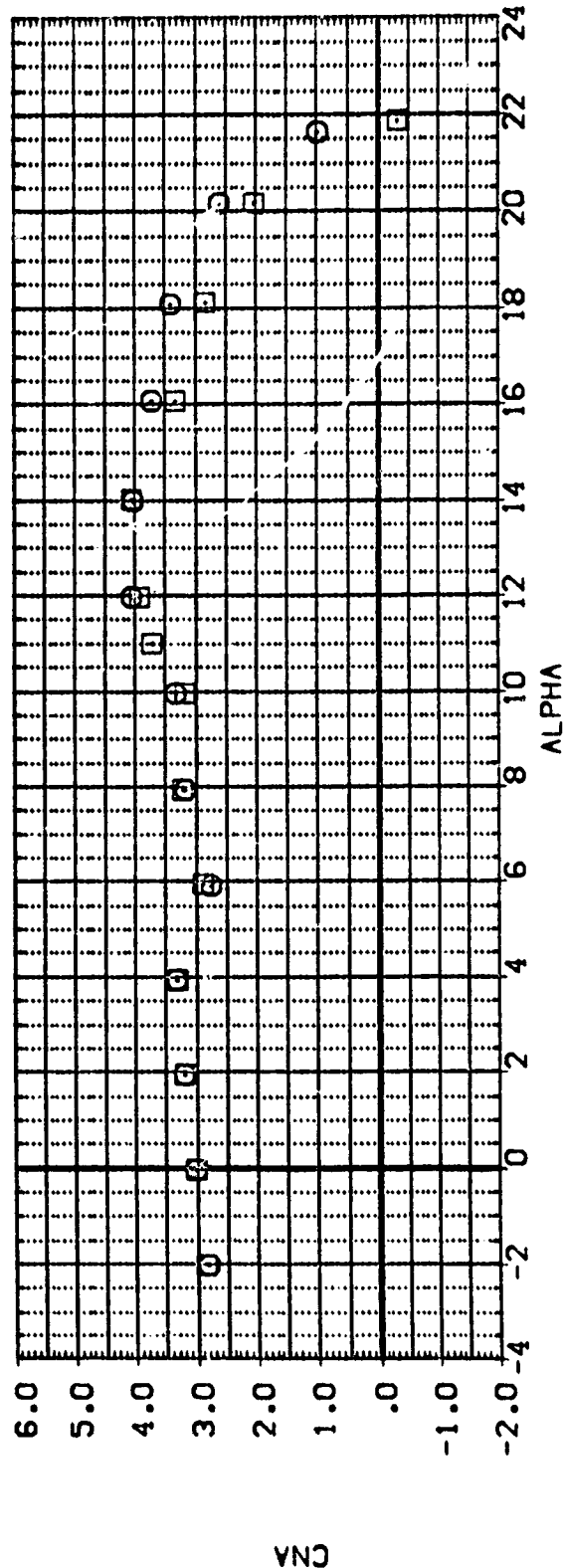
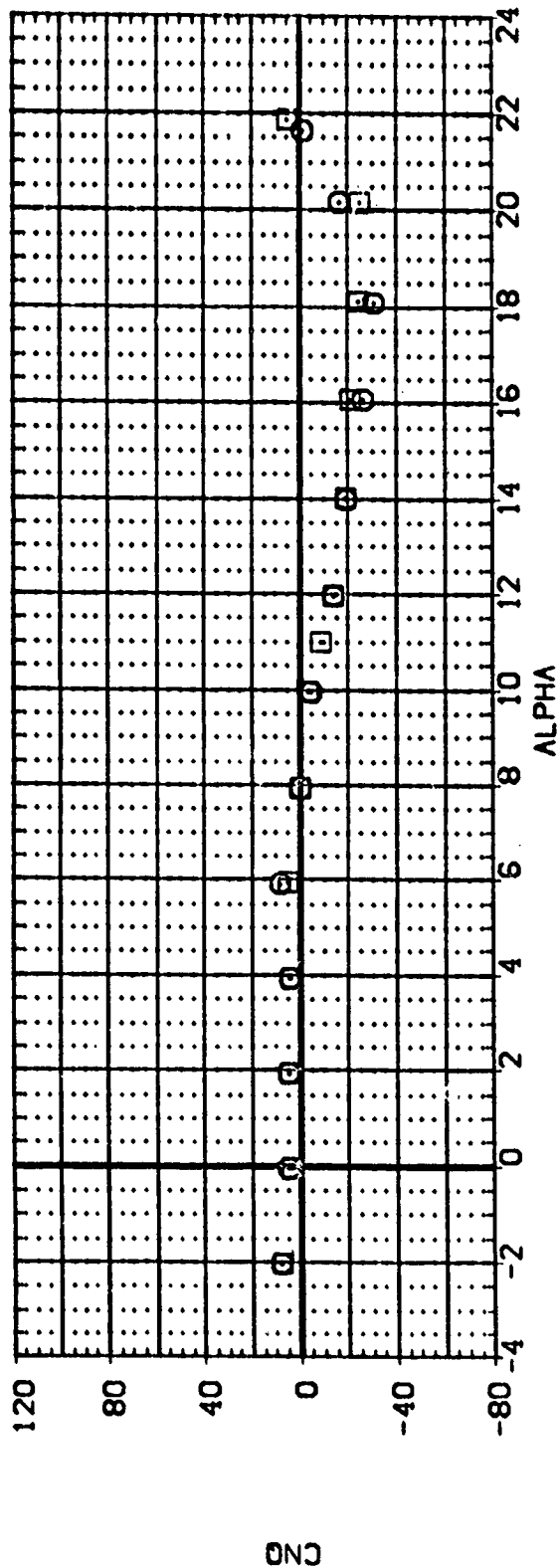


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

(B)MACH = .80

DATA SET: SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 [RPAPO4] LA-20: ROCKWELL 0693 ORB V/HOO NOSE (BNVHF) 1.000 .000 10.000
 [RPAPO5] LA-20: ROCKWELL 0693 ORB V/HOO NOSE (BNVHF) 2.000 .000 10.000

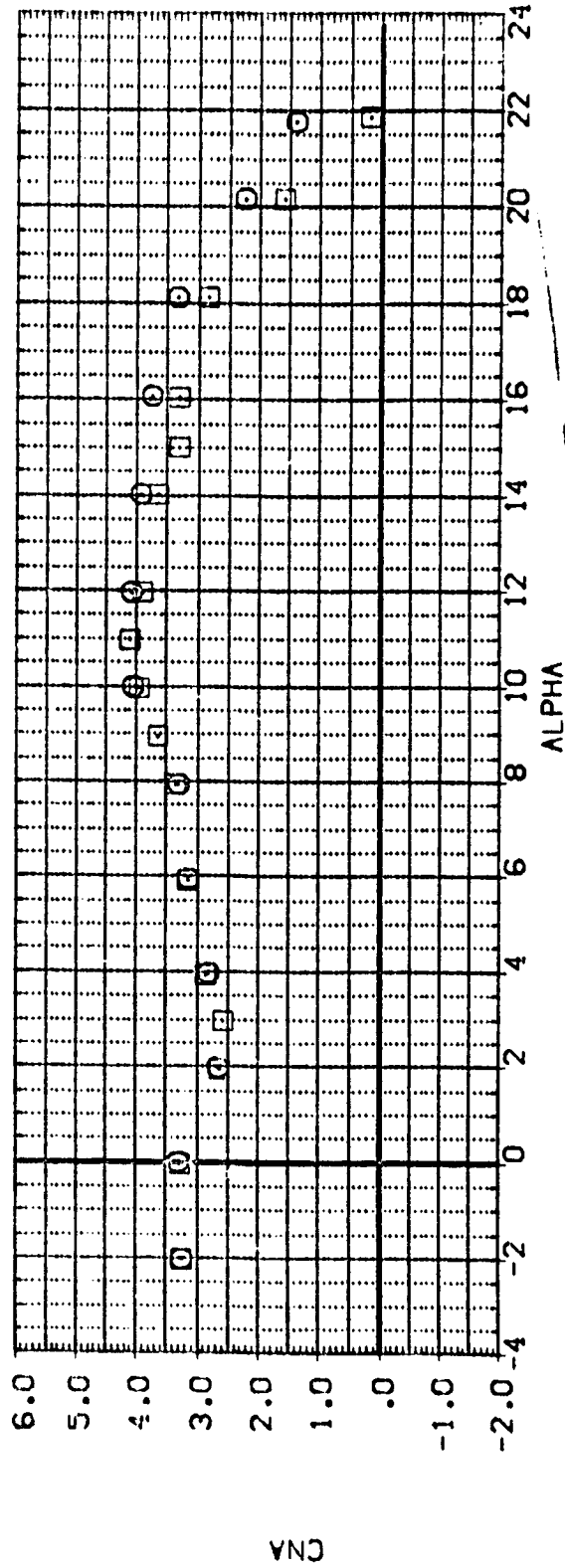
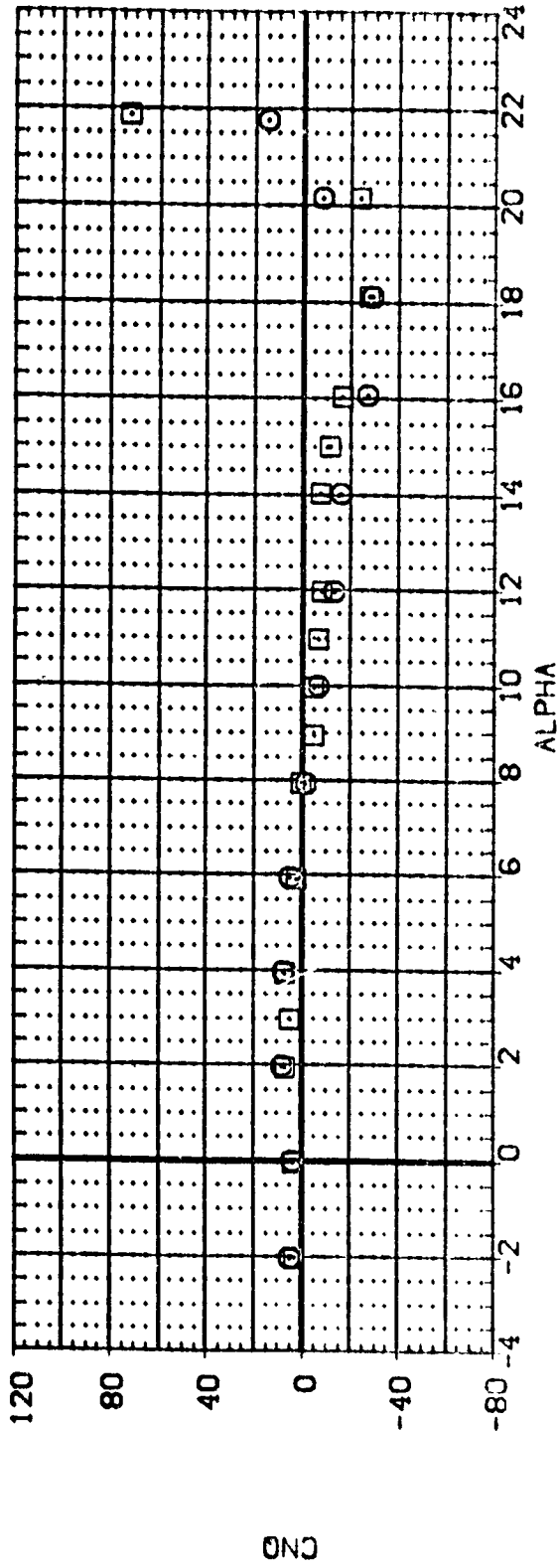


FIGURE 4. EFFECT OF C.G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR
 (RPMPO1) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMWF) 1.000 .000 .000 10.000
 (RPMPO6) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMWF) 2.000 .000 .000 10.000

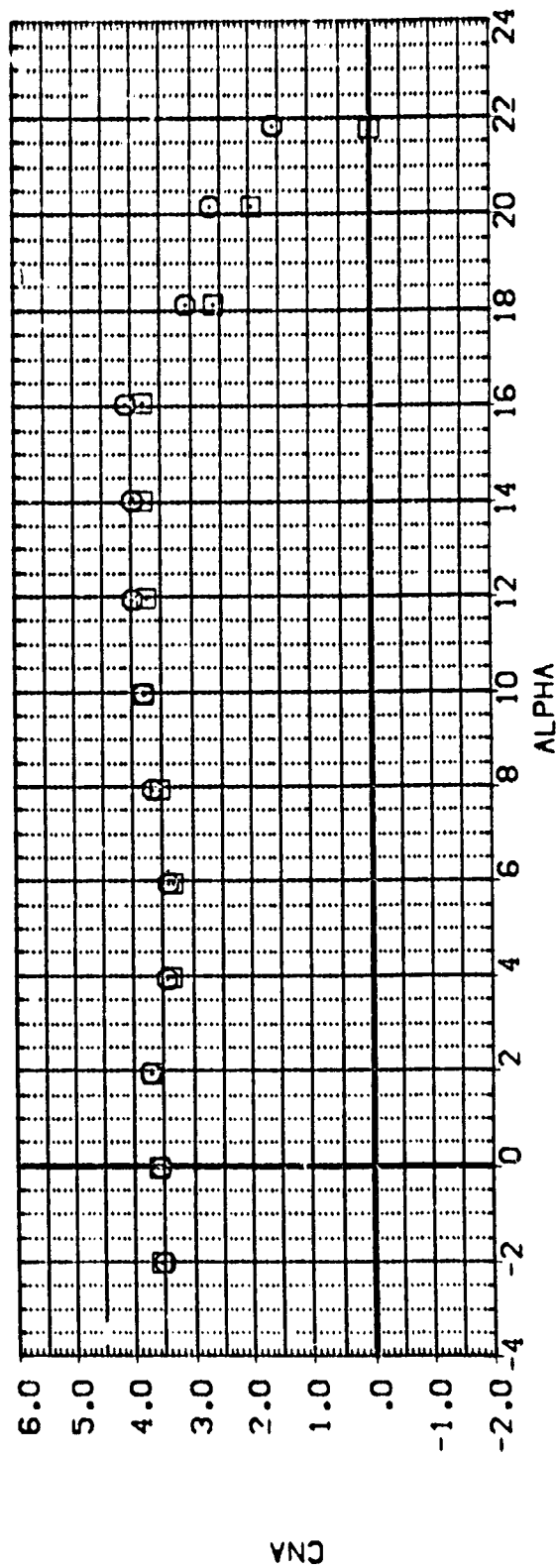
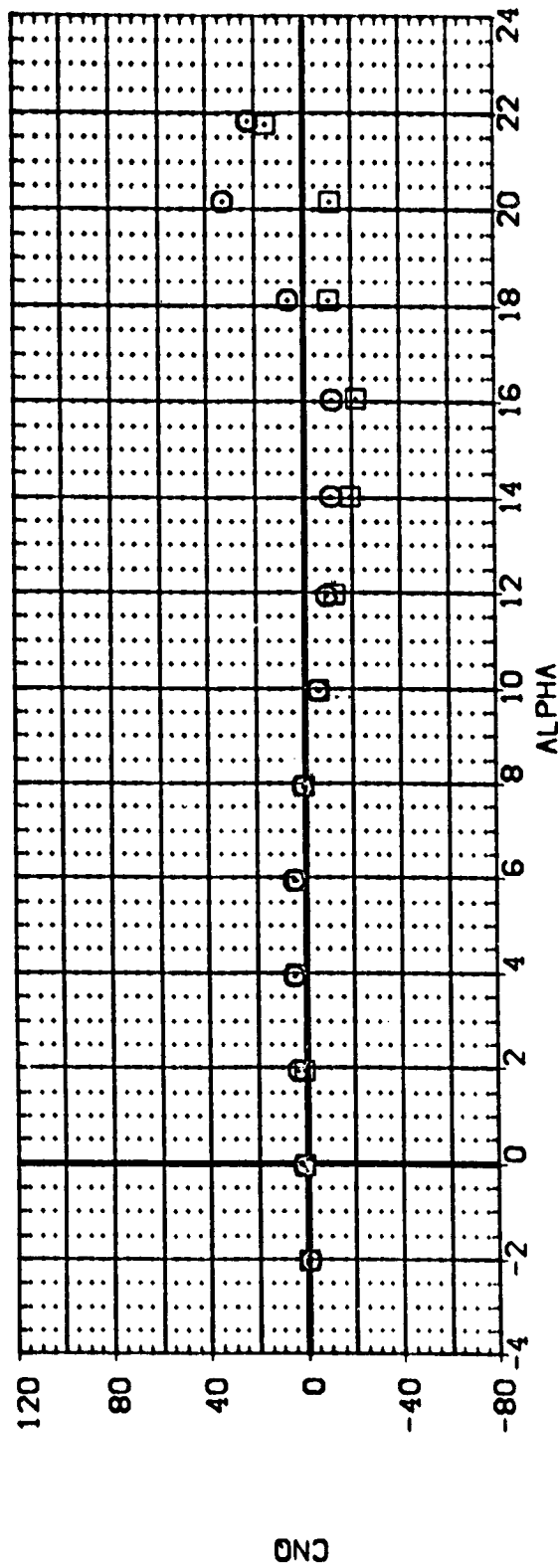


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

(C)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUFLR

[RKP04] LA-20, ROCKWELL 0899 ORB V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

[RKP06] LA-20, ROCKWELL 0899 ORB V/MOD NOSE (SVMF) 2.000 .000 .000 10.000

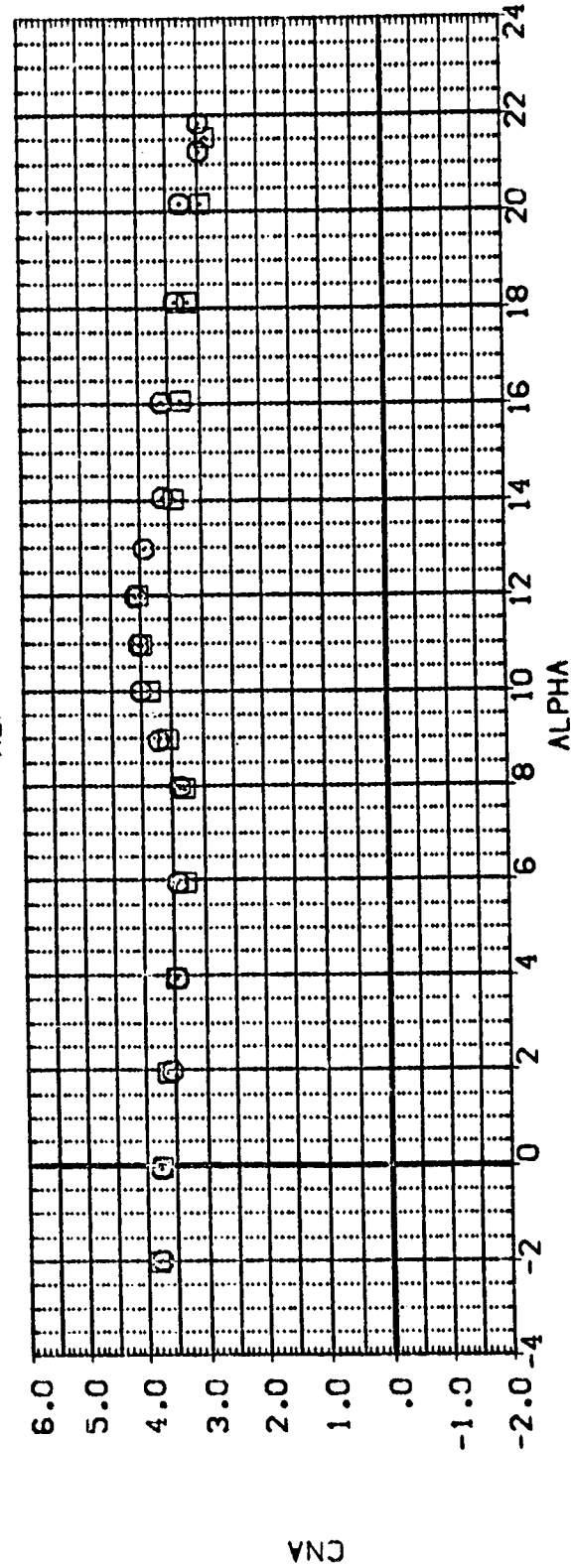
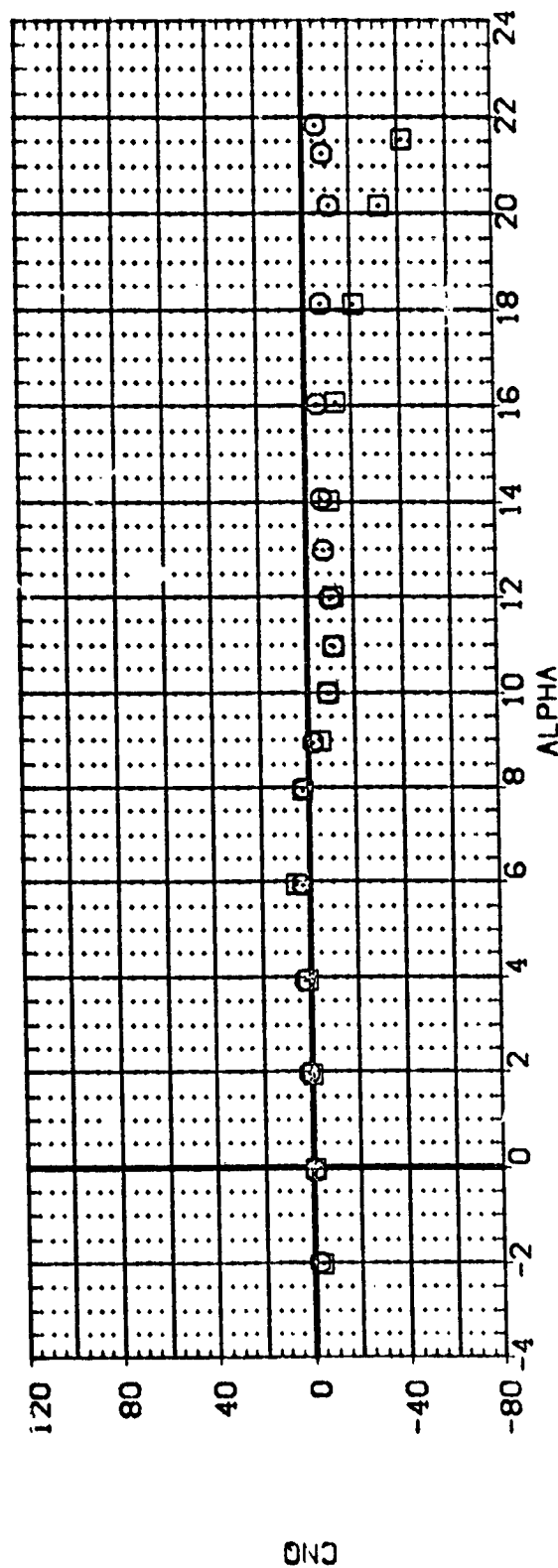


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

(C)MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOF LR

[RPKPO4] LA-20, ROCKWELL 089B ORB V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

[RPKPO6] LA-20, ROCKWELL 089B ORB V/MOD NOSE (BVMF) 2.000 .000 .000 10.000

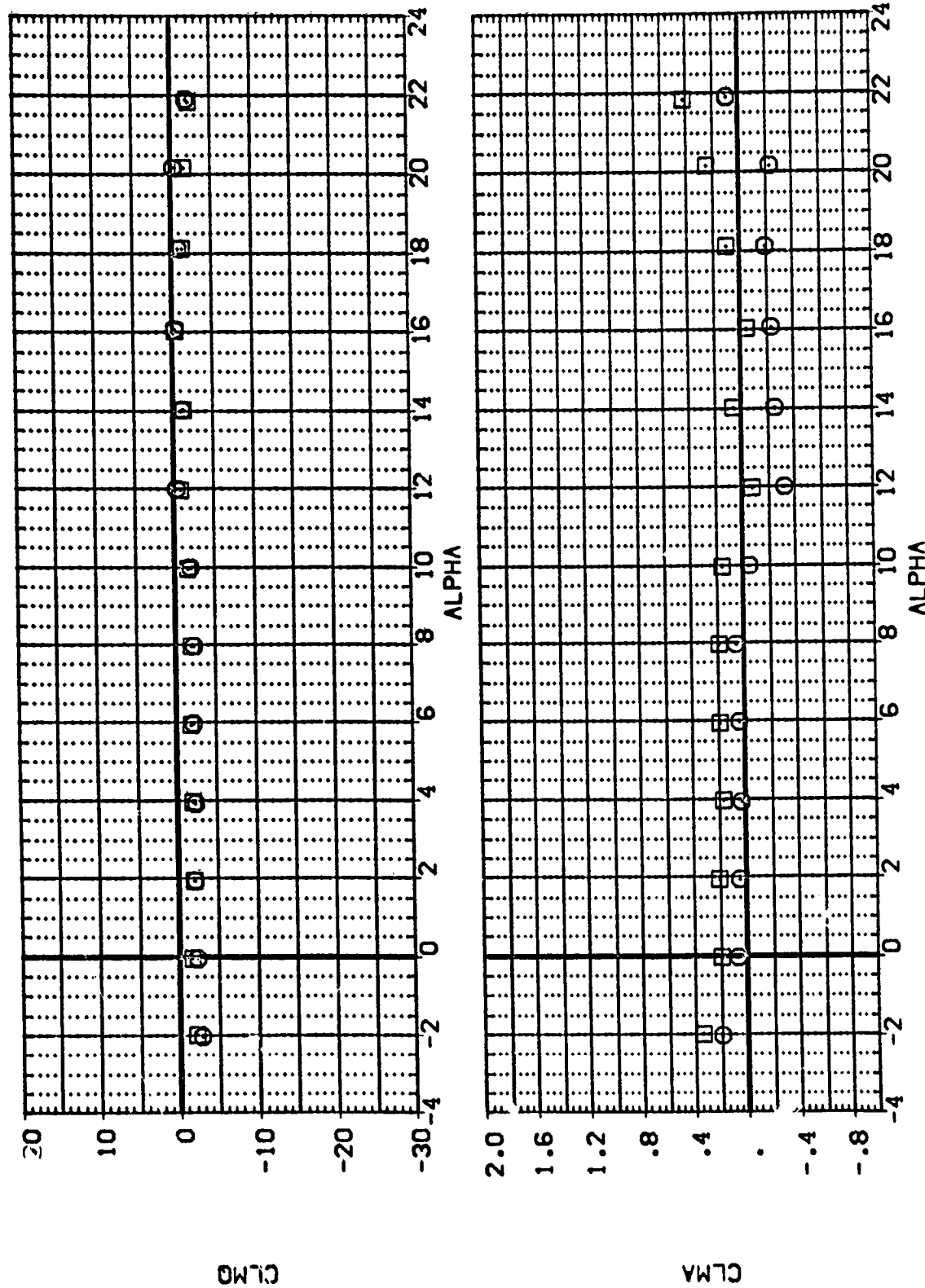


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

(MACH = .30

DATA SET SYMBOL: [RKIP04] [RKIP05] CONFIGURATION DESCRIPTION: LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMVF) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMVF) CG-LOC: 1.000 2.000 ELEVTR: .000 .000 BOFLAP: .000 .000 RUOFIR: 10.000 10.000

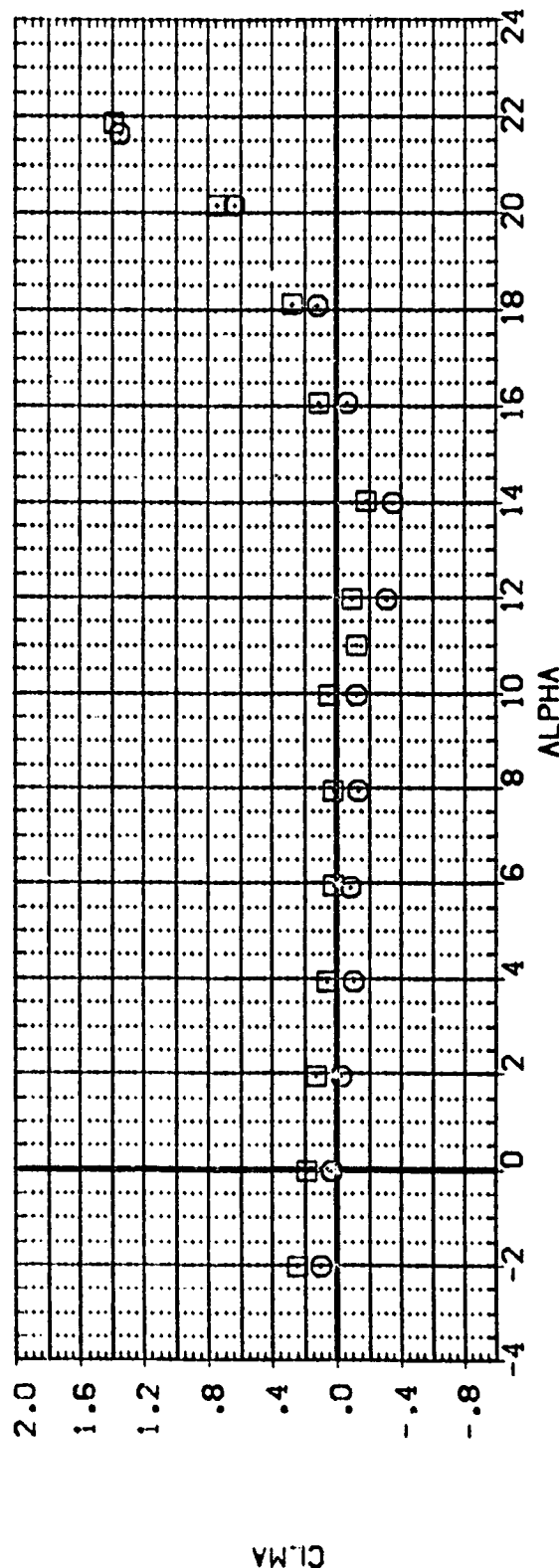
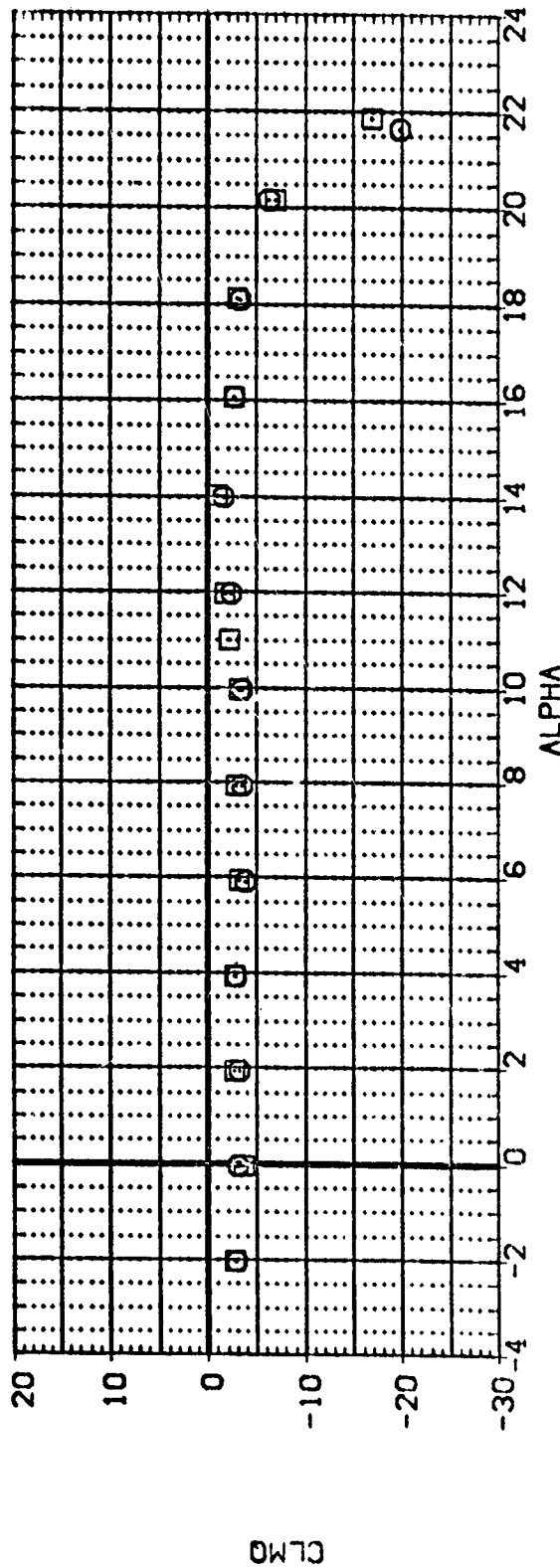


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH
[B]MACH = .80 PAGE

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 [RPKPO4] LA-20, ROCKWELL 085B ORB V/MOD NOSE (BWVF) 1.000 .000 .000 10.000
 [RPKPOS] LA-20, ROCKWELL 085B ORB V/MOD NOSE (BWVF) 2.000 .000 .000 10.000

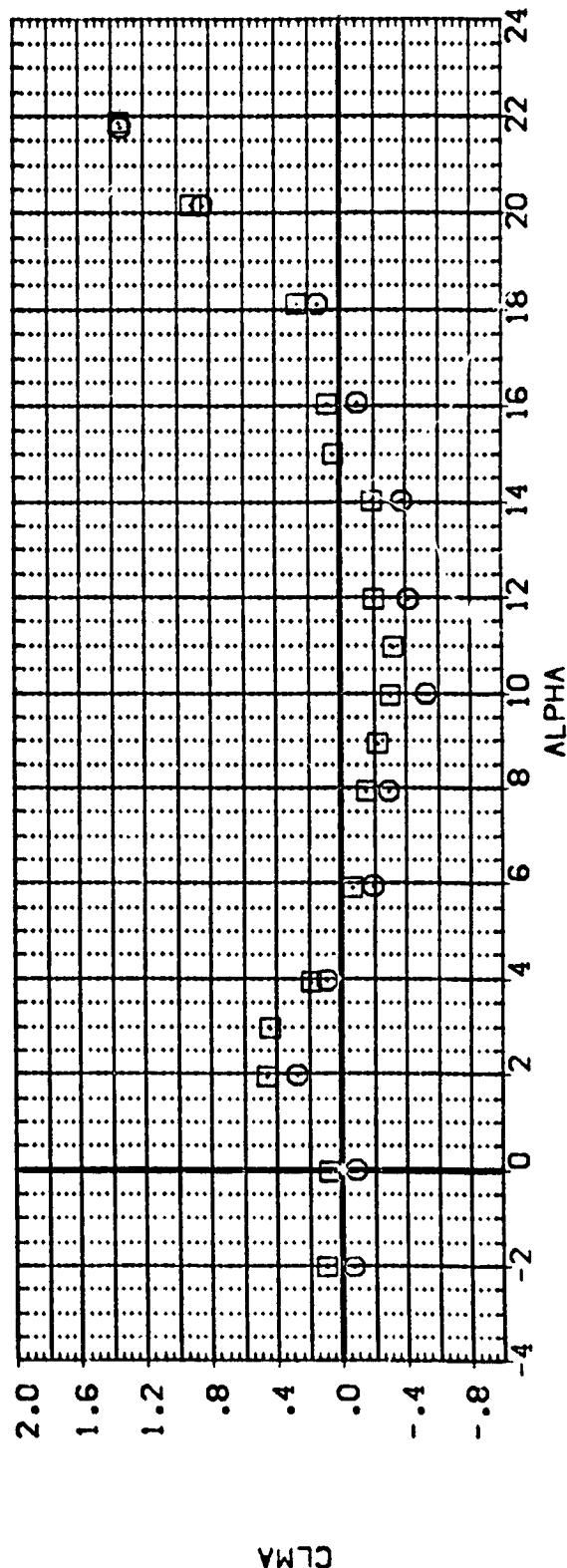
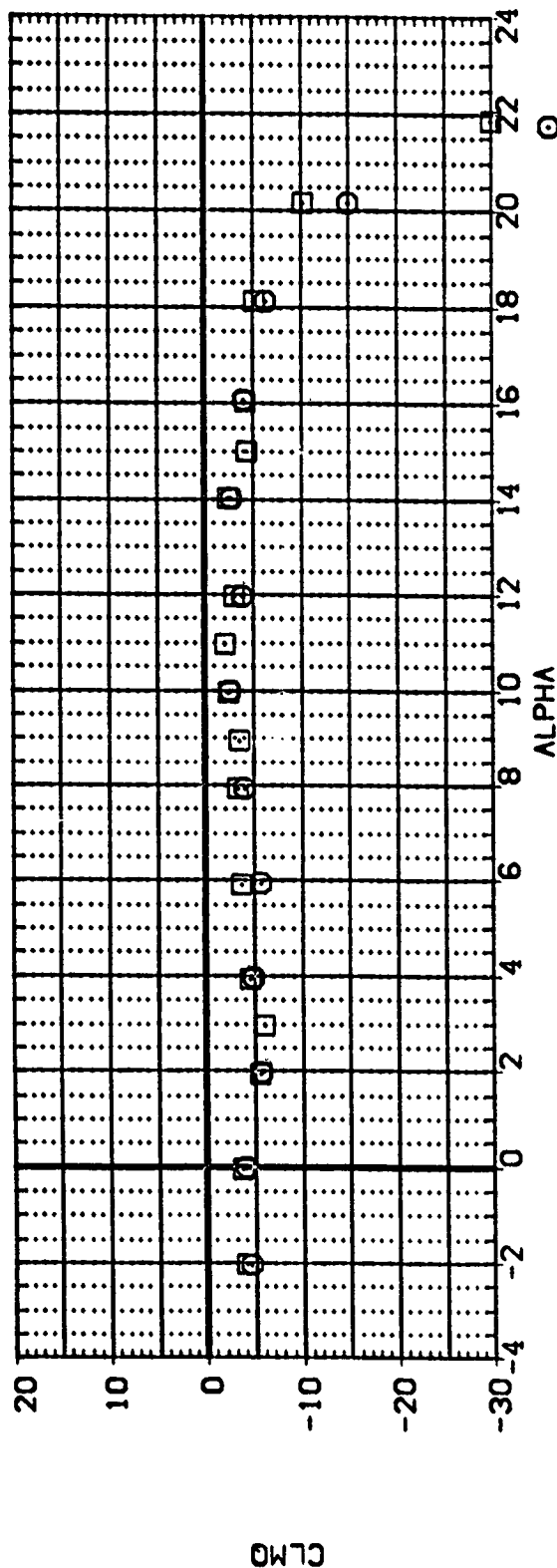


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

(C)MACH = .90

DATA SET SYMBOL: [Symbol] CONFIGURATION DESCRIPTION: LA-20, ROCKWELL 089B DRB V/MOD NOSE (BVMVF) CG-LOC: 1.000 ELEVTR: .000 BDFLAP: .000 RUDEL R: 10.000
 [Symbol] LA-20, ROCKWELL 089B DRB V/MOD NOSE (BVMVF) CG-LOC: 2.000 ELEVTR: .000 BDFLAP: .000 RUDEL R: 10.000

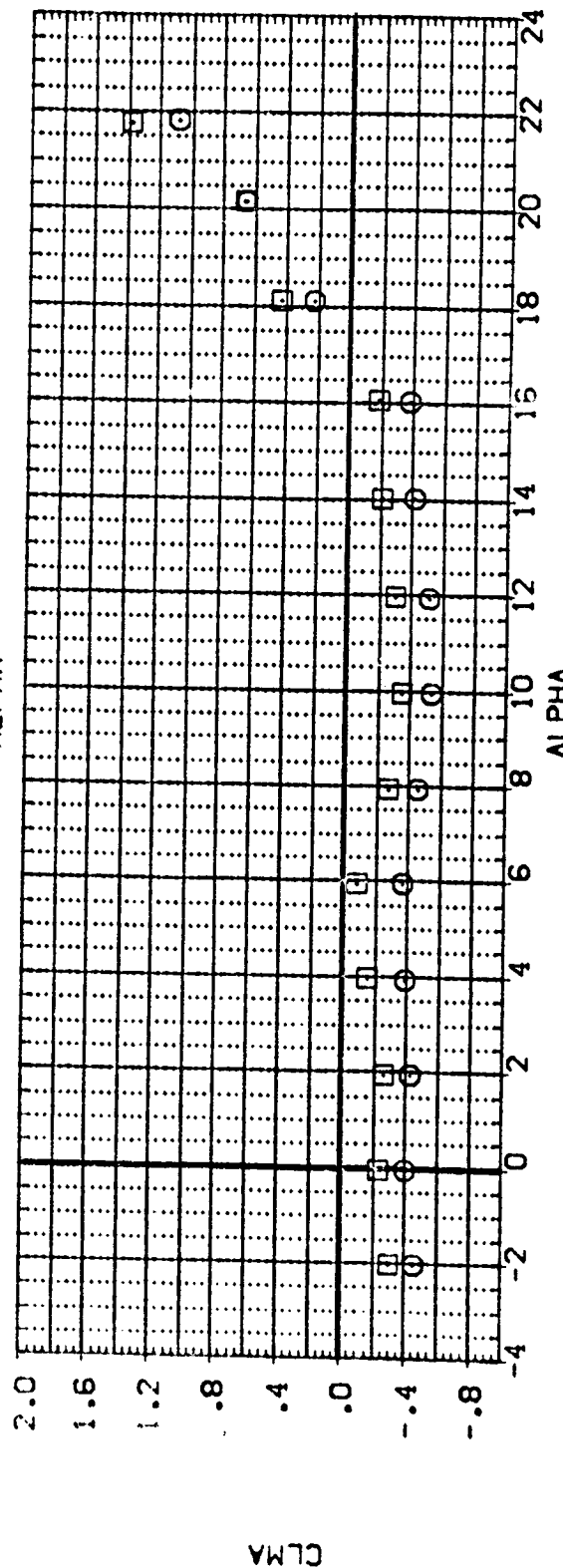
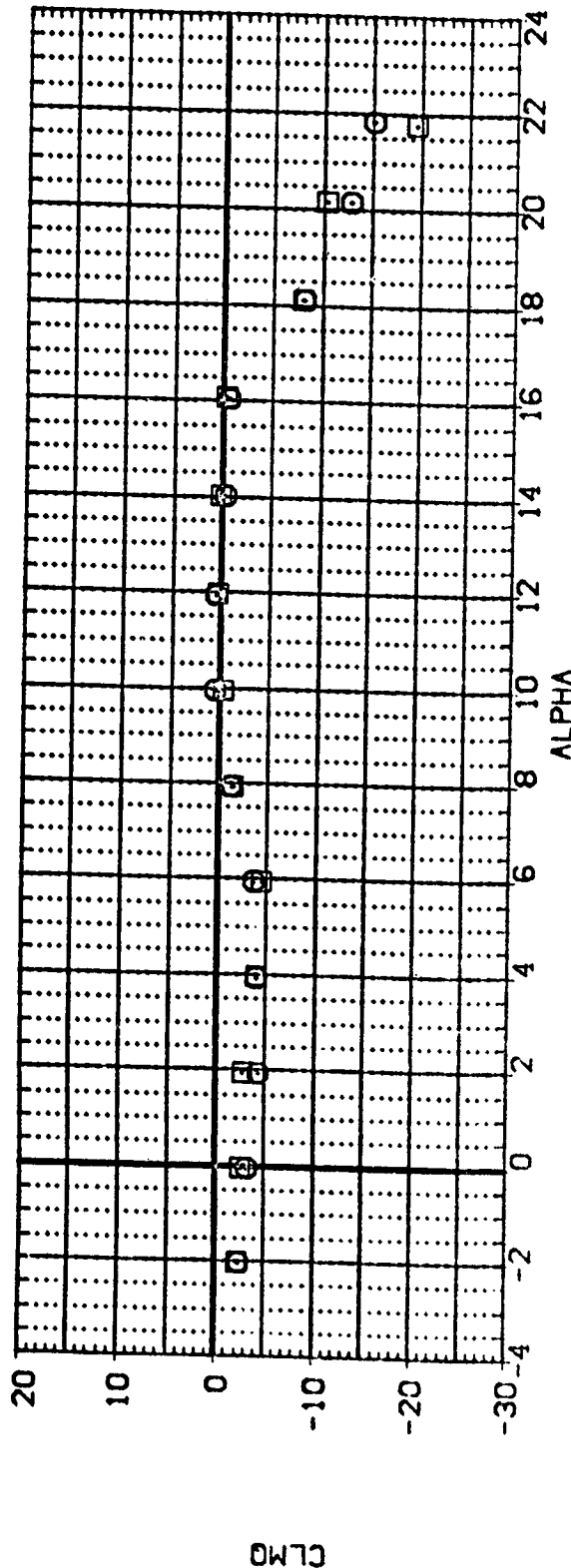


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

CLMACH = .98

CG-LOC ELEVTR ROTLAP RUOLFR
1.000 .000 .000 10.000
2.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
LA-20: ROCKWELL 0898 ORB V/MOD NOSE (BVMF)
LA-20: ROCKWELL 0898 ORB V/MOD NOSE (BVMF)

[RKP04]
[RKP06]

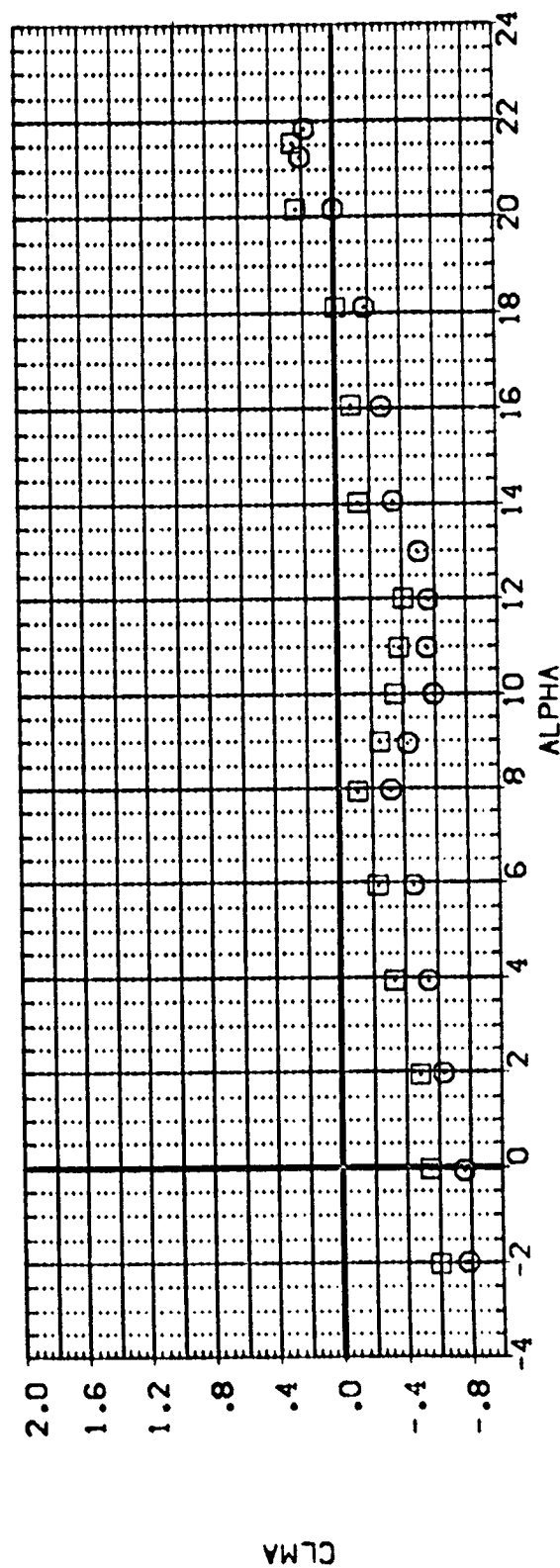
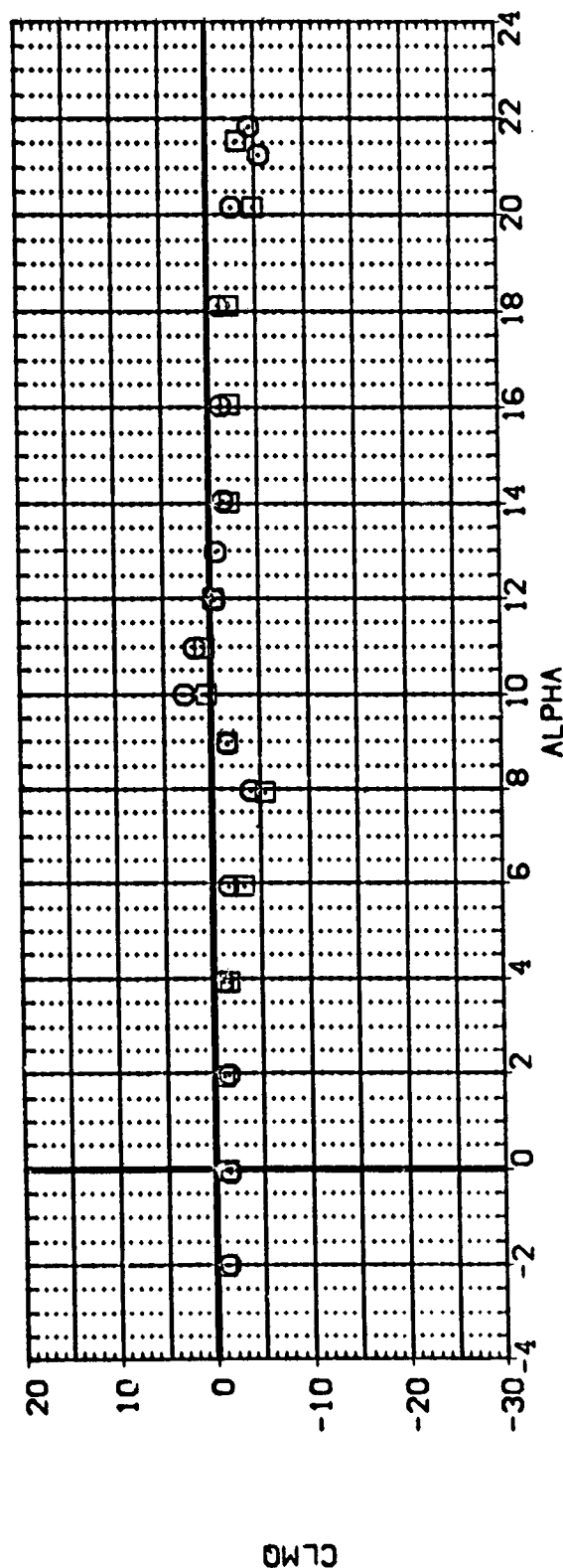


FIGURE 4. EFFECT OF C. G. LOCATION ON DYNAMIC STABILITY PARAMETERS IN PITCH

(E)MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR

(RPK202) LA-20: ROCKWELL 0898 ORB V/MOD NOSE (BNV.F) 1.000 .000 .000 10.000

(RPK204) LA-20: ROCKWELL 0898 ORB V/MOD NOSE (BNV.F) 1.000 .000 .000 10.000

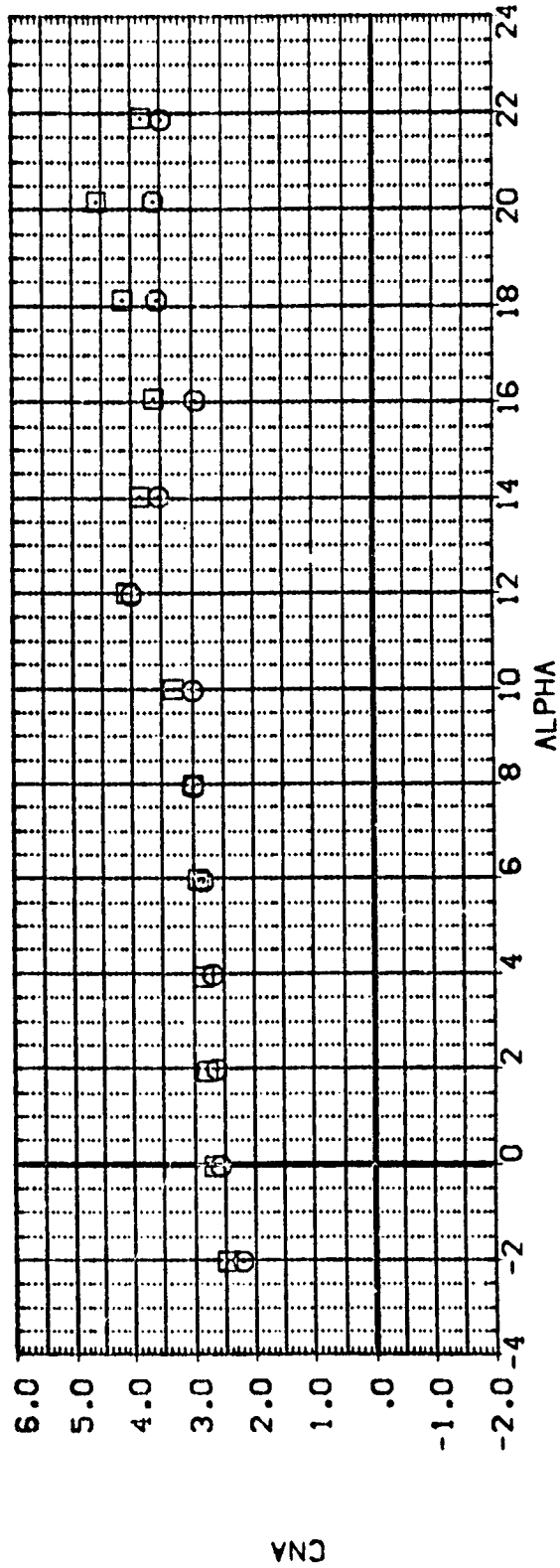
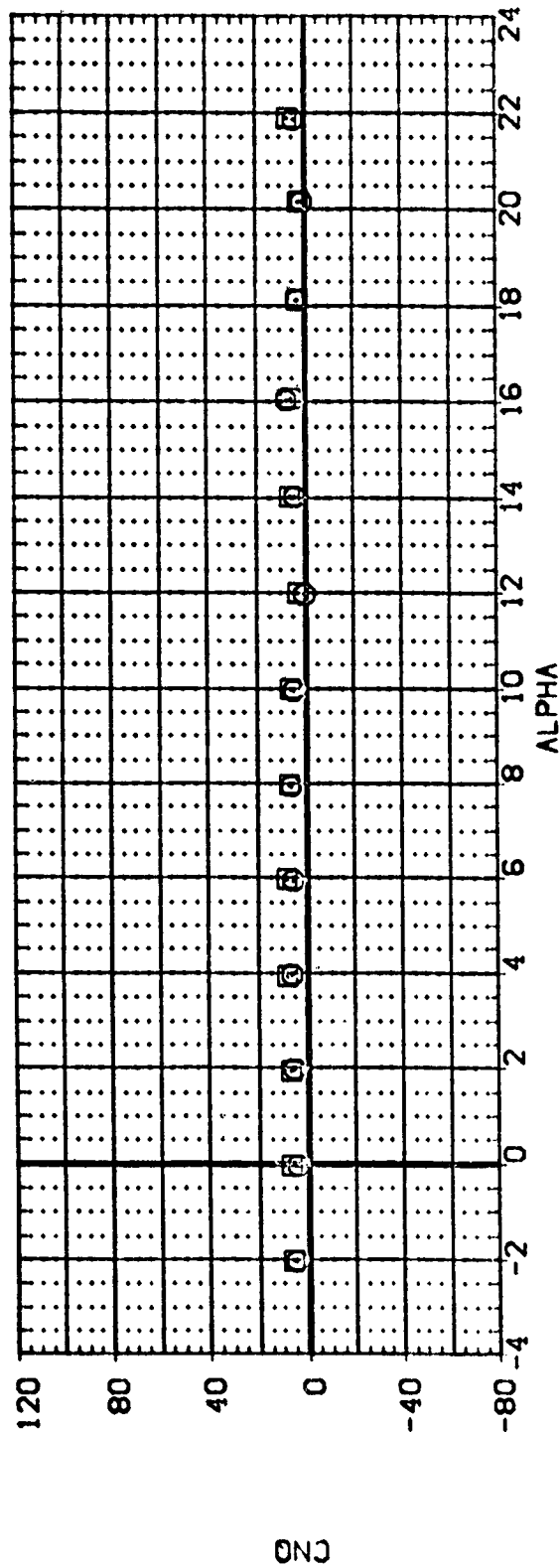


FIGURE 5. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN PITCH

CALMACH = .30

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUFLR

[RAMPG2] LA-20, ROCKWELL 0899 DRB V/MOD NOSE (BVA F) 1.000 .000 .000 10.000

[RAMPG3] LA-20, ROCKWELL 0899 DRB V/MOD NOSE (BVA F) 1.000 .000 .000 10.000

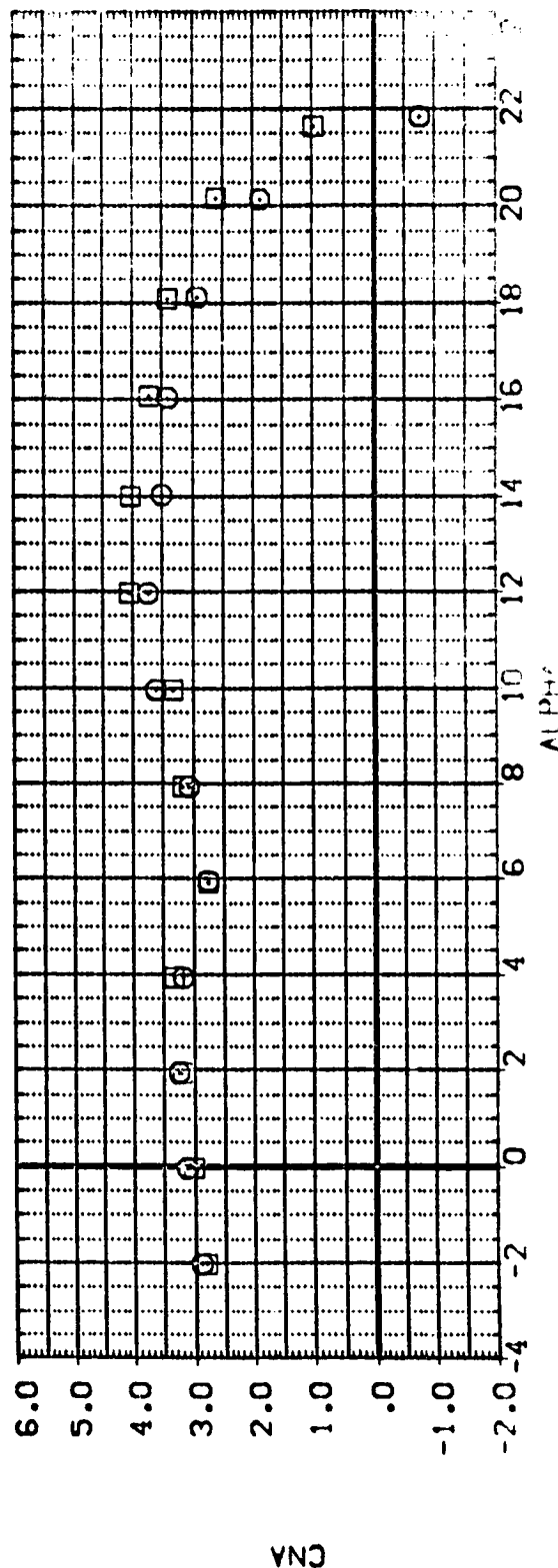
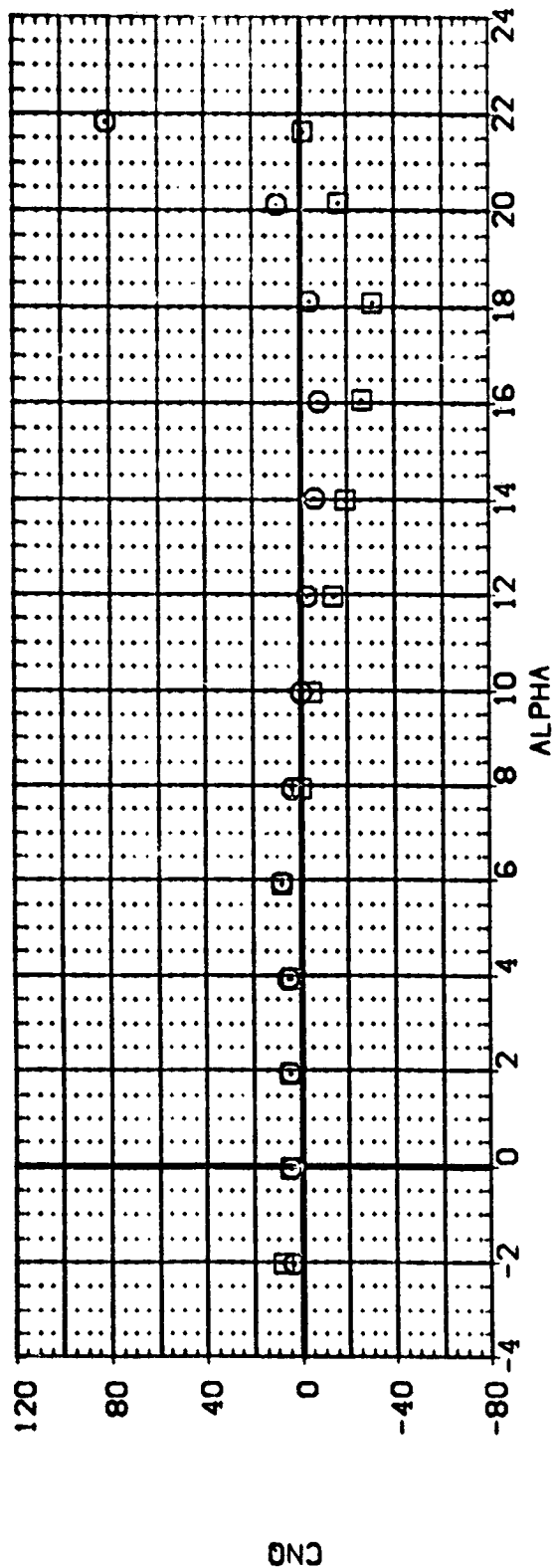


FIGURE 5. EFFECT OF DR PDS ON DYNAMIC STABILITY PARAMETERS IN PITCH

DATA SET SYMBOL CONFIGURATION DESCRIPTION CU-LOC ELEVTR BOFLAP RUFLR

[RAMP01] LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVF F) 1.000 .000 .000 10.000

[RAMP04] LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVFH) 1.000 .000 .000 10.000

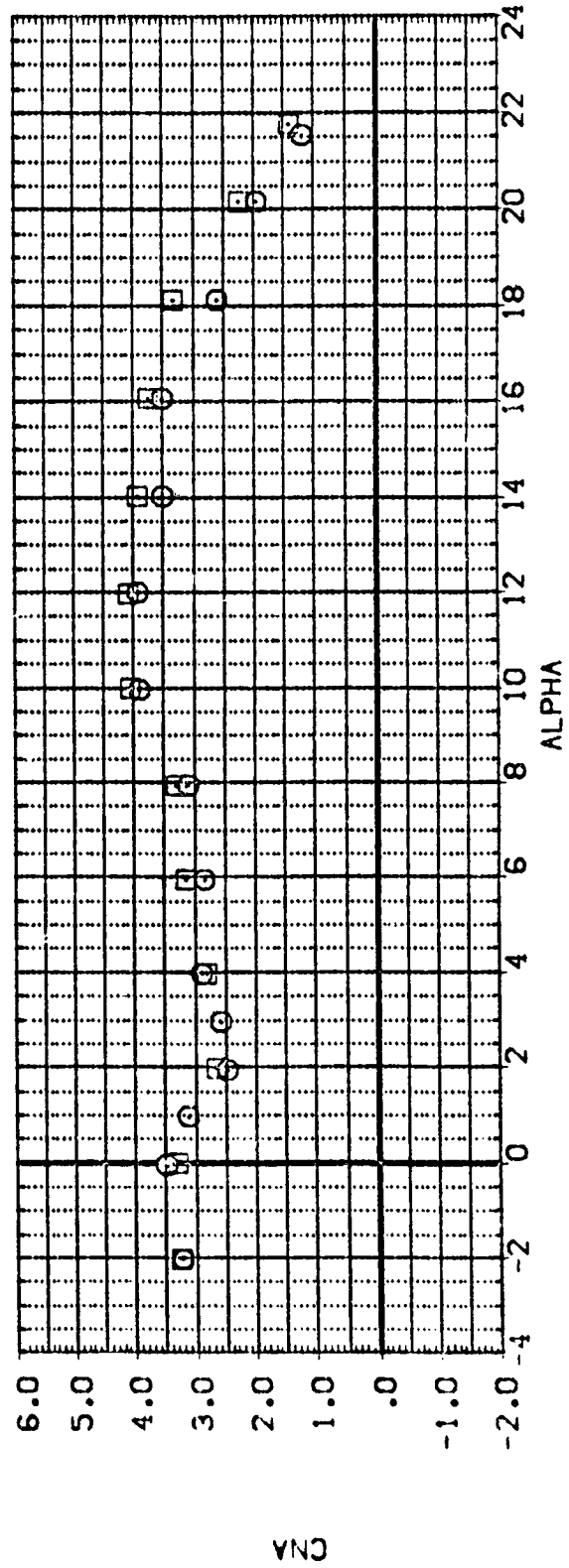
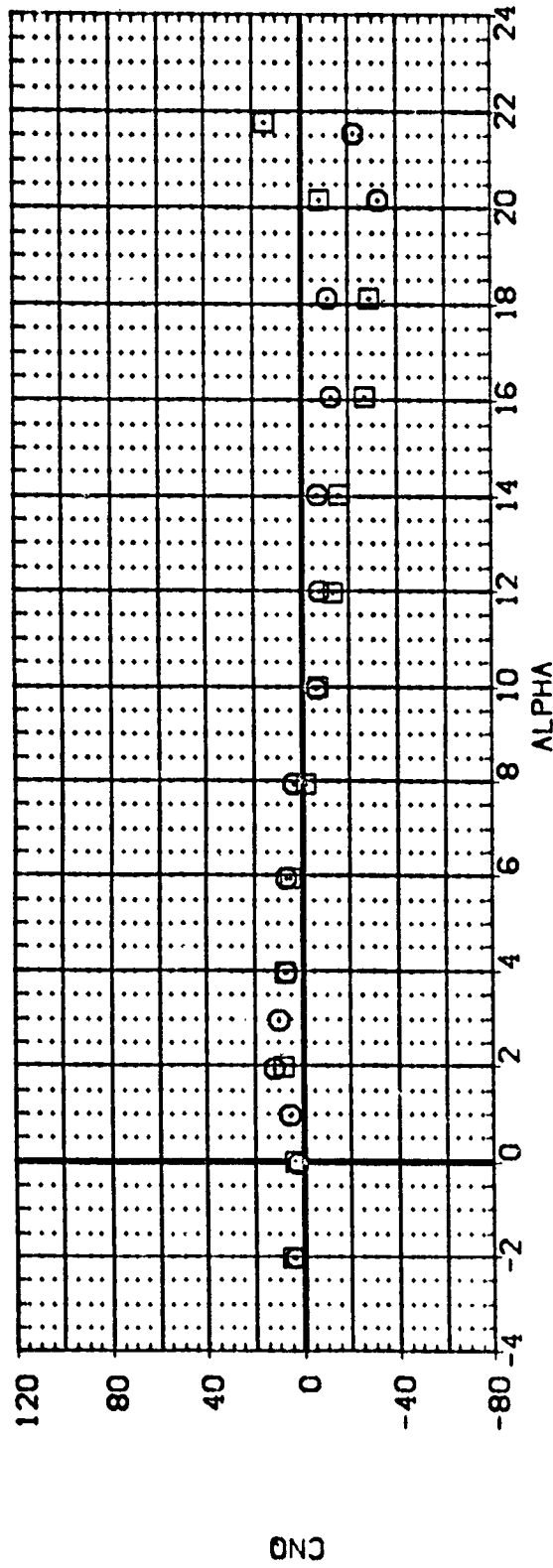
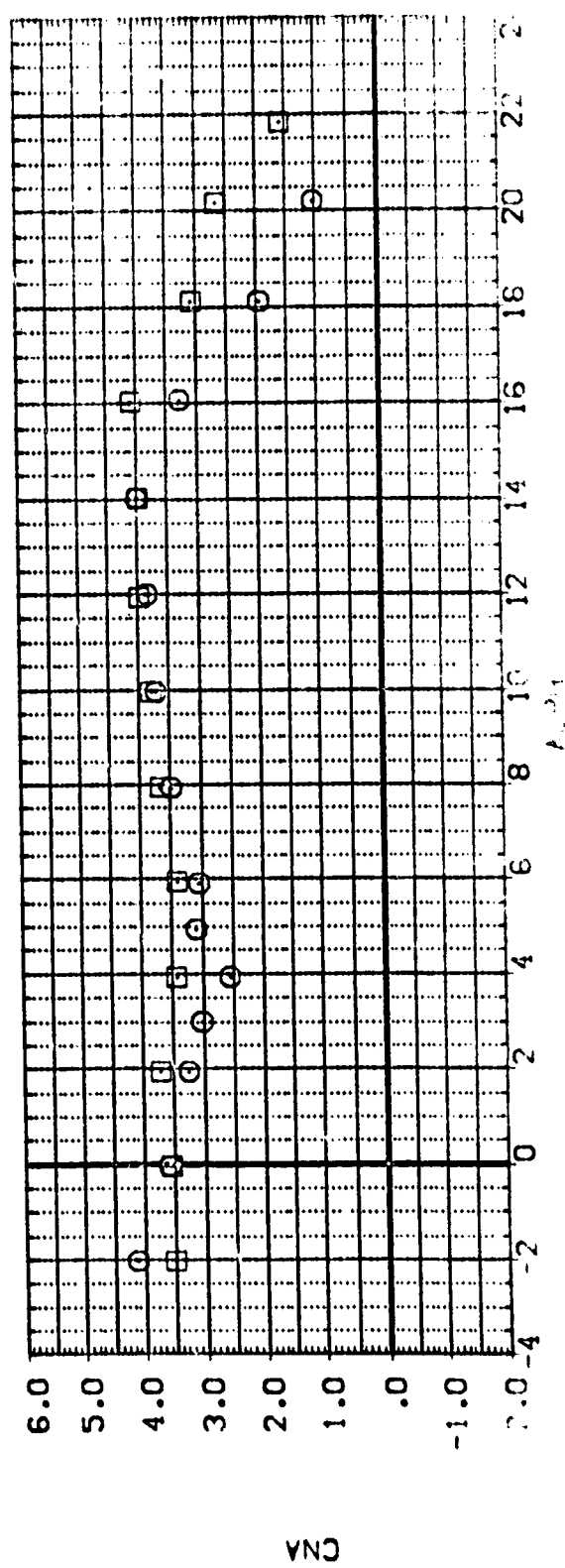
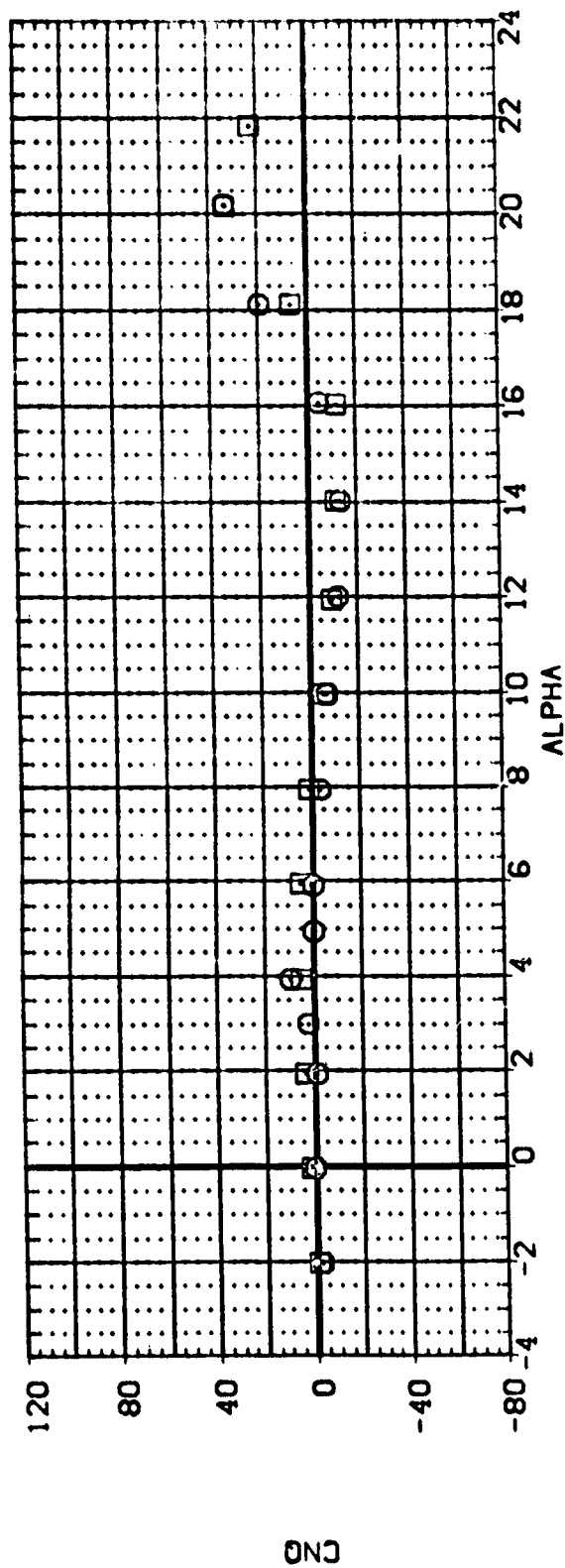


FIGURE 5. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN PITCH

COM MACH = .90

14-0

QAT. S. T. SYMBOL	CONFIGURATION	DESCRIPTION
(RMPD3)	LA-20, ROCKWELL	0698 DRB V/MOD NOSE (BHV F)
(RMPD4)	LA-20, ROCKWELL	0698 DRB V/MOD NOSE (BHV F)



5. EFFECT OF AN PDS ON DRYING OF WET FIBERS

DATA SET SYMBOL: [RPA-P02] [RPA-P04] CONFIGURATION DESCRIPTION: LA-20, ROCKWELL O898 OR8 V/MOD NOSE (BVM F) LA-20, ROCKWELL O898 OR8 V/MOD NOSE (BVM F) CG-LOC: 1.000 1.000 ELEVTR: .000 .000 BDFLAP: .000 .000 RUOFLR: 10.000 10.000

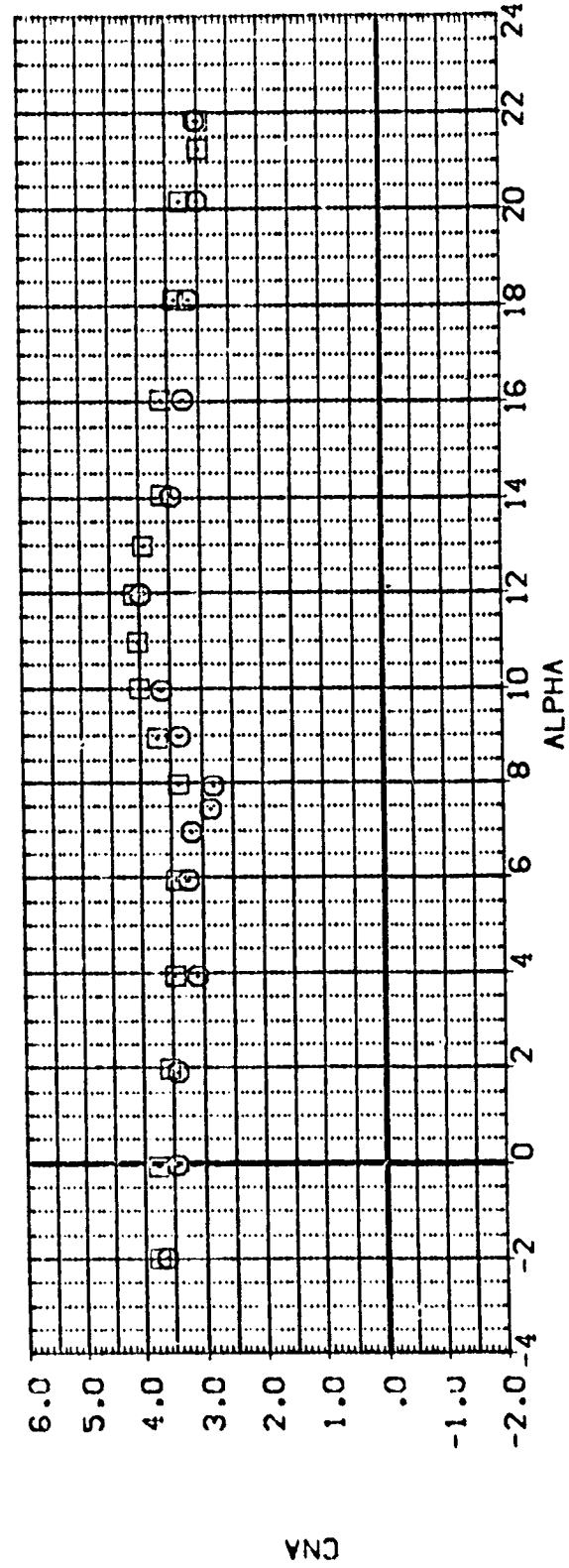
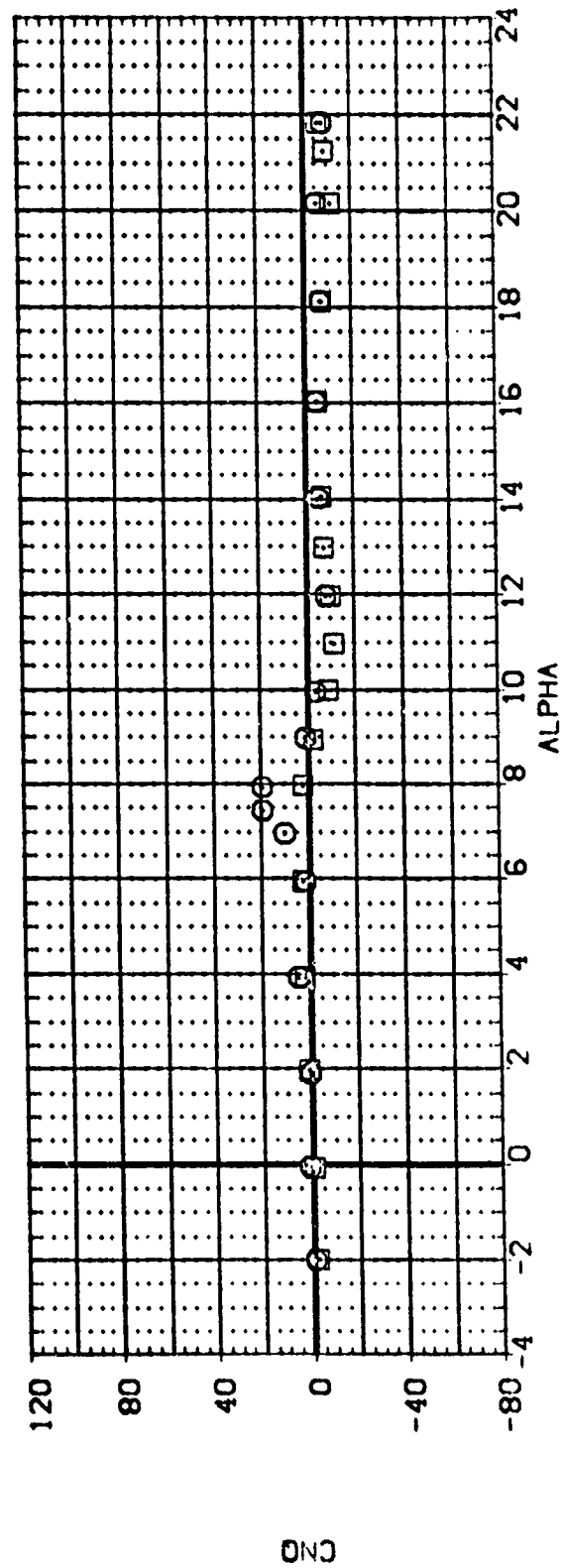


FIGURE 5. EFFECT OF OMS POOS ON DYNAMIC STABILITY PARAMETERS IN PITCH

(MACH = 1.20)

DATA SET SYMBOL CONFIGURATION DESCRIPTION CS-LOC ELEVTR BOFLAP RUOFLR

(RPM002) LA-20; ROCKWELL 0898 ORB V/MOD NOSE (BNV F) 1.000 .000 .000 10.000

(RPM004) LA-20; ROCKWELL 0898 ORB V/MOD NOSE (BNV F) 1.000 .000 .000 10.000

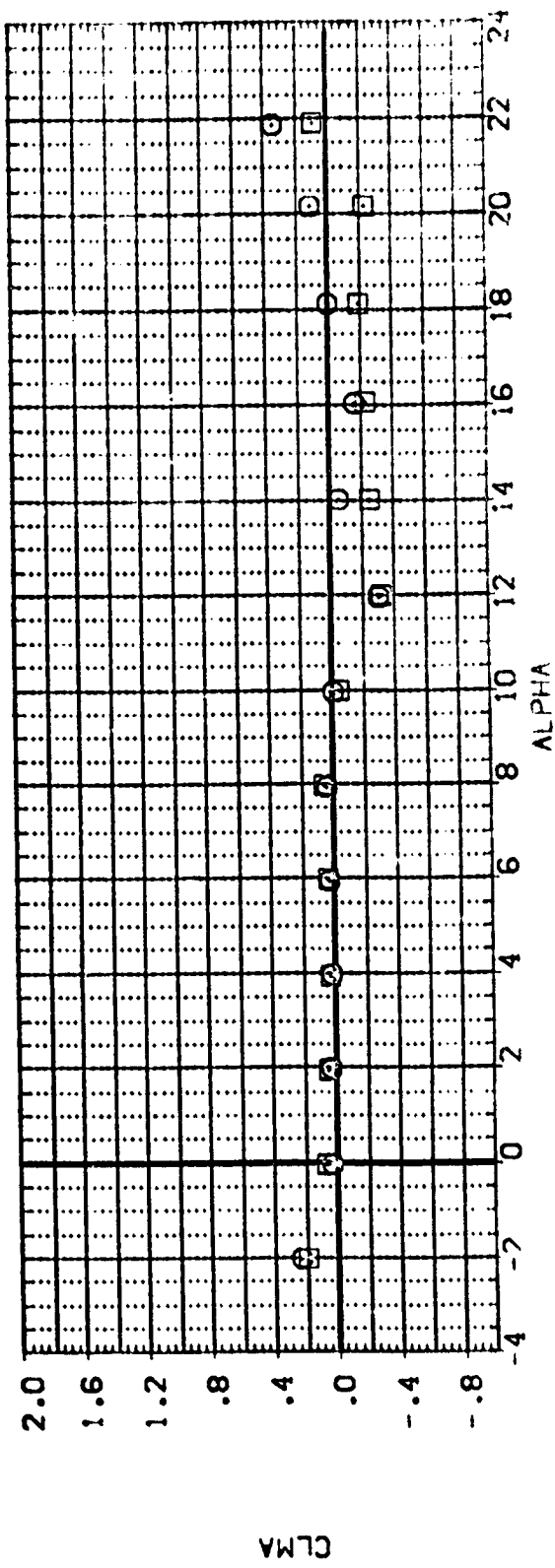
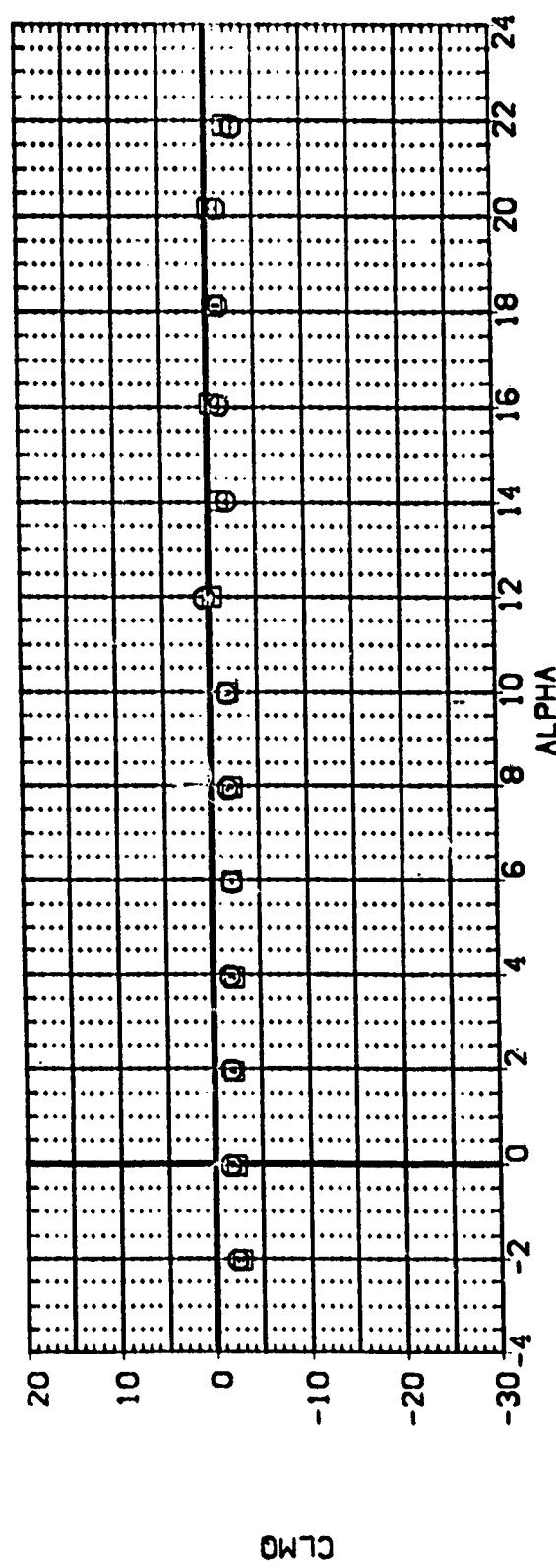


FIGURE 5. EFFECT OF RMS RMS IN DYNAMIC STABILITY PARAMETERS IN PITCH

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG LOC ELEVTR BOFLAP ROFLR

[RPMQ2] LA-20, ROCKWELL O898 ORB V/MOD NOSE (BWW F) 1.000 .000 .000 10.000

[RPMQ4] LA-20, ROCKWELL O898 ORB V/MOD NOSE (BWW F) 1.000 .000 .000 10.000

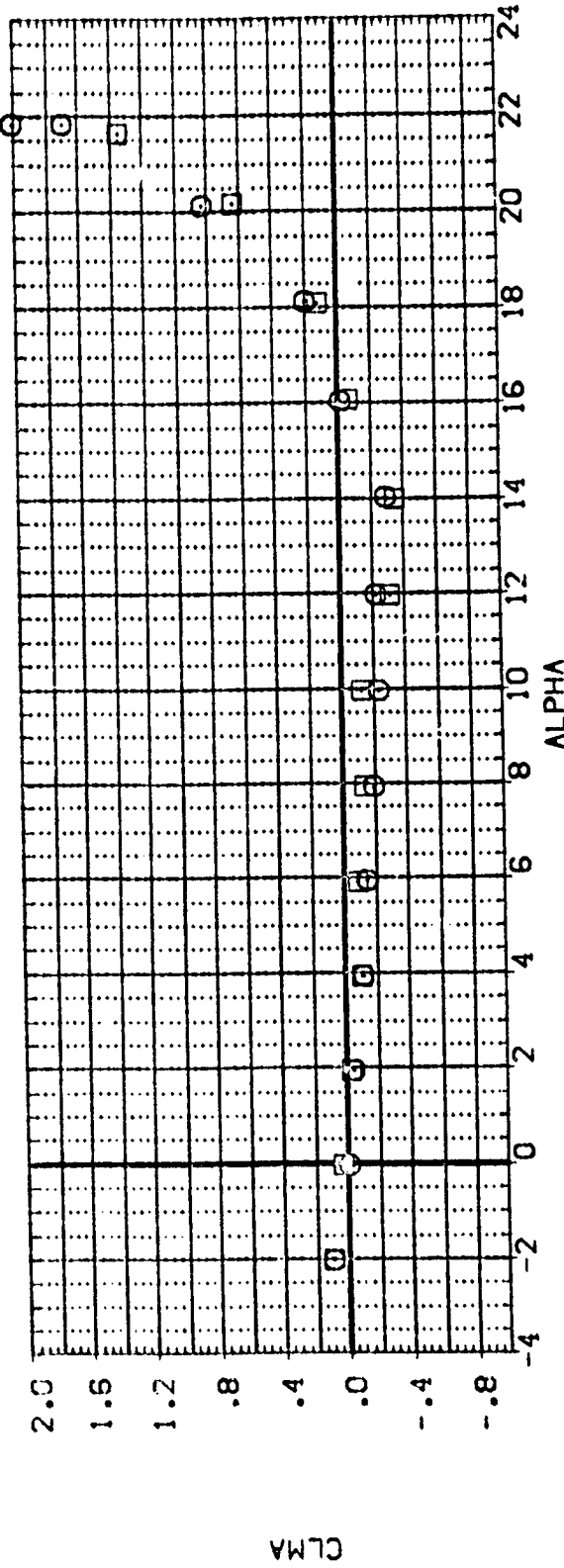
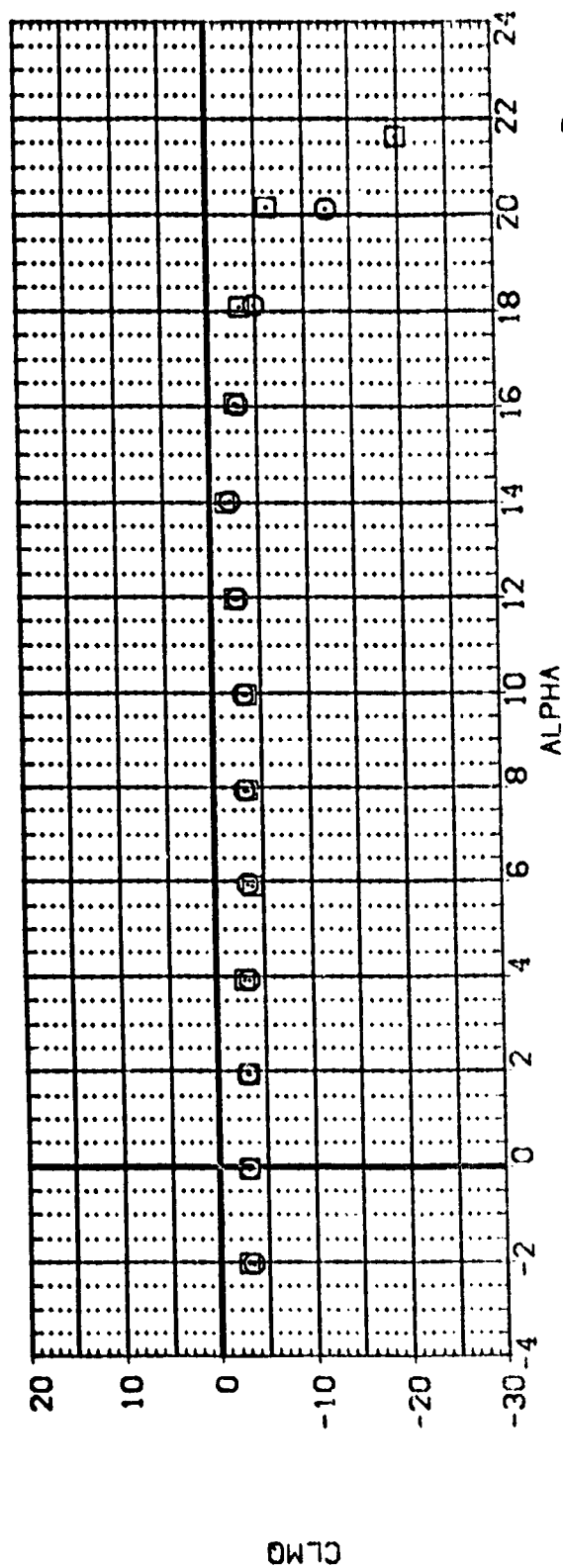


FIGURE 5. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN PITCH

(B)MACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR

[RPKPD2] LA-20, ROCKWELL 0998 GRB V/MOD NOSE (BVM F) 1.000 .000 .000 10.000

[RPKPD4] LA-20, ROCKWELL 0892 GRB V/MOD NOSE (BVM F) 1.000 .000 .000 10.000

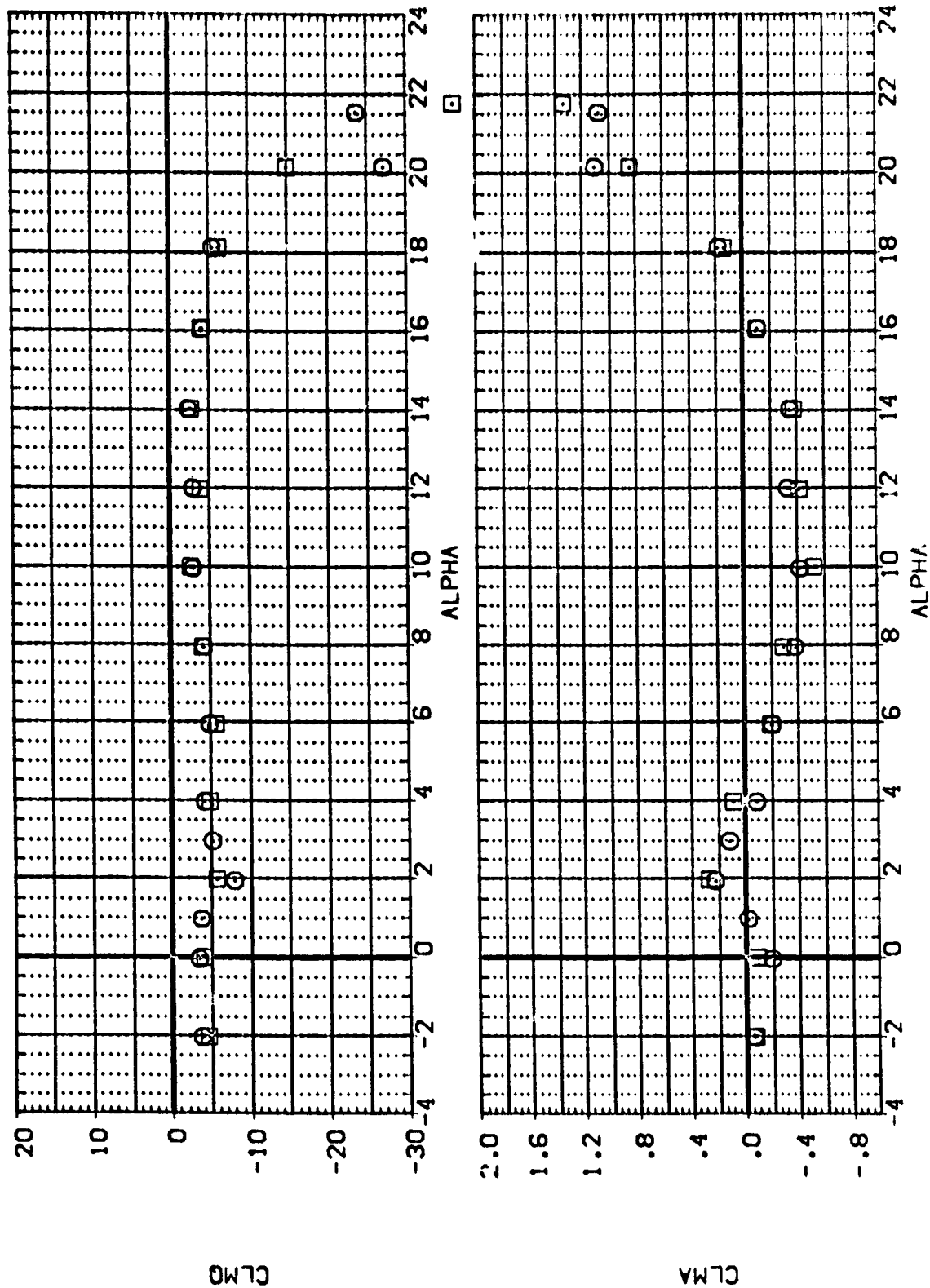


FIGURE 5. EFFECT OF RMS PODS ON DYNAMIC STABILITY PARAMETERS IN PITCH

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	CG-LOC	ELEVTR	BOFLAP	RUDFLR
(RPM-P02)	LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVM-F)	1.000	.000	.000	10.000
(RPM-P01)	LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVM-F)	1.000	.000	.000	10.000

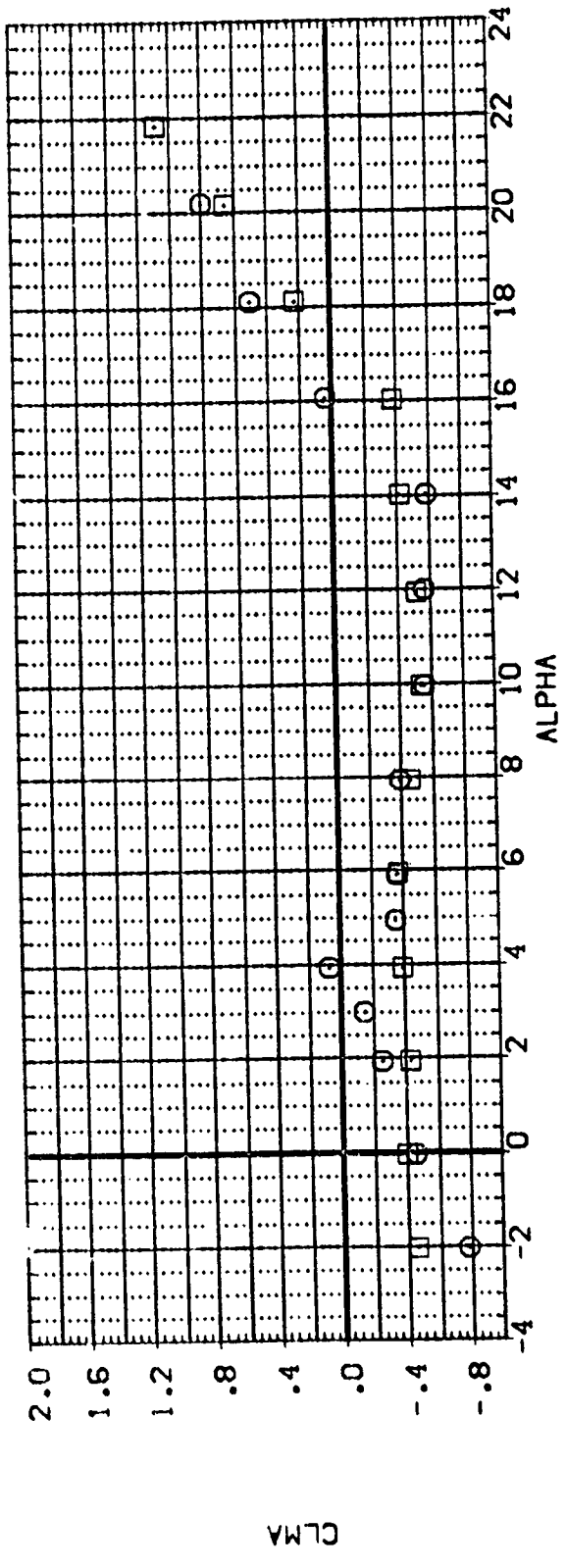
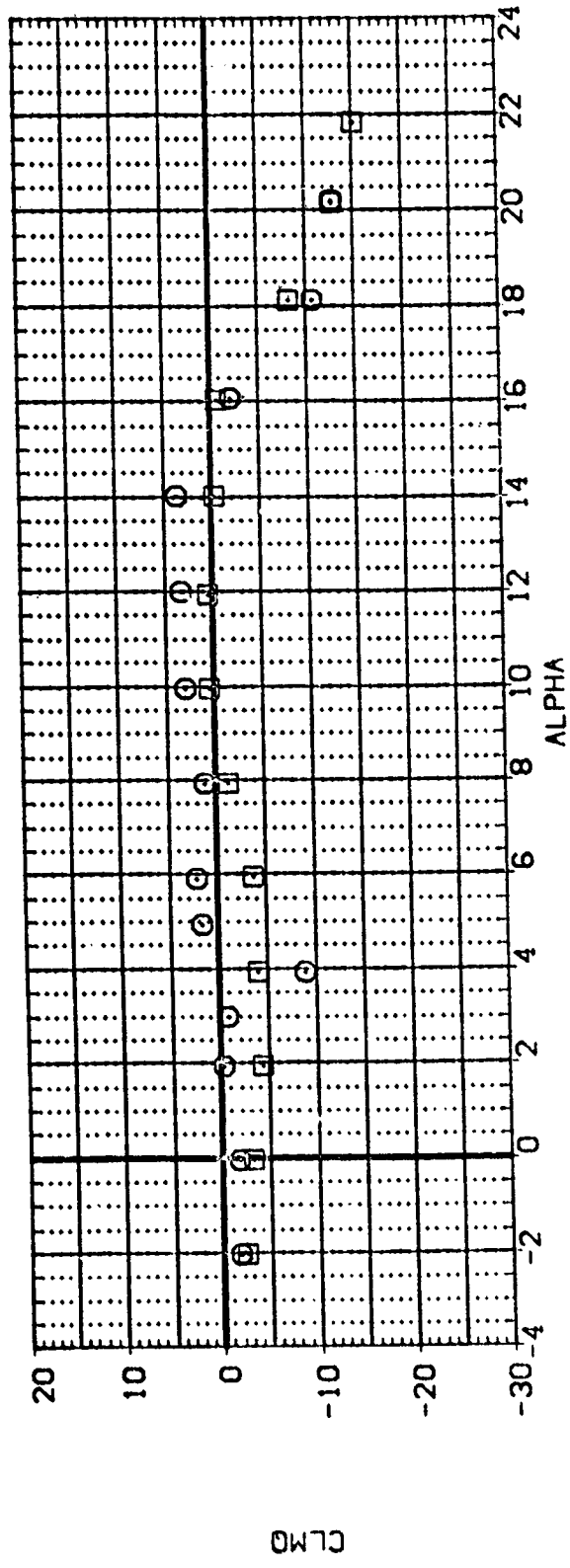


FIGURE 5. EFFECT OF QMS PODS ON DYNAMIC STABILITY PARAMETERS IN PITCH

(D)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

(RPMPO2) LA-20, ROCKWELL 0898 0R8 V/MCD NOSE (BNV F)

(RPMPO1) LA-20, ROCKWELL 0898 0R8 V/MCD NOSE (BNVHF)

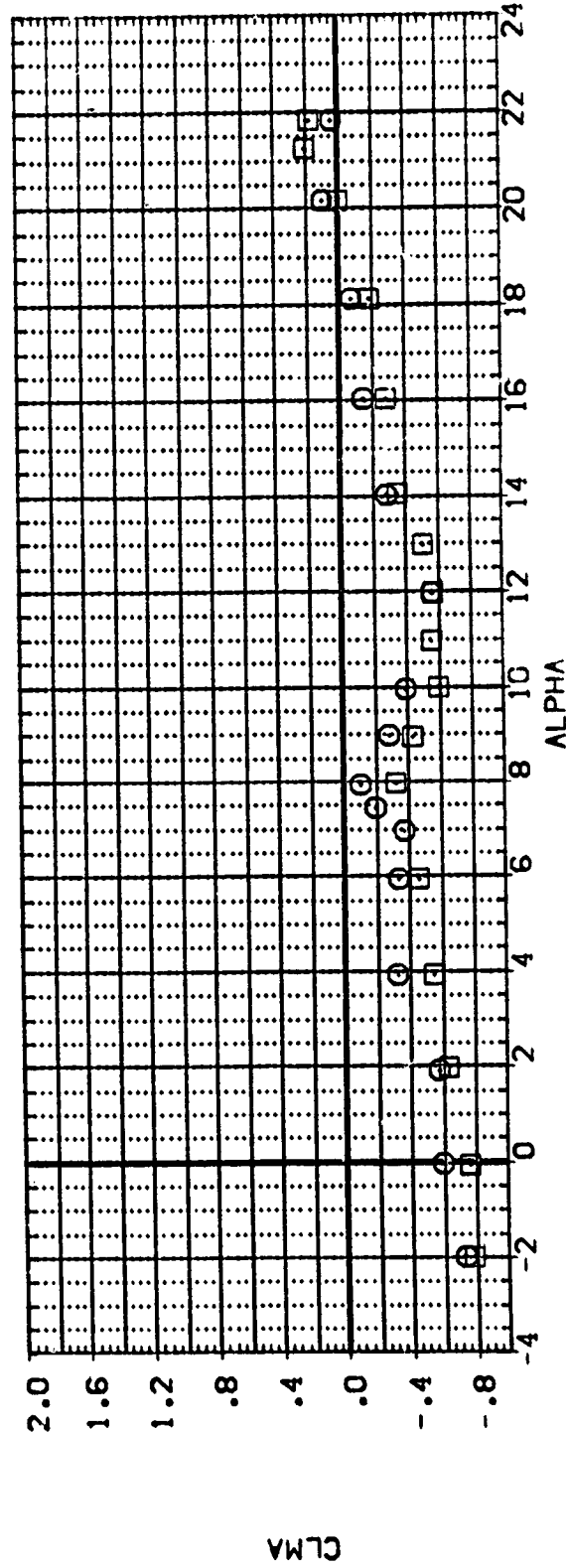
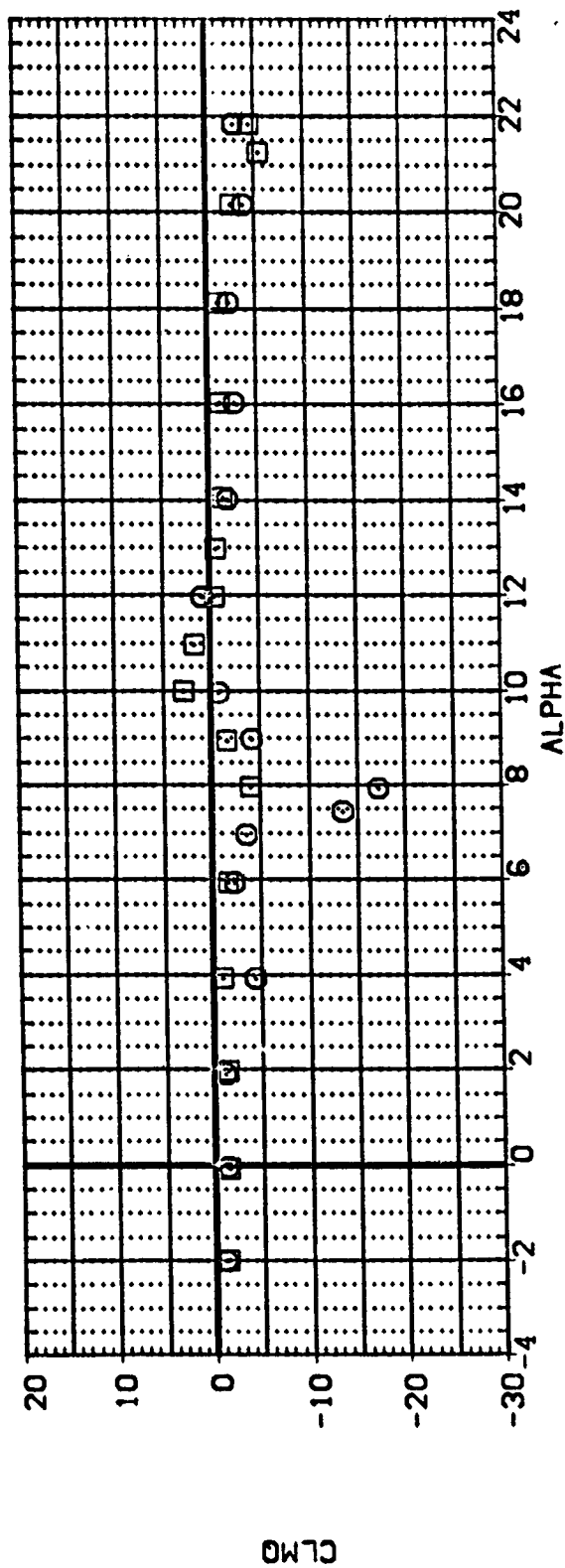


FIGURE 5. EFFECT OF RMS PODS ON DYNAMIC STABILITY PARAMETERS IN PITCH

CG-LOC ELEVTR BOFLAP RUOFLR
1.000 .000 .000 10.000
1.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
[RPKY02] LA-20. ROCKWELL ORB 0898 V/MOD. NOSE (BWW F)
[RPKY04] LA-20. ROCKWELL ORB 0898 V/MOD. NOSE (BWW F)

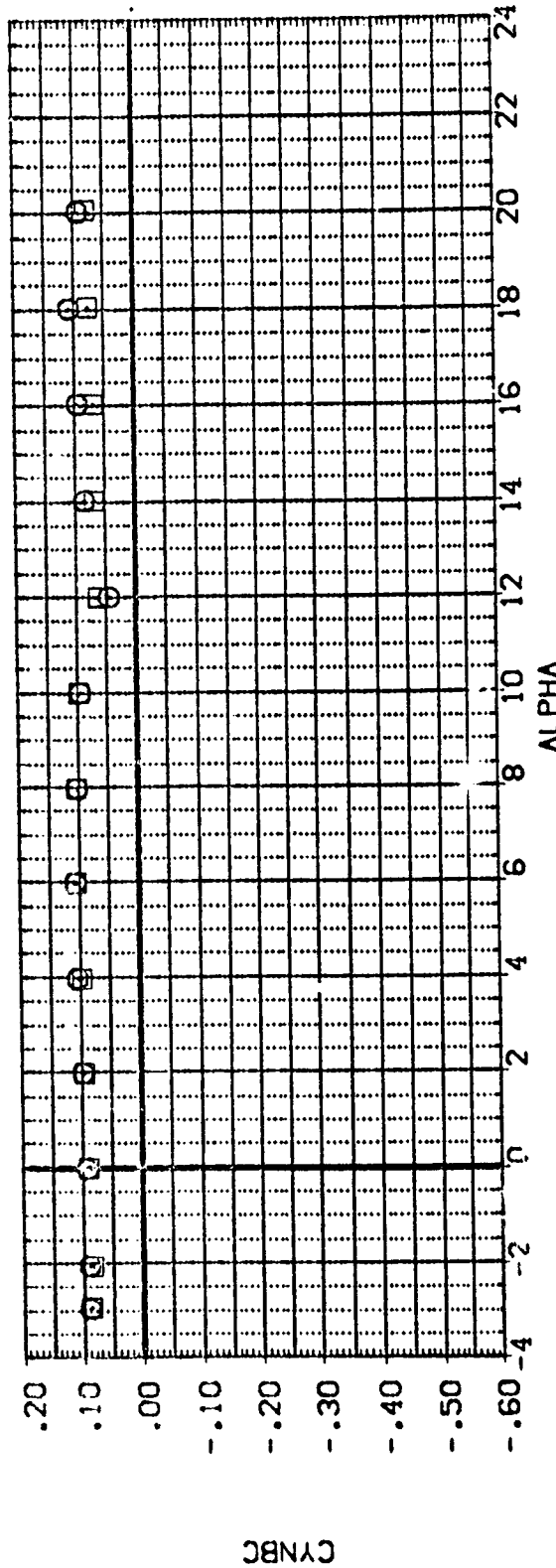
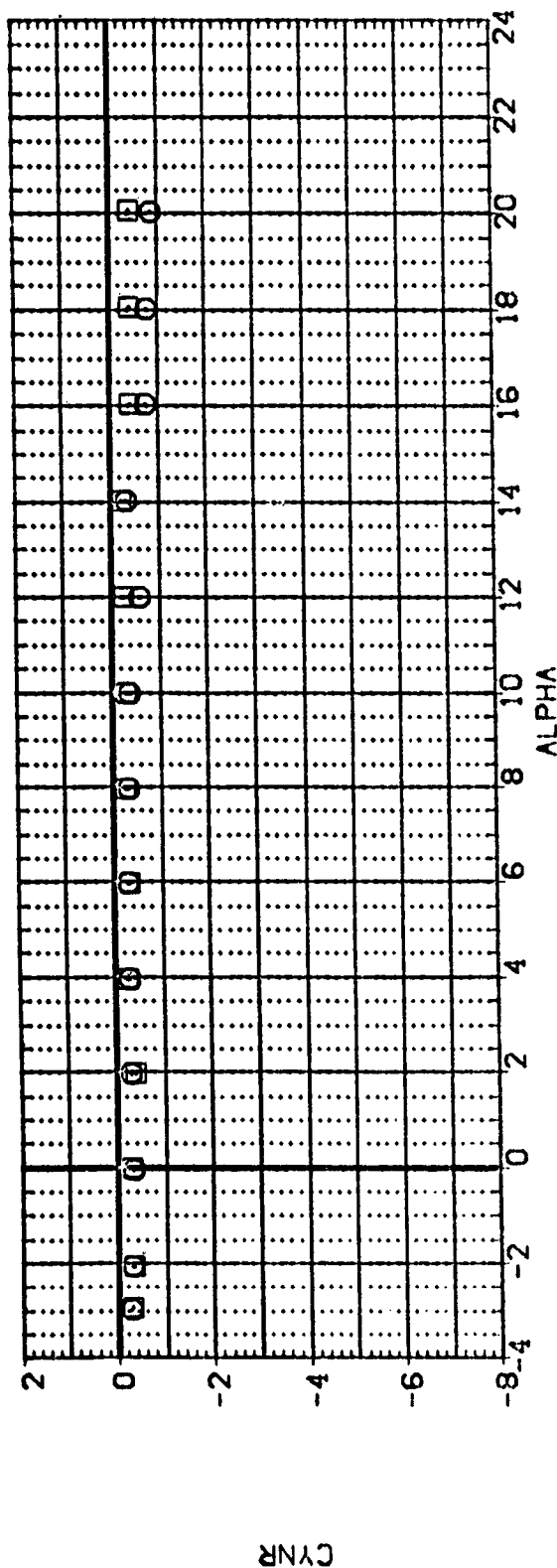


FIGURE 6. EFFECT OF OMS POOS ON DYNAMIC STABILITY PARAMETERS IN YAW

(A)MACH = .30

CG-LOC ELEVTR BOFLAP RUOFLR
1.000 .000 .000 10.000
1.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
LA-20: ROCKWELL ORB 0898 V/HOO. NOSE (BNV F)
LA-20: ROCKWELL ORB 0898 V/HOO. NOSE (BNVNF)

[RPMYC2]
[RPMYC4]

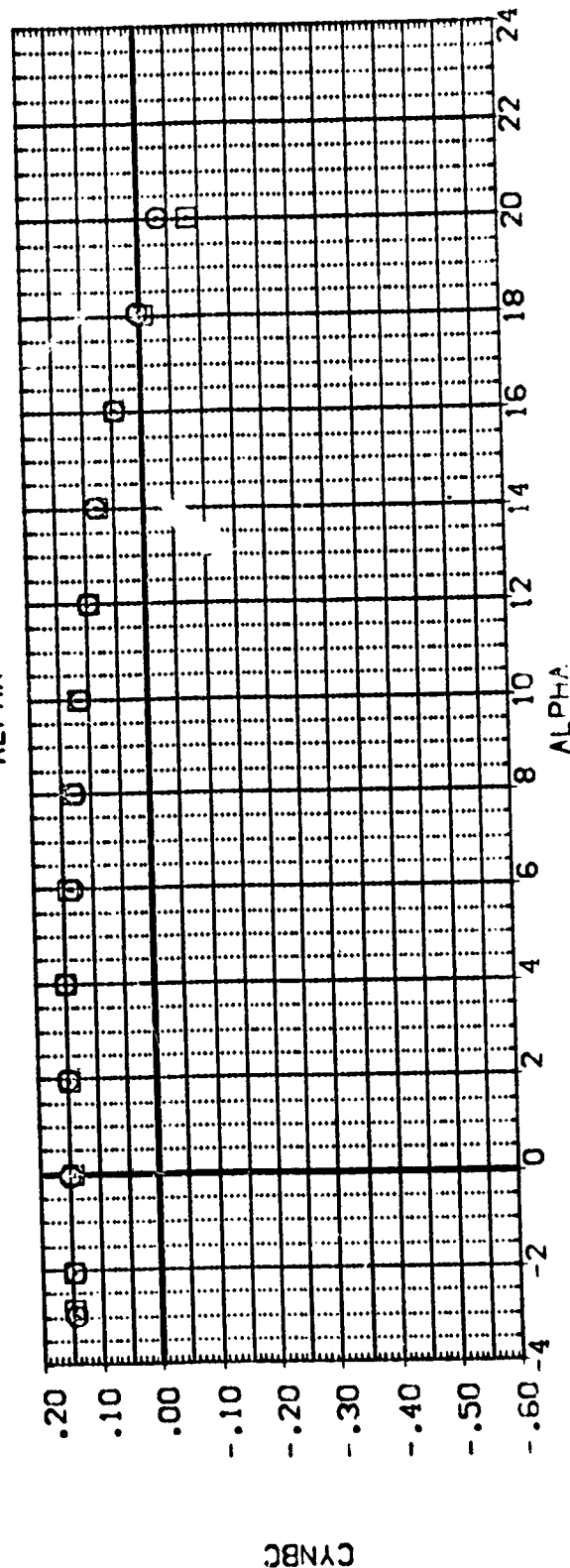
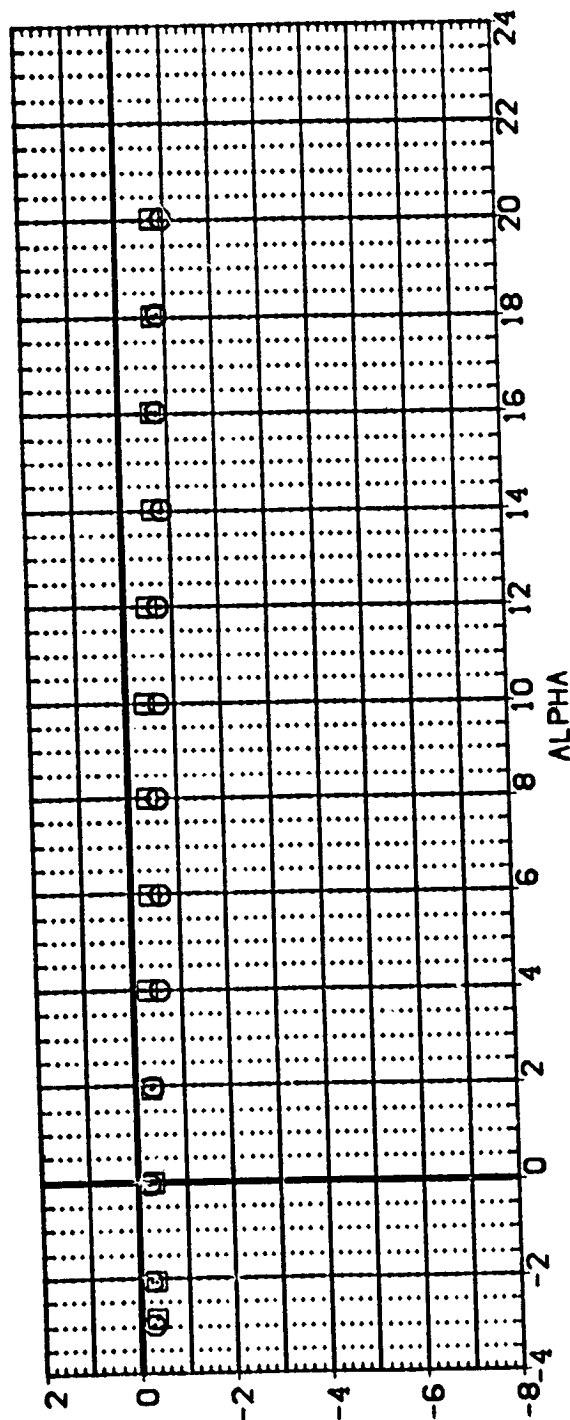


FIGURE 6. EFFECT OF RMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	CG-LOC	ELEVTR	BOFLAP	RUOFLR
[RPKY02]	LA-20, ROCKWELL ORB 089B V/MOD, NOSE (BNAV F)	1.000	.000	.000	10.000
[RPKY04]	LA-20, ROCKWELL ORB 089B V/MOD, NOSE (BNAV F)	1.000	.000	.000	10.000

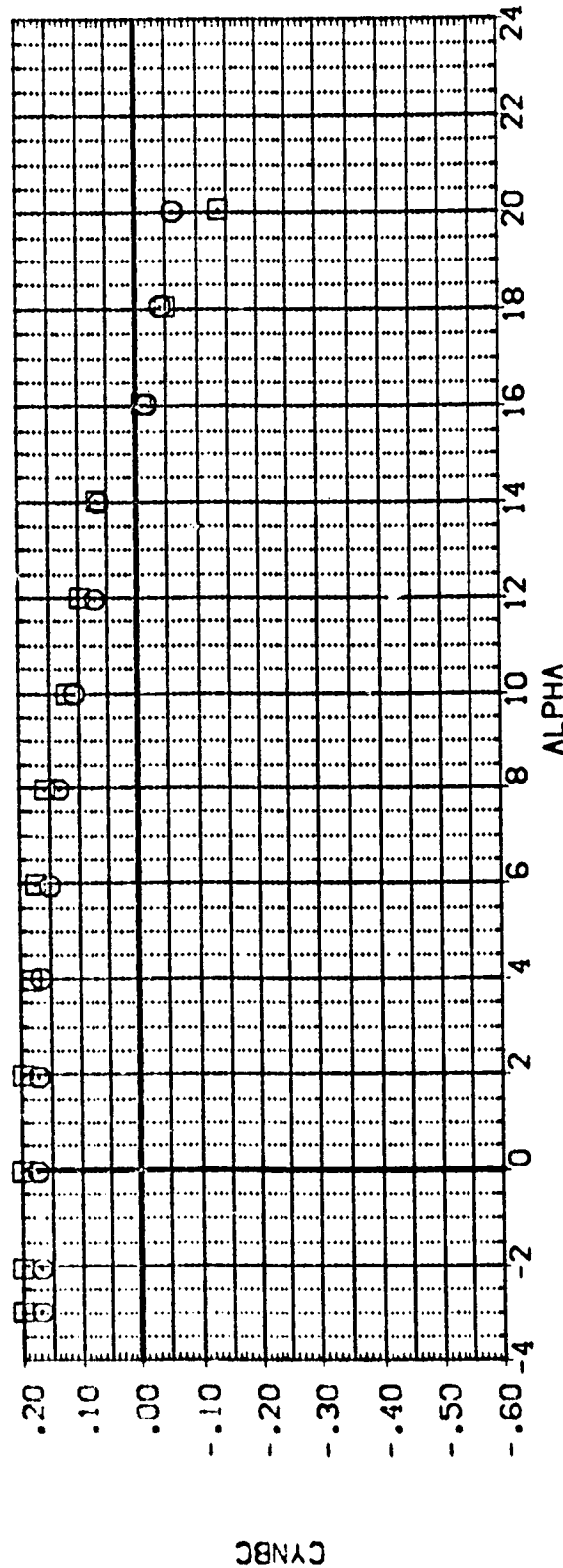
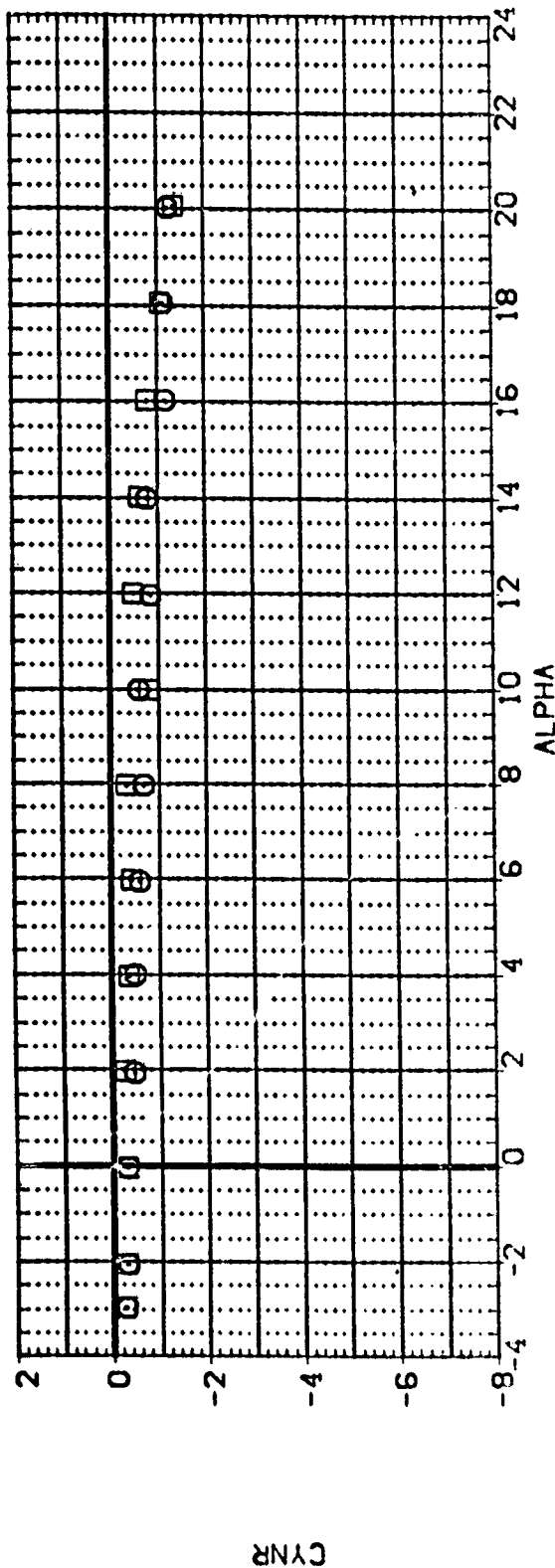


FIGURE 6. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW
 (C)MACH = .90

CG-LOC ELEVTR BOFLAP RUOFLR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [RPNV03] A LA-20, ROCKWELL CRB 0898 V/MOD, NOSE (BNV F)
 [RPNV04] LA-20, ROCKWELL CRB 0898 V/MOD, NOSE (BNV F)

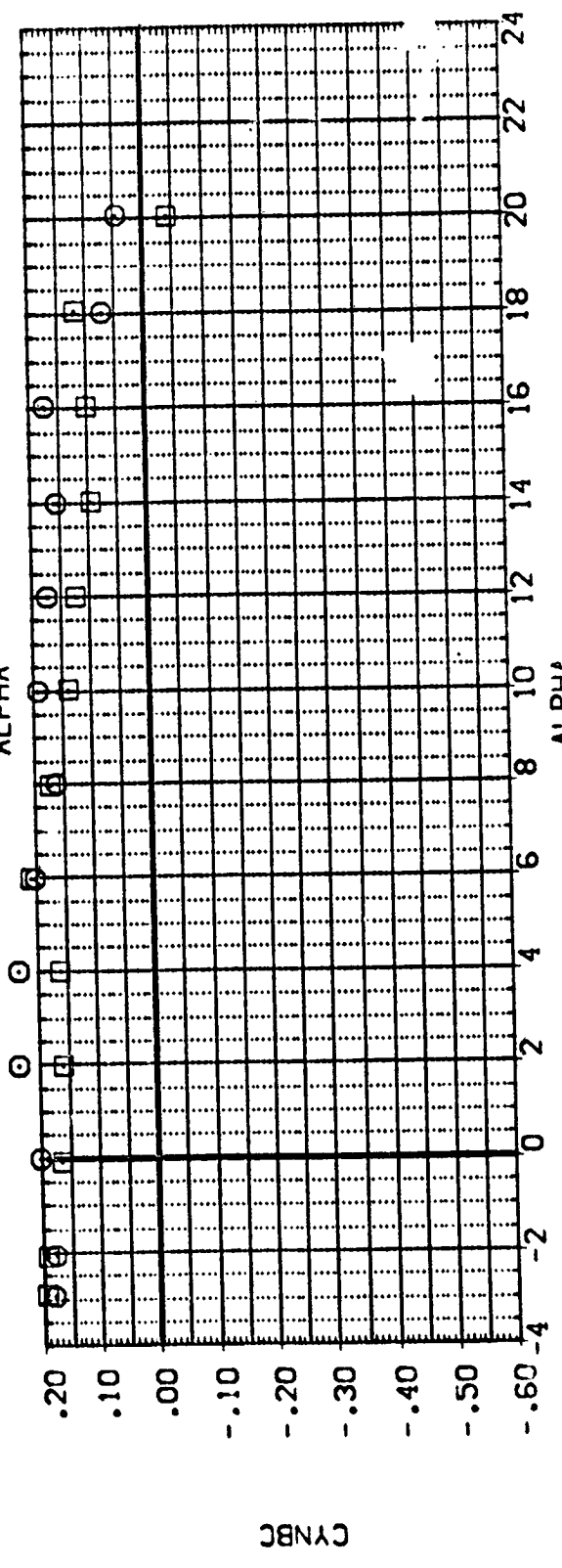
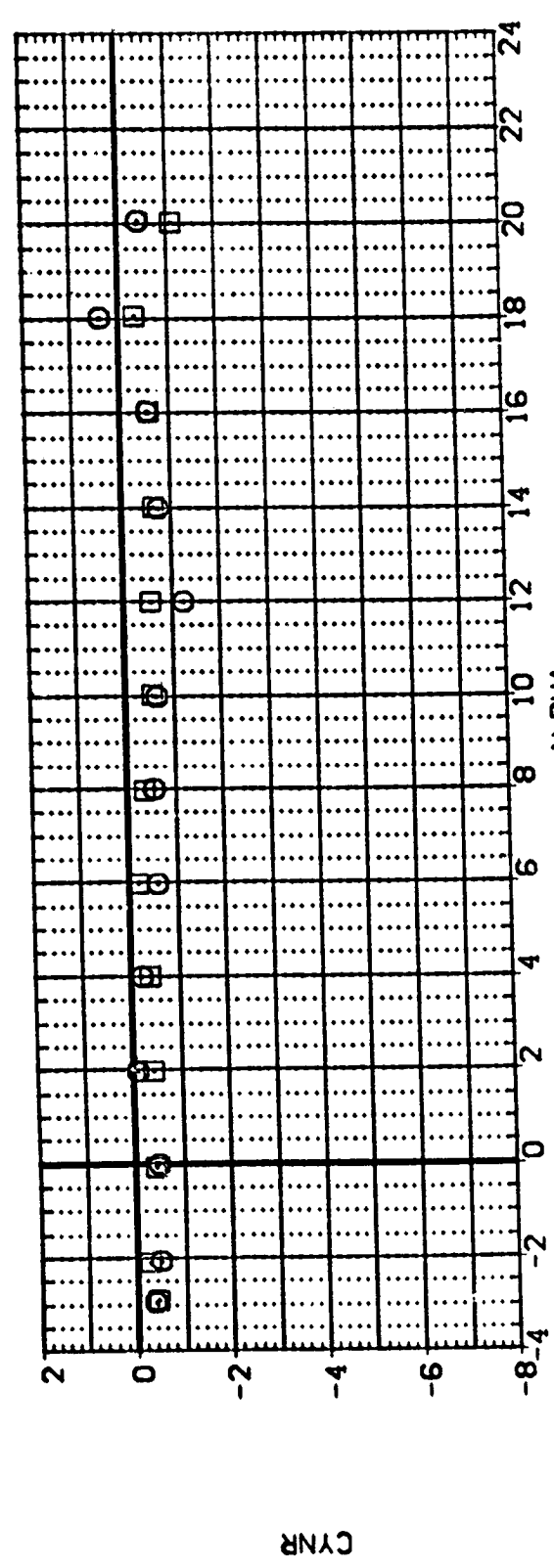


FIGURE 6. EFFECT OF RMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPNY02) LA-20: ROCKWELL ORB 0858 V/MOD. NOSE (BVM F) 1.000 .000 .000 10.000
 (RPNY04) LA-20: ROCKWELL ORB 0858 V/MOD. NOSE (BVM F) 1.000 .000 .000 10.000

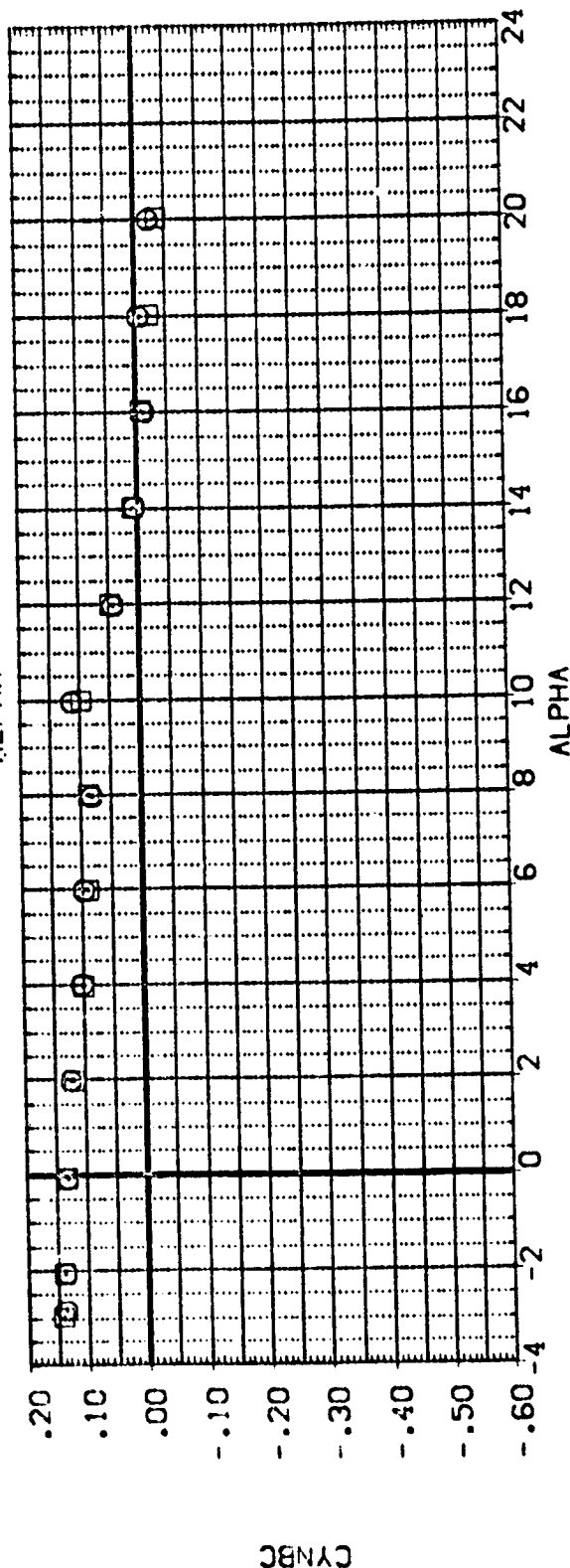
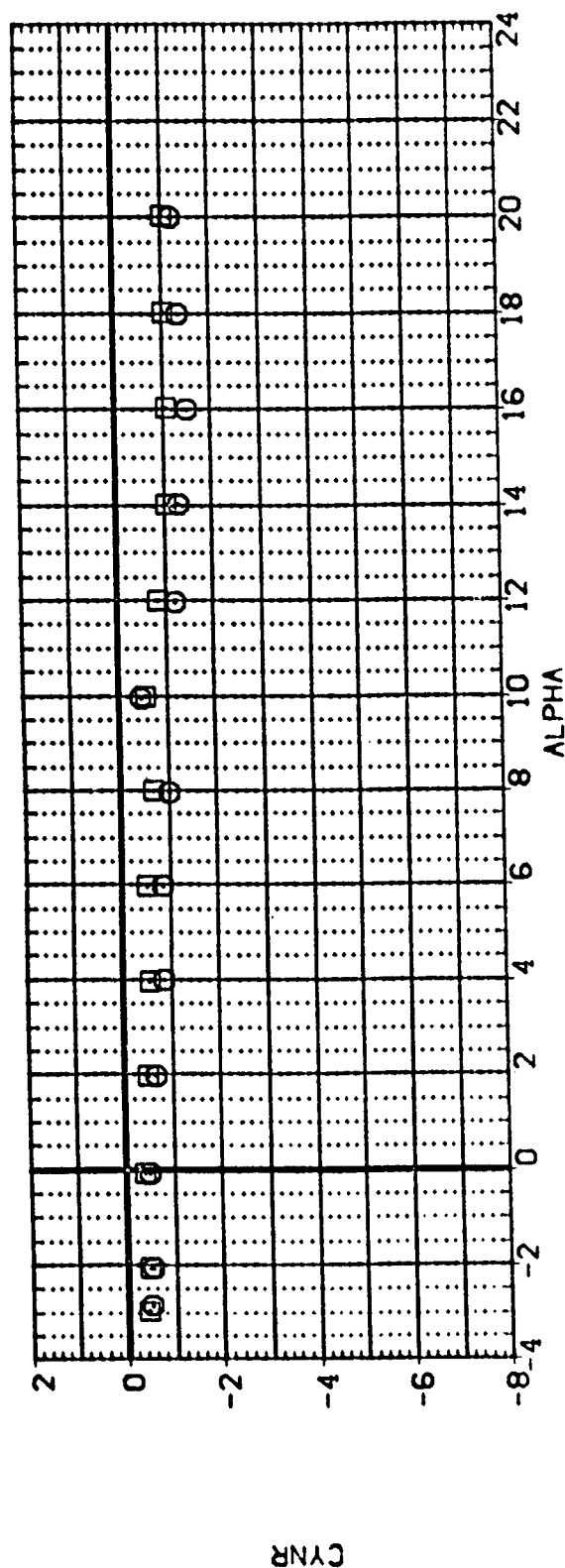


FIGURE 6. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW

(E)MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPNY02) LA-20, ROCKWELL DRB 0898 V/MOD, NOSE (BVM F) 1.000 .000 .000 10.000
 (RPNY03) LA-20, ROCKWELL DRB 0898 V/MOD, NOSE (BVM F) 1.000 .000 .000 10.000

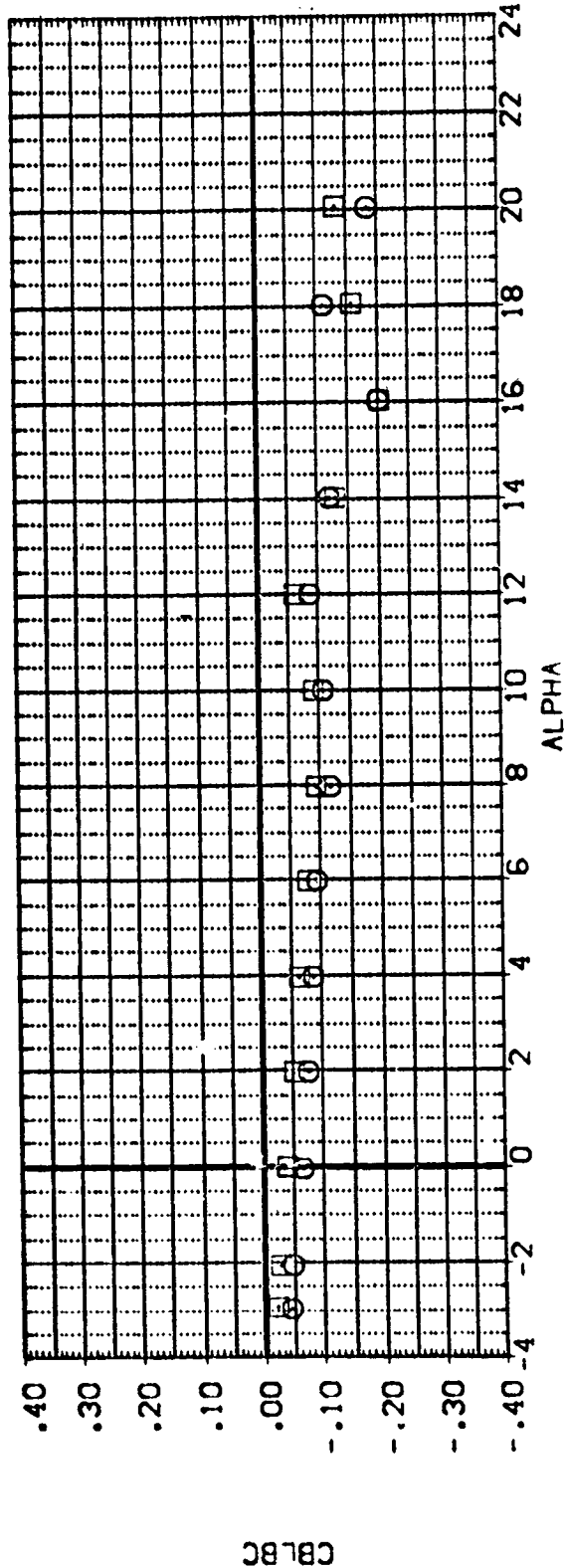
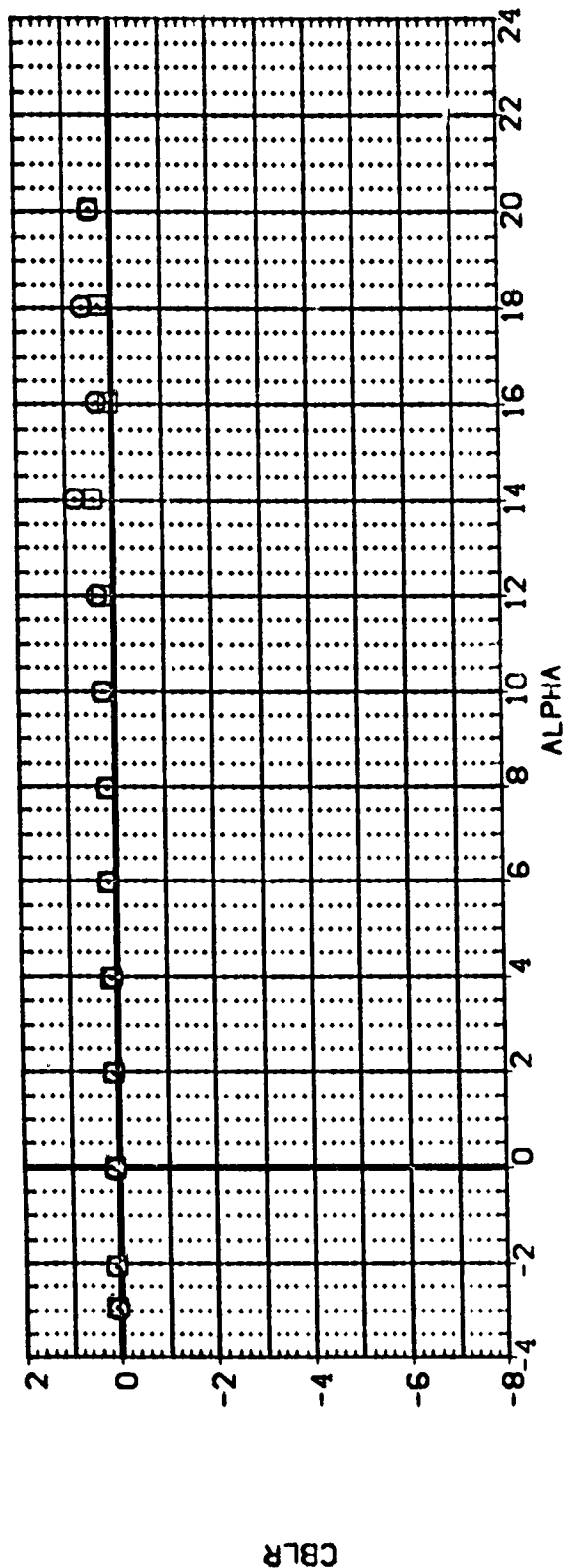


FIGURE 6. EFFECT OF OMS POOS ON DYNAMIC STABILITY PARAMETERS IN YAW

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPKY02) LA-20: ROCKWELL OR8 0898 V/MOD. NOSE (BVM F) 1.000 .000 .000 10.000
 (RPKY04) LA-20: ROCKWELL OR8 0898 V/MOD. NOSE (BVM F) 1.000 .000 .000 10.000

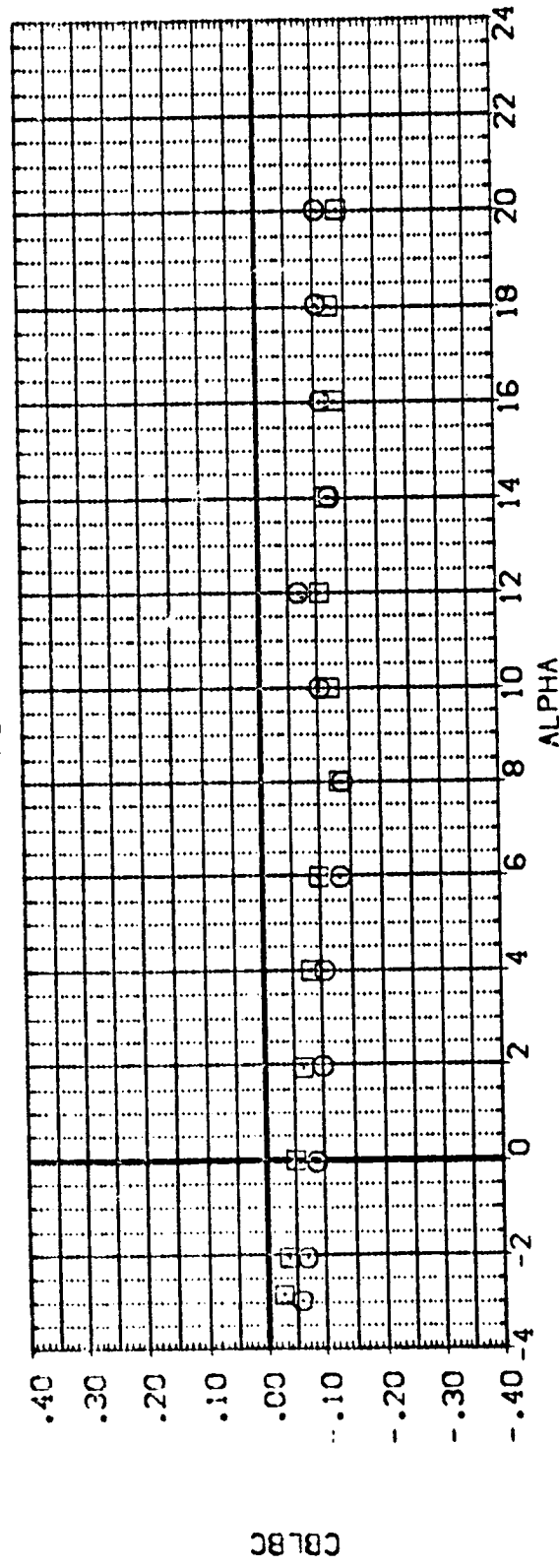
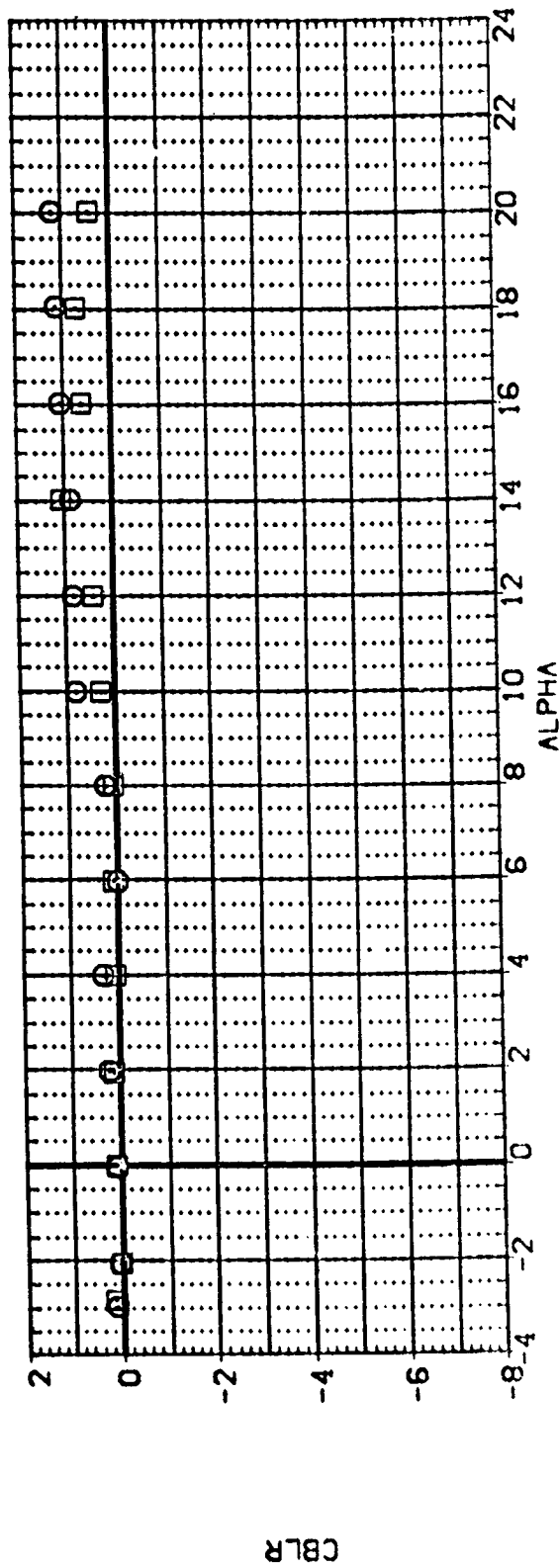


FIGURE 6. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW

(B) MACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPNV02) LA-20; ROCKWELL DRB 0898 V/MOD; NOSE (BNV F)
 (RPNV04) LA-20; ROCKWELL DRB 0898 V/MOD; NOSE (BNVHF)

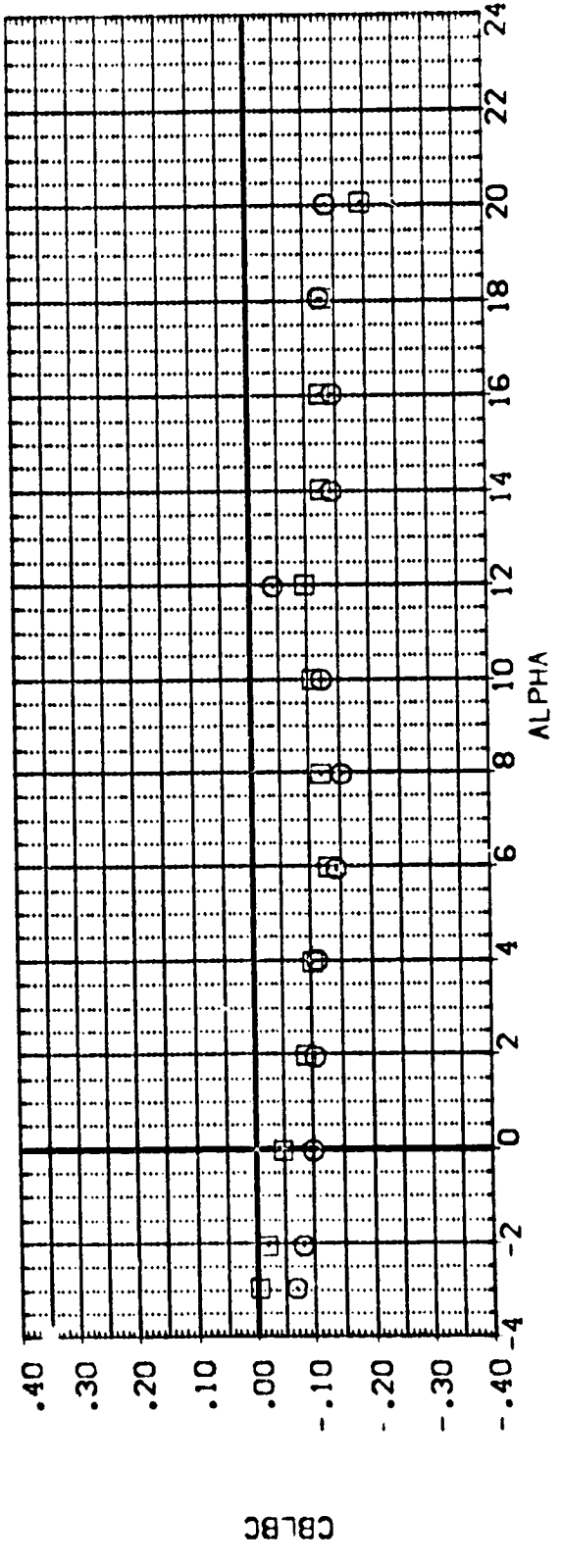
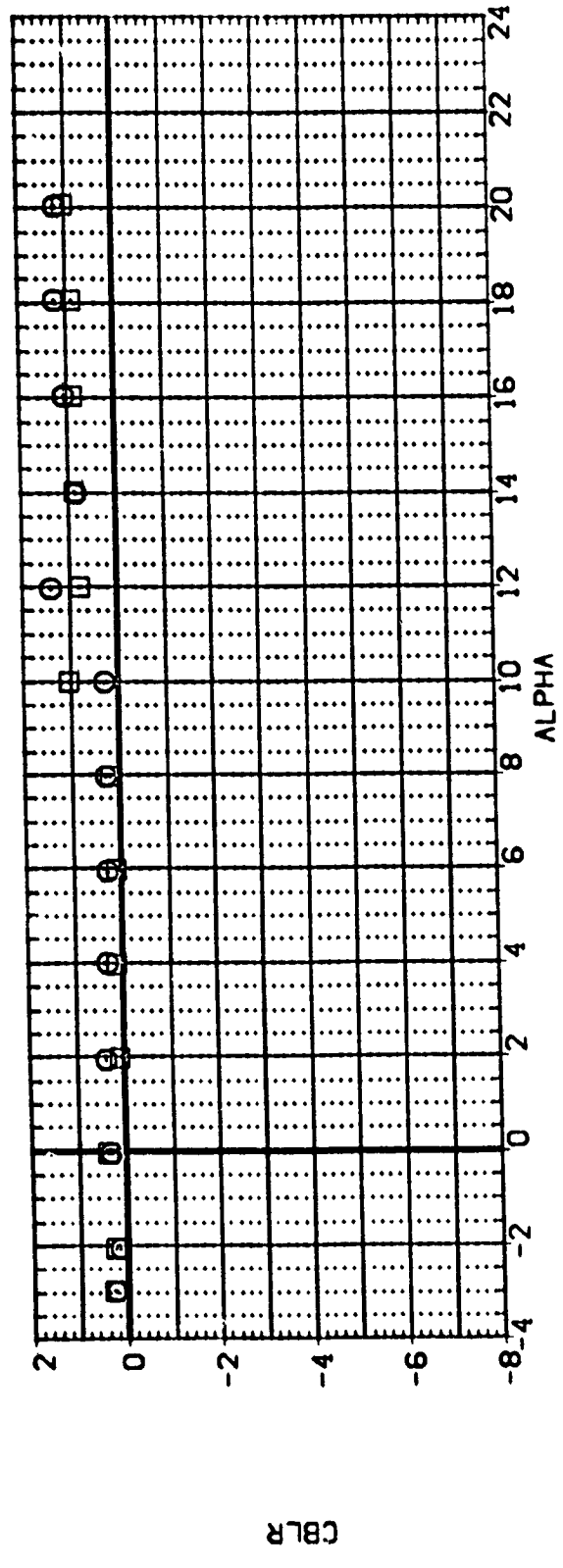


FIGURE 6. EFFECT OF QMS POOS ON DYNAMIC STABILITY PARAMETERS IN YAW

REVISION 1 - .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

[RPKY02] LA-20. ROCKWELL ORB 0898 V/MOD. NOSE (BNV F)

1.000 .000 .000 10.000

[RPKY04] LA-20. ROCKWELL ORB 0898 V/MOD. NOSE (BNV F)

1.000 .000 .000 10.000

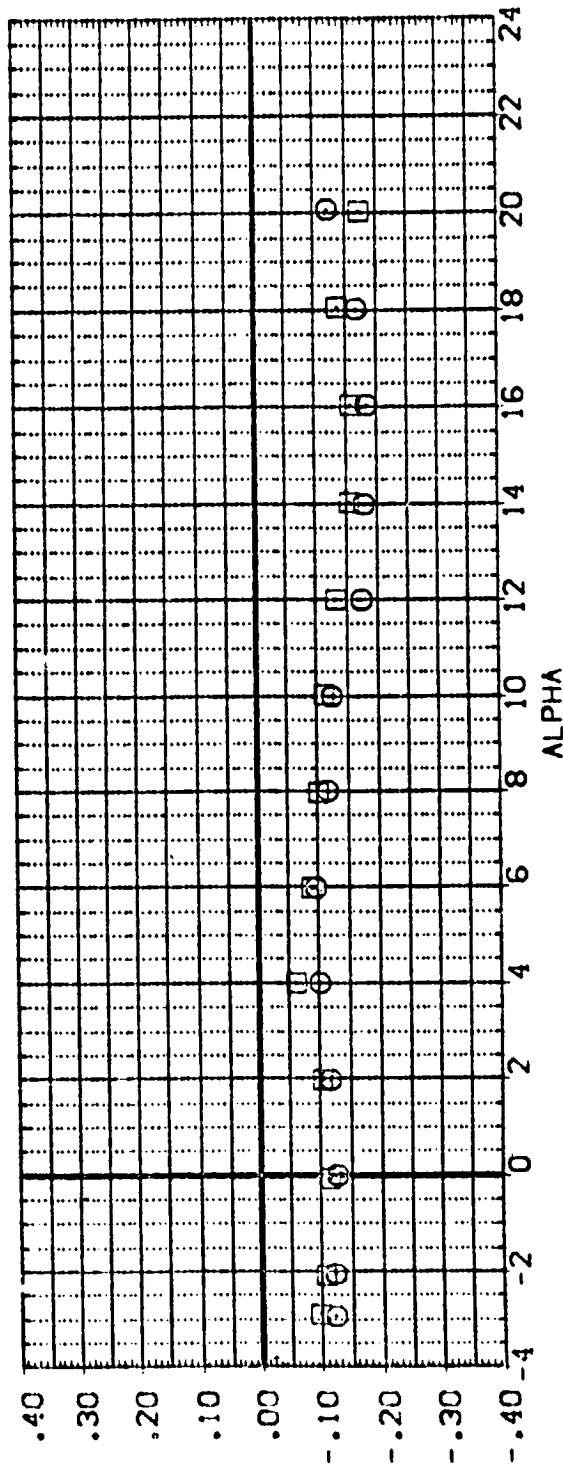
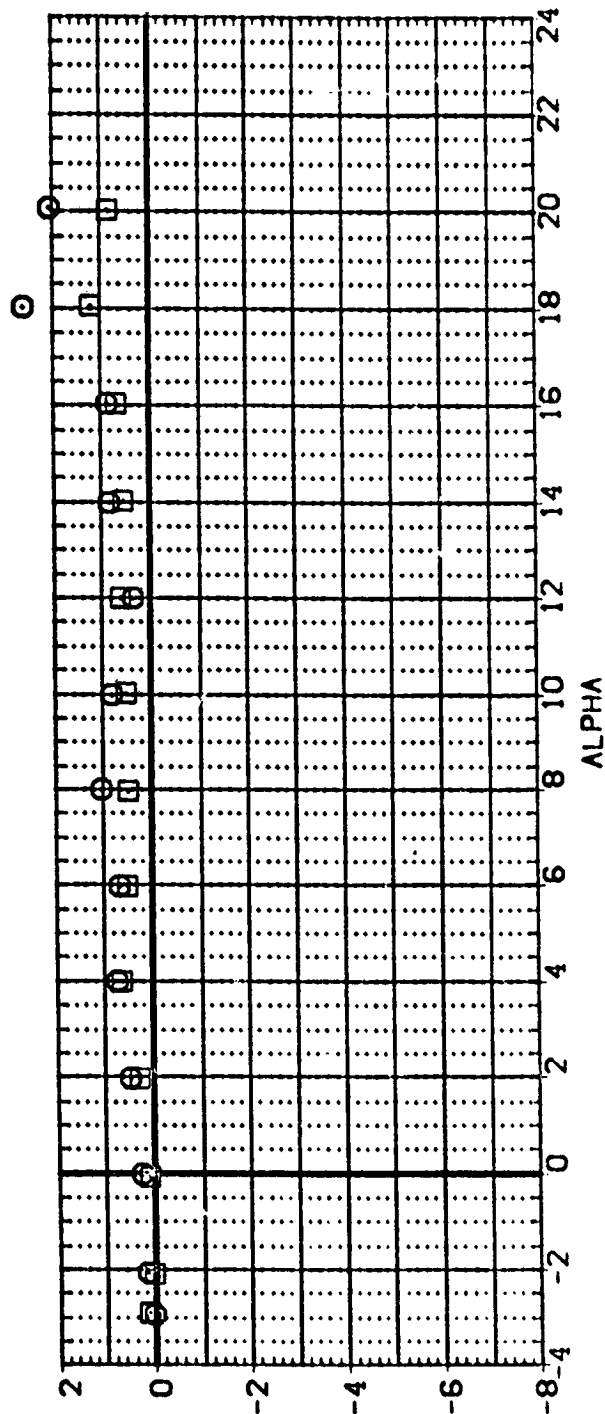


FIGURE 6. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW

(O)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPKY02) LA-20, ROCKWELL CRB 0858 V/MOD, N°SE (BHV F)
 (RPKY01) LA-20, ROCKWELL CRB 0858 V/MOD, NL (BHVNF)

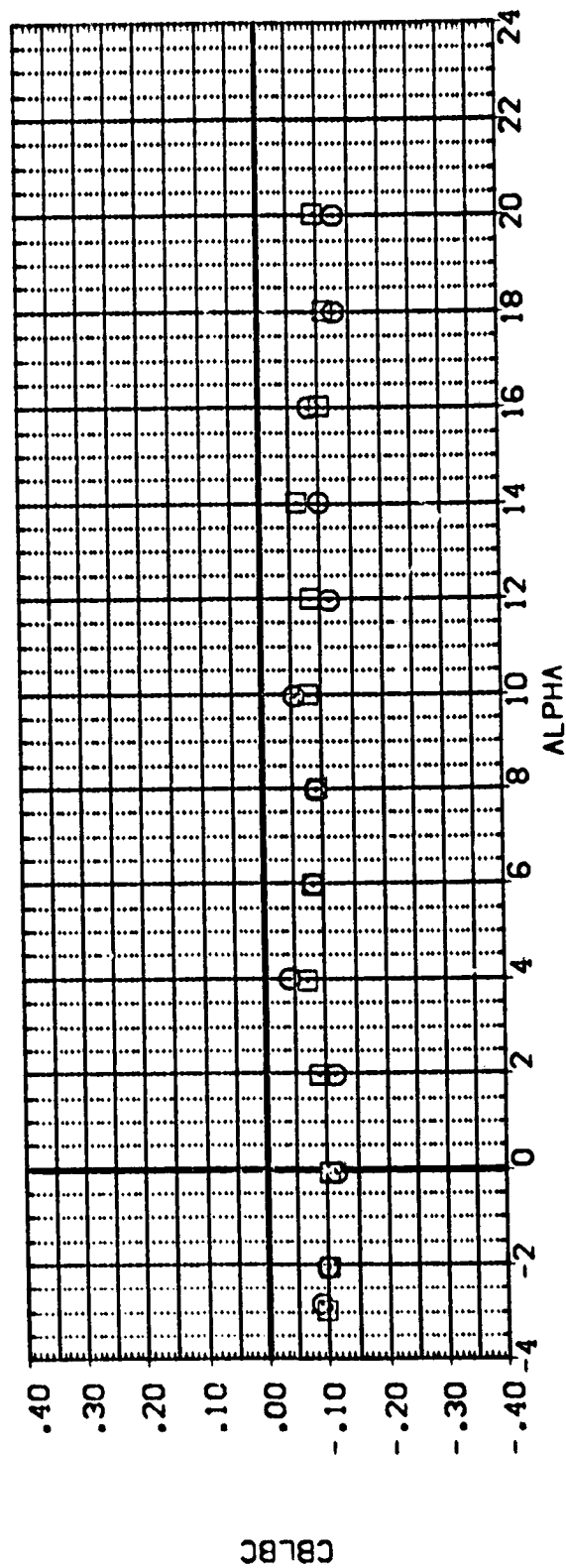
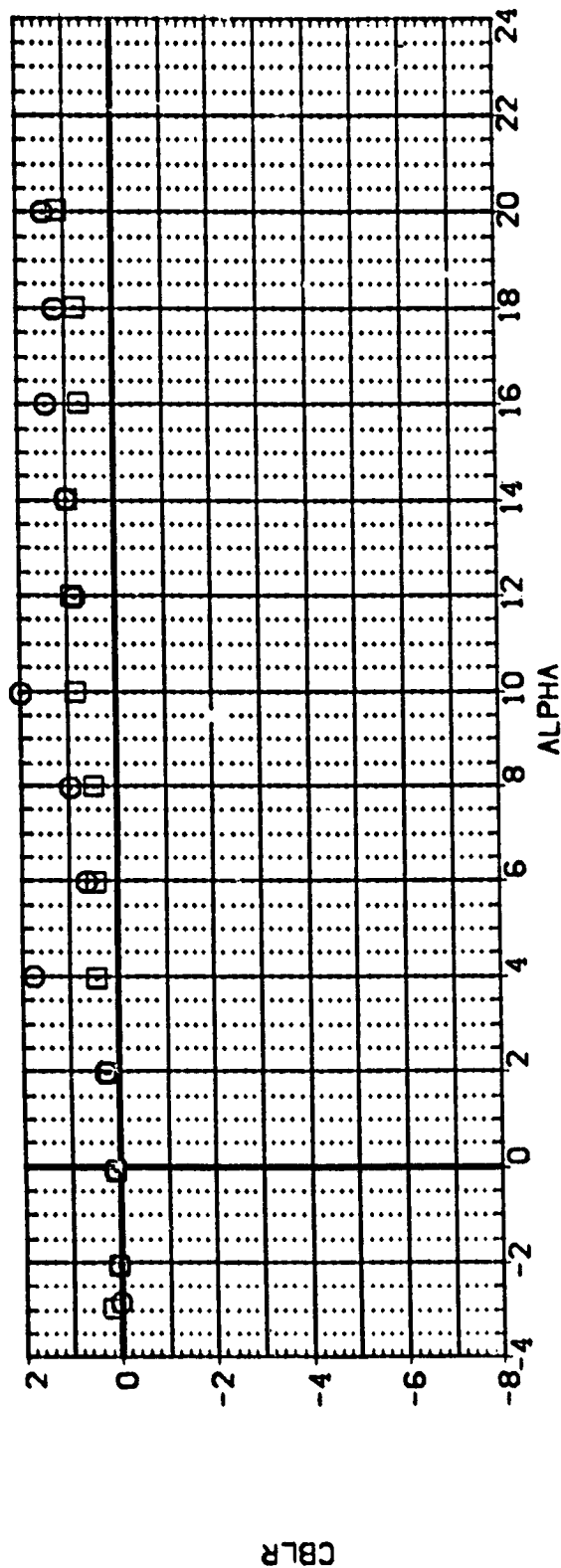


FIGURE 6. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN YAW

CF MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 [RPKRO2] LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVF F)
 [RPKRO4] LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMF)

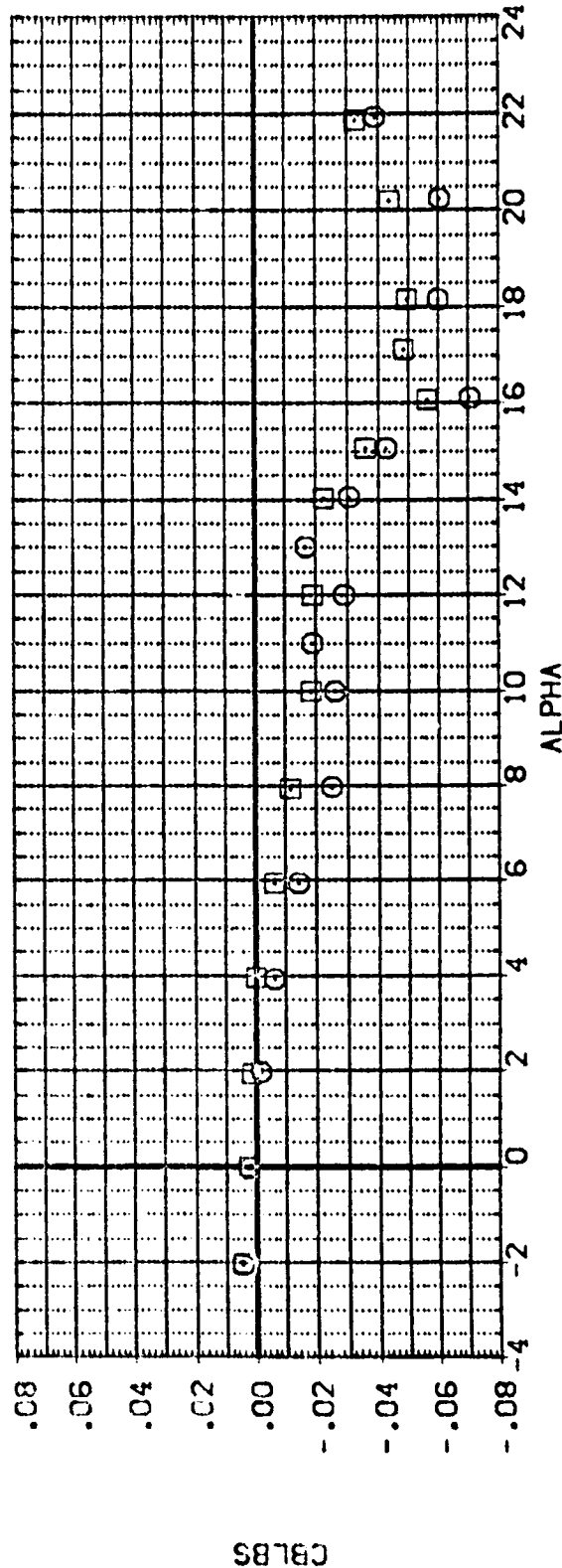
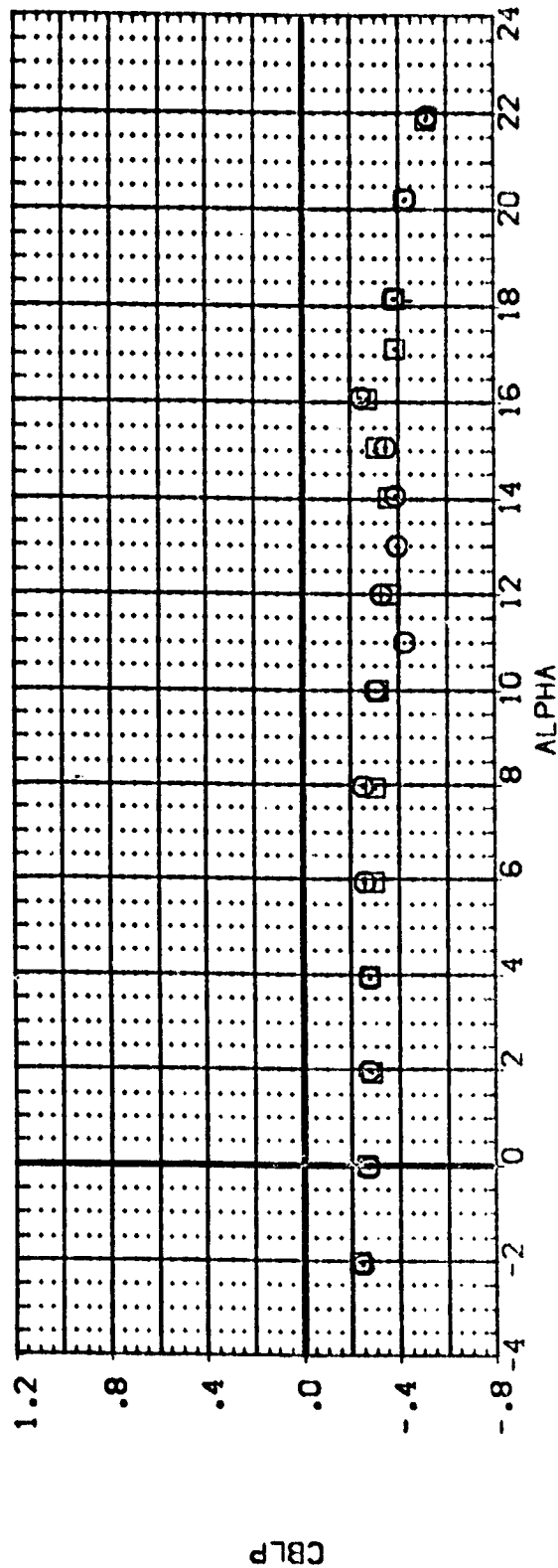


FIGURE 7. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL

[A]MACH = .30

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 [RPMR2] LA-20. ROCKWELL 0898 OR8 V/MOD NOSE (BVF F) 1.000 .000 10.000
 [RPMR2] LA-20. ROCKWELL 0898 OR8 V/MOD NOSE (BVF F) 1.000 .000 10.000

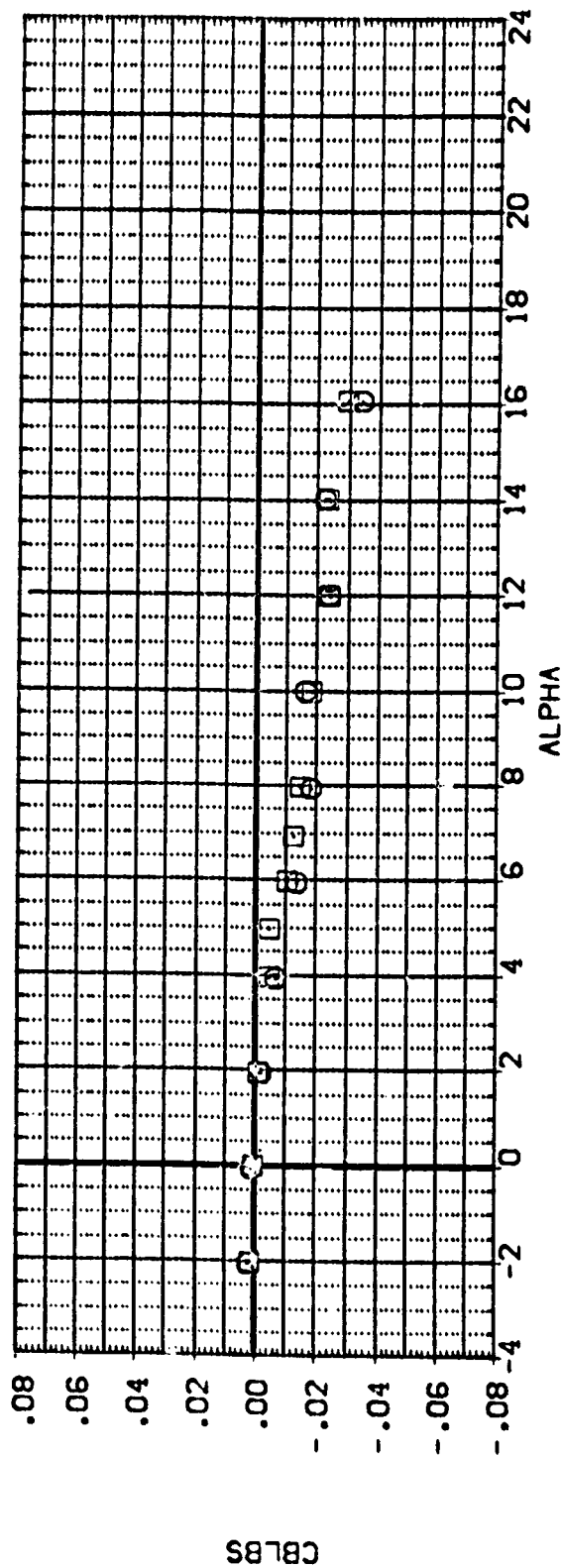
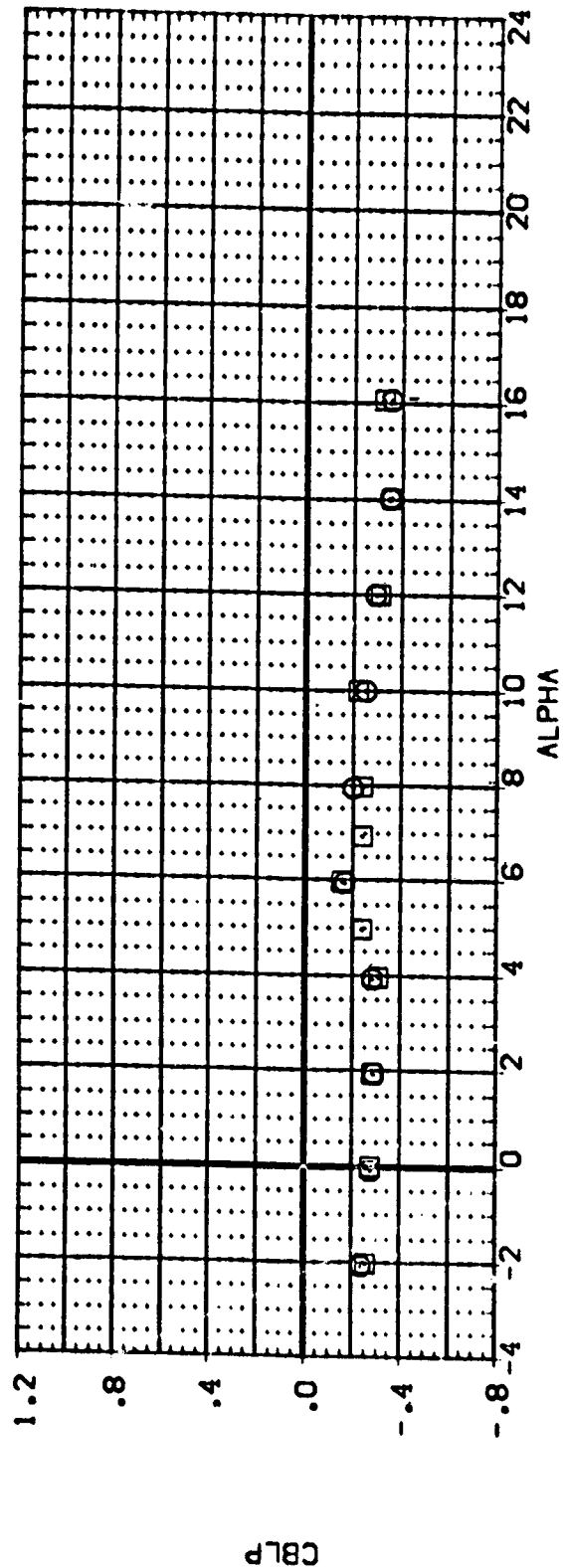


FIGURE 7. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL

(B) MACH = .80

CG-LOC ELEVTR BOFLAP RUOFLR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

DATA SET SYMB. CONFIGURATION DESCRIPTION
 (RPAK02) LA-20: ROCKWELL O898 ORB V/MOD NOSE (BNV F)
 (RPAK04) LA-20: ROCKWELL O898 ORB V/MOD NOSE (BNV F)

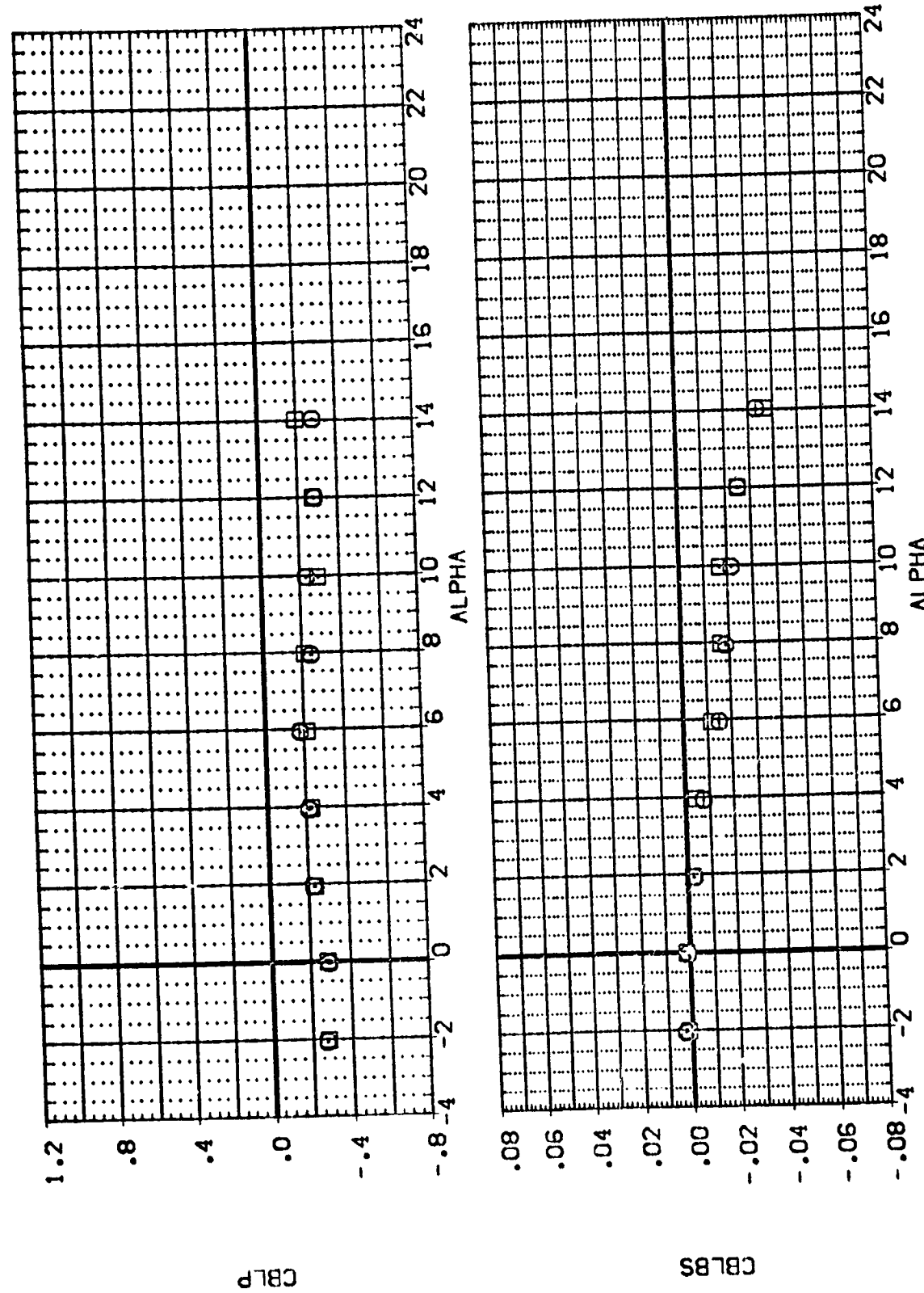


FIGURE 7. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL

COMACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 [RPRC02] LA-20, ROCKWELL 085B ORB V/MOD NOSE (BVV F) 1.000 .000 .000 10.000
 [RPRC04] LA-20, ROCKWELL 085B ORB V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

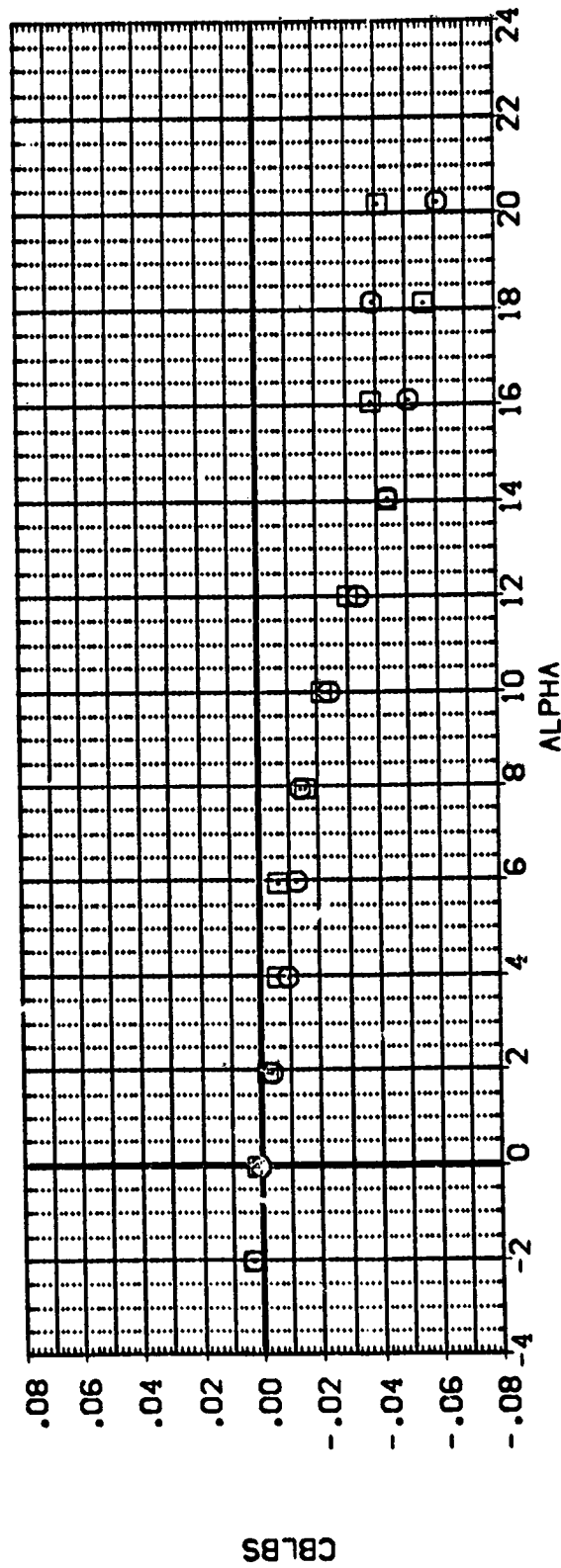
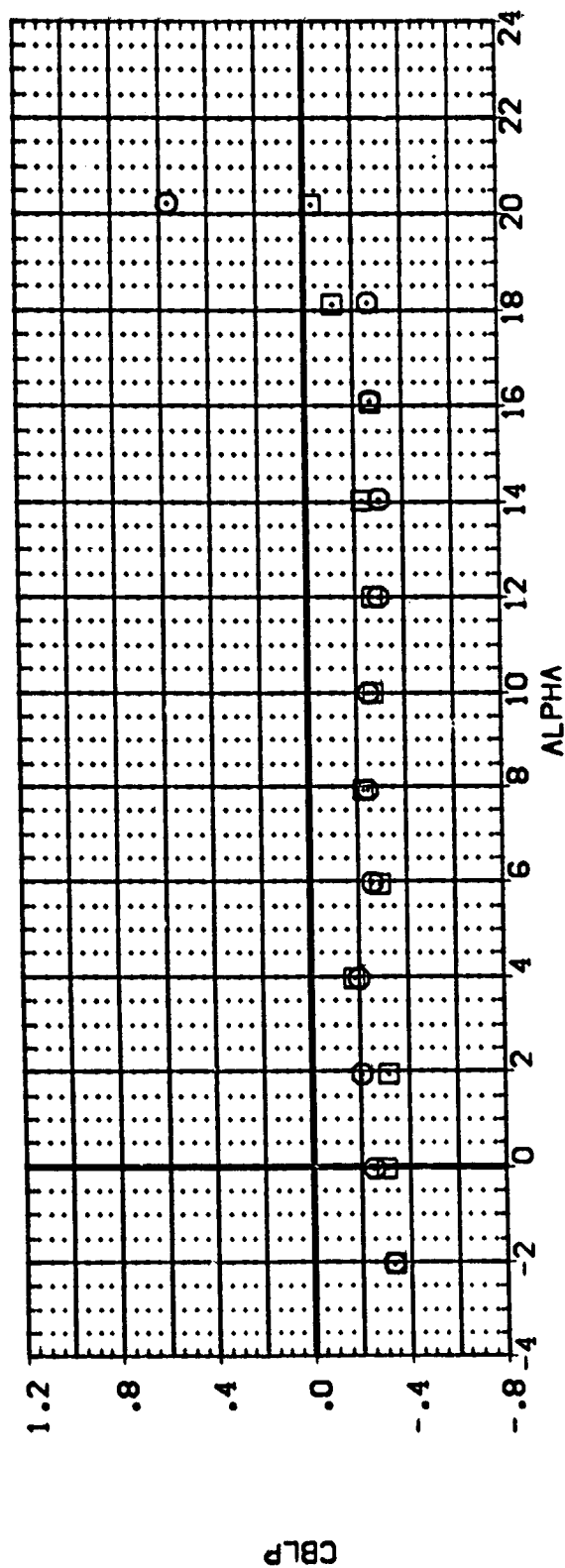


FIGURE 7. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL

(O)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPMR02) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BNV F) 1.000 .000 .000 10.000
 (RPMR04) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BNV F) 1.000 .000 .000 10.000

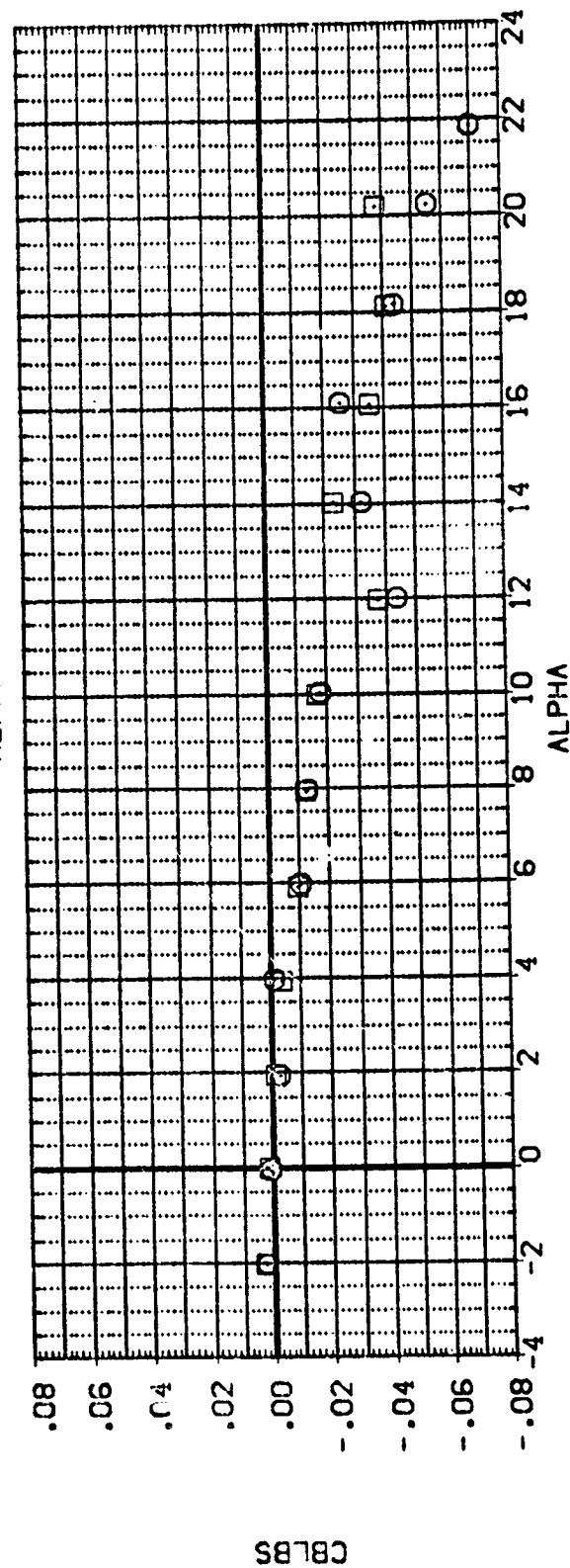
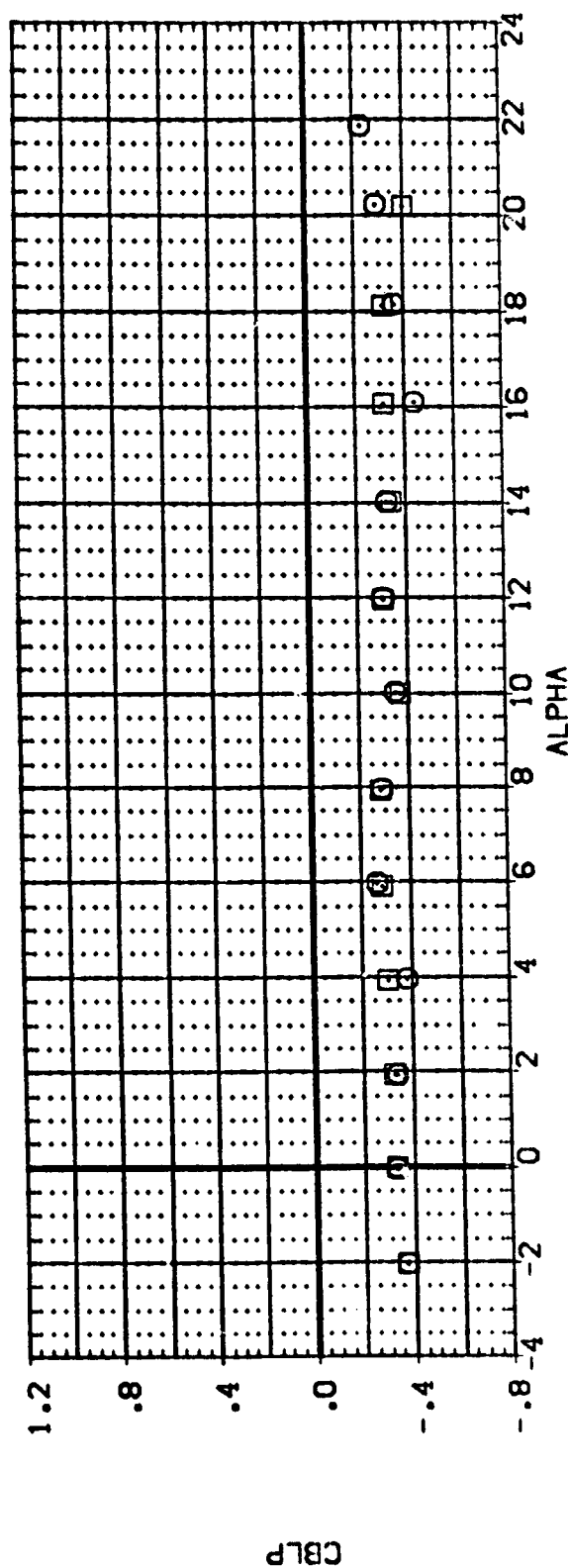


FIGURE 7. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL

(E)MACH = 1.20

DATA SET SYMBOL: [RPKR02] [RPKR04] CONFIGURATION DESCRIPTION: LA-20, ROCKWELL O898 ORB V/MOD NOSE (BVM F) LA-20, ROCKWELL O898 ORB V/MOD NOSE (BVM F) CG-LOC: 1.000 1.000 ELEVTR: .000 .000 ROFLAP: .000 .000 RUOFLR: 10.000 10.000

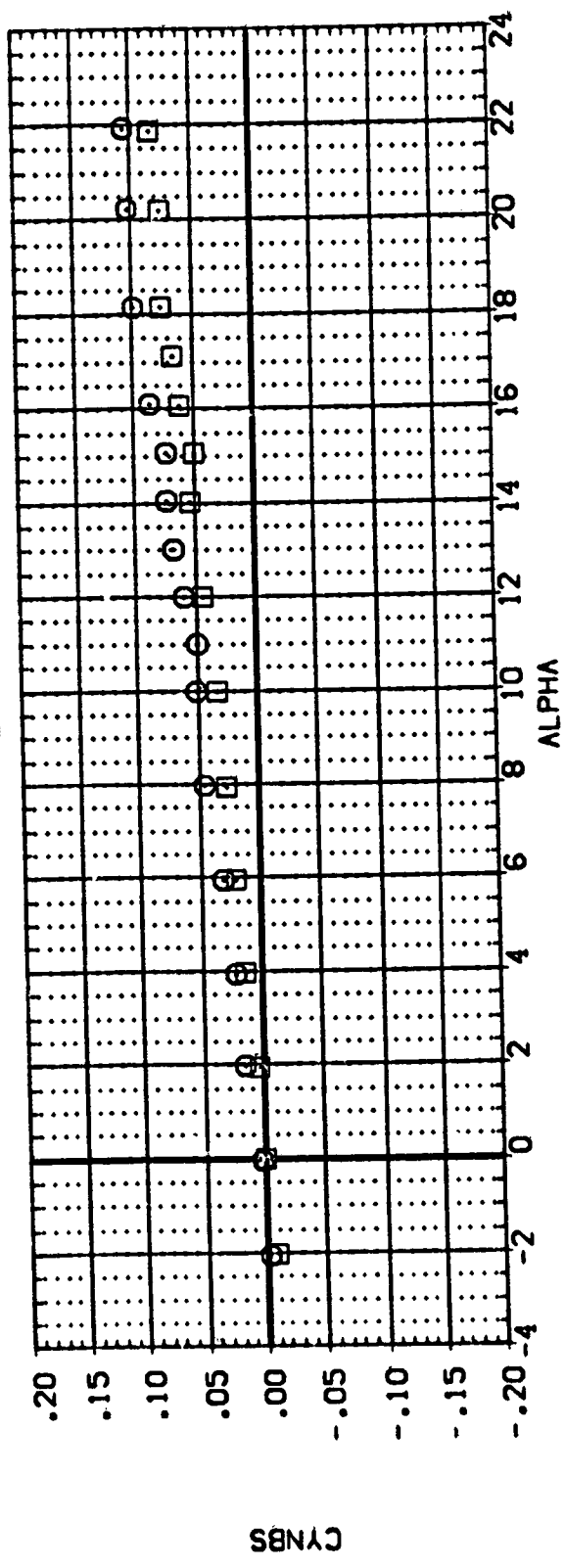
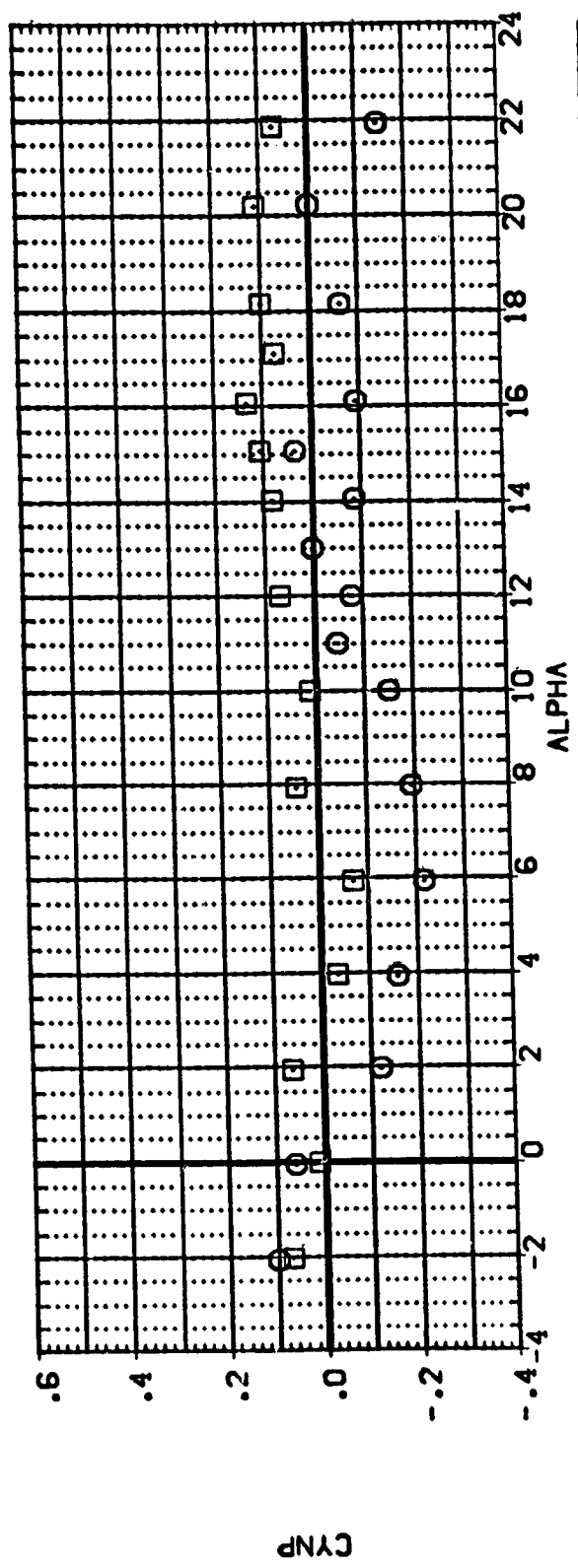


FIGURE 7. EFFECT OF OMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL.

[A] MACH = .30

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

[RPKNR02] LA-20, ROCKWELL 0888 ORB V/MOD NOSE (BNV F) 1.000 .000 .000 10.000

[RPKNR04] LA-20, ROCKWELL 0888 ORB V/MOD NOSE (BNV F) 1.000 .000 .000 10.000

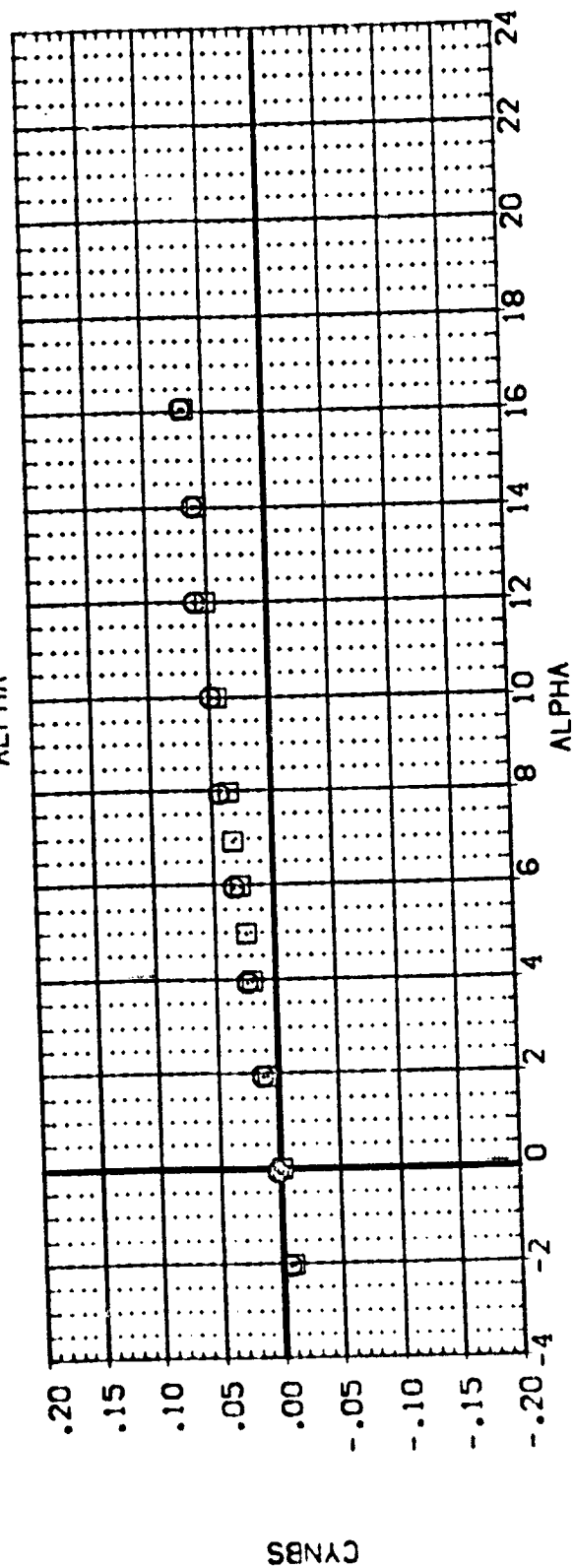
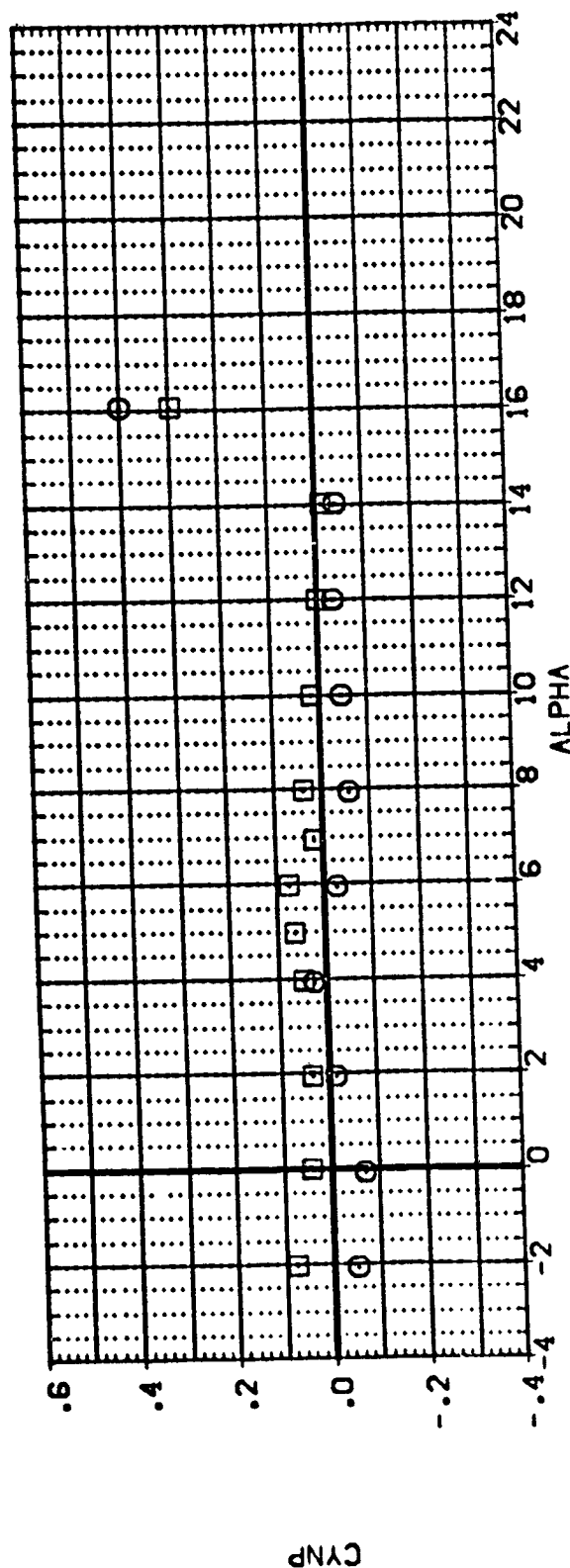


FIGURE 7. EFFECT OF GMS PODS ON DYNAMIC STABILITY PARAMETERS IN ROLL

(B)MACH = .80

DATA SET SYMBOL. CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR
 [RPM002] LA-20: ROCKWELL 0898 DRB V/MOD NOSE (BVM F) 1.000 .000 .000 10.000
 [RPM004] LA-20: ROCKWELL 0898 DRB V/MOD NOSE (BVM F) 1.000 .000 .000 10.000

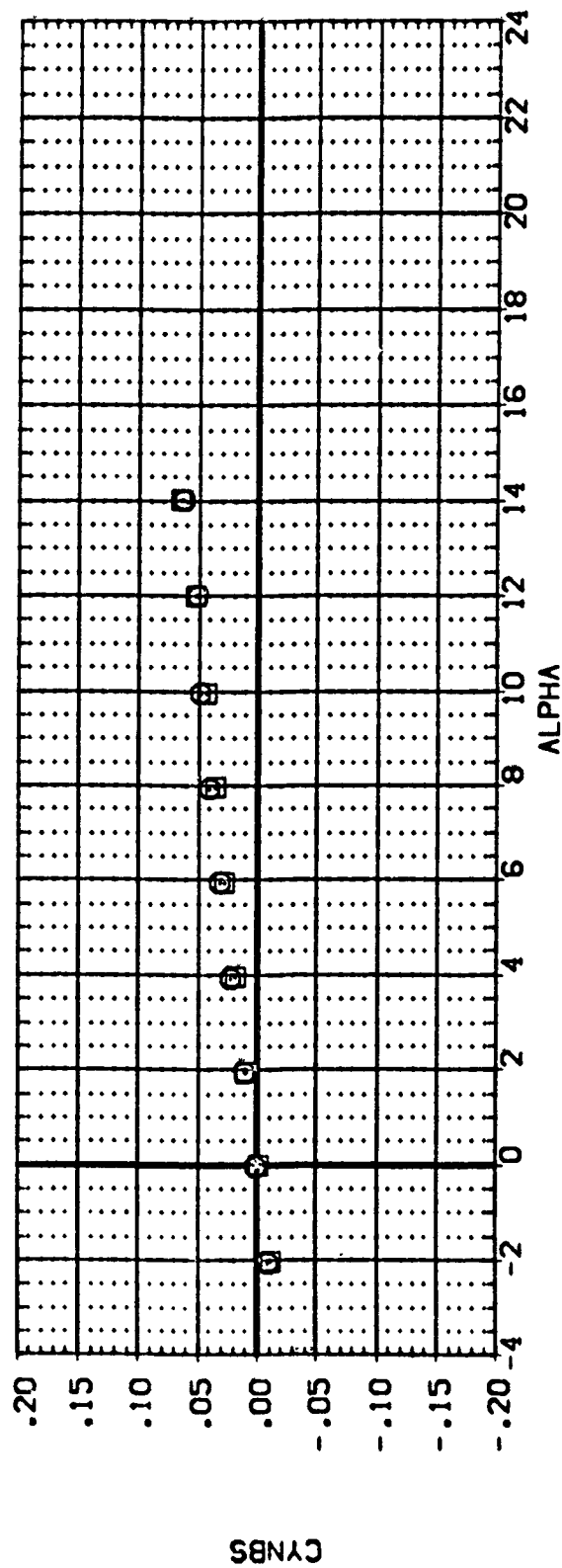
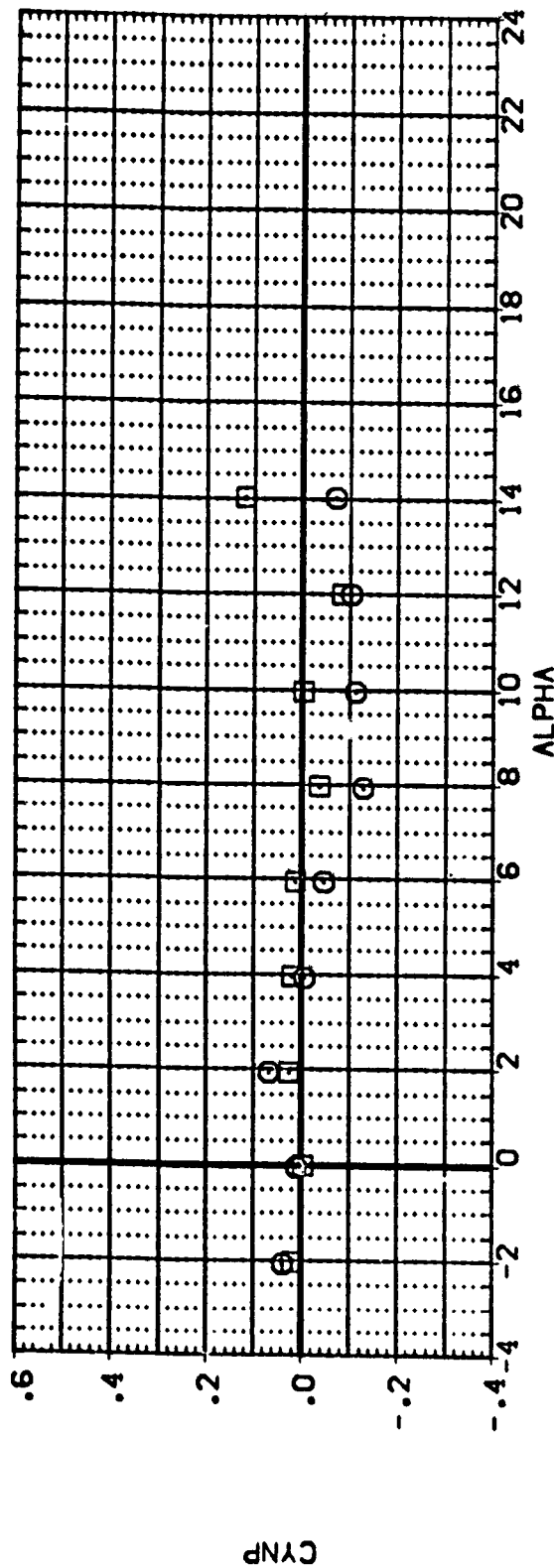


FIGURE 7. EFFECT OF OMS POOS ON DYNAMIC STABILITY PARAMETERS IN ROLL

(C)MACH = .90

DATA SET SYMBOL. CONFIGURATION DESCRIPTION
 (RPMR02) LA-20. ROCKWELL O898 O88 V/MOD NOSE (BVM F)
 (RPMR04) LA-20. ROCKWELL O898 O88 V/MOD NOSE (BVM F)

CG-LOC ELEVTR BDFLAP RUOFLR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

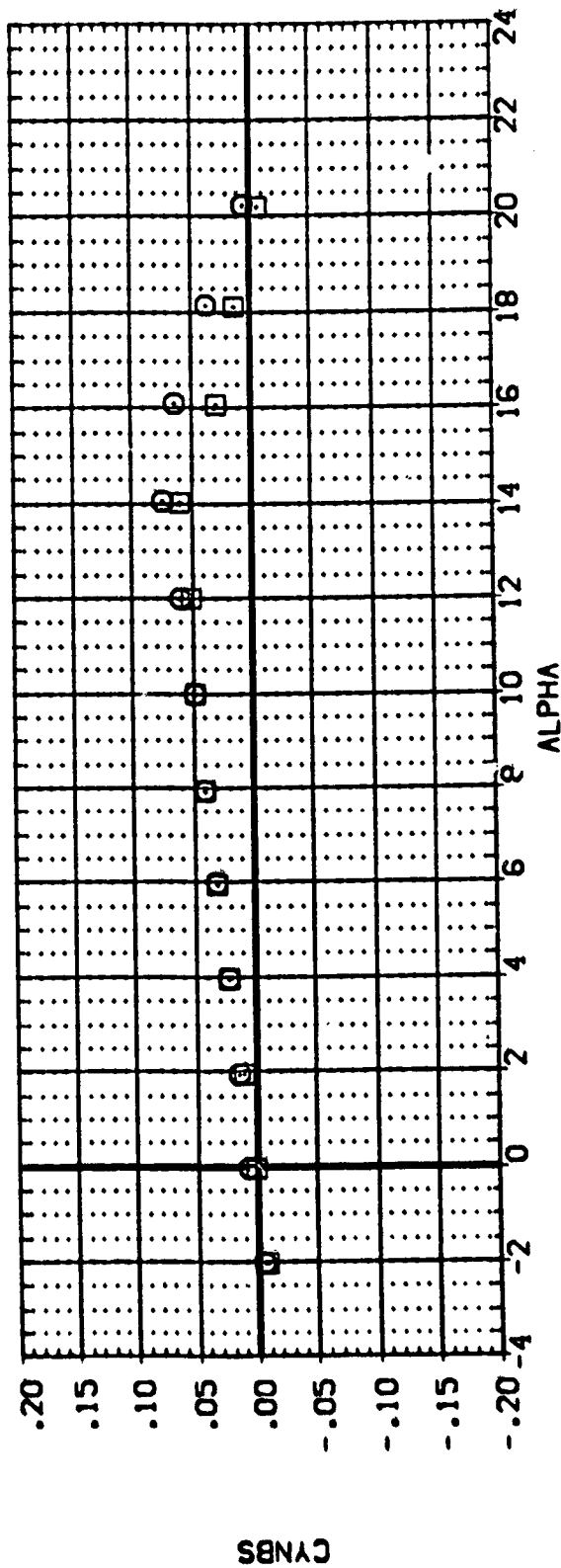
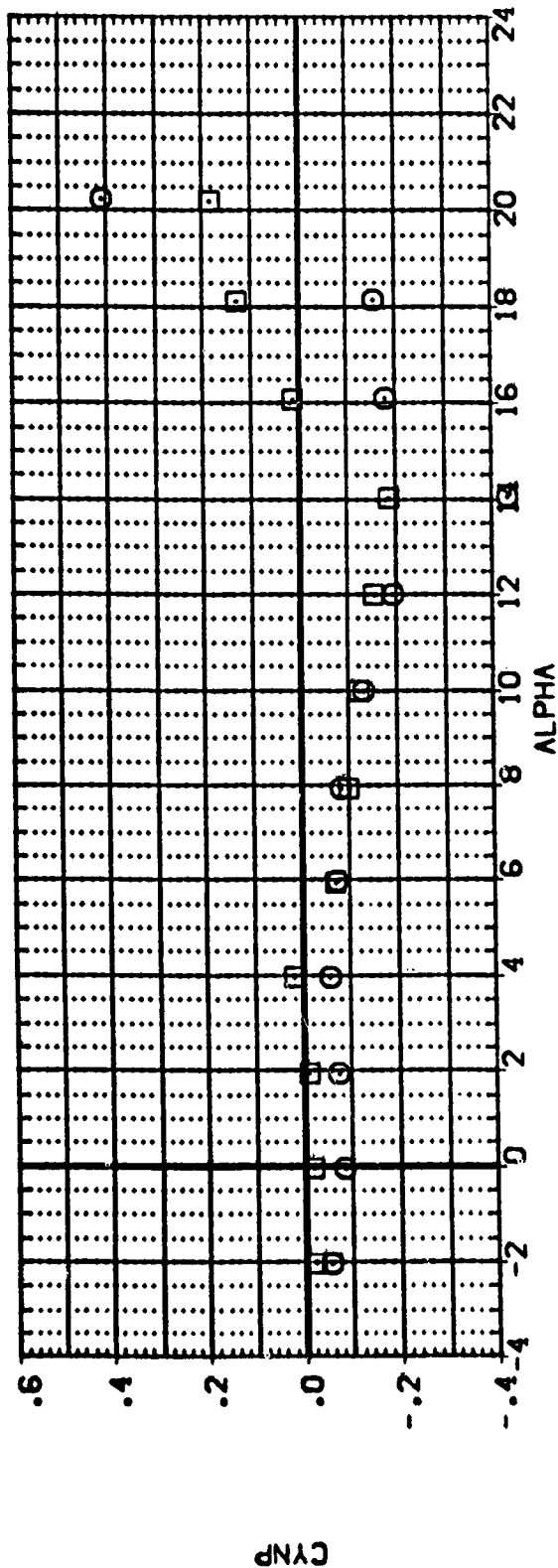


FIGURE 7. EFFECT OF OMS POOS ON DYNAMIC STABILITY PARAMETERS IN ROLL

(O)MACH = .98

CG-LOC FLEVIR OFFLAP RUFLR
1.000 .000 .000 10.000
1.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
[RPMR02] Q LA-20: ROCKWELL 0898 ORB W/MOD NOSE (BNV F)
[RPMR03] LA-20: ROCKWELL 0899 ORB W/MOD NOSE (BNVNF)

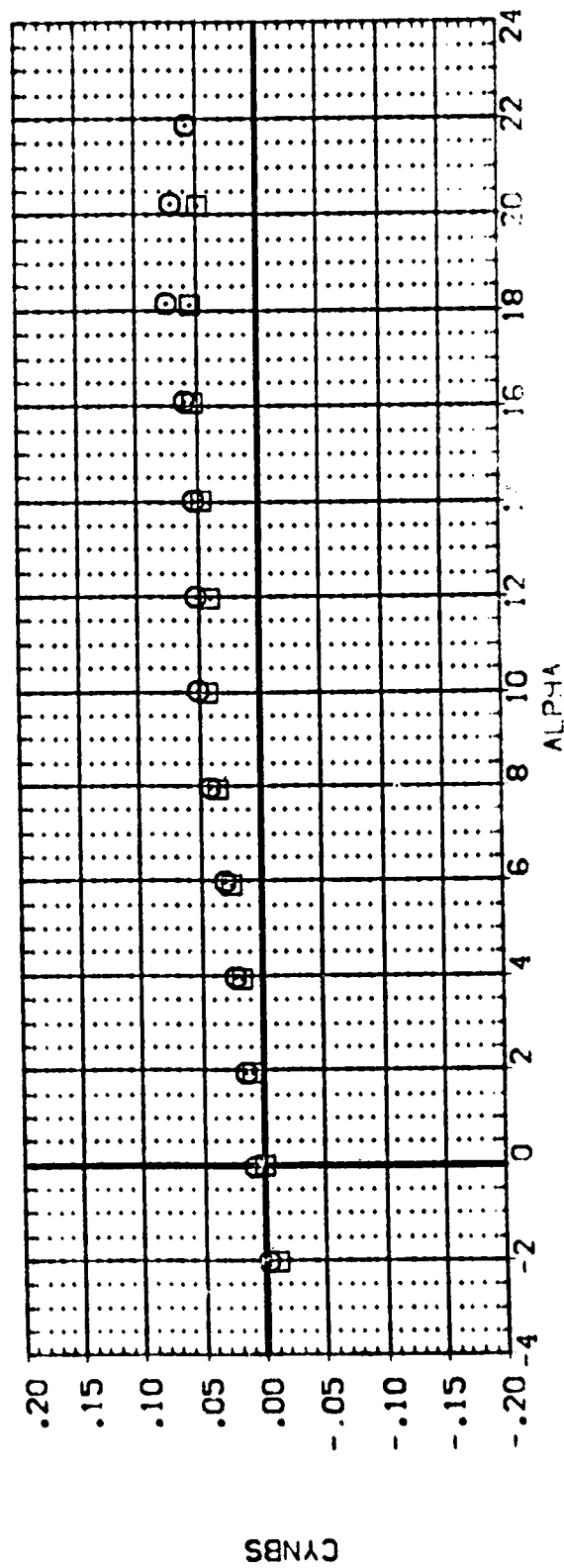
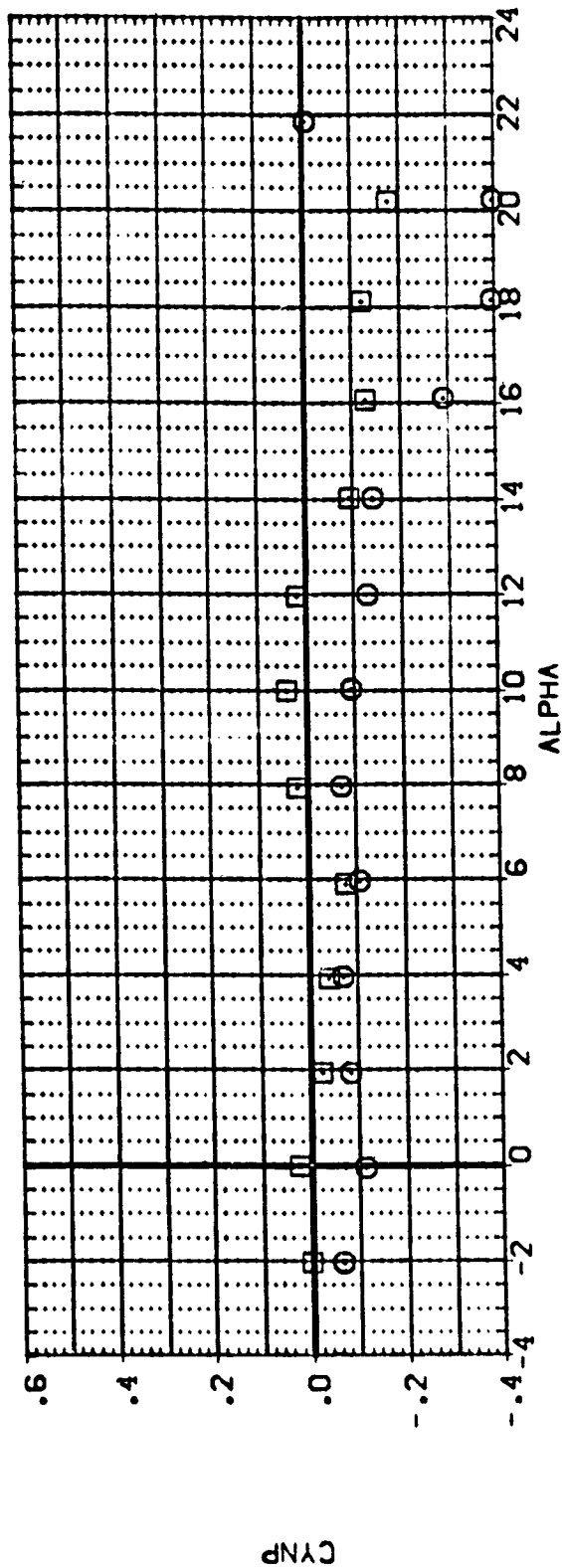


FIGURE 7. EFFECT OF RMS POOS ON DYNAMIC STABILITY PARAMETERS IN RC

CEMACH = 1.0

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVIR BDFLAP RUOFLR

[RPKY03] LA-20: ROCKWELL ORB 0858 V/MOD. NOSE (BN MF) 1.000 .000 .000 10.000

[RPKY04] LA-20: ROCKWELL ORB 0858 V/MOD. NOSE (BVMF) 1.000 .000 .000 10.000

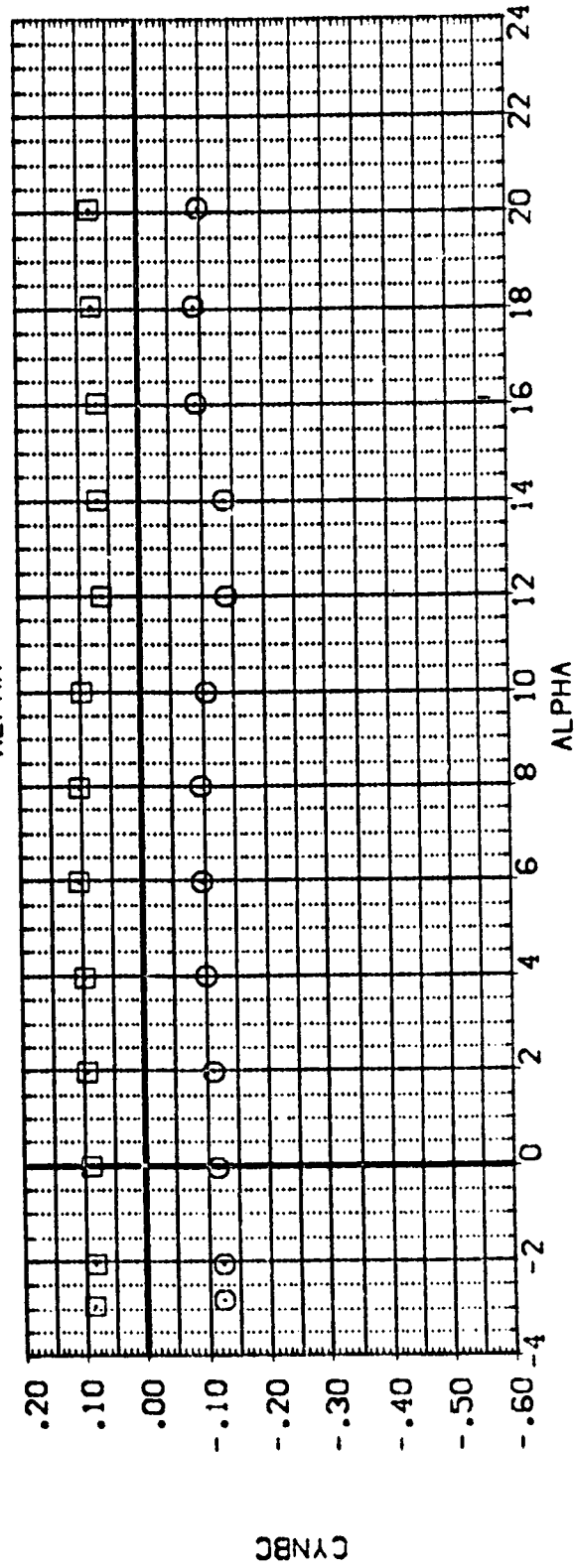
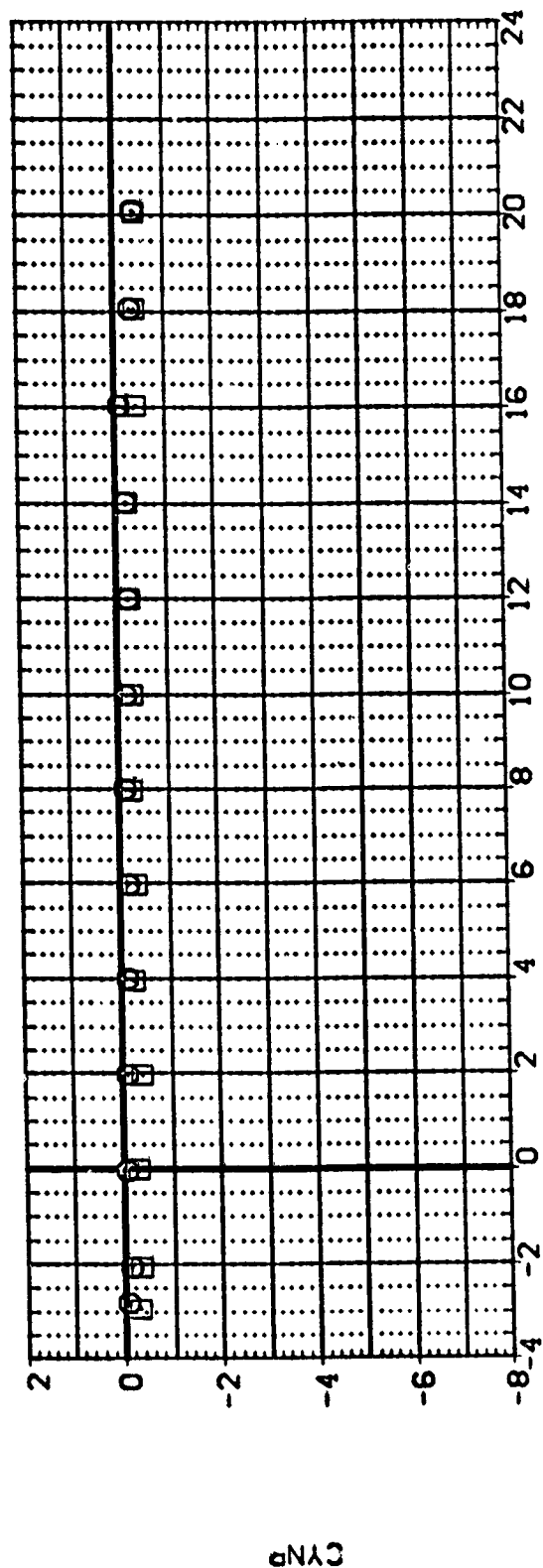


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

[A] WACH = .30

DATA SET SYMBOL: CONFIGURATION DESCRIPTION CG-LOC ELEVTR RDEFLAP RUDEFLR
 (RPNV33) 9 LA-20. ROCKWELL CRB 0898 V/MOD. NOSE (EV MF) 1.000 .000 .000 10.000
 (RPNV34) LA-20. ROCKWELL CRB 0898 V/MOD. NOSE (BWMF) 1.000 .000 .000 10.000

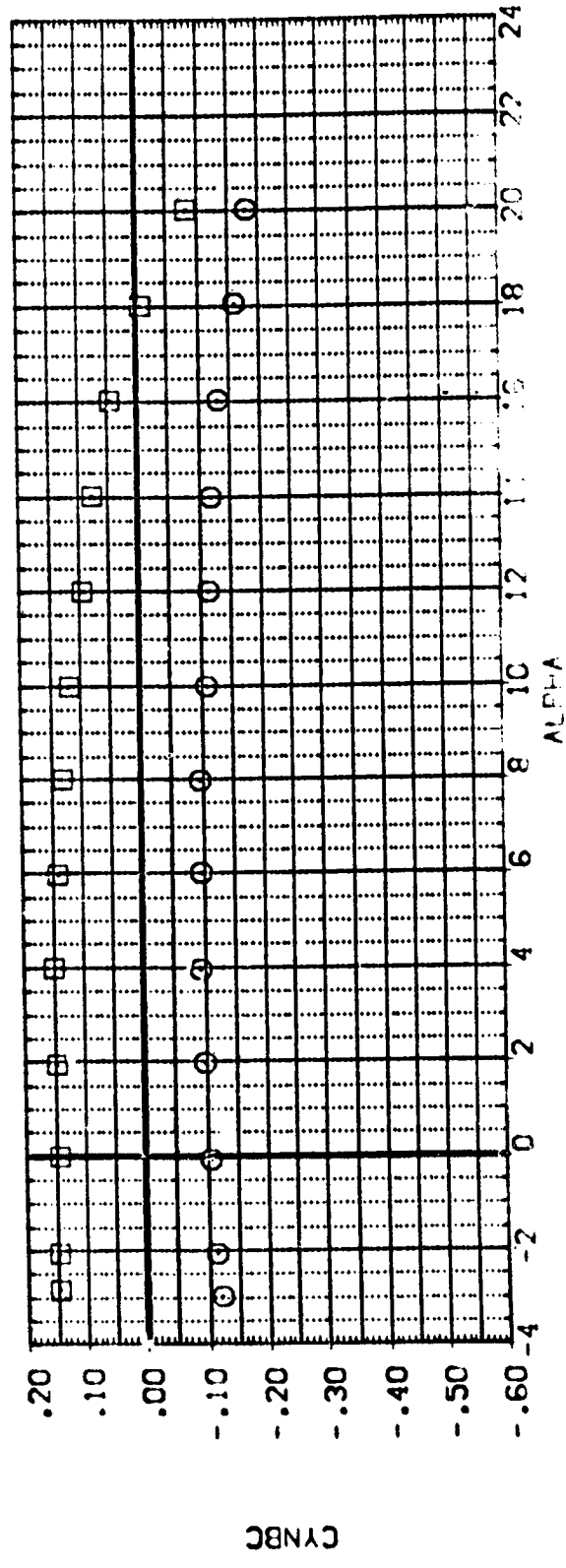
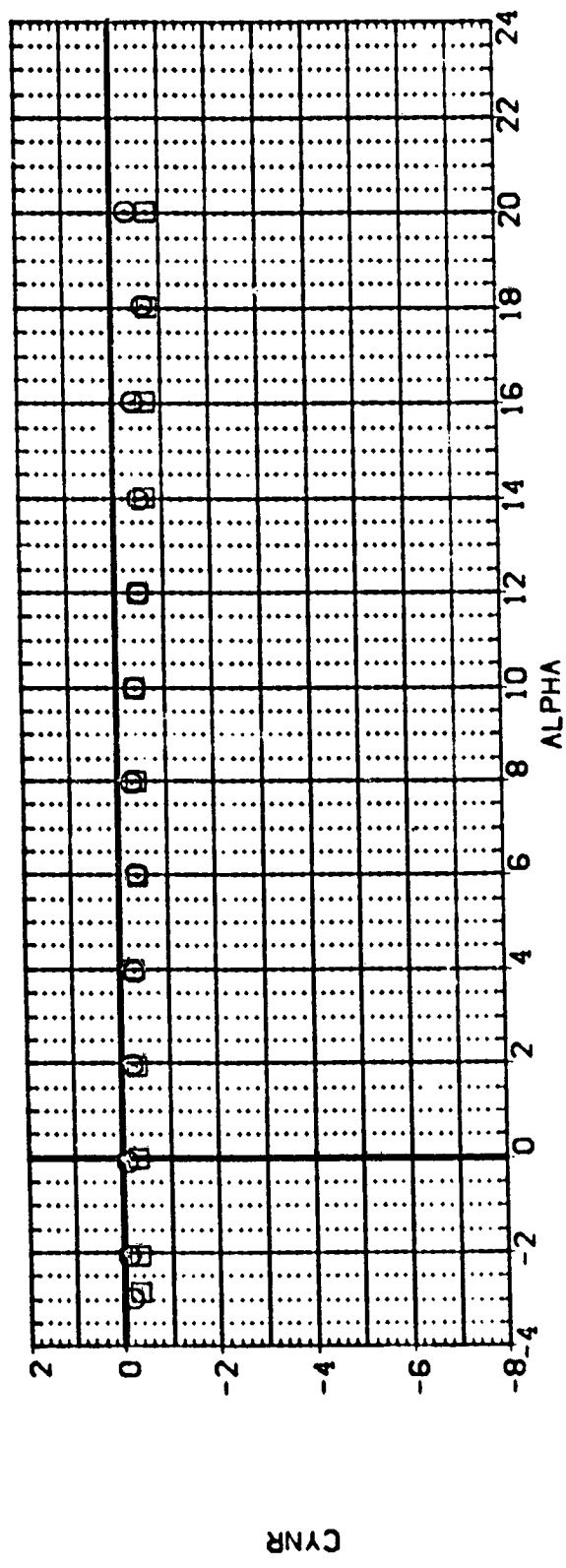


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN VAW

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVIR BOFLAP RUOFLR
 [RPMY03] LA-20, ROCKWELL ORB 0898 V/MOD, NOSE (BN MF) 1.000 .000 .000 10.000
 [RPMY04] LA-20, ROCKWELL ORB 0898 V/MOD, NOSE (BNMF) 1.000 .000 .000 10.000

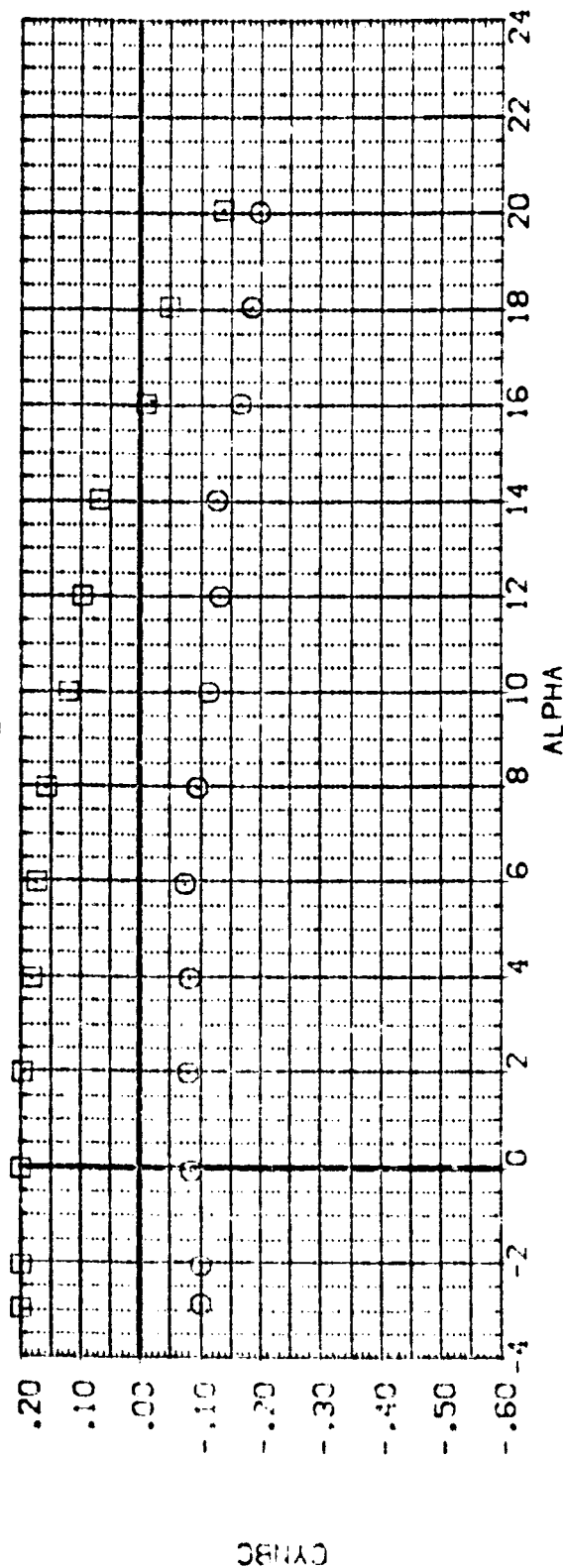
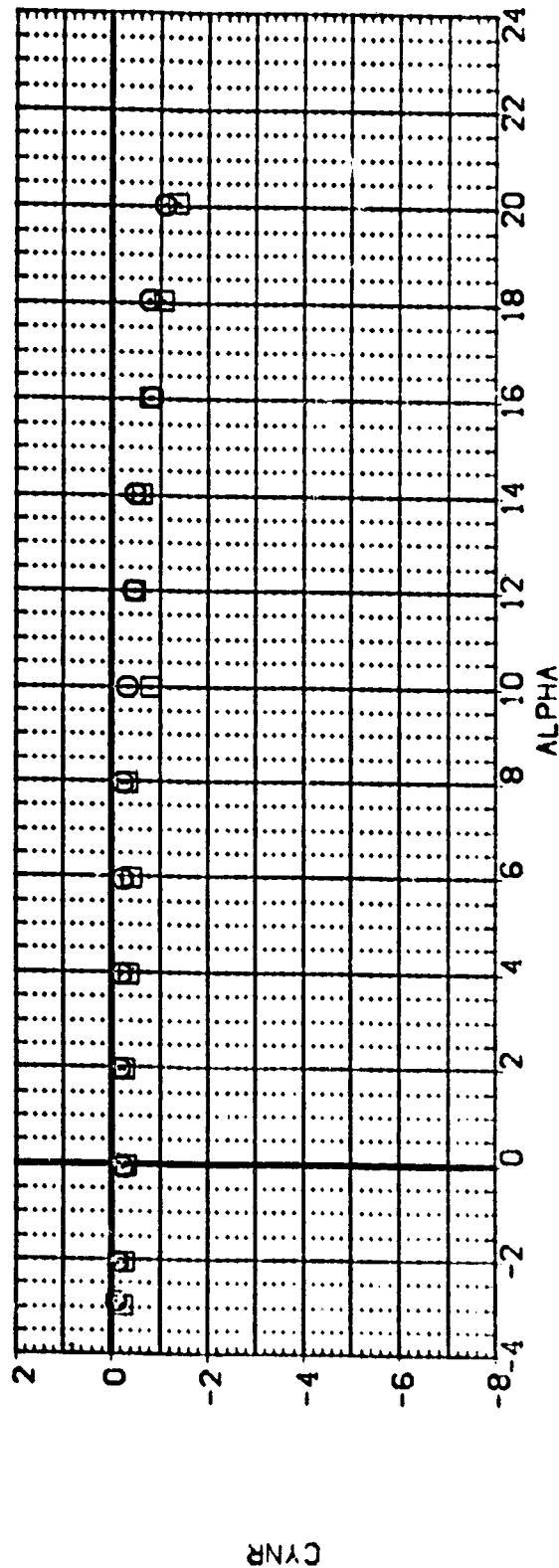


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

(C)MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUJFLR
 [RPKY03] LA-20. ROCKWELL ORB 0898 W/MOD. NOSE (BV MF) 1.000 .000 .000 10.000
 [RPKY04] LA-20. ROCKWELL ORB 0898 V/MOD. NOSE (BVVF) 1.000 .000 .000 10.000

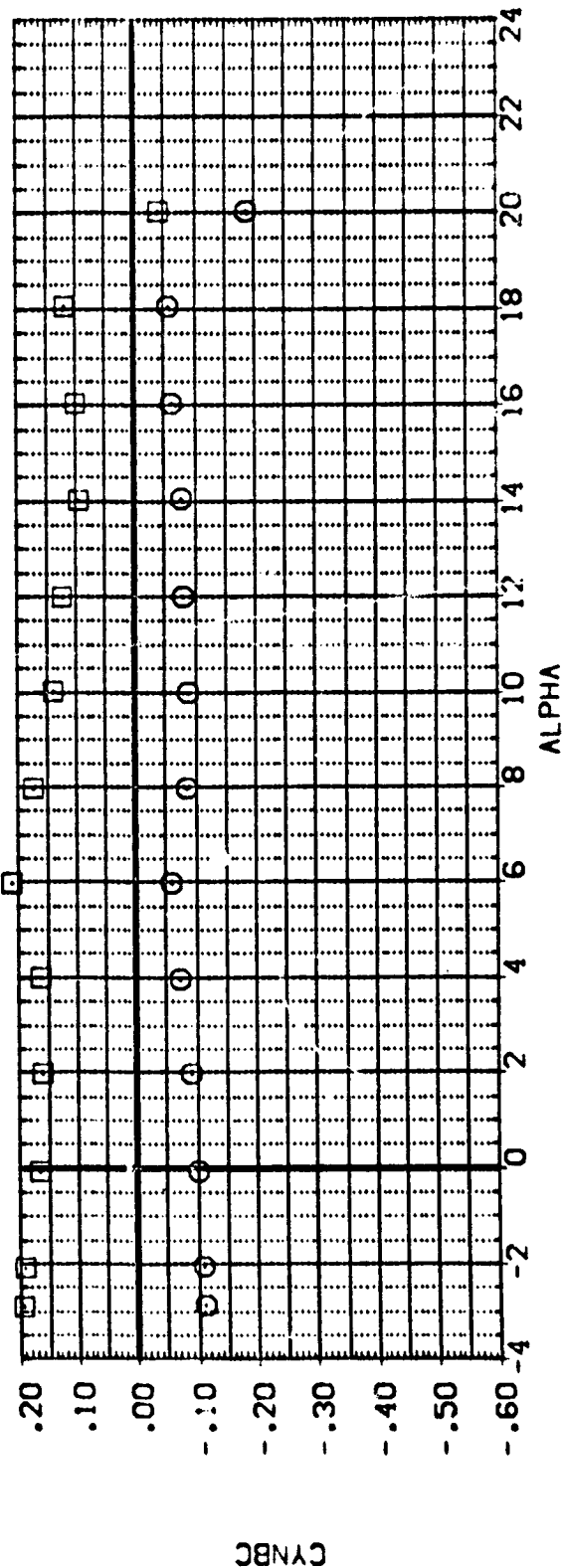
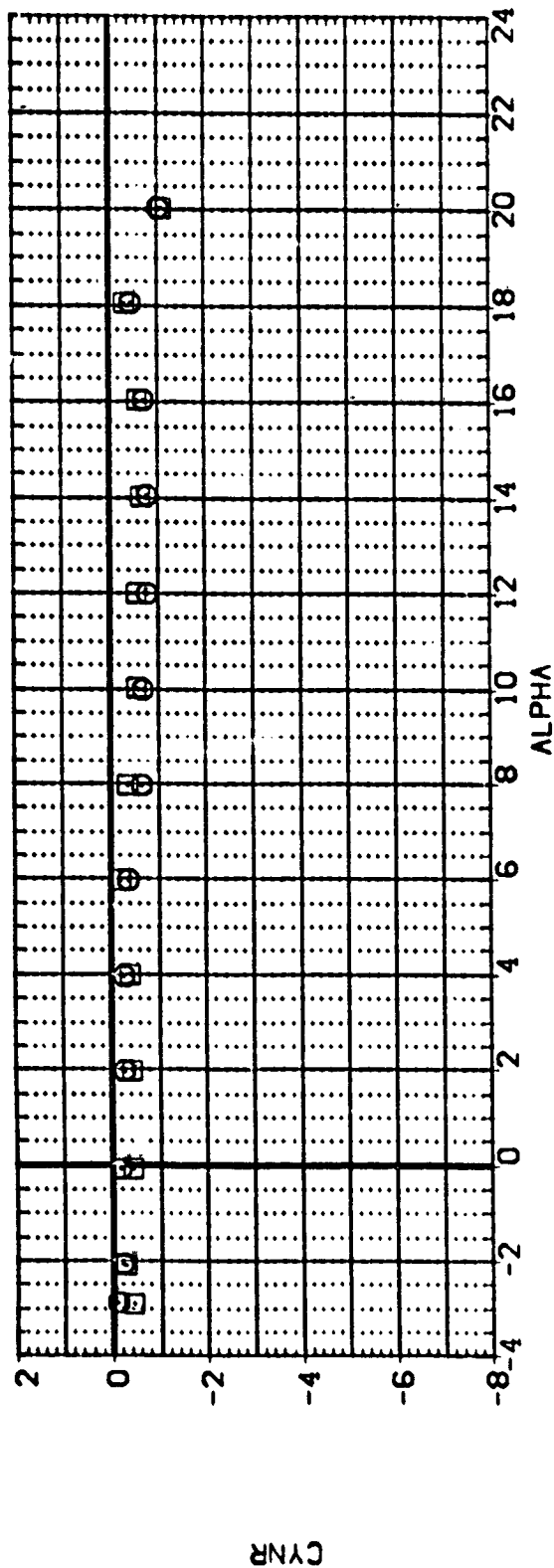


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

CONFIDENTIAL

CG-LOC ELEVTR BOFLAP RUOFIR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RPKY03) LA-20, ROCKWELL CRB 0898 V/MOD, NOSE (BV MF)
 (RPKY04) LA-20, ROCKWELL CRB 0898 V/MOD, NOSE (BV MF)

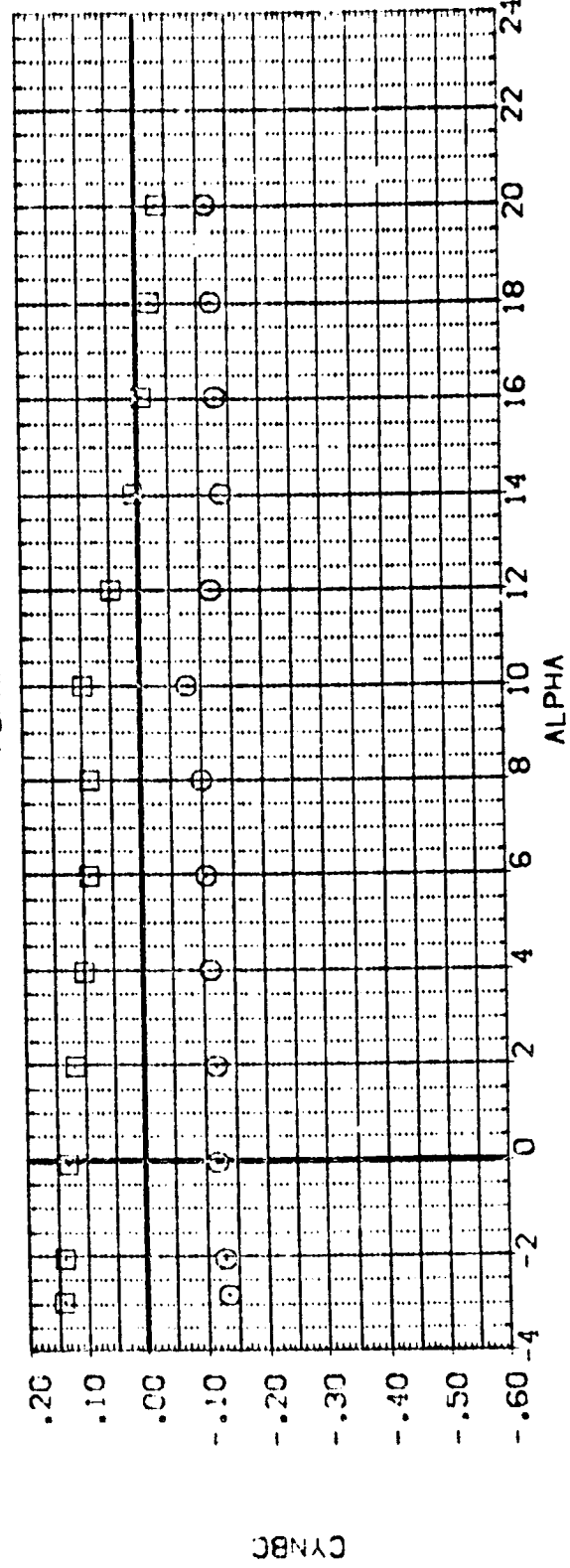
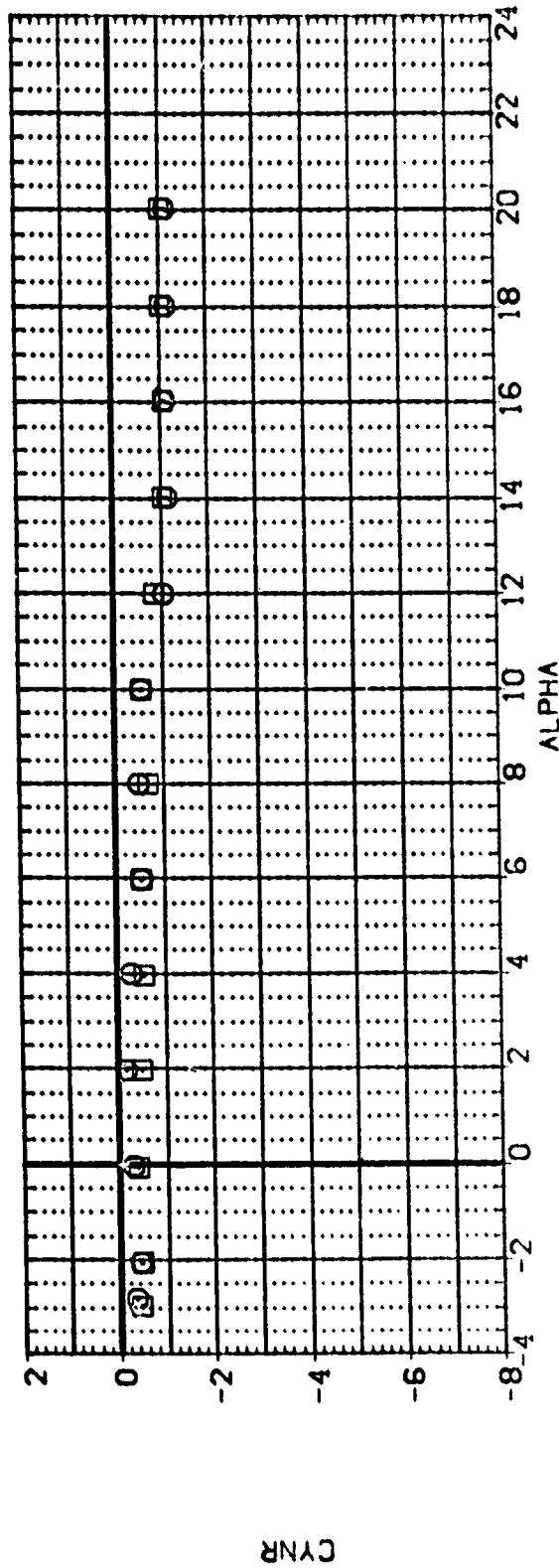


FIGURE 8- EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

CEMACH = 1.20

DATA SYMBOL CONFIGURATION DESCRIPTION
 (RPMY03) LA-20. ROCKWELL OPR 0858 V/MOD. NOSE (BV MF)
 (RPMY04) LA-20. ROCKWELL OPR 0858 V/MOD. NOSE (BV MF)

CG-LOC ELEVTR HDEFAP RUOFLR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

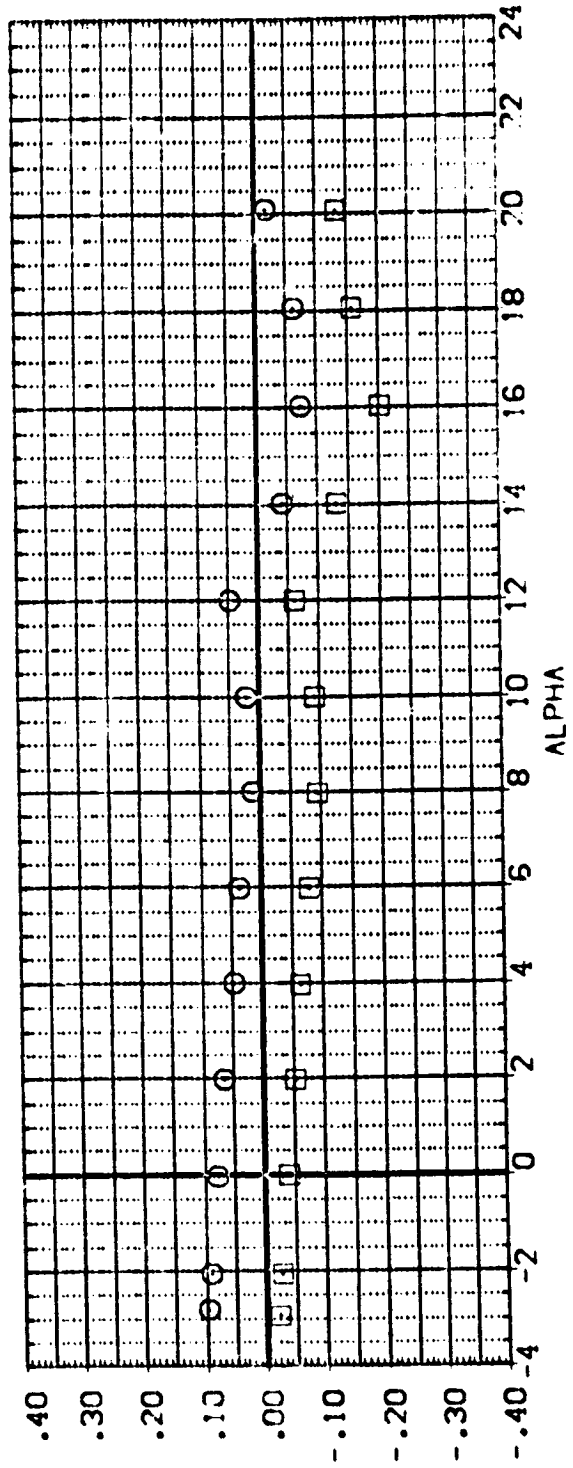
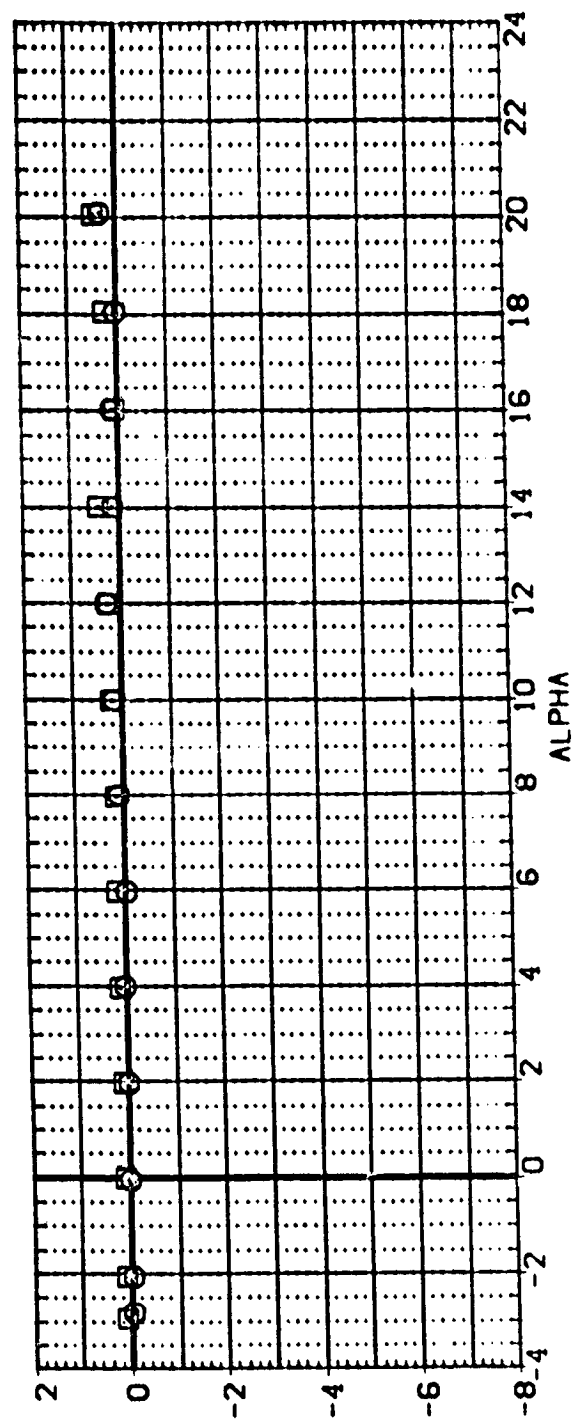


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

CG-LOC ELEVTR BOFLAP RUOFLR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RPM Y03) LA-20. ROCKWELL QRB 0828 V/MCO. NOSE (BN MF)
 (RPM YC4) LA-20. ROCKWELL QRB 0828 V/MCO. NOSE (BNMF)

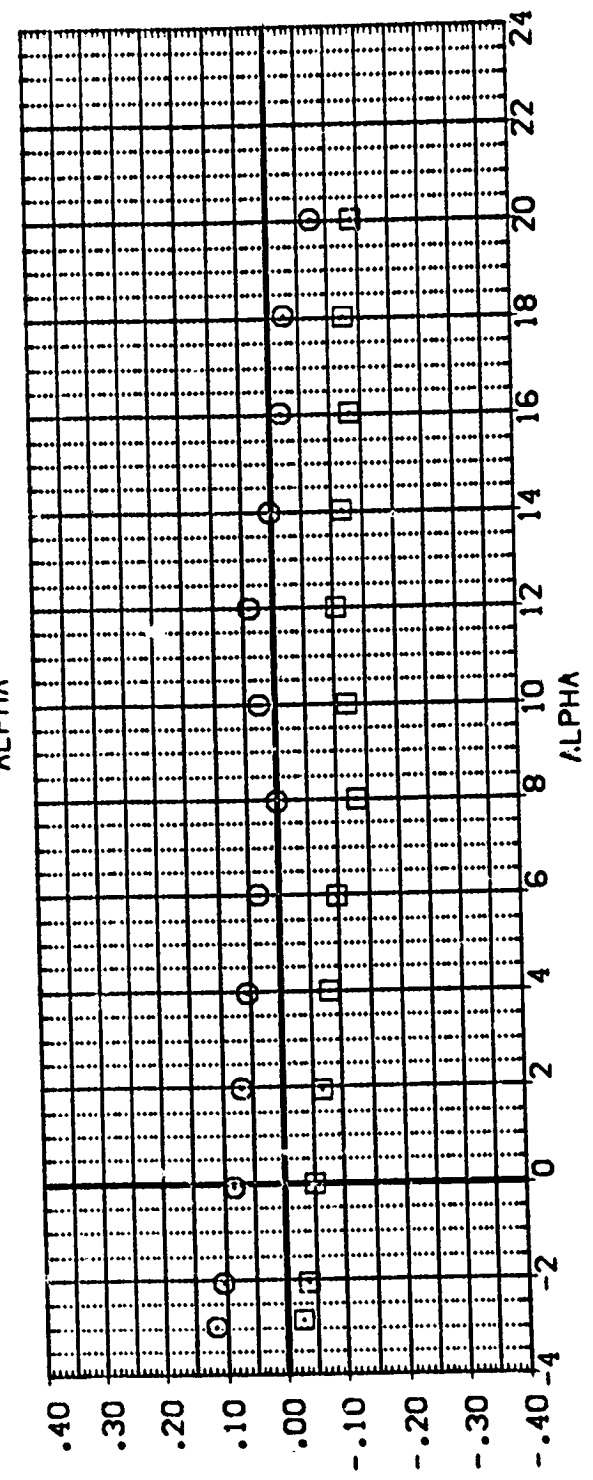
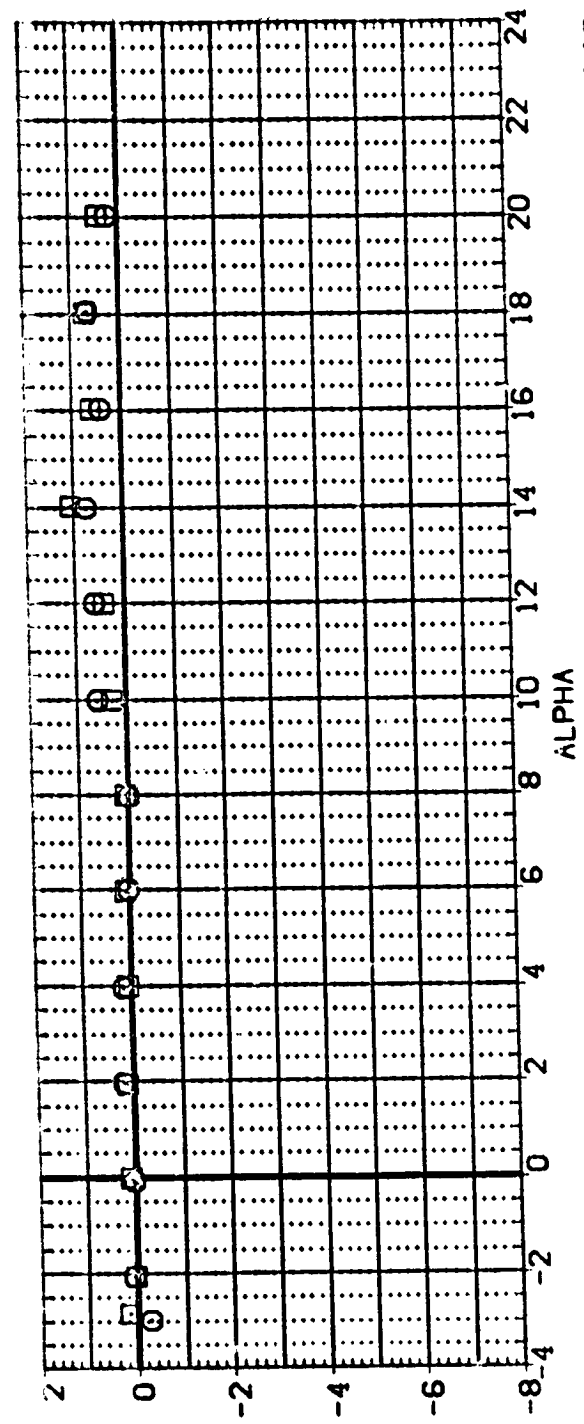


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

BJMACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

[RPKY03] LA-20, ROCKWELL ORB 0858 V/MOD, NOSE (8V MF) 1.000 .000 .000 10.000

[RPKY04] LA-20, ROCKWELL ORB 0858 V/MOD, NOSE (8V MF) 1.000 .000 .000 10.000

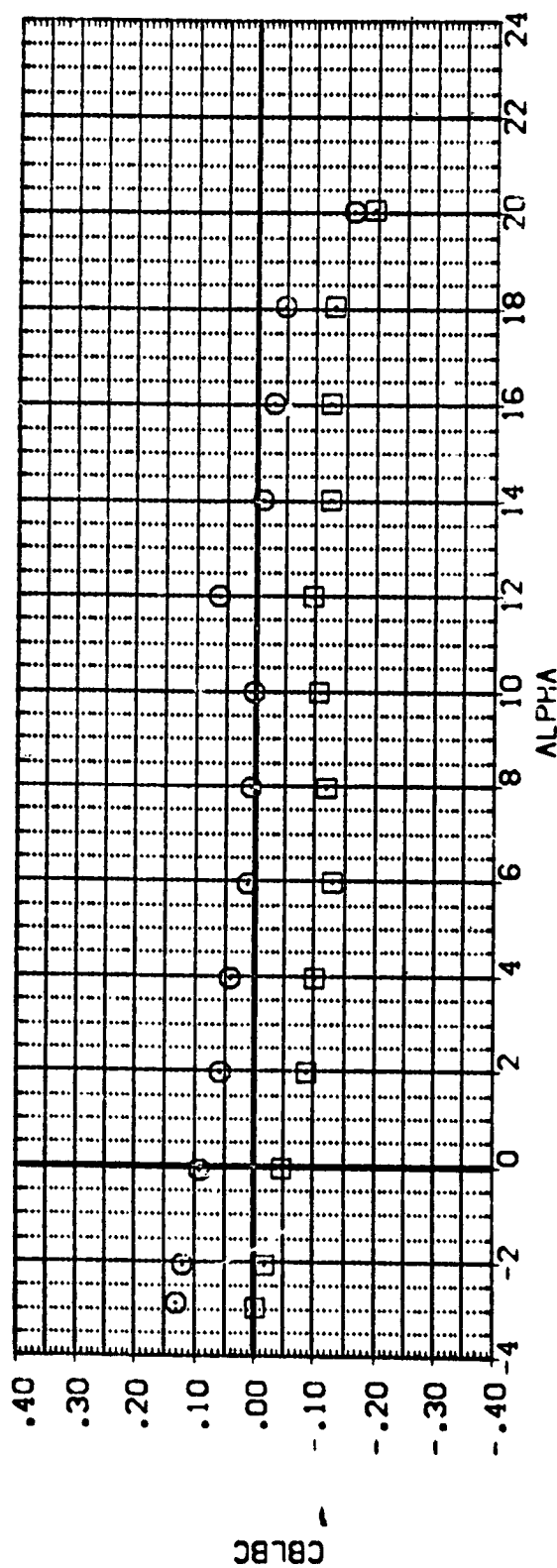
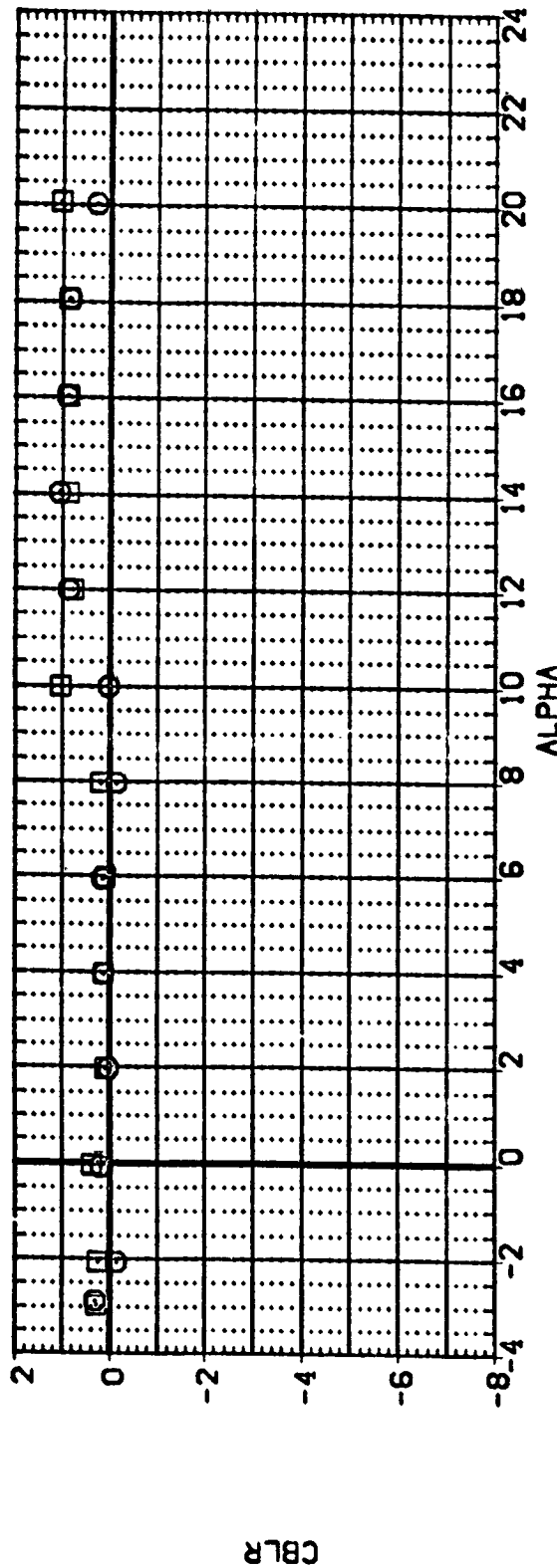


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

(CJ MACH = .90)

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48

DATA SET 3130L CONFIGURATION DESCRIPTION
 (PPL103) LA-20: ROCKWELL ORB 0800 V.1000 NOSE (BY MF)
 (PPL103) LA-20: ROCKWELL ORB 0800 V.1000 NOSE (BY MF)

CS-LOC ELEVTR EDOFLAP RUOFLR
 1.000 .000 .000 10.000
 1.000 .000 .000 10.000

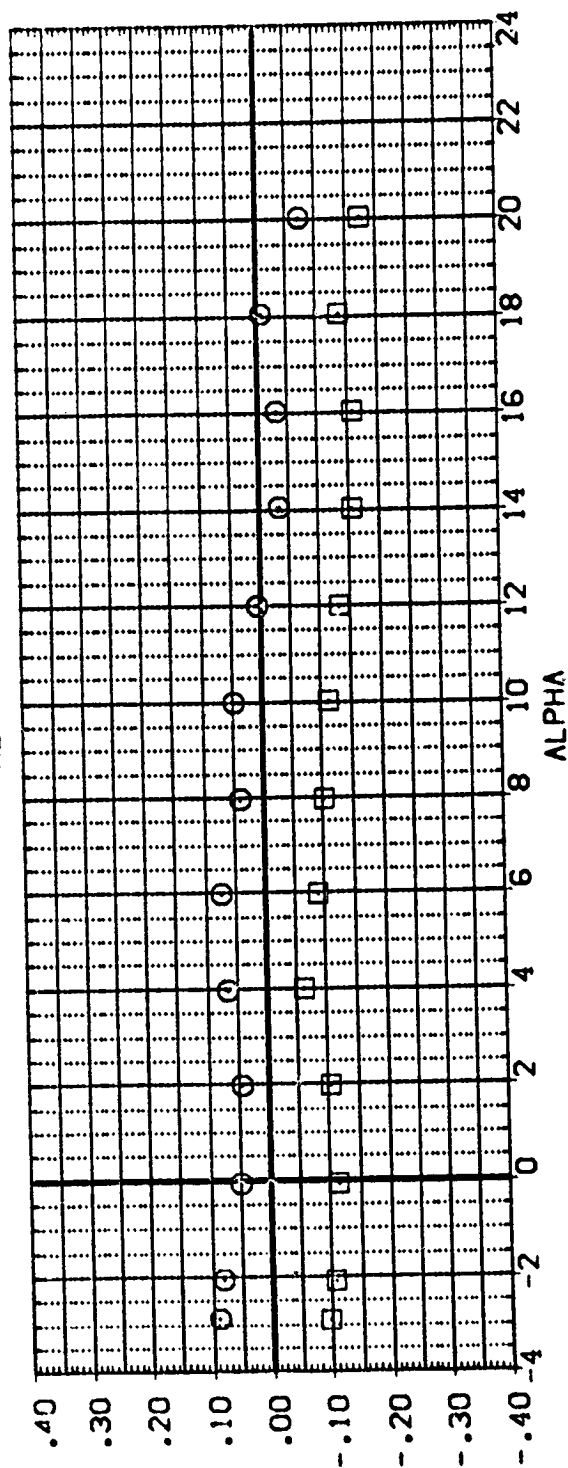
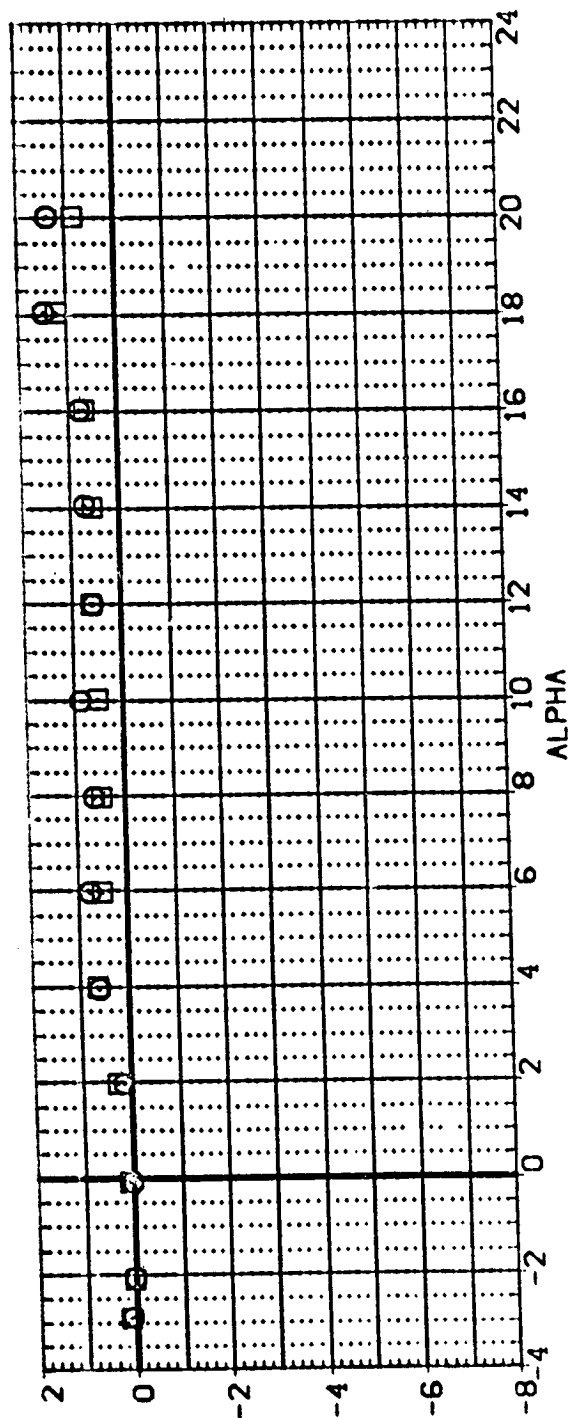


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

(C)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVIR BOT'LAP RUOFLR
 [RPKY03] LA-20, ROCKWELL ORB 0898 V/MOD, NOSE (BV MF) 1.000 .000 .000 10.000
 [RPKY04] LA-20, ROCKWELL ORB 0898 V/MOD, NOSE (BV MF) 1.000 .000 .000 10.000

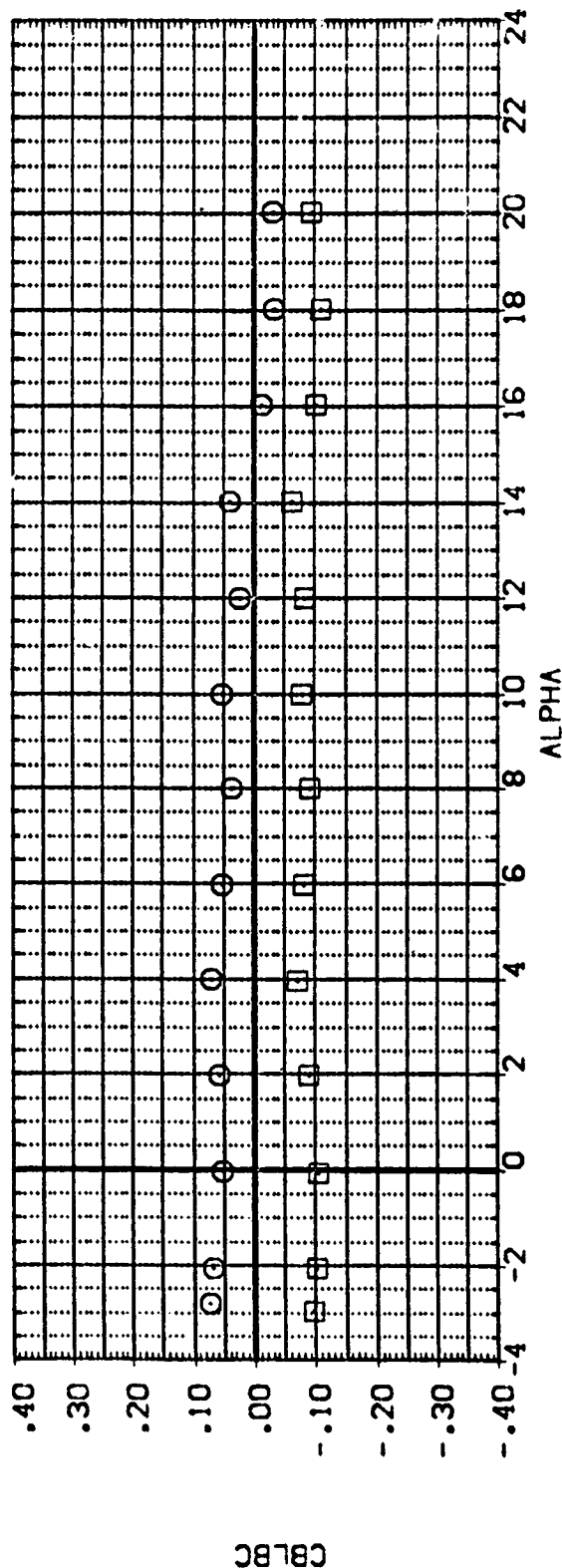
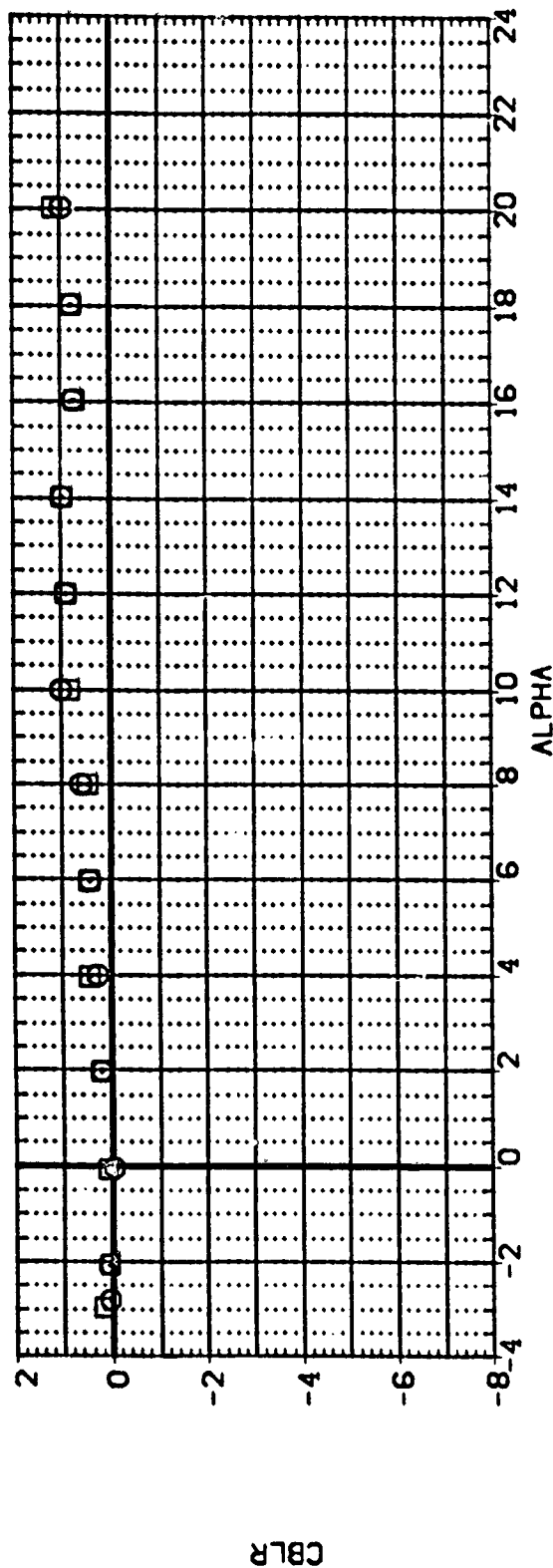


FIGURE 8. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN YAW

GEOMACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR RNF LAP RUOFLR
 (PREFIG) (REMARKS) LA-20: ROCKWELL CB88 CB8 V/MOD NOSE (BV MF) 1.000 .000 .000 10.000
 LA-20: ROCKWELL CB88 CB8 V/MOD NOSE (BV MF) 1.000 .000 .000 10.000

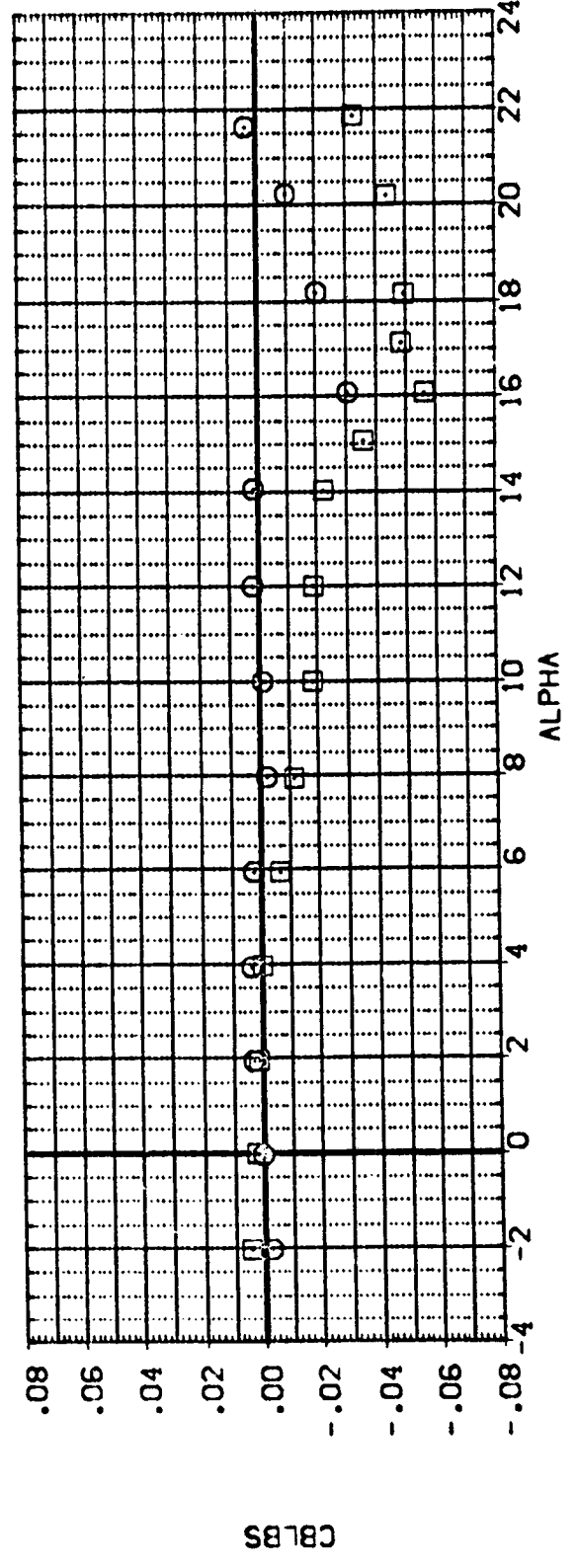
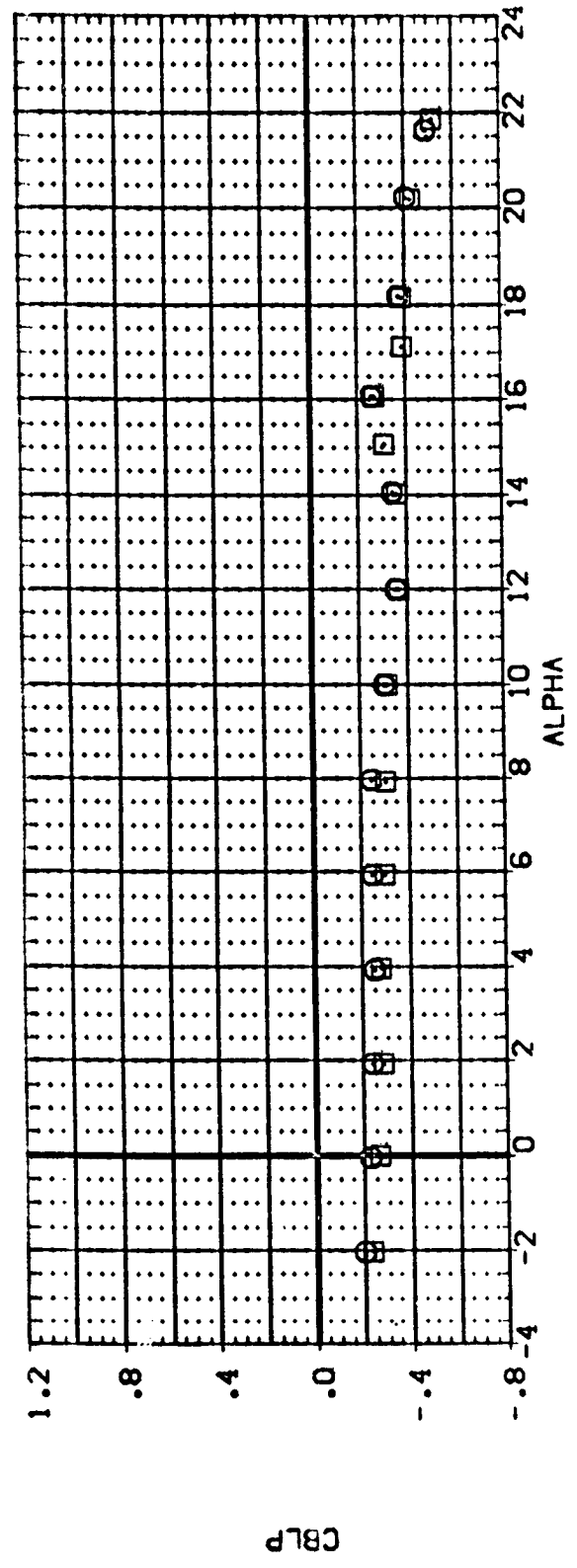


FIGURE 9. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN ROLL

(A)MACH = .30

CG-LOC ELEVTR BOFLAP RUOFLR
1.000 .000 .000 10,000

DATA SET SYMBOL CONFIGURATION DESCRIPTION
(RPR03) LA-20: ROCKWELL 0898 OR8 V/MOD NOSE (BV MF)
(RPR04) LA-20: ROCKWELL 0898 OR8 V/MOD NOSE (BV MF)

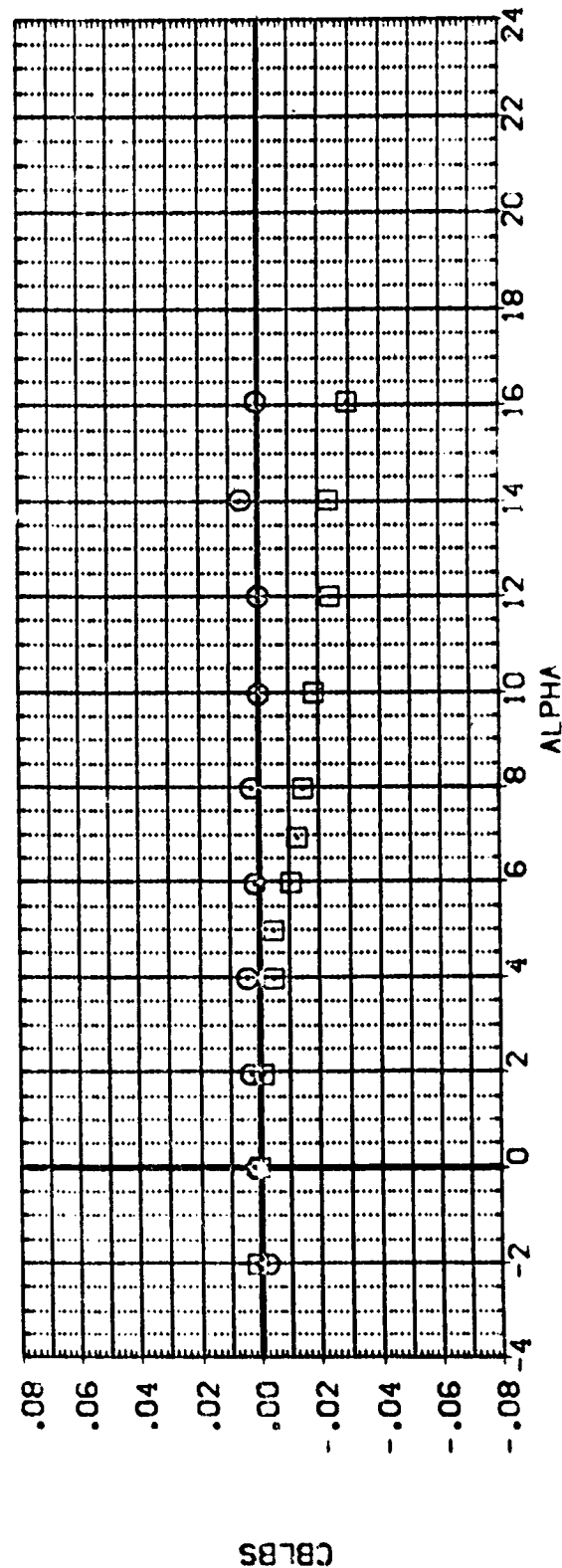
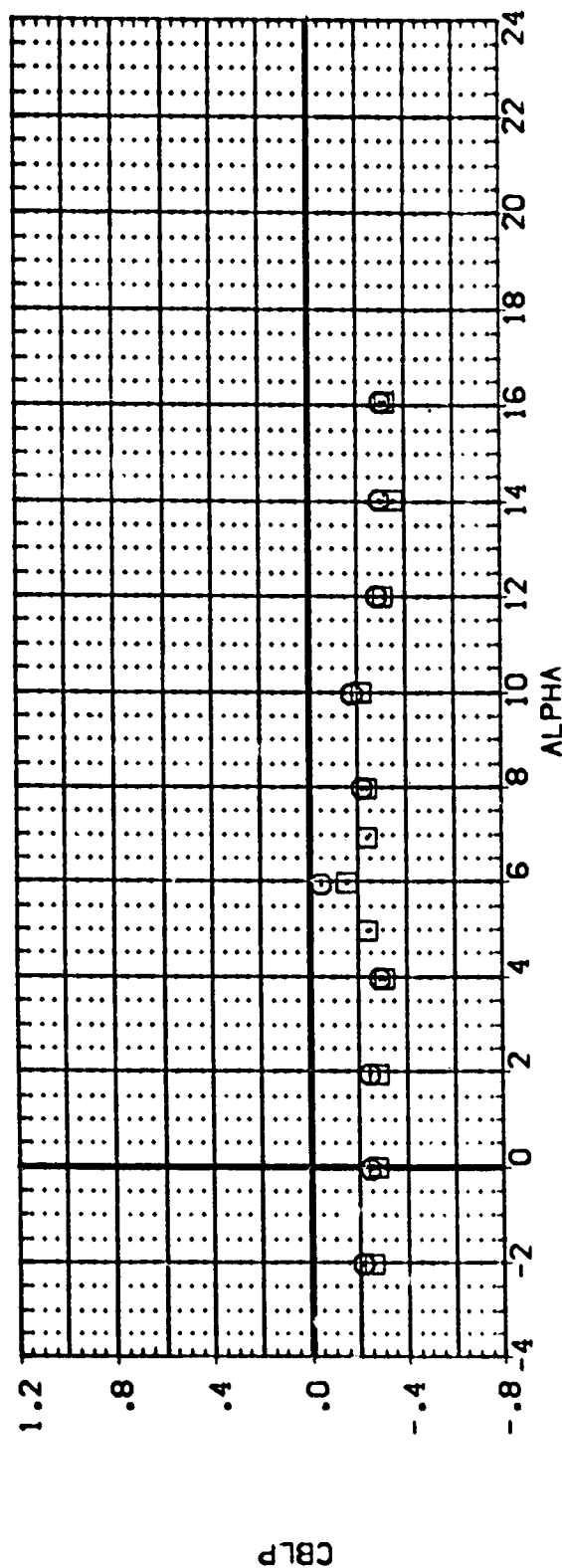


FIGURE 9. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN ROLL

(B)MACH = .80

CG-LOC ELEVTR BOFLAP RUOFLR
1.000 .000 .000 10.000

DATA SET SYMBOL CDF IGRATION DESCRIPTION
(RPM03) LA-20. ROCKWELL 0898 OR8 V/MOD NOSE (BV MF)
(RPM04) LA-20. ROCKWELL 0898 OR8 V/MOD NOSE (BV MF)

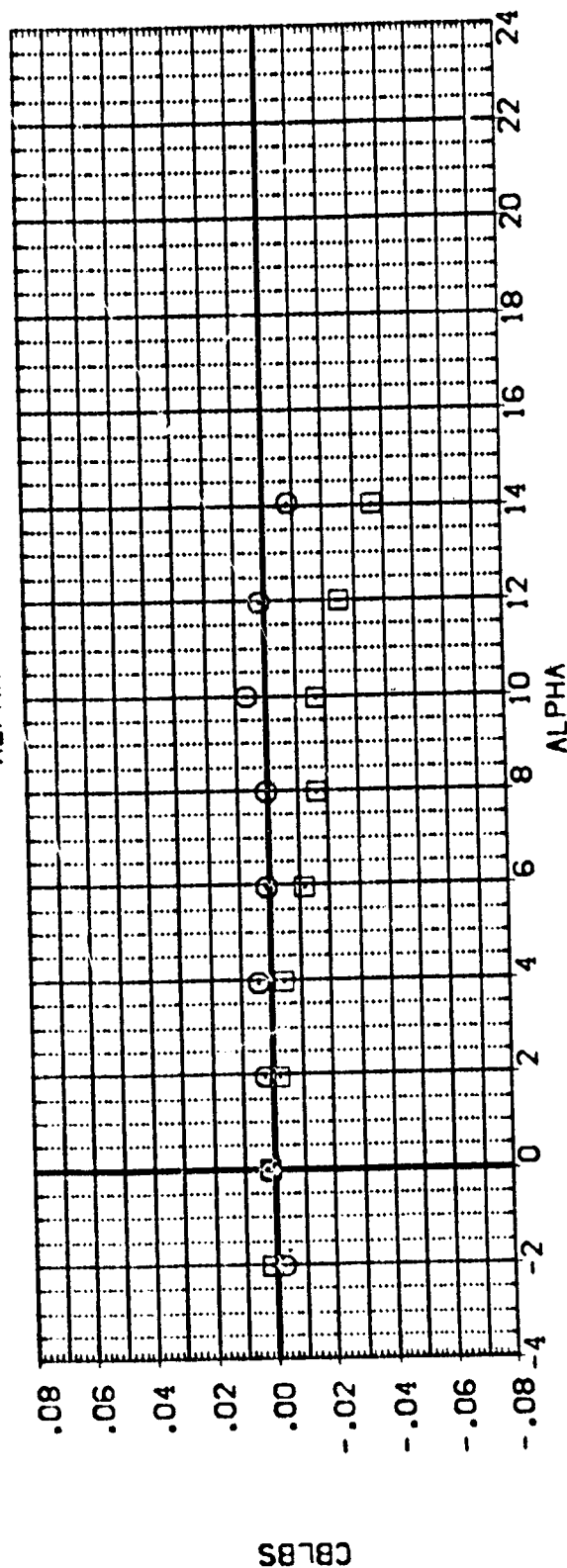
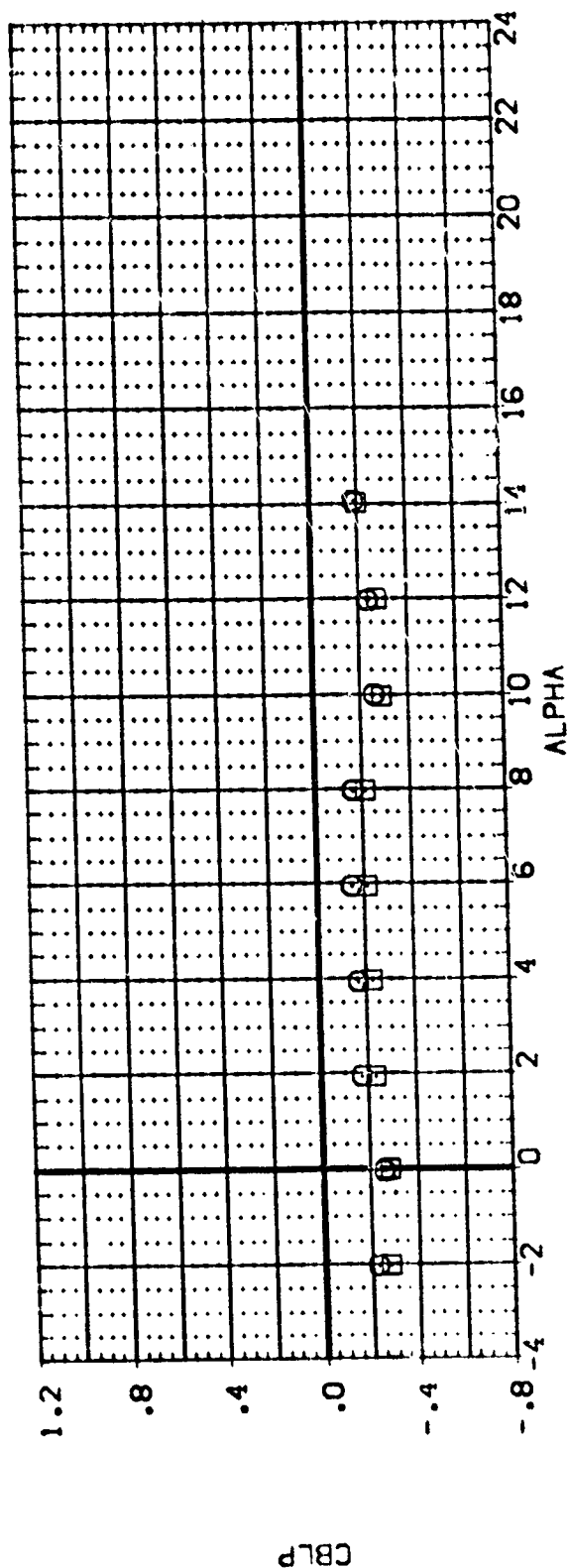


FIGURE 9. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN ROLL

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 [RPMR03] [RPMR03] LA-20, ROCKWELL C888 ORB V/MOD NOSE (BV MF)
 [RPMR03] [RPMR03] LA-20, ROCKWELL C888 ORB V/MOD NOSE (BVMF)

CG-LOC 1.000
 ELEVIR .000
 BOFLAP .000
 RUOFLR 0.000

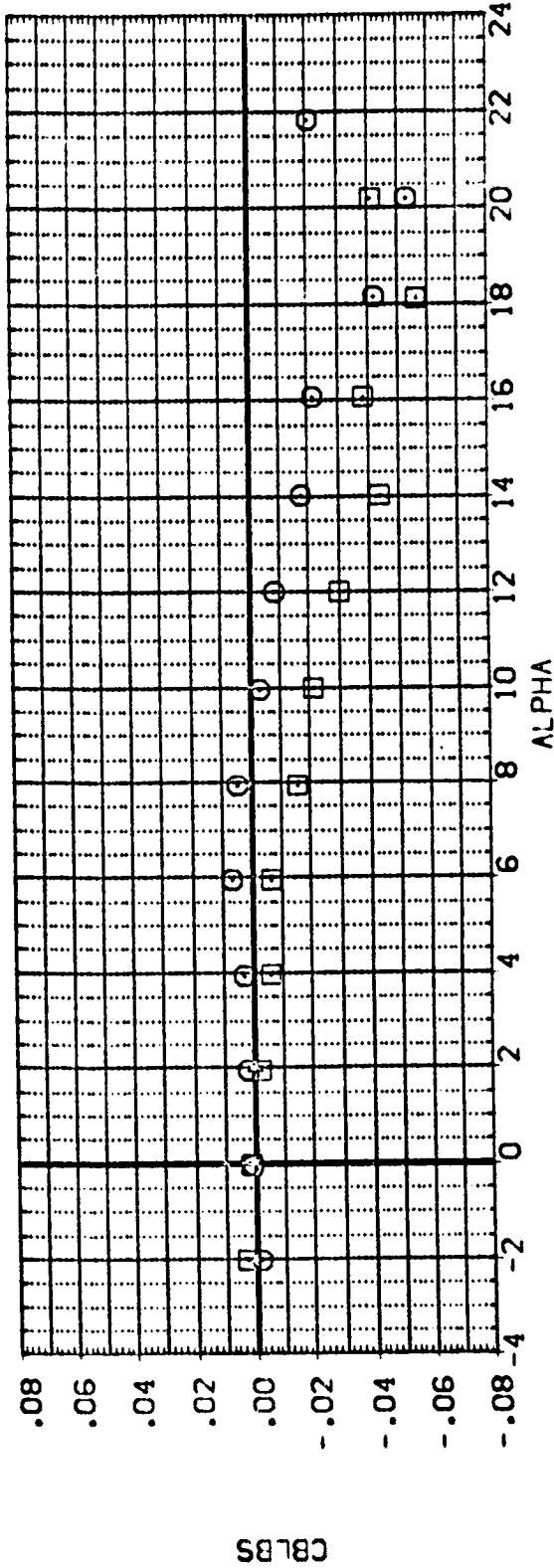
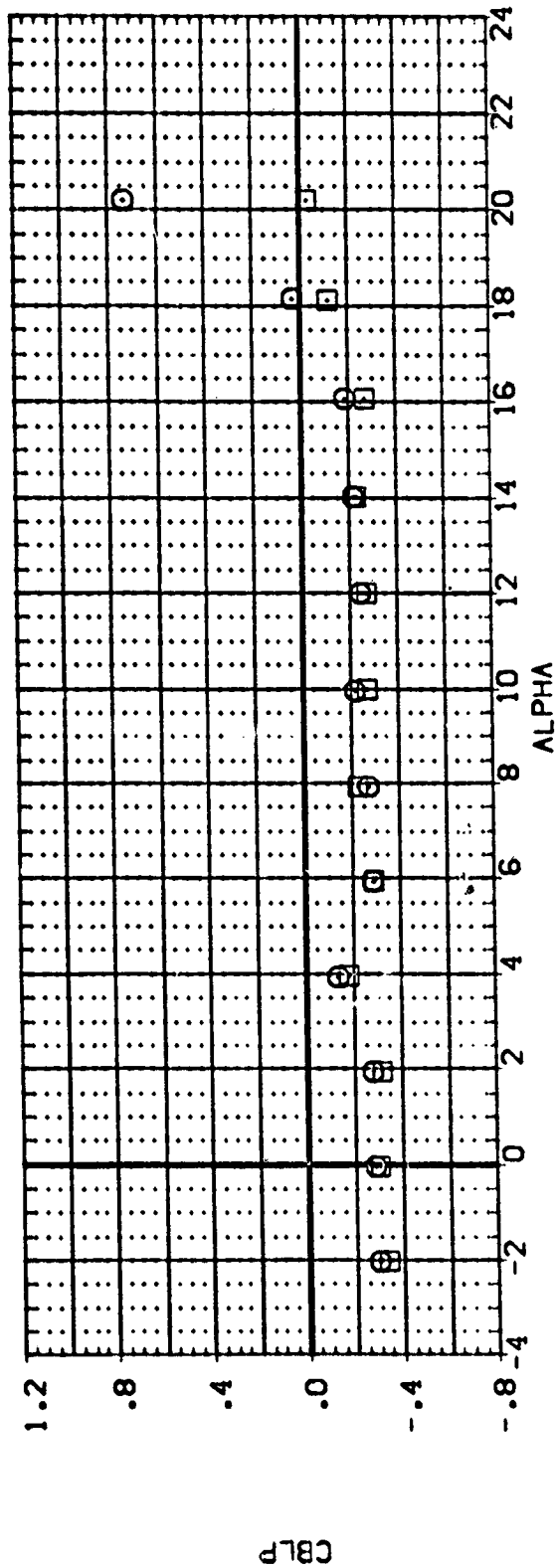


FIGURE 9. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN ROLL

LA-20 SYMBOL DESCRIPTION
 LA-20: RECONSTRUCTION OF V AND W (BY ME)
 LA-20: RECONSTRUCTION OF V AND W (BY ME)
 LA-20: RECONSTRUCTION OF V AND W (BY ME)

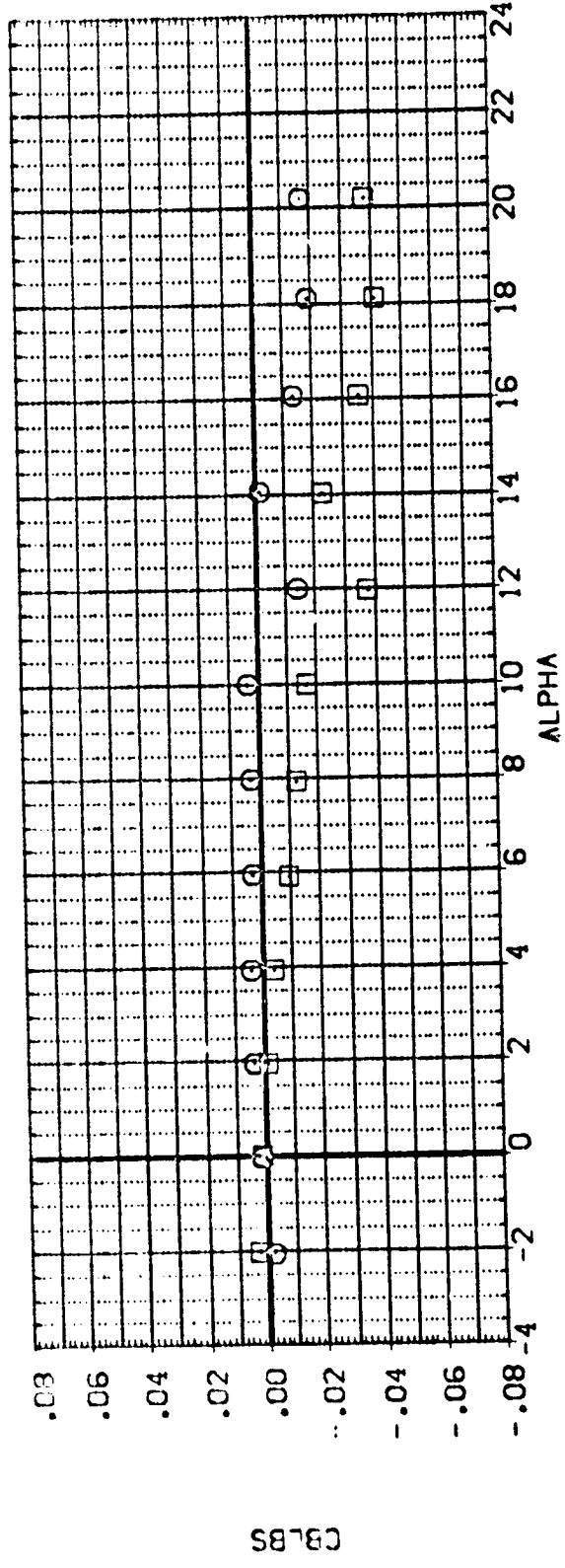
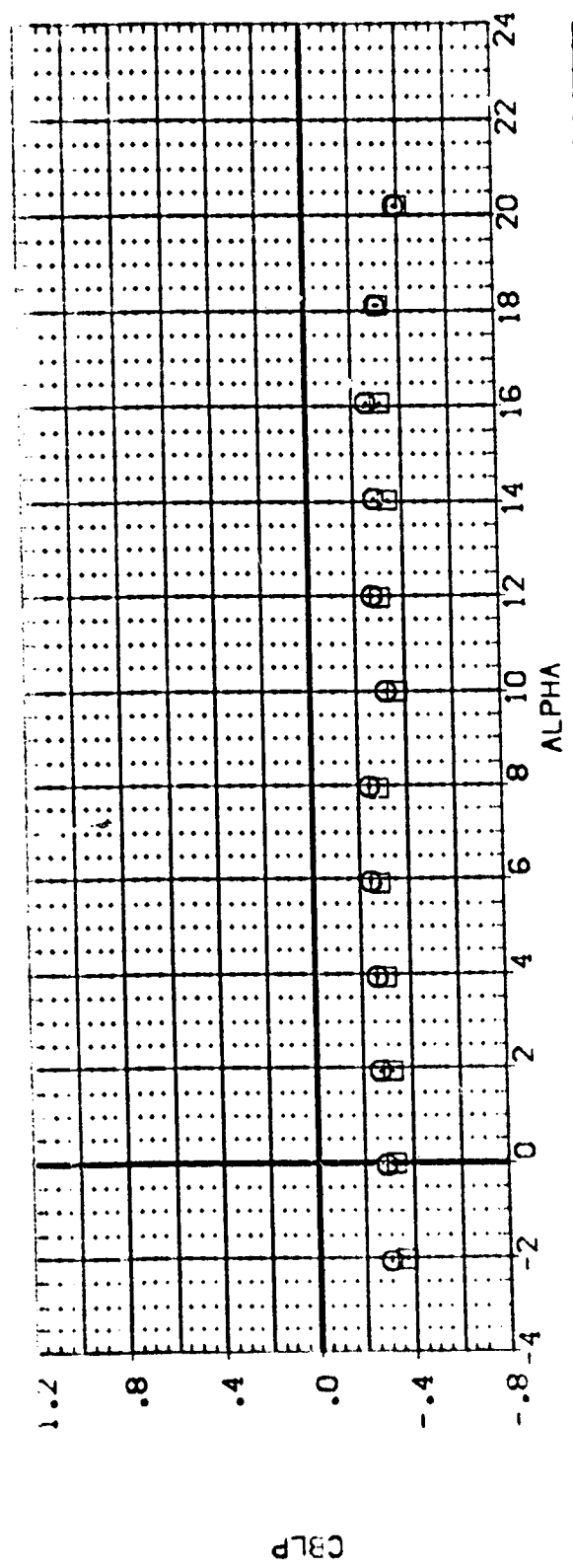



FIGURE 9. EFFECT OF VERTICAL TAIL ON DYNAMIC STABILITY PARAMETERS IN ROLL

(C)MACH = 1.20

DATA SET SYMBOL:  CONFIGURATION DESCRIPTION: LA-20: ROCKWELL 0898 OR8 V/HOD NOSE (BWWF) CG-LOC: 1.000 ELEVTR: .000 BDFLAP: .000 RUOFLR: 10.000
 (RPN/P04) LA-20: ROCKWELL 0898 OR8 V/HOD NOSE (BWWF) 1.000 .000 13.000 85.000
 (RPN/P05)

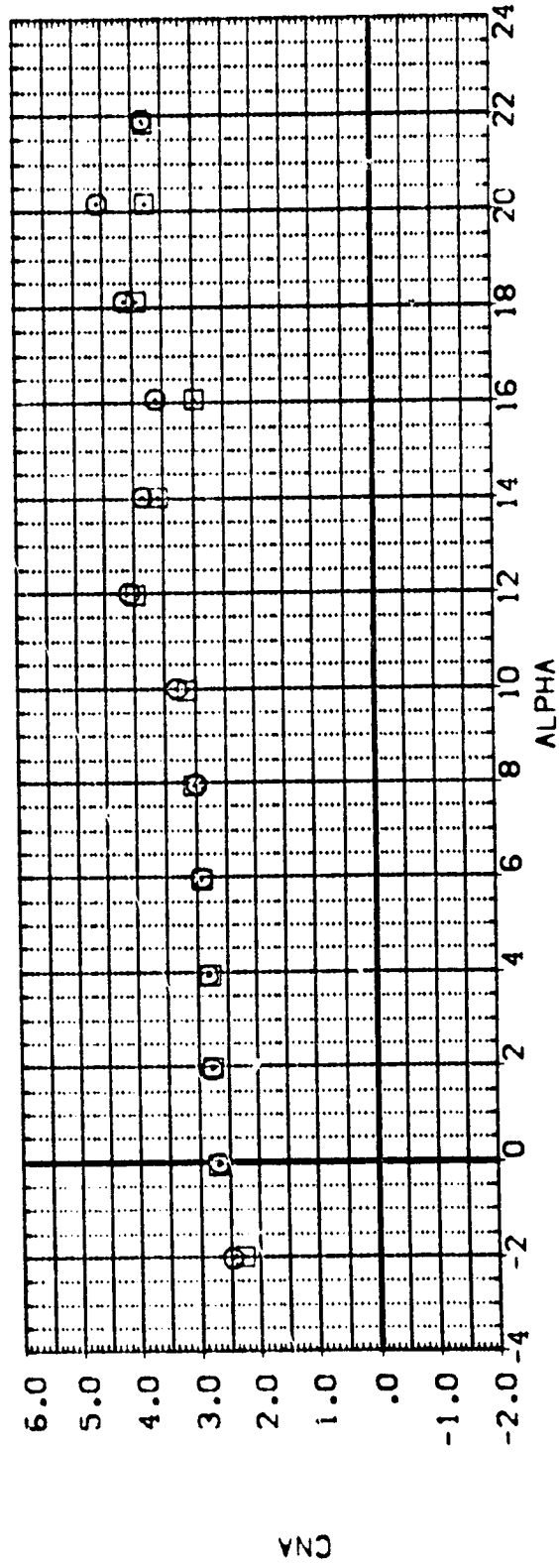
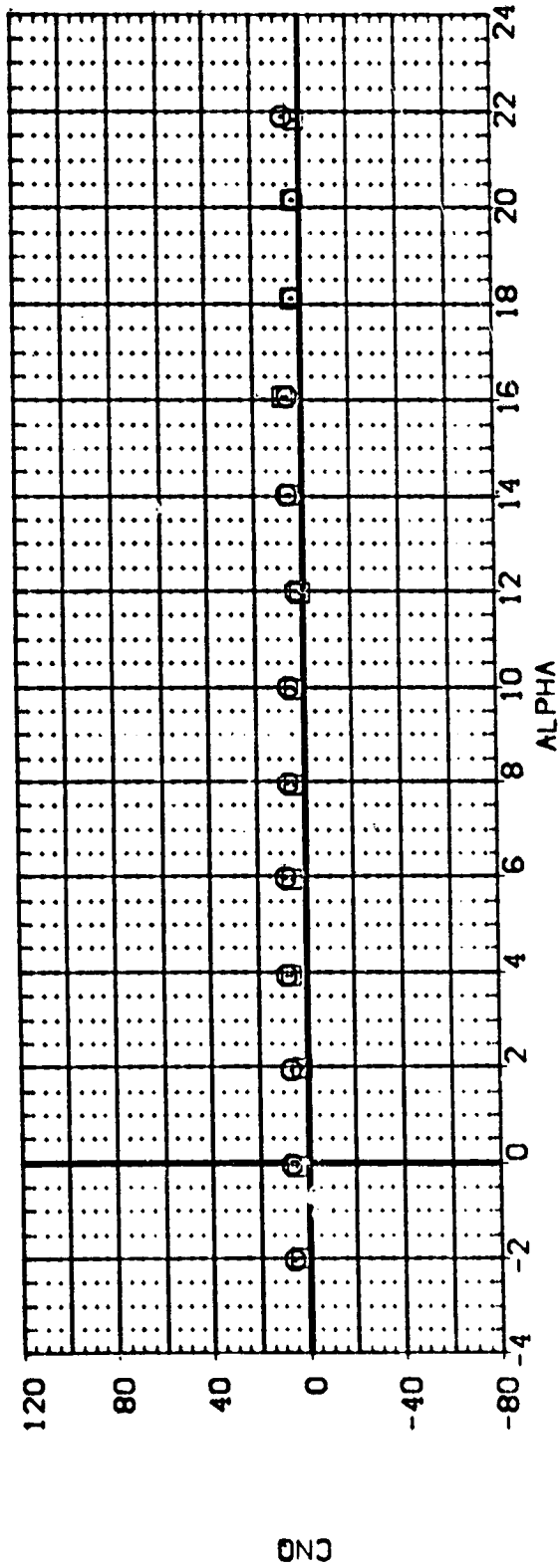


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH
 (M)MACH = .30 PAGE 56

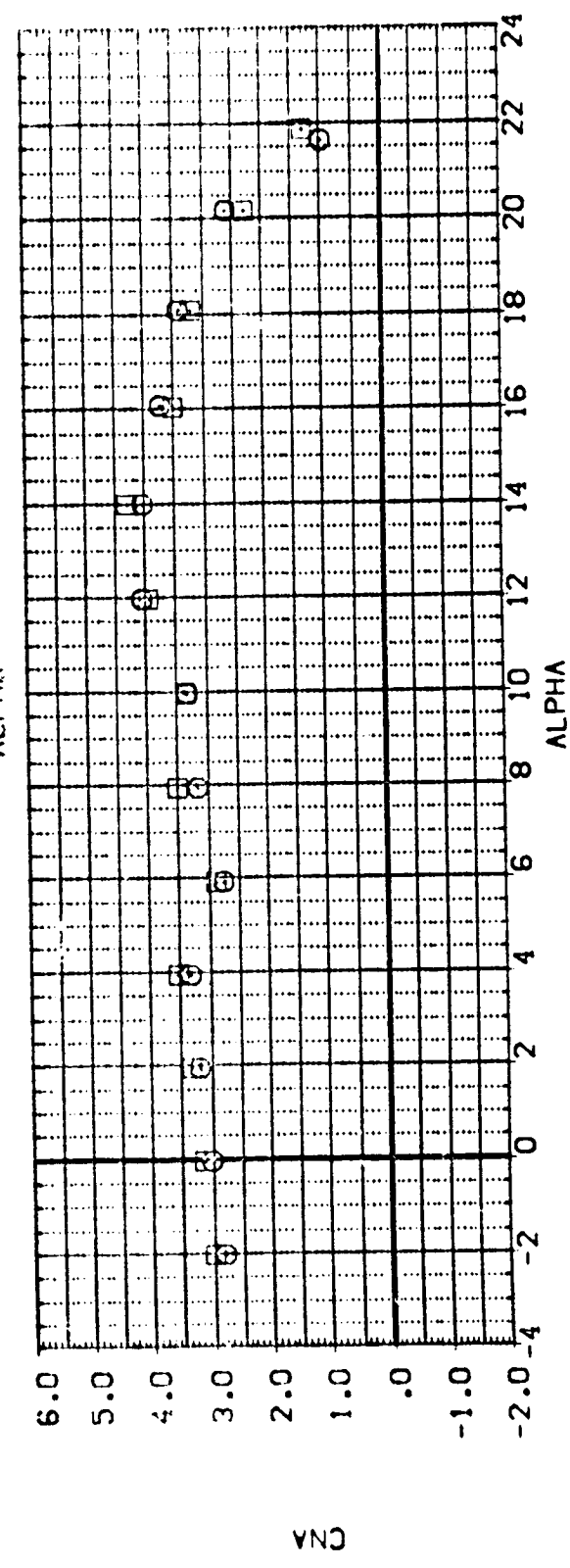
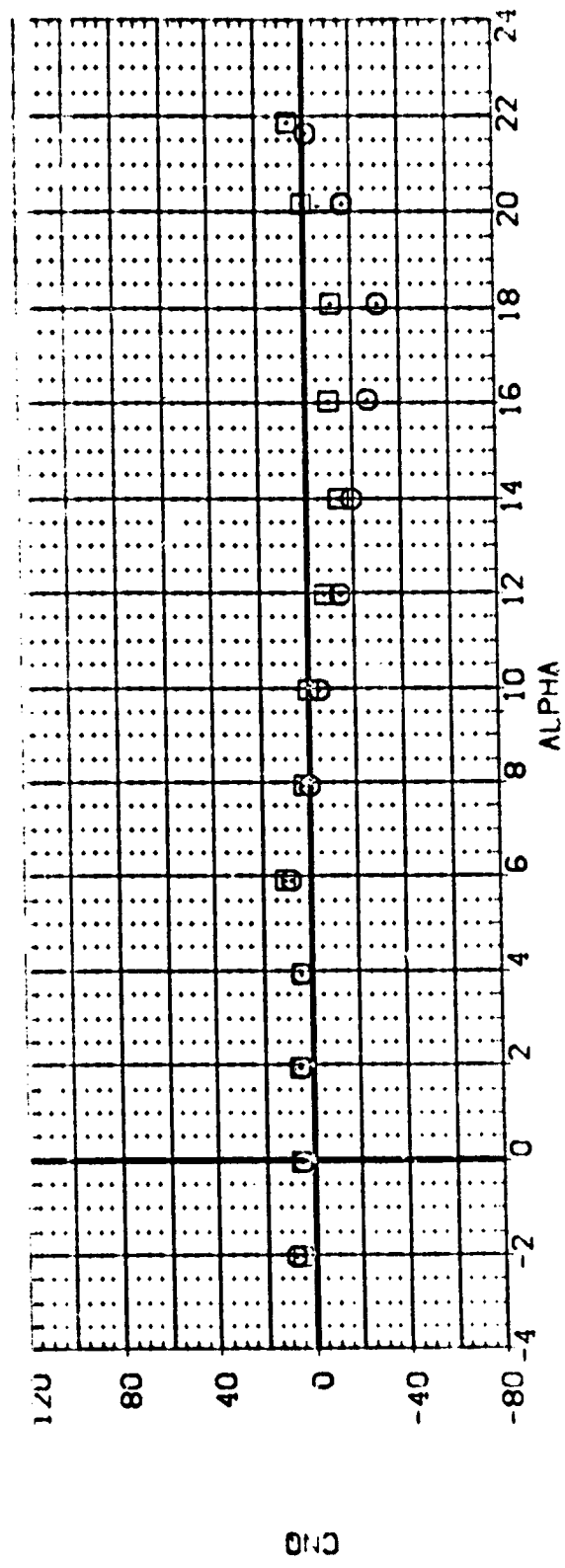
[illegible]

FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. IN PITCH

$$[E]_{\text{max}} = .80$$

CG-LOC ELEVIR 80FLAP RUOFLR
1.000 .000 10.000
1.000 .000 13.000 85.000

DATA SLT SYMBOL CONFIGURATION DESCRIPTION
(RPMPO4) LA-20, ROCKWELL 0898 DRB V/MOD NOSE (BVMVF)
(RPMPOS) LA-20, ROCKWELL 0898 DRB V/MOD NOSE (BVMVF)

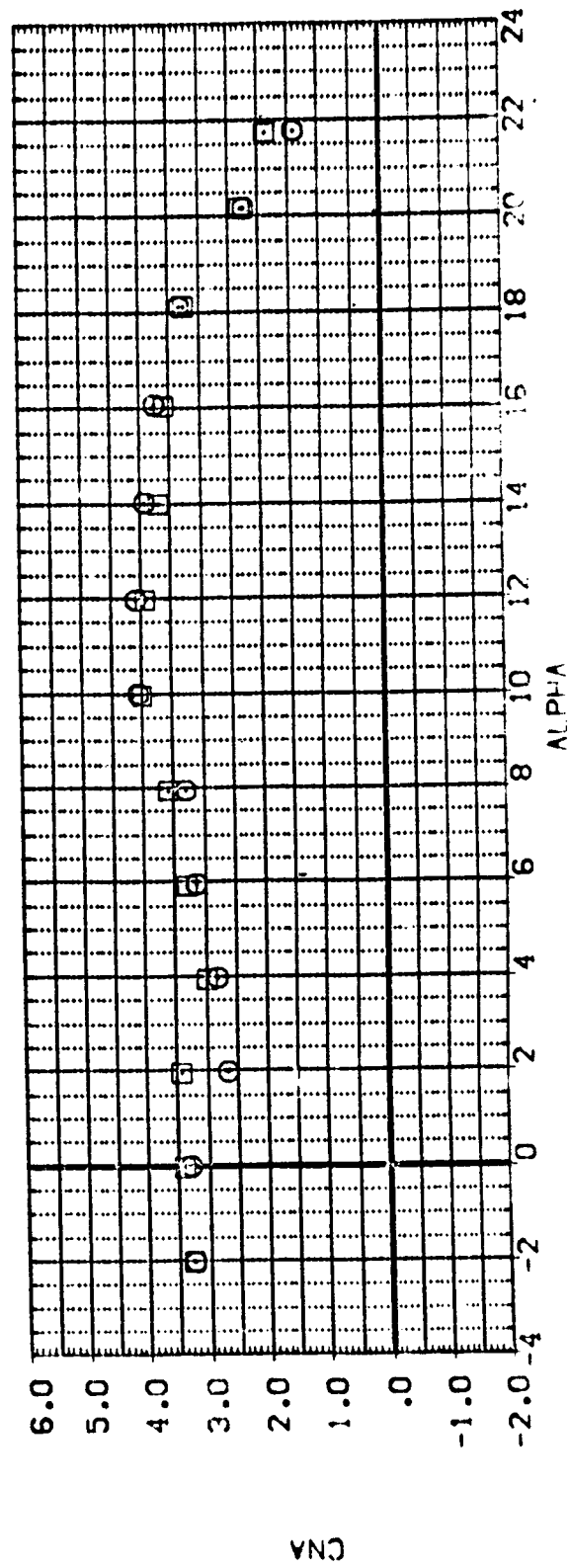
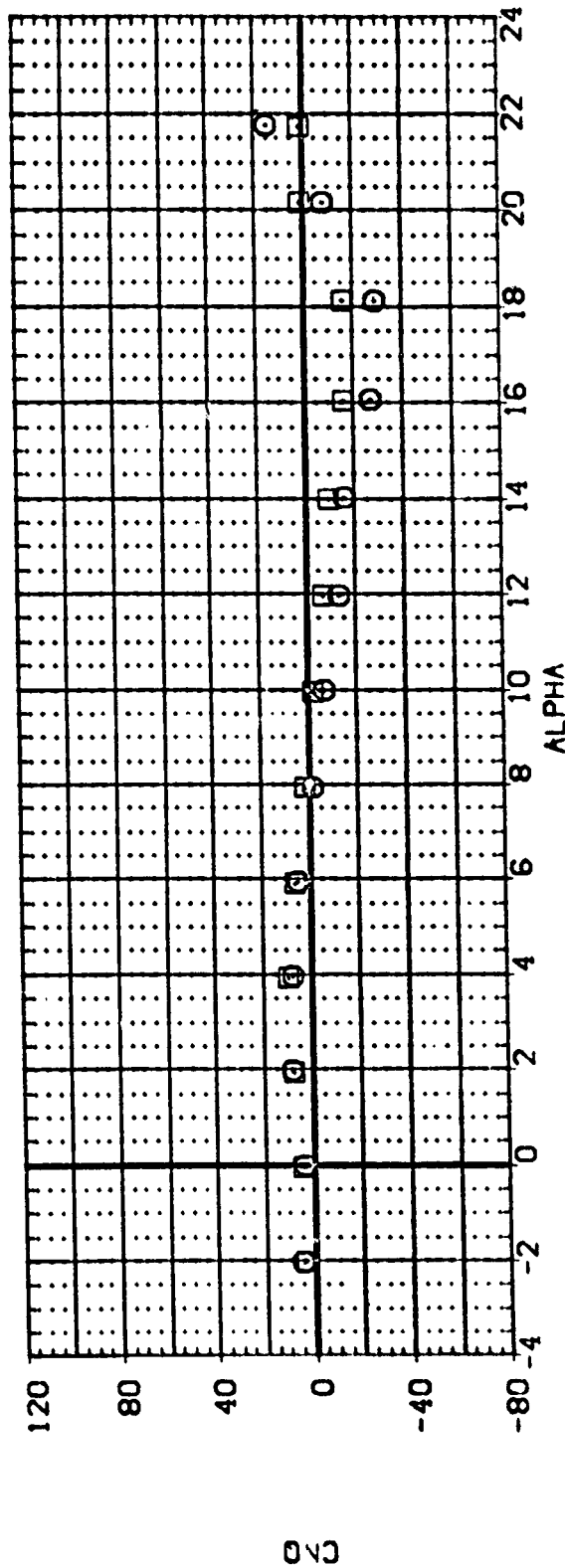


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH

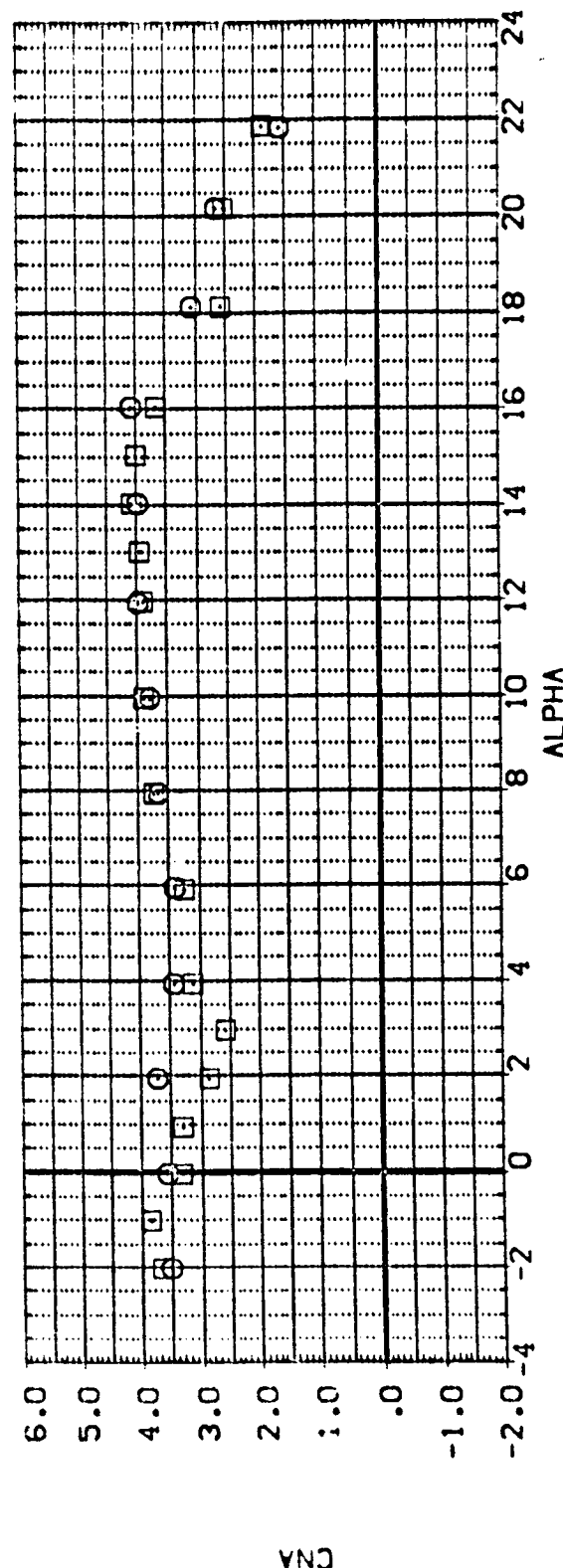
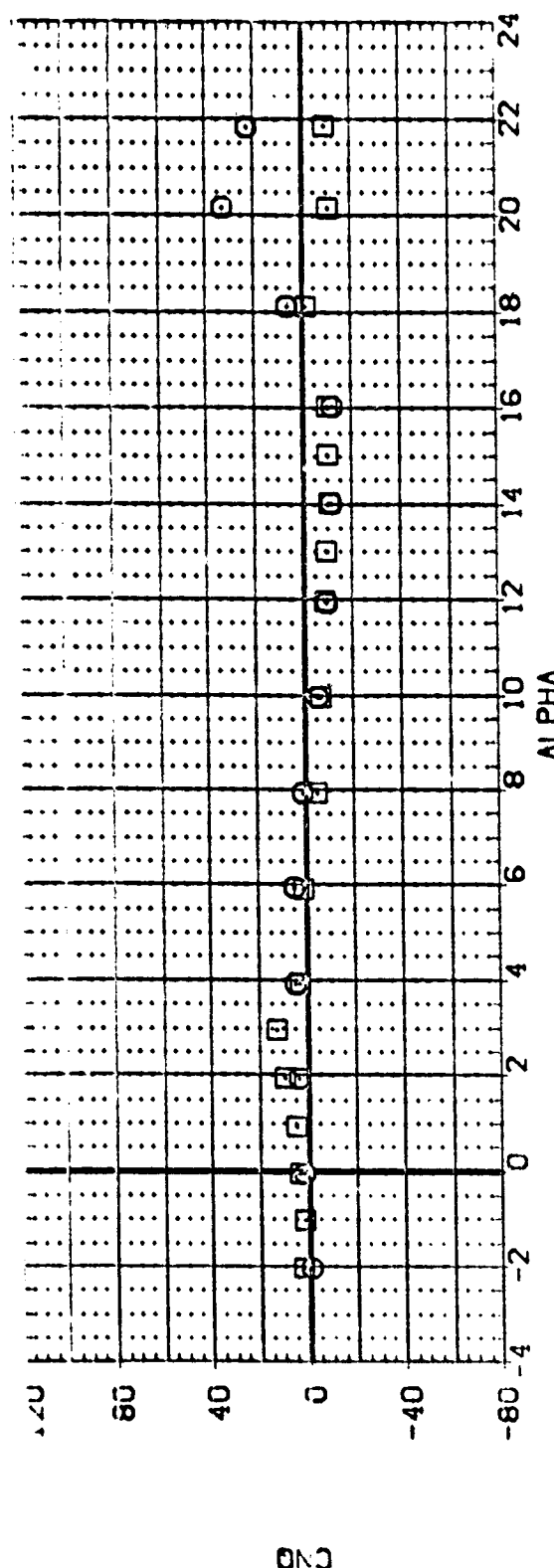
[illegible]

FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH

COMACH = .98

DATA SET SYMBOL: [RKMP04] [RKMP05] CONFIGURATION DESCRIPTION: LA-20, ROCKWELL 0858 DRB V/MOD NOSE (BVMF) LA-20, ROCKWELL 0858 DRB V/MOD NOSE (BVMF) CG-LOC: 1.000 1.000 ELEVIR: .000 .000 BOFLAP: .000 13.000 RUOFLR: 10.000 85.000

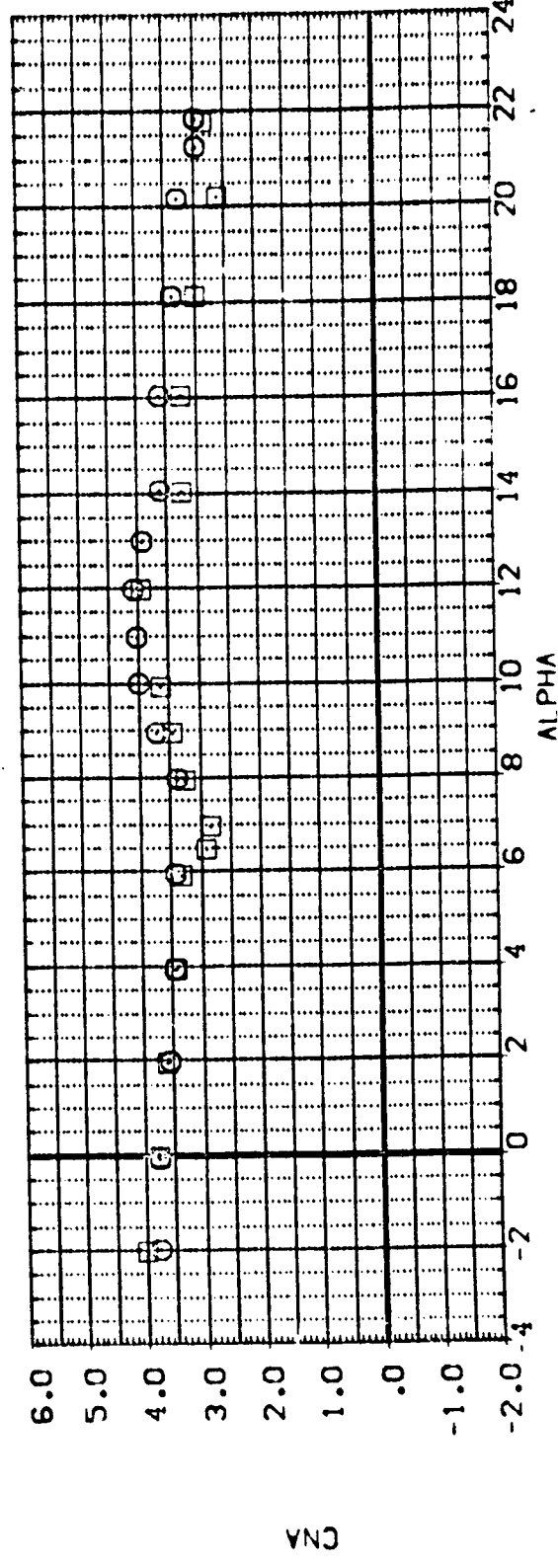
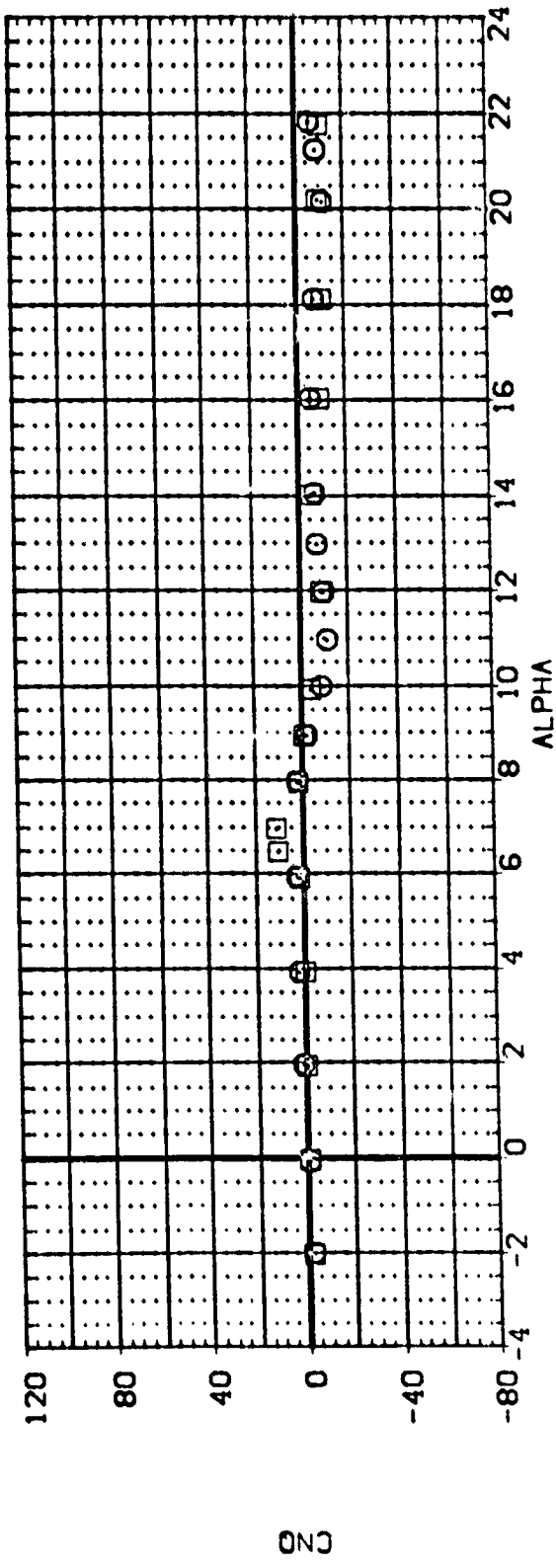


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH
CFMACH = 1.20 PAGE 60

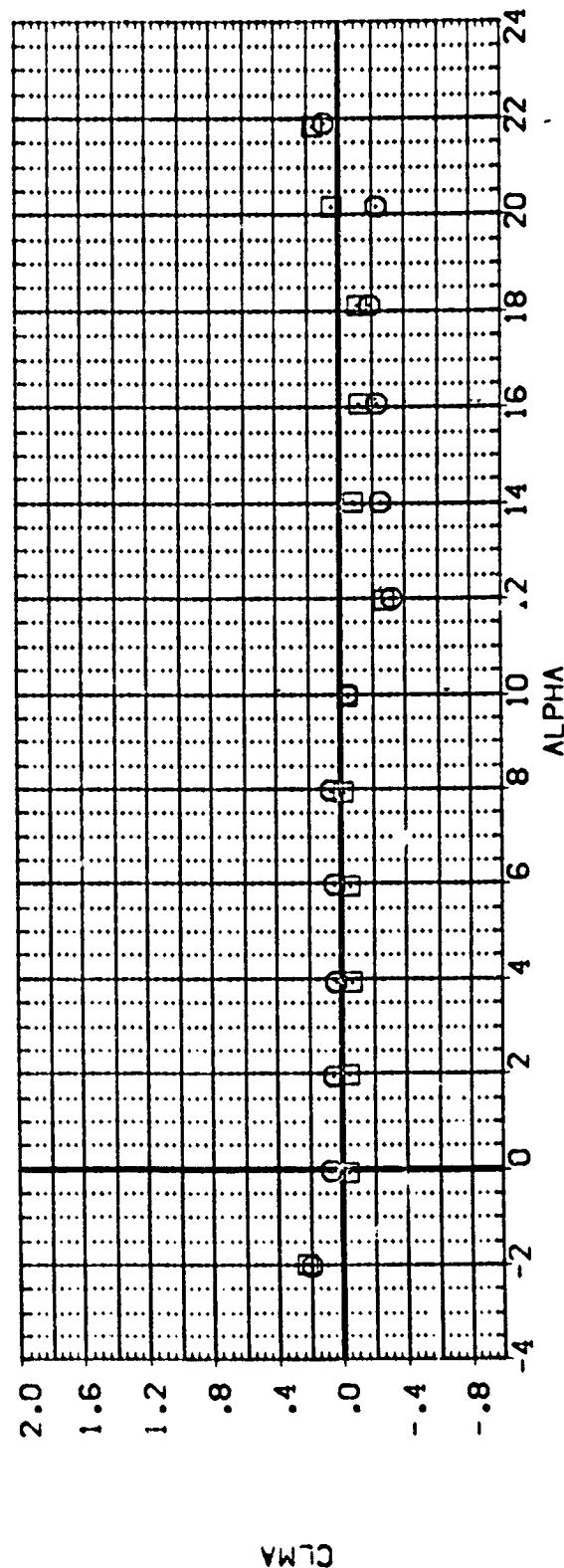
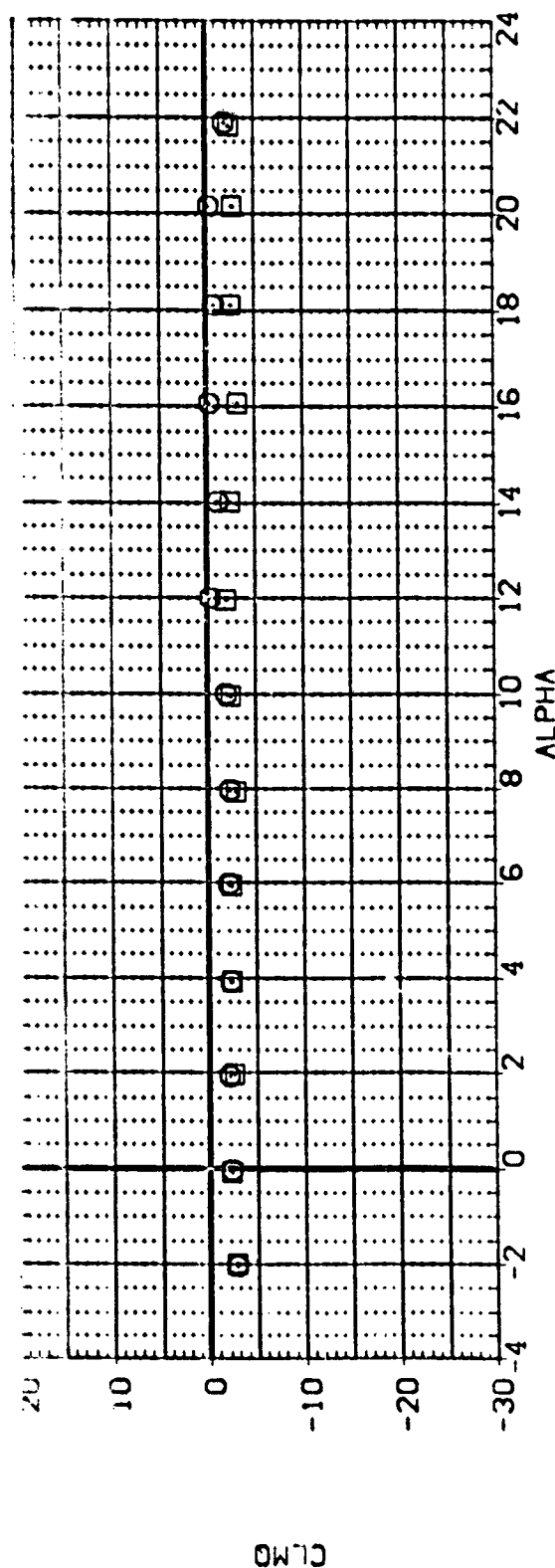
[illegible]

FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH

$$[A]_{MACH} = .30$$

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DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

(RPKPC1) LA-20, ROCKWELL O83B ORB V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

(RPKPO5) LA-20, ROCKWELL O83B ORB V/MOD NOSE (BVMF) 1.000 .000 13.000 65.000

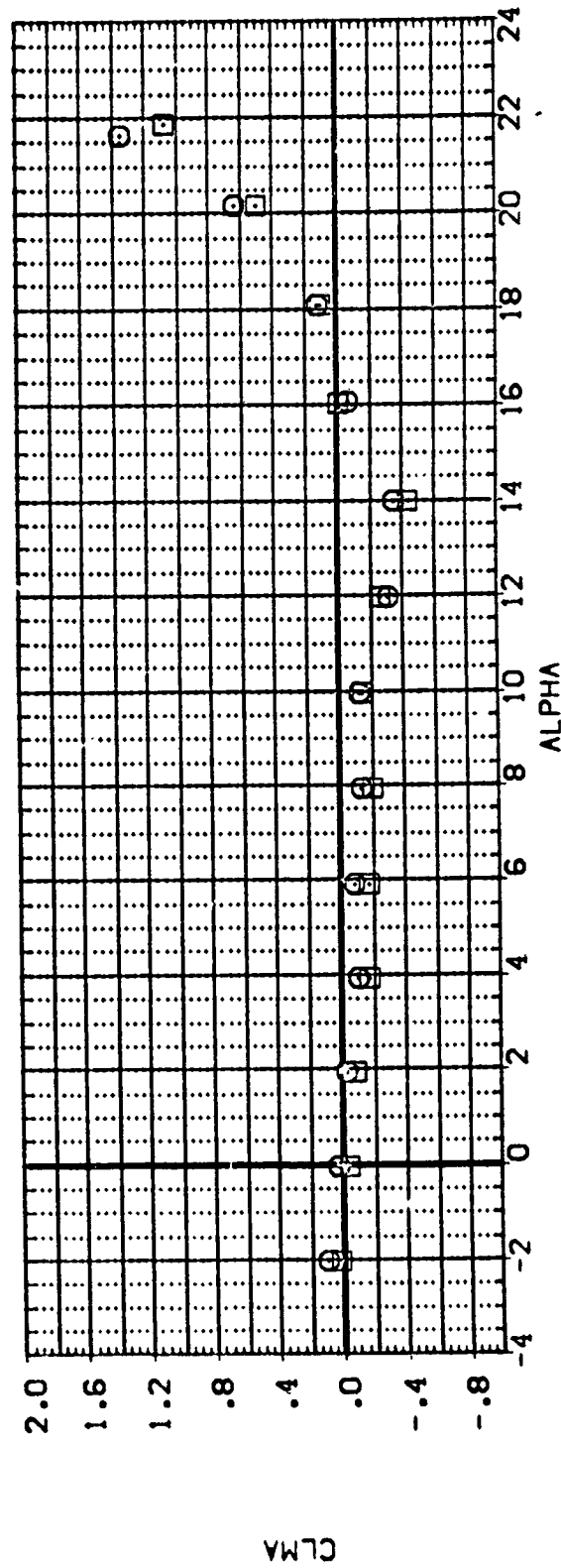
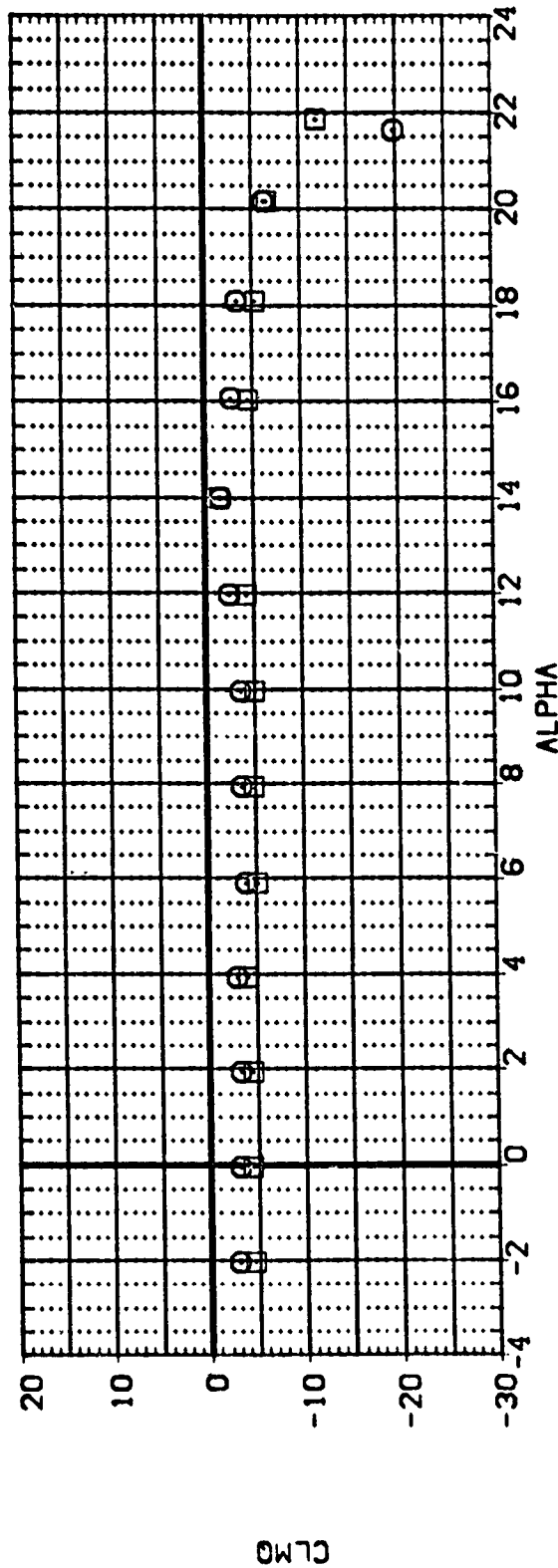


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH

(B)MACH = .80

DATA SET SYMBOL: CO-LOC ELEVTR BODY FLAP RUDDER
 LA-20: ROCKWELL C803 D98 41000 NOSE (B/MACH)
 LA-20: ROCKWELL C803 D98 41000 NOSE (B/MACH)

LA-20: ROCKWELL C803 D98 41000 NOSE (B/MACH)
 LA-20: ROCKWELL C803 D98 41000 NOSE (B/MACH)

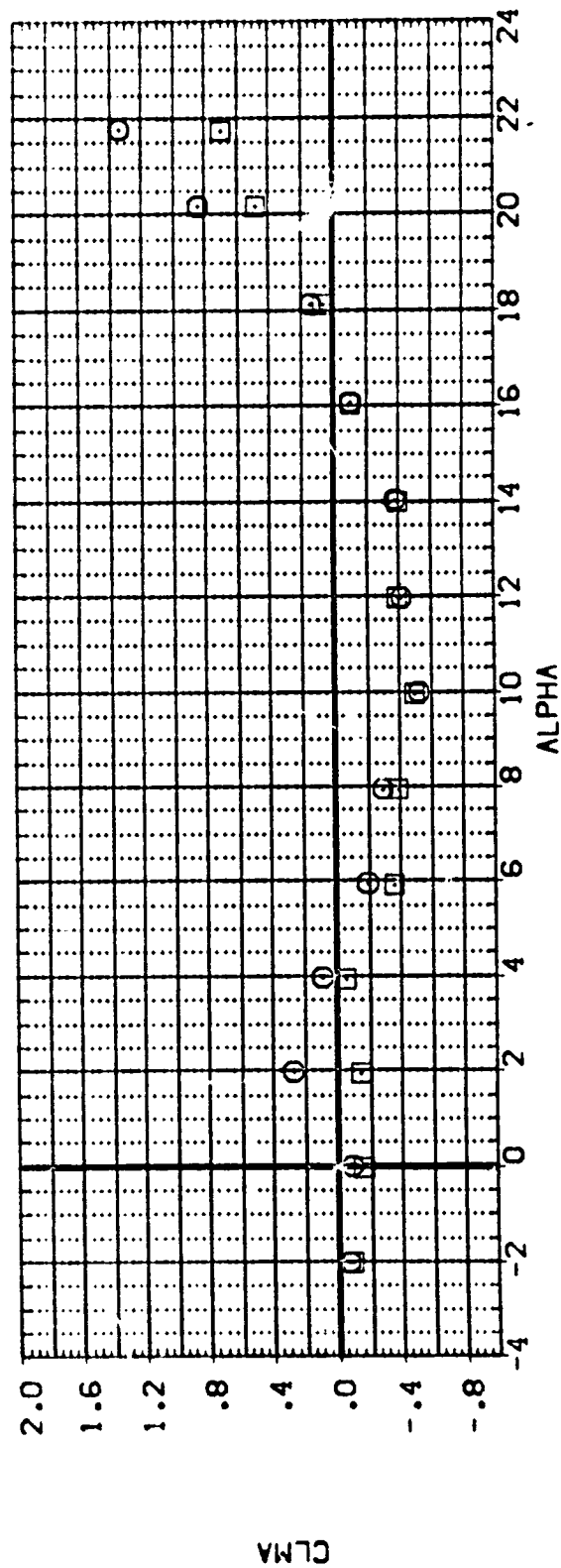
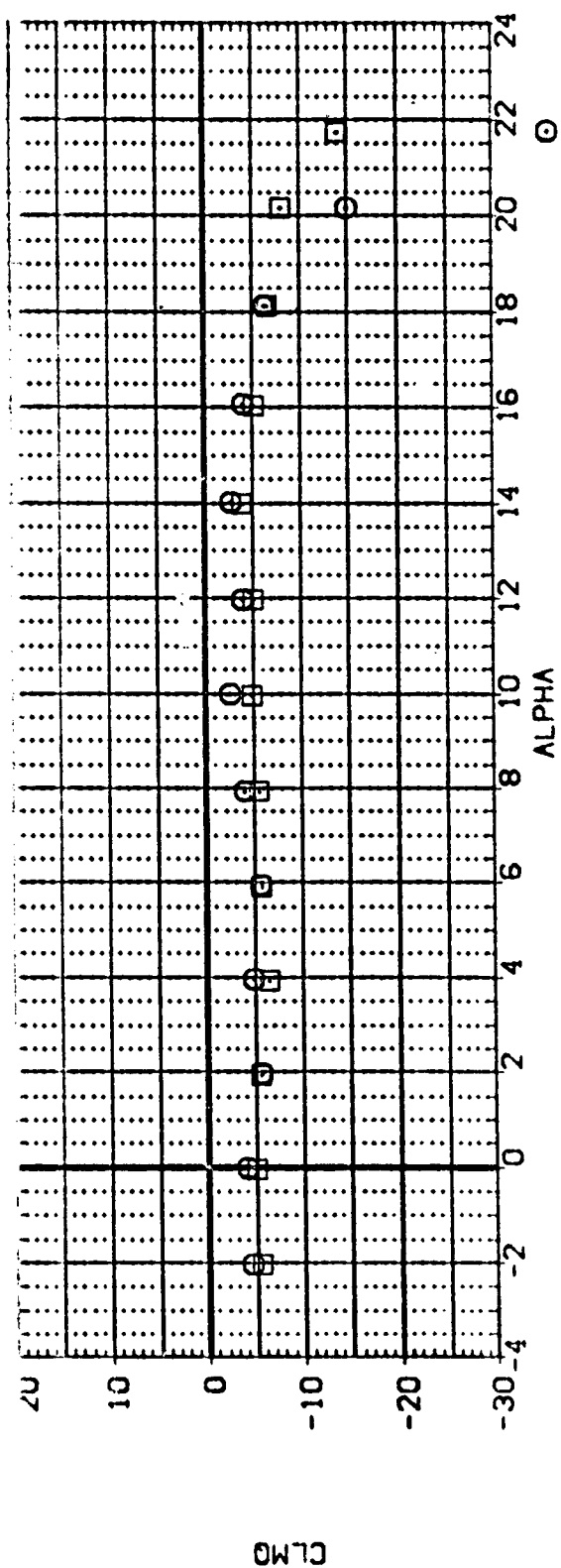


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH
 (CJMACH = .90)

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR

[RPKPCA] LA-20, ROCKWELL D89B DRB V/MOD NOSE (BVNF)

[RPKPCS] LA-20, ROCKWELL D89B DRB V/MOD NOSE (BVNF)

1.000 .000 .000 10.000

1.000 .000 13.000 85.000

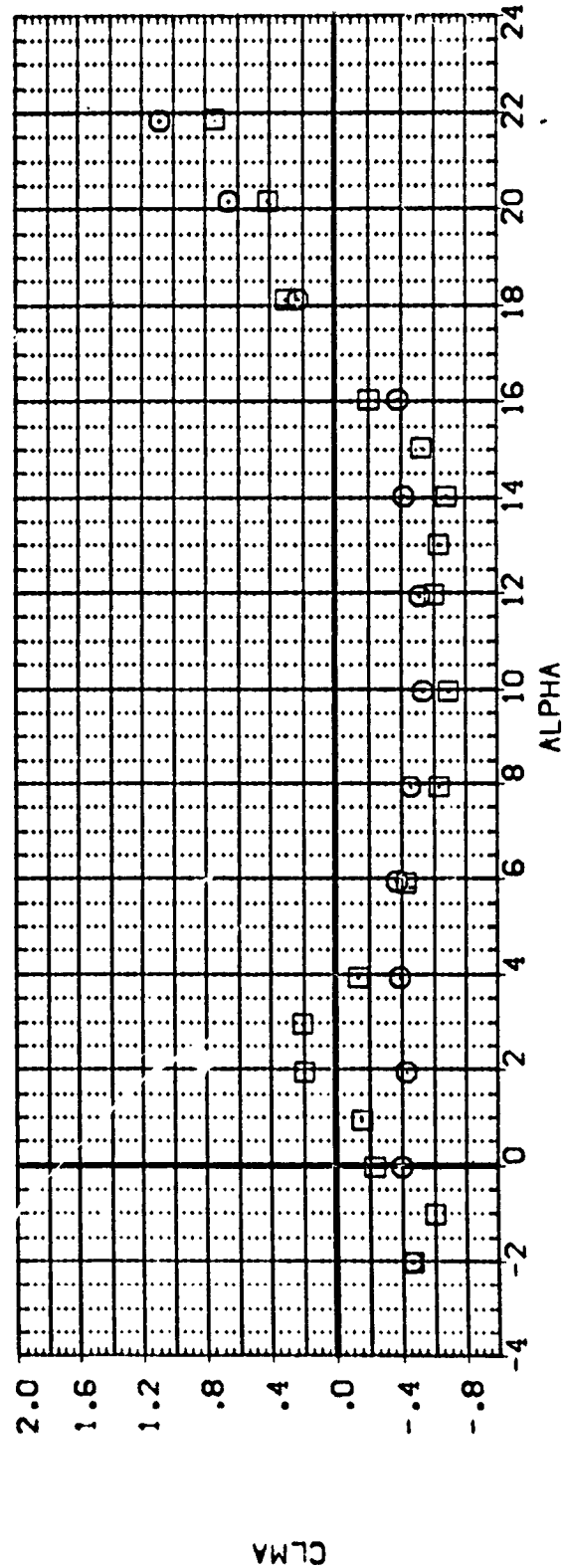
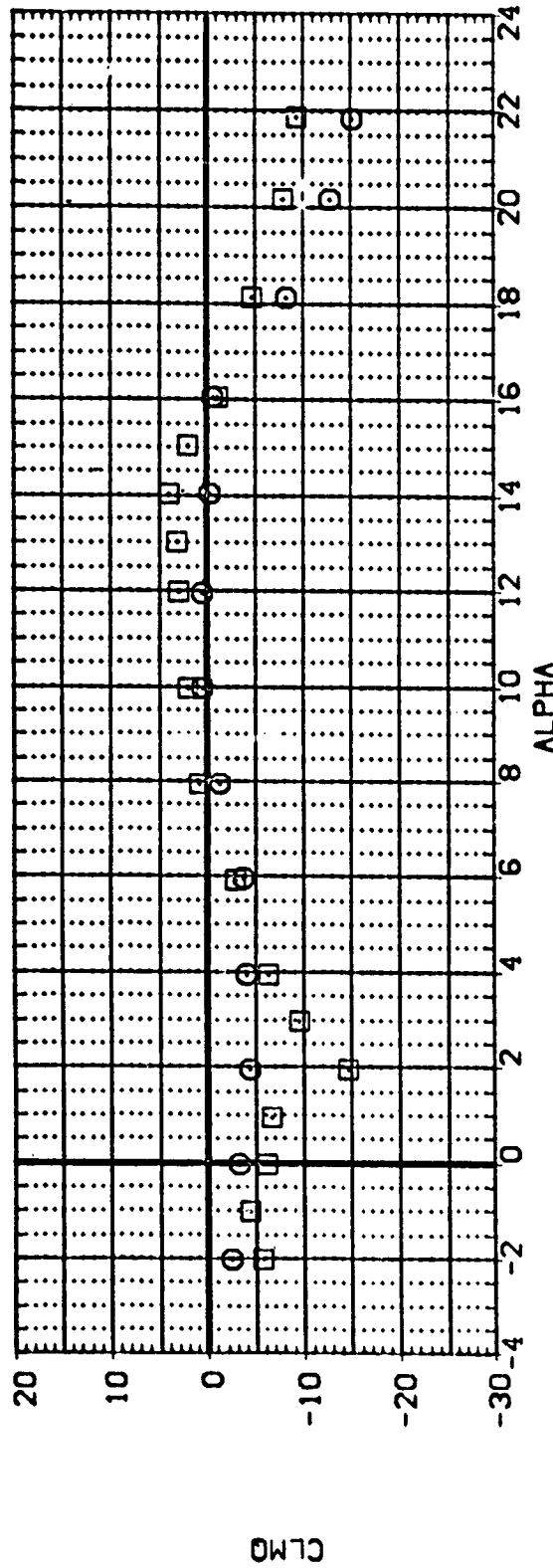


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH

[O]MACH = .98

DATA SET SYMBOL: CONFIGURATION DESCRIPTION: CG-LOC ELEVTR BODYLAP RUDDER

LA-20: ROCKWELL C89B CR8 V-MOD NOSE (BVMF) 1.000 0.000 0.000 10.000

LA-20: ROCKWELL C89B CR8 V-MOD NOSE (BVMF) 1.000 0.000 0.000 85.000

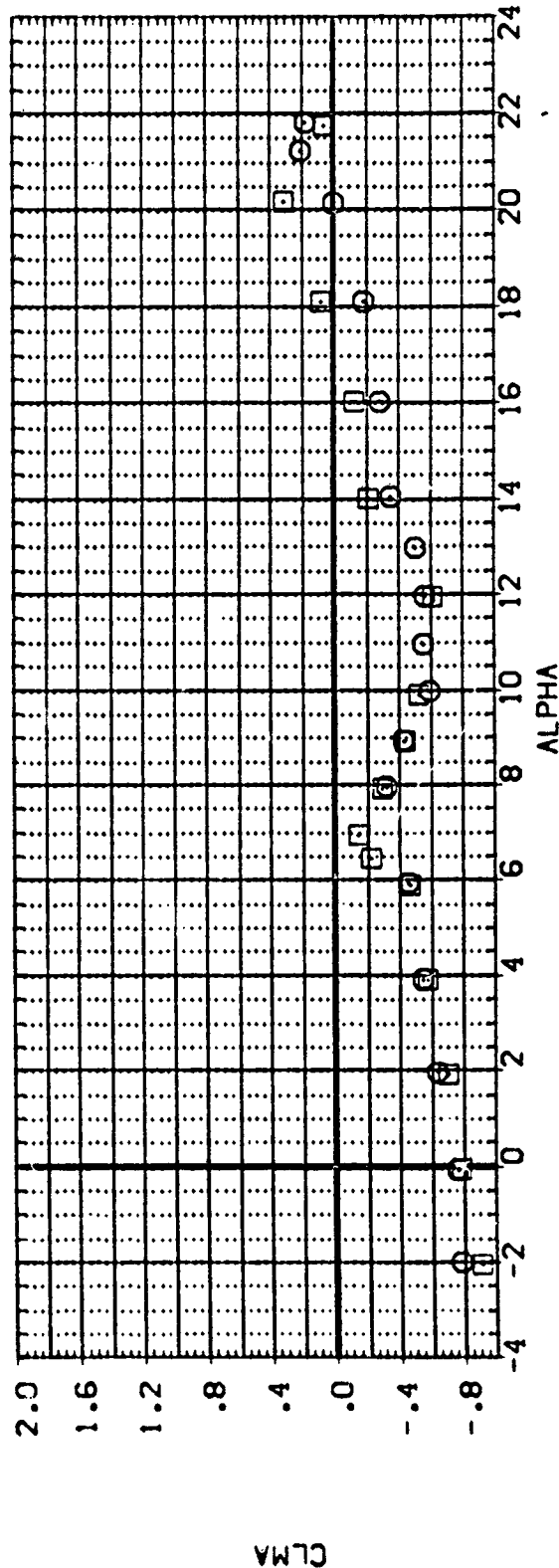
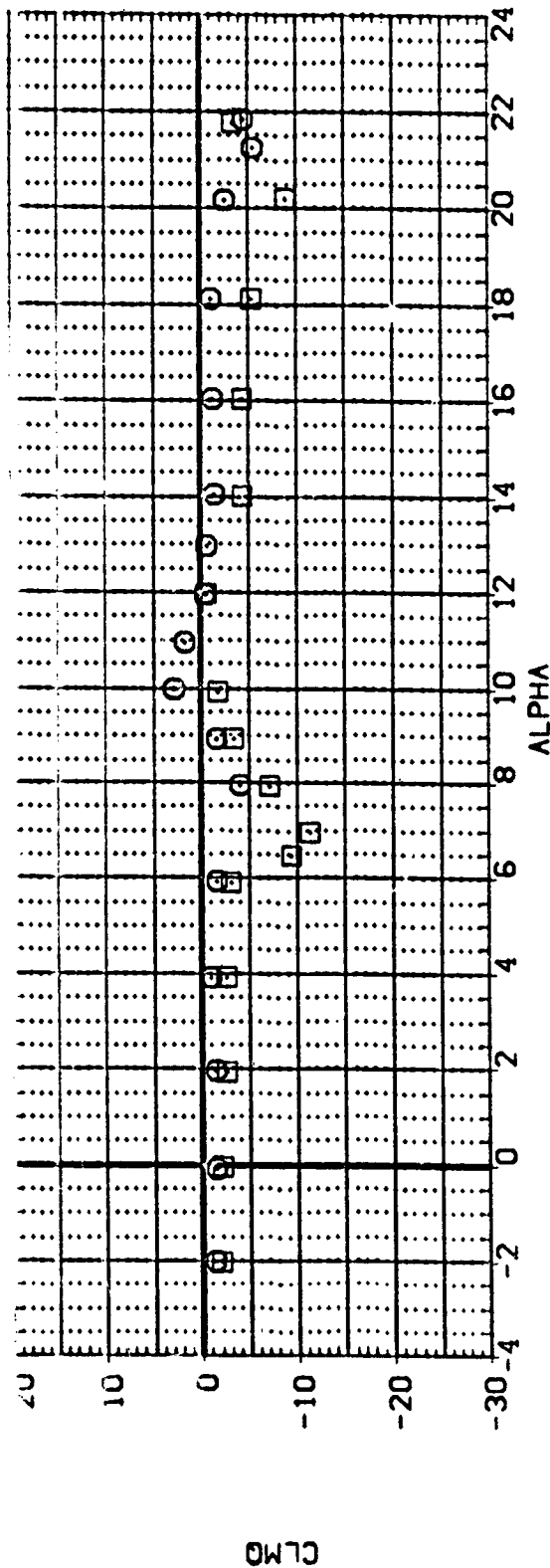


FIGURE 10. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN PITCH

(E)MACH = 1.20

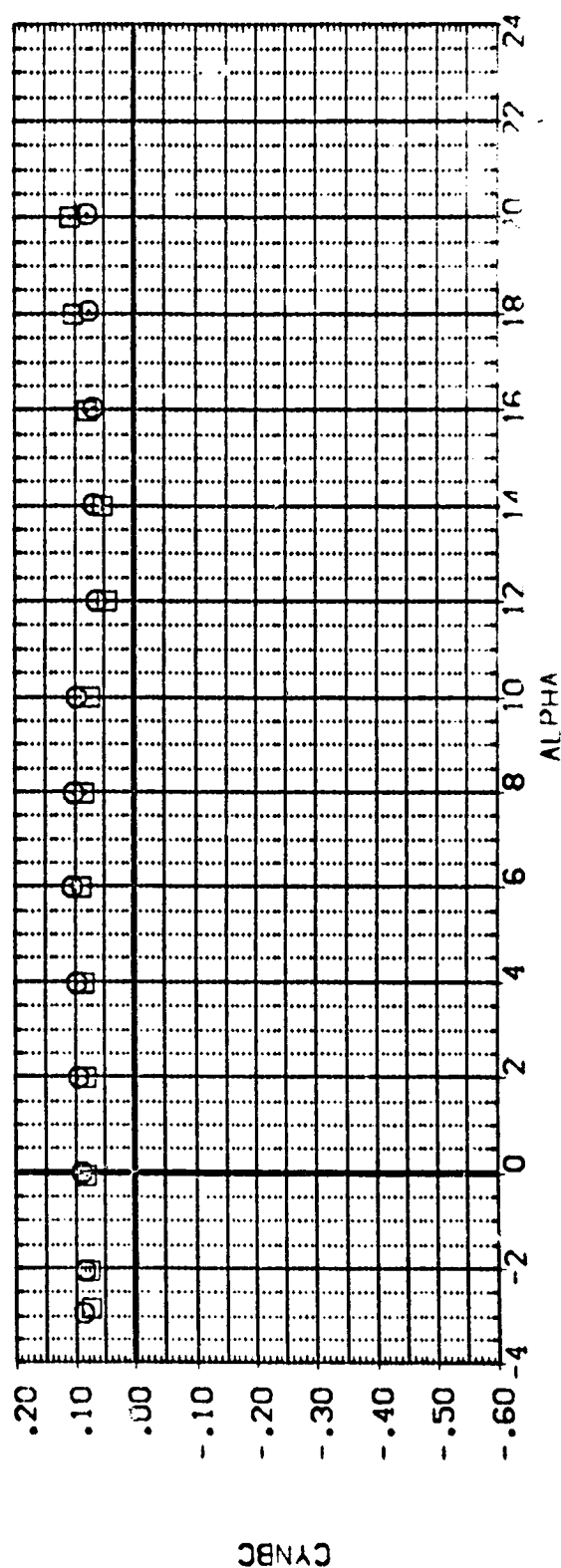
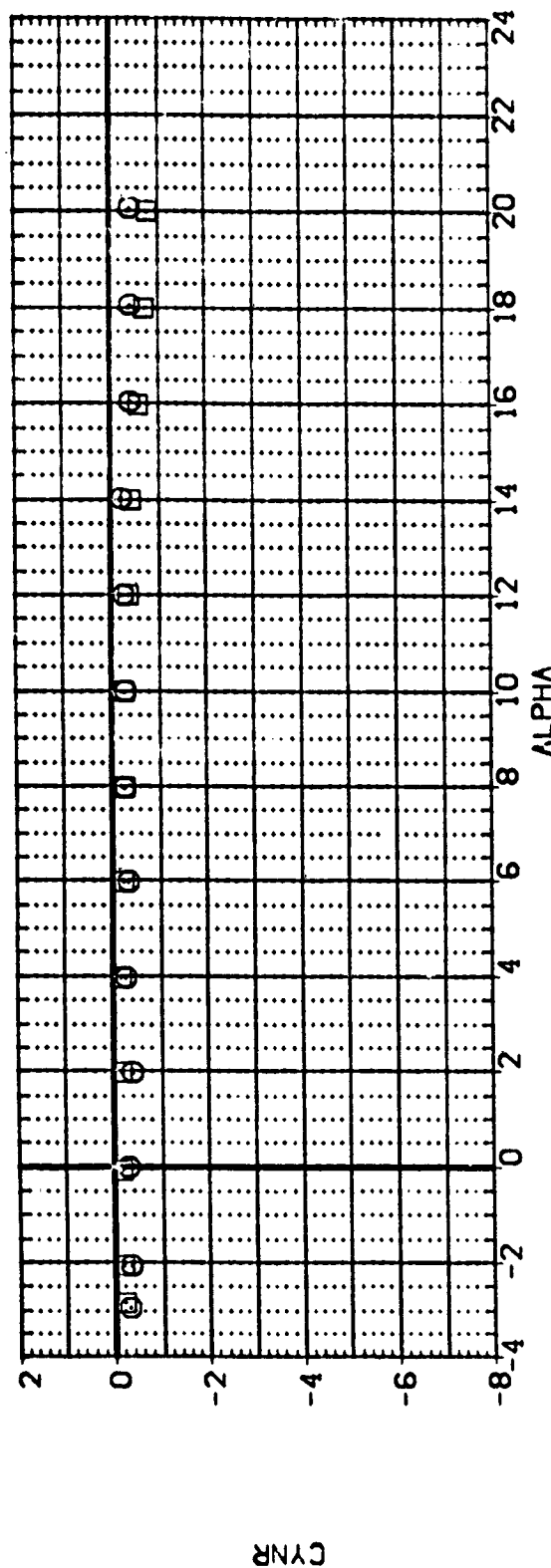
[illegible]

FIGURE 11. EFFECT OF BODY FLAP AND RUNNER FLAP ON PULLING FORCE. (1) 100% FLAP, (2) 50% FLAP, (3) 0% FLAP.

$$29,949.1 = 3.$$

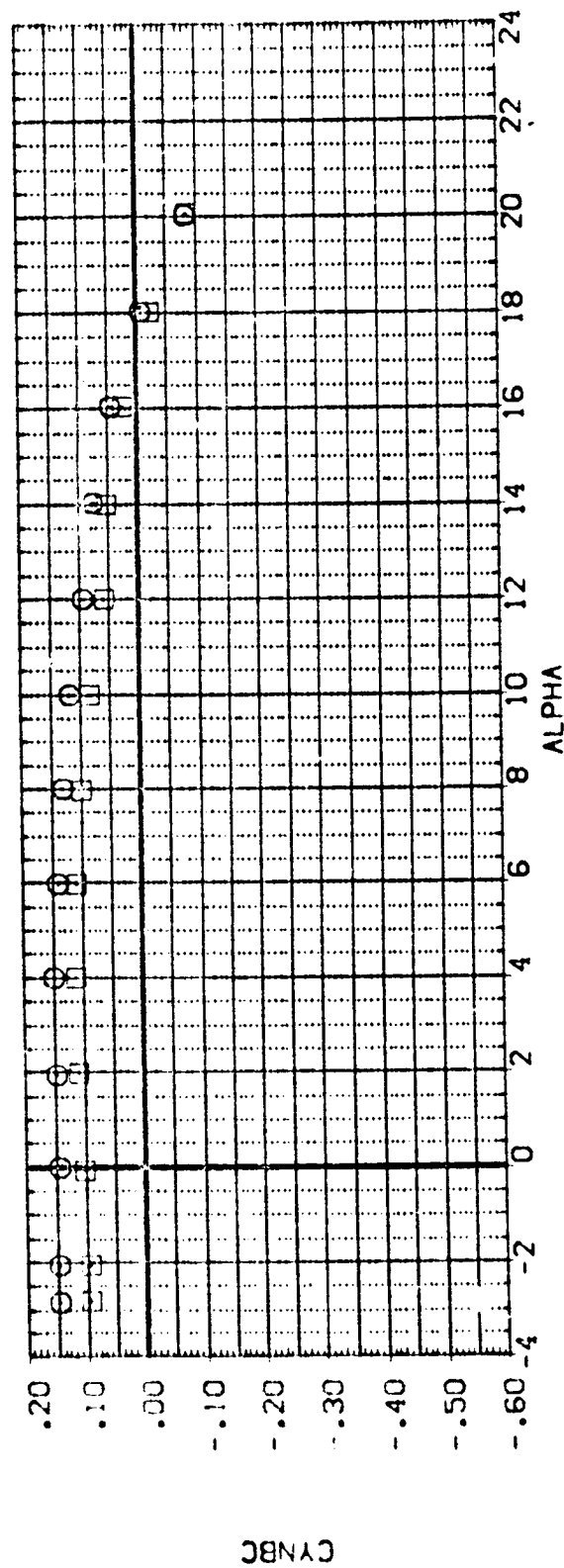
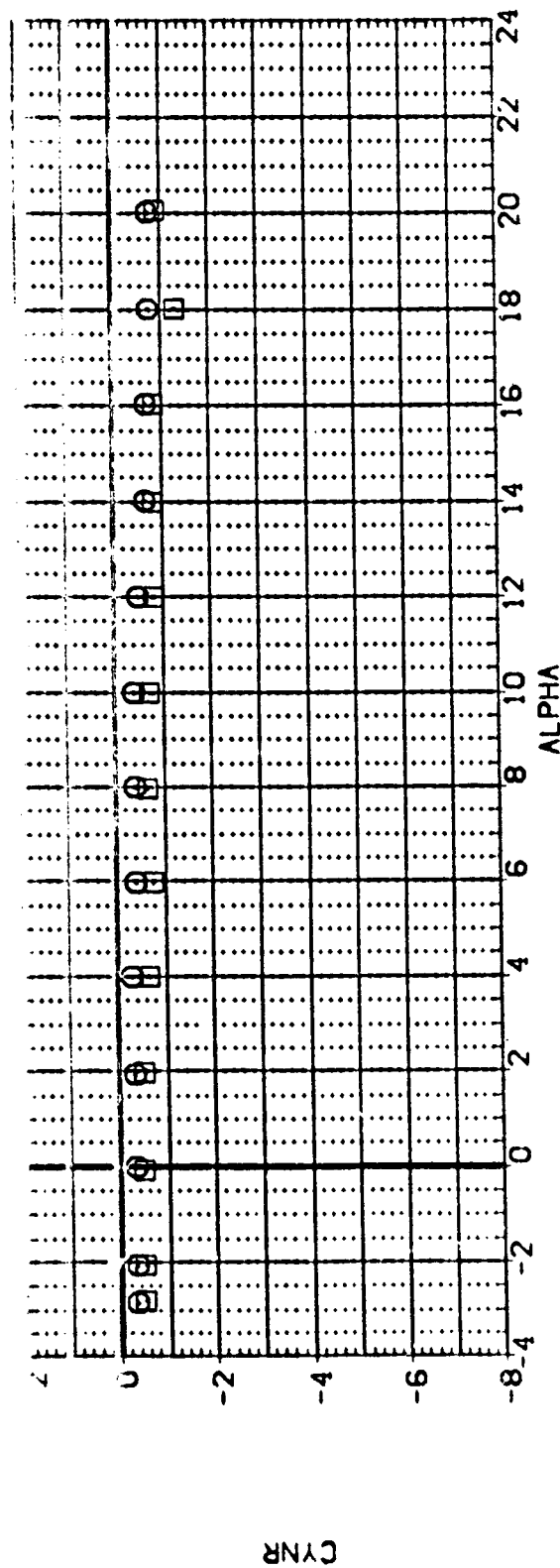


FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. IN YAW

$$[B]_{ACH} = .80$$

SCA 1000

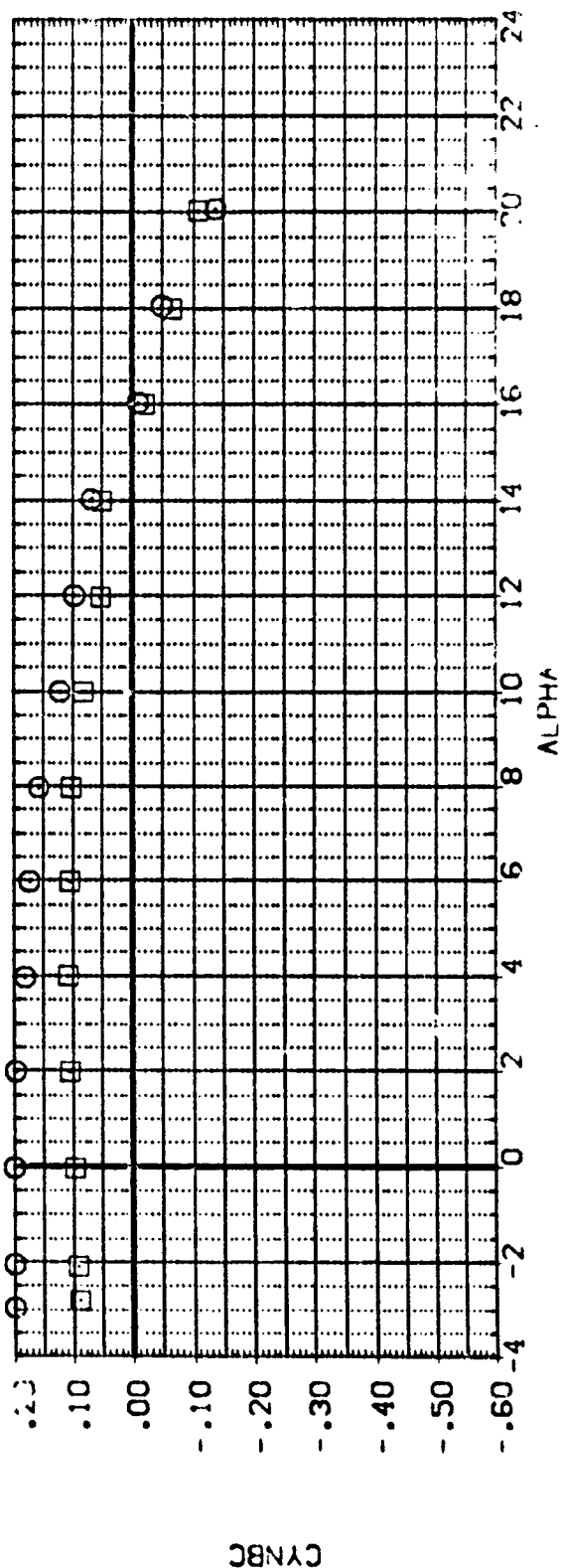
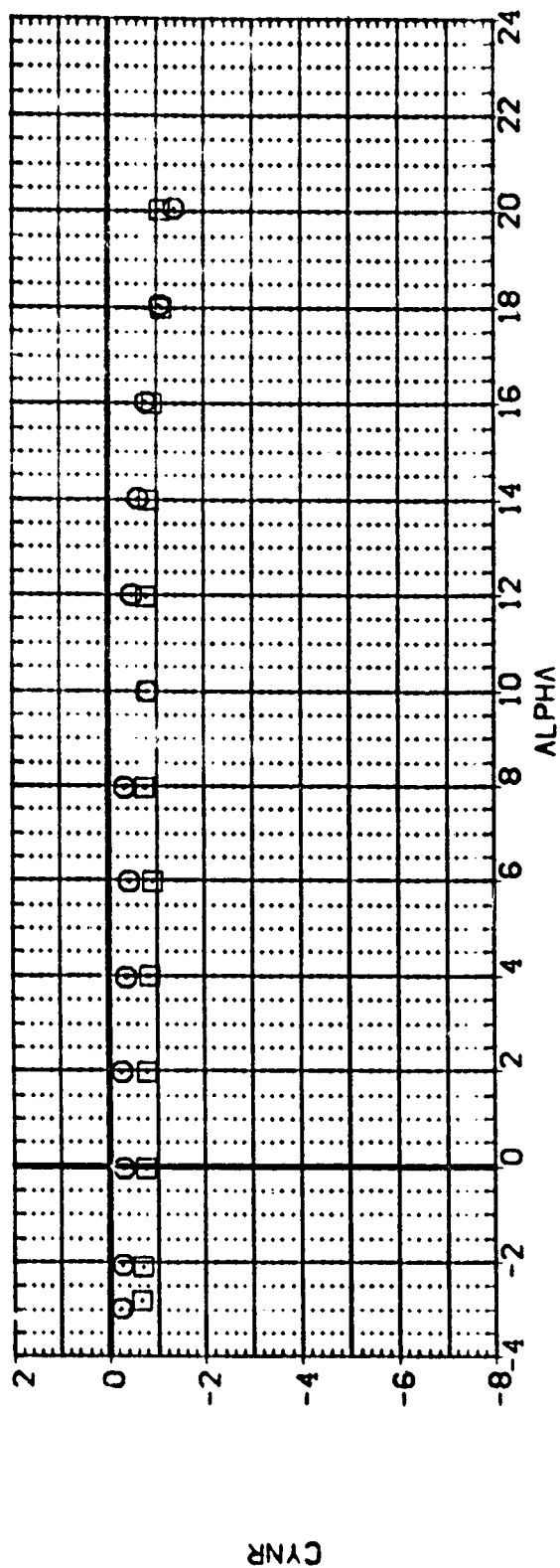


FIGURE 11. EFFECT OF BOOY FLAP AND KUDDEK FLAP ON DOWN DRAG PLANE COEFFICIENT

DATA SET SYMBOL. CONFIGURATION DESCRIPTION
 (RPAVC4) LA-20. ROCKWELL CRB 0893 V/HOO. NOSE (BVVNF)
 (RPAVC5) LA-20. ROCKWELL CRB 0893 V/HOO. NOSE (BVVNF)

CG-LDC ELEVTR BOFLAP ROFLR
 1.000 .000 .000 10.000
 1.000 .000 13.000 85.000

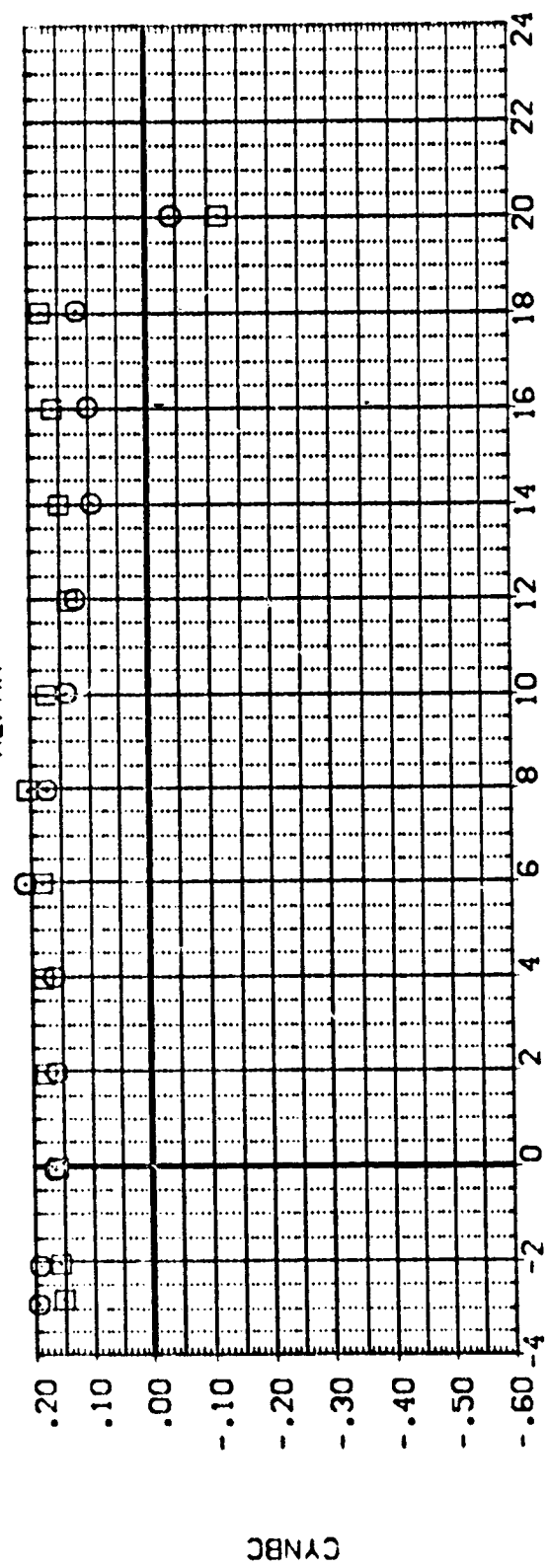
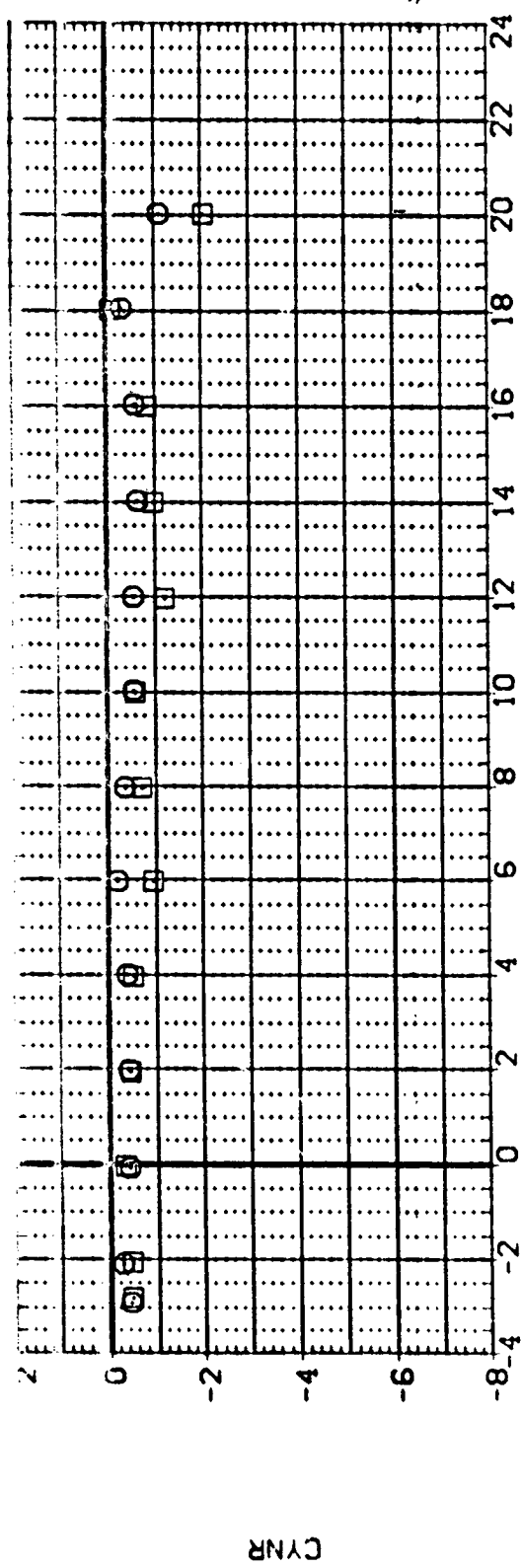


FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN YAW

DATA SET: DYN5C CONFIGURATION: 2. DESCRIPTION: 305.0
 (RKNYCA) [] LA-20, ROCKWELL CRB 0898 V/MOD. NOSE (BNVNF) 1.000 .000 300 10.000
 (RKNYCS) [] LA-20, ROCKWELL CRB 0898 V/MOD. NOSE (BNVNF) 1.000 .000 13.000 85.000

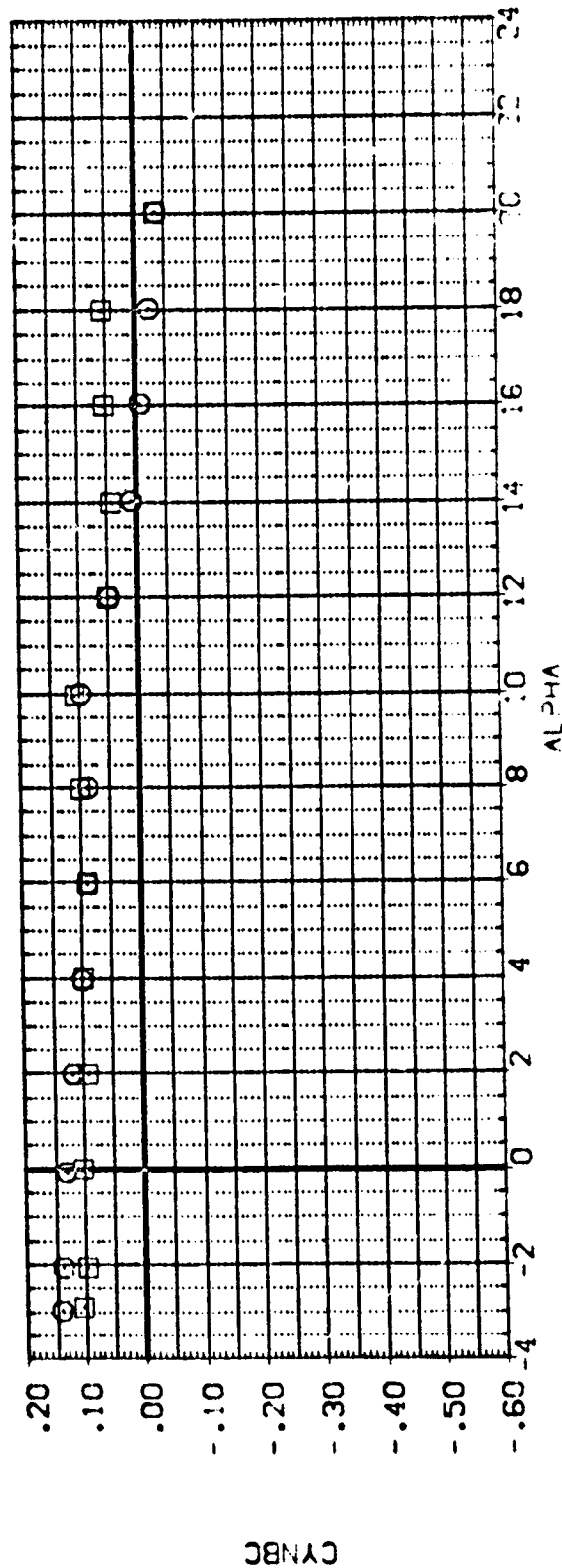
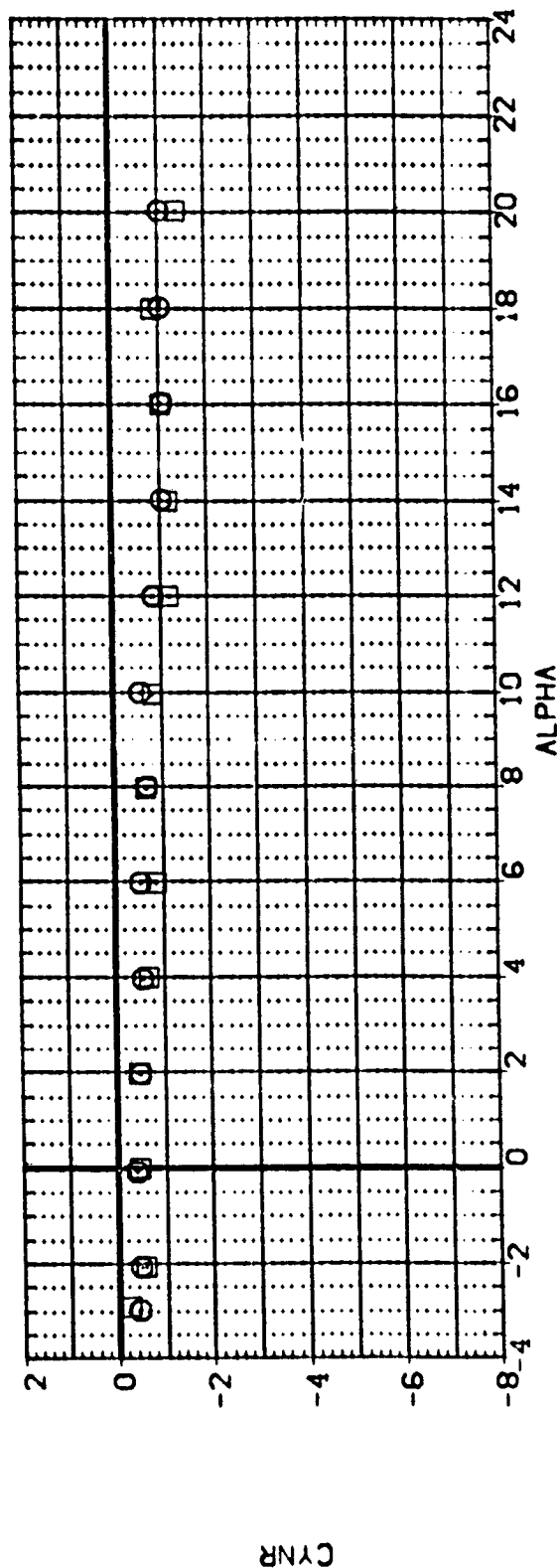


FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYNAMIC STABILITY PARAM. IN YAW

REYNOLDS = 1.20

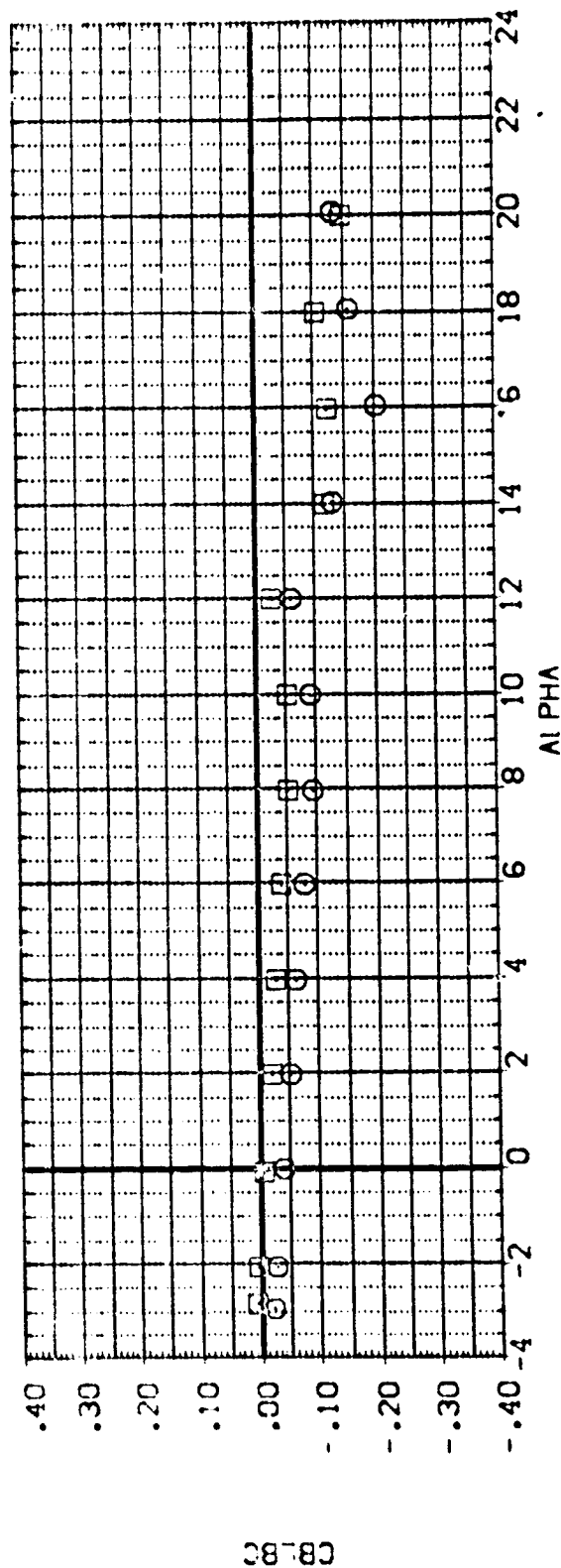
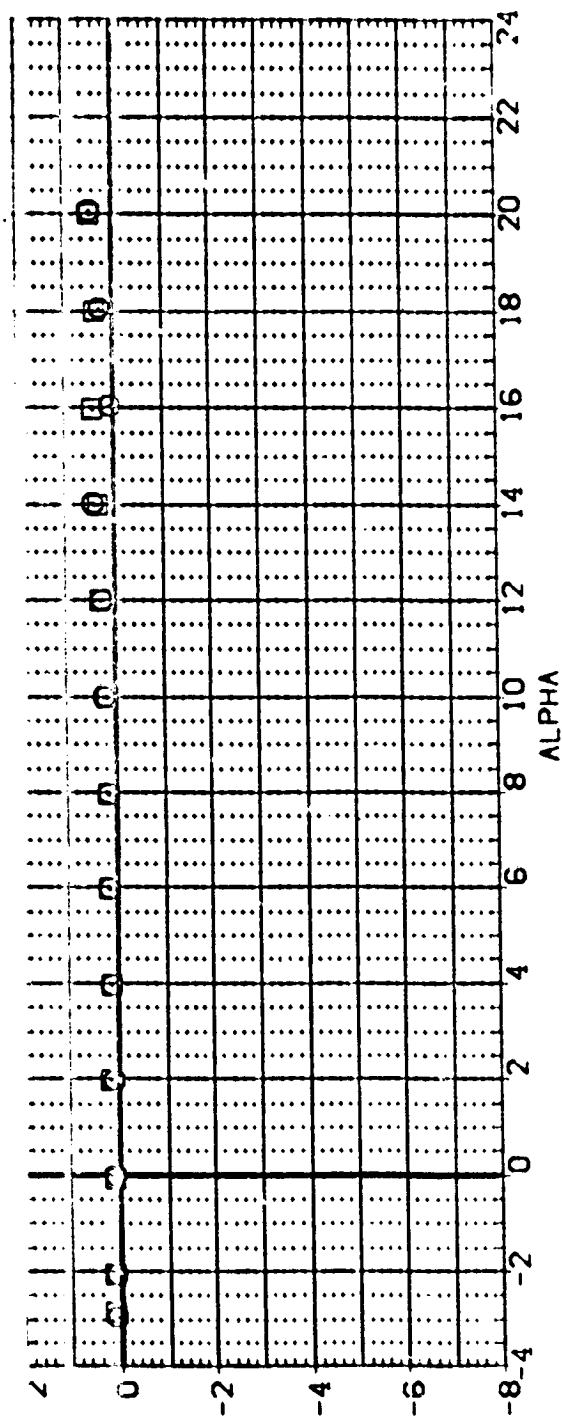


FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARM. IN YAW

(C)MAG: = .30

PAGE:

DATA 34 30000 CONF IGURE 10N DESCRIP 10:
 [RPMVCS] [RPMVCS] LA-20, ROCKWELL DRB 0893 V/MOD, NOSE (BVMF) 1.000 10.000
 [RPMVCS] [RPMVCS] LA-20, ROCKWELL DRB 0893 V/MOD, NOSE (BVMF) 1.000 13.000 85.000

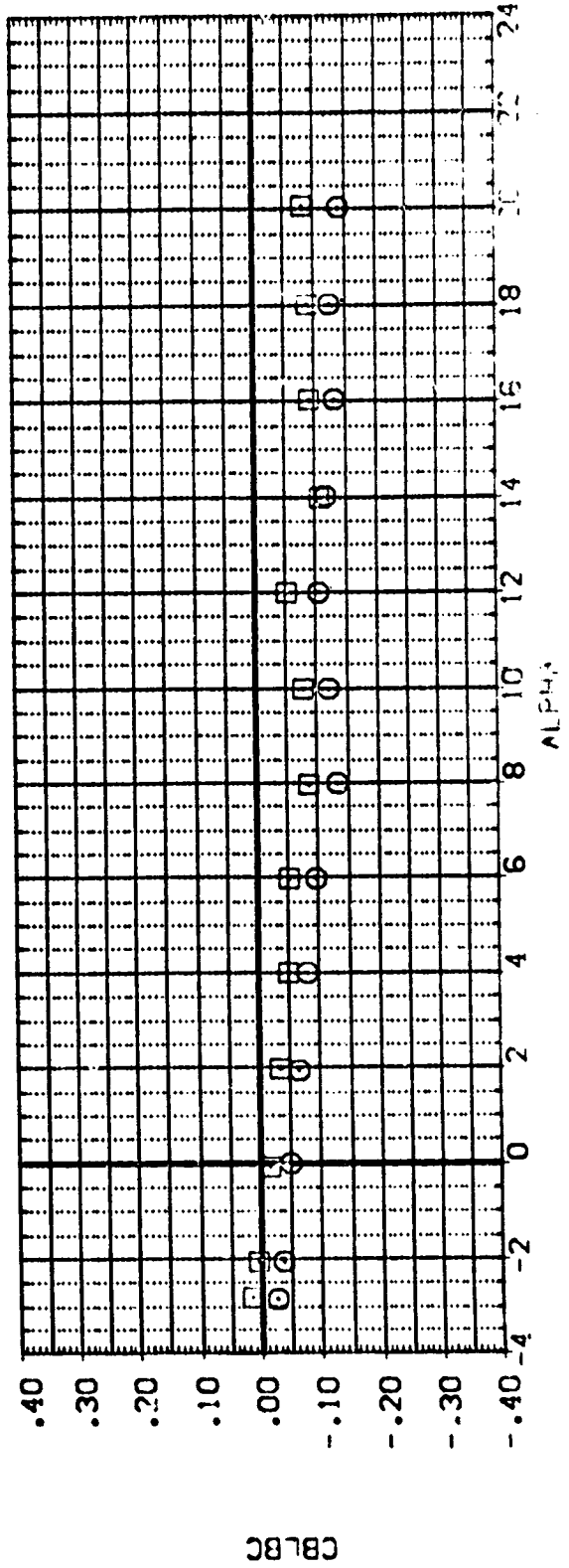
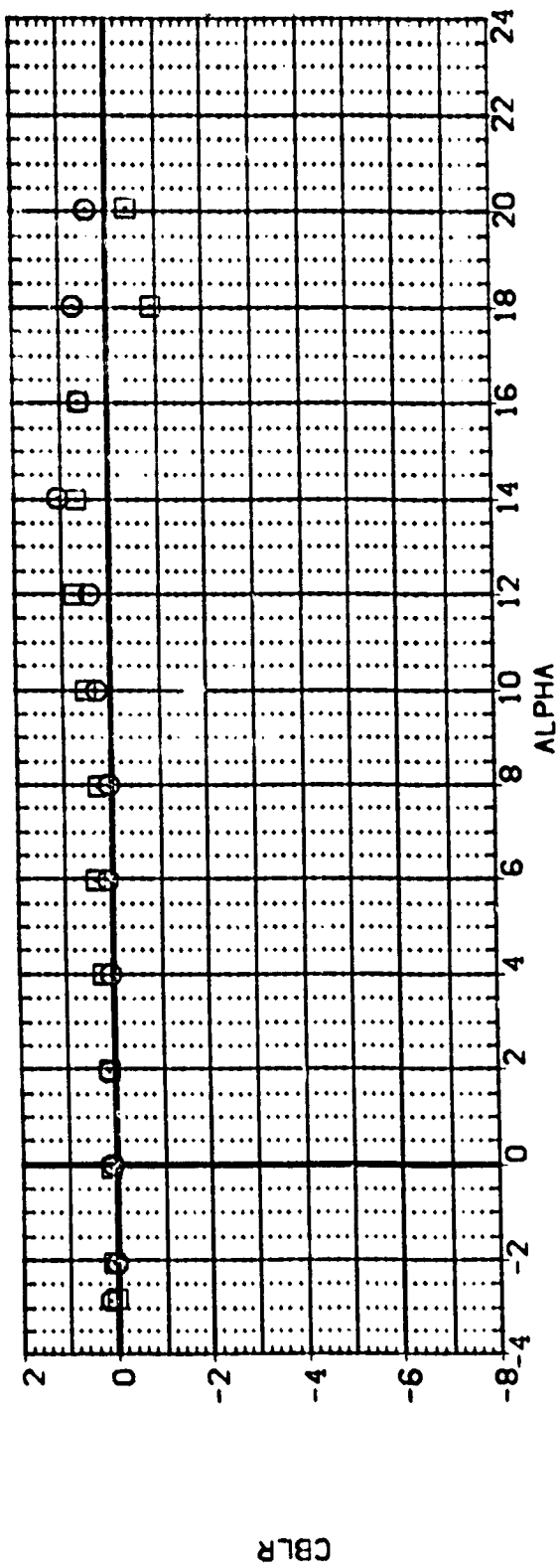


FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLAP ON DYNAMIC PRESSURE COEFFICIENT
 (B/MACH = .80)

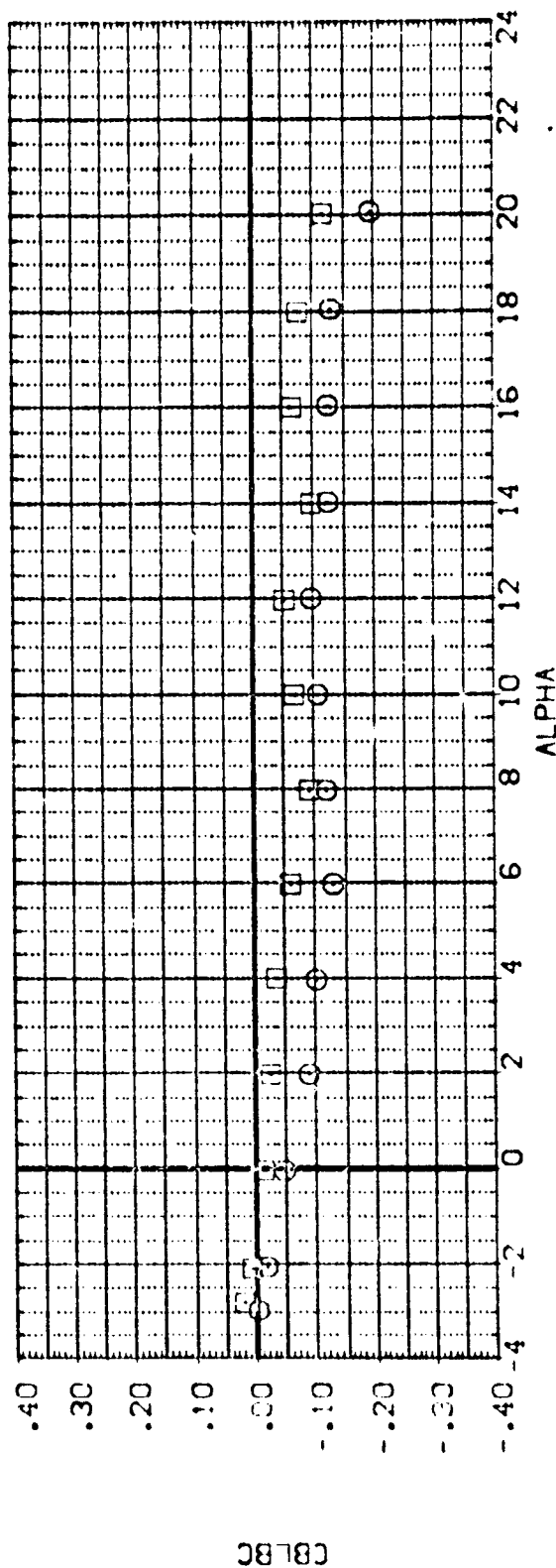
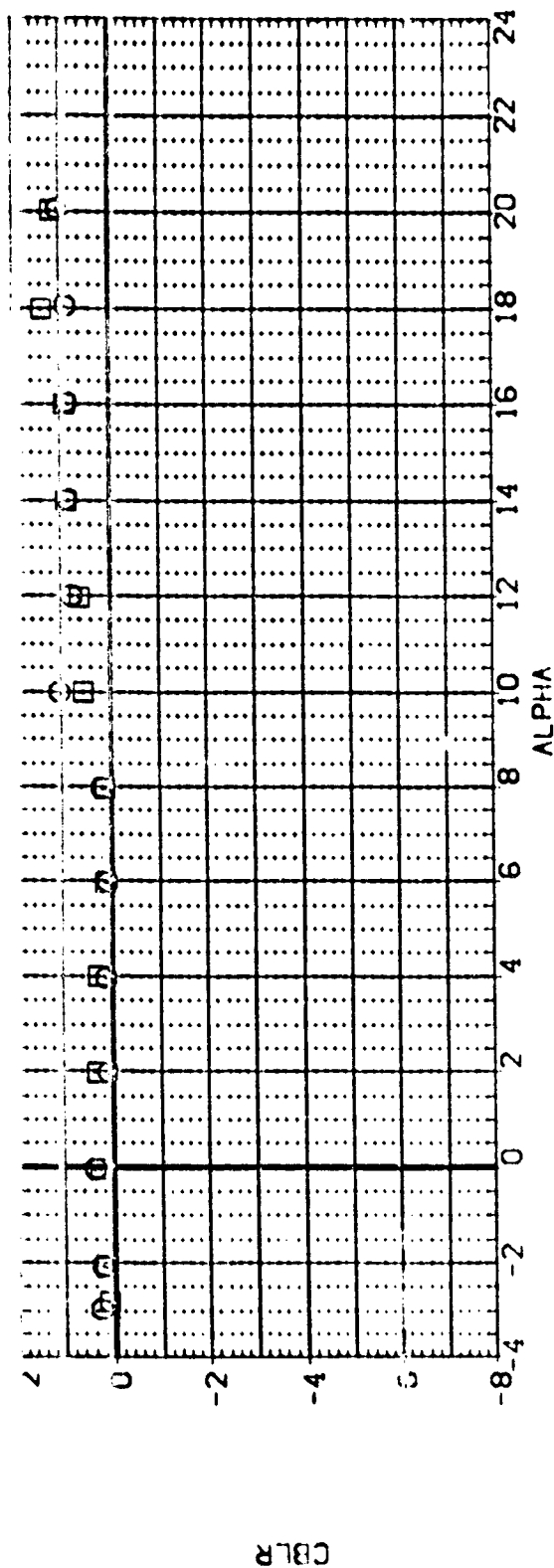
[illegible]

FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN YAW

COWING	.90	PAGE

CONFIG: LA-20: ROCKWELL ORB 0893 W/MOD: NOSE (BAVNF) 1.000 10.000 85.000
 [REMOVED] LA-20: ROCKWELL ORB 0893 W/MOD: NOSE (BAVNF) 1.000 10.000 85.000

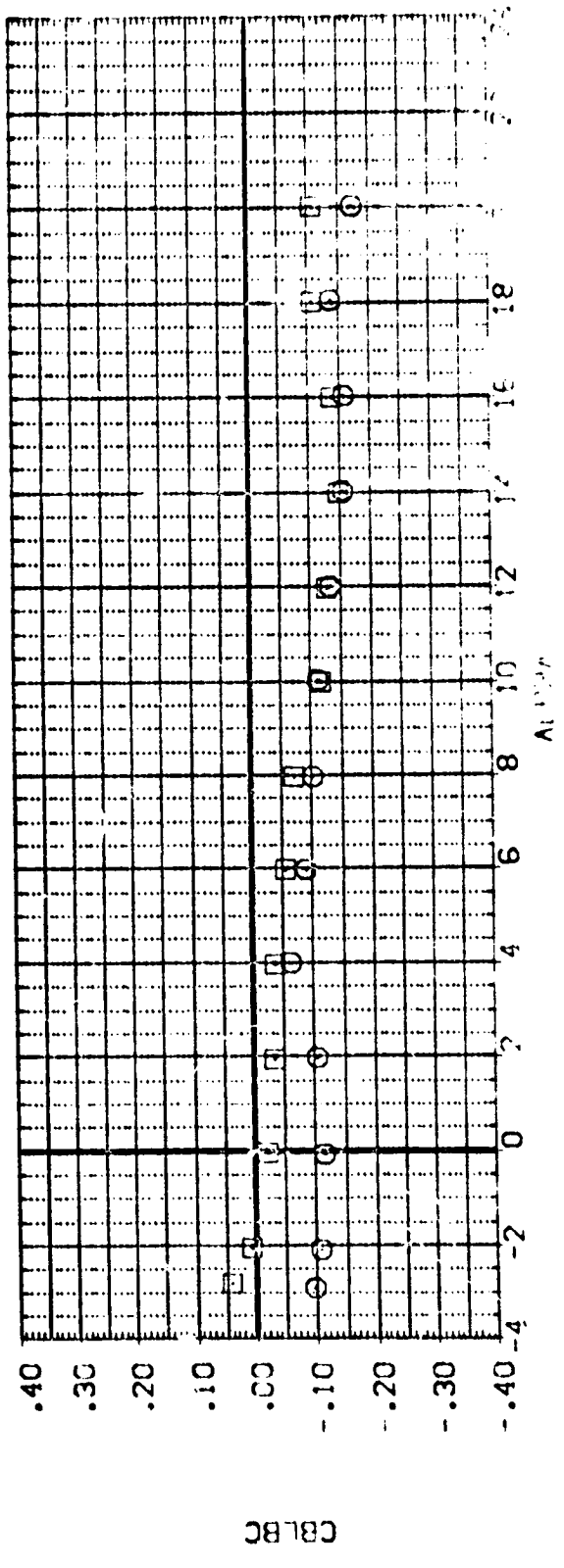
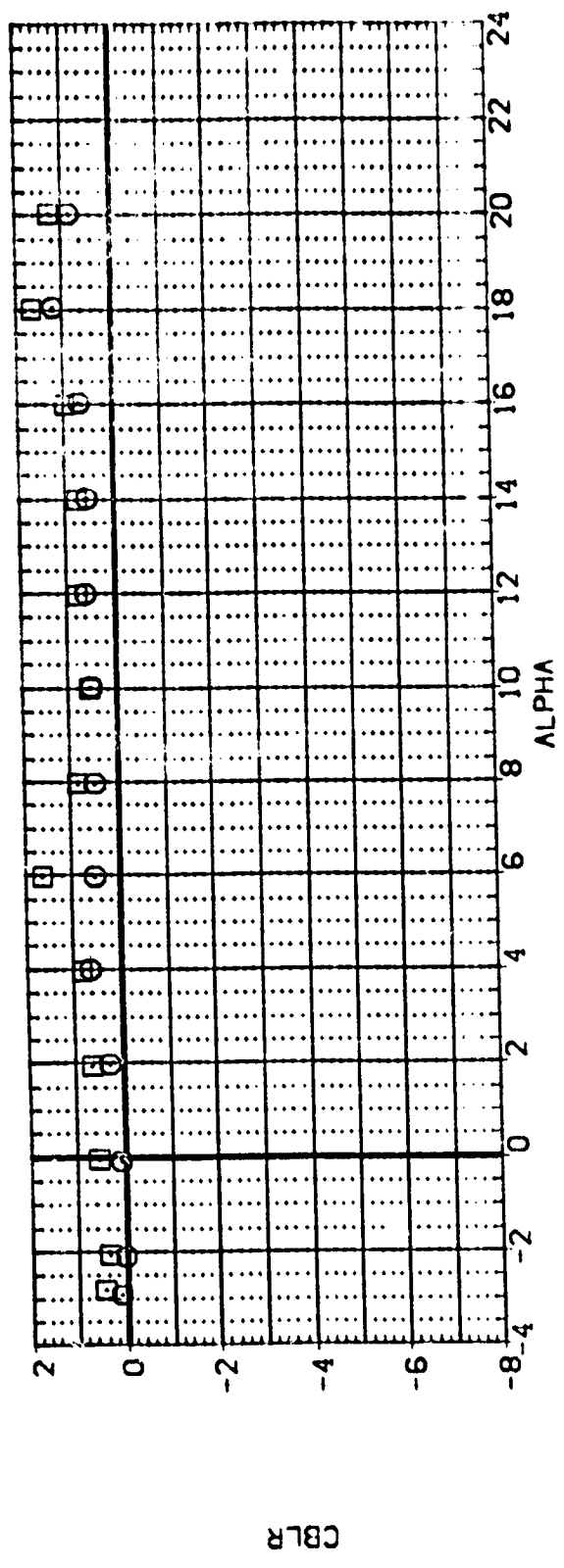


FIGURE 11. EFFECT OF BODY FLAP AND RODDED TAIL ON CBLR/CBLBC = 1.0

DATA SET SYMBOL: LA-20: ROCKWELL ORB 0898 W/MOD: NOSE (DYNMF) NOSE (BWMF)
 (RENTS) (RENTS)
 (RENTS) (RENTS)

CG-LCC: 1.000 1.000
 ELEVTR: .000 .000
 BOFLAP: .000 13.000
 RUDEL R: 10.000 85.000

CONFIGURATION DESCRIPTION

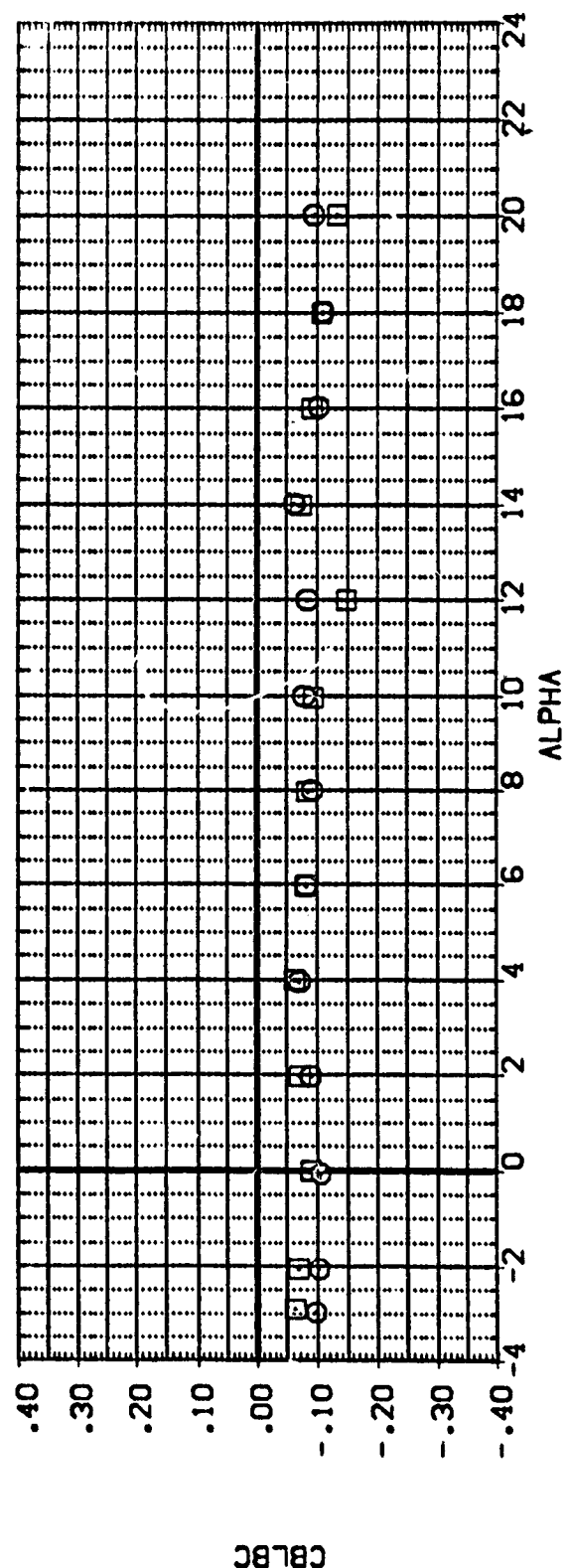
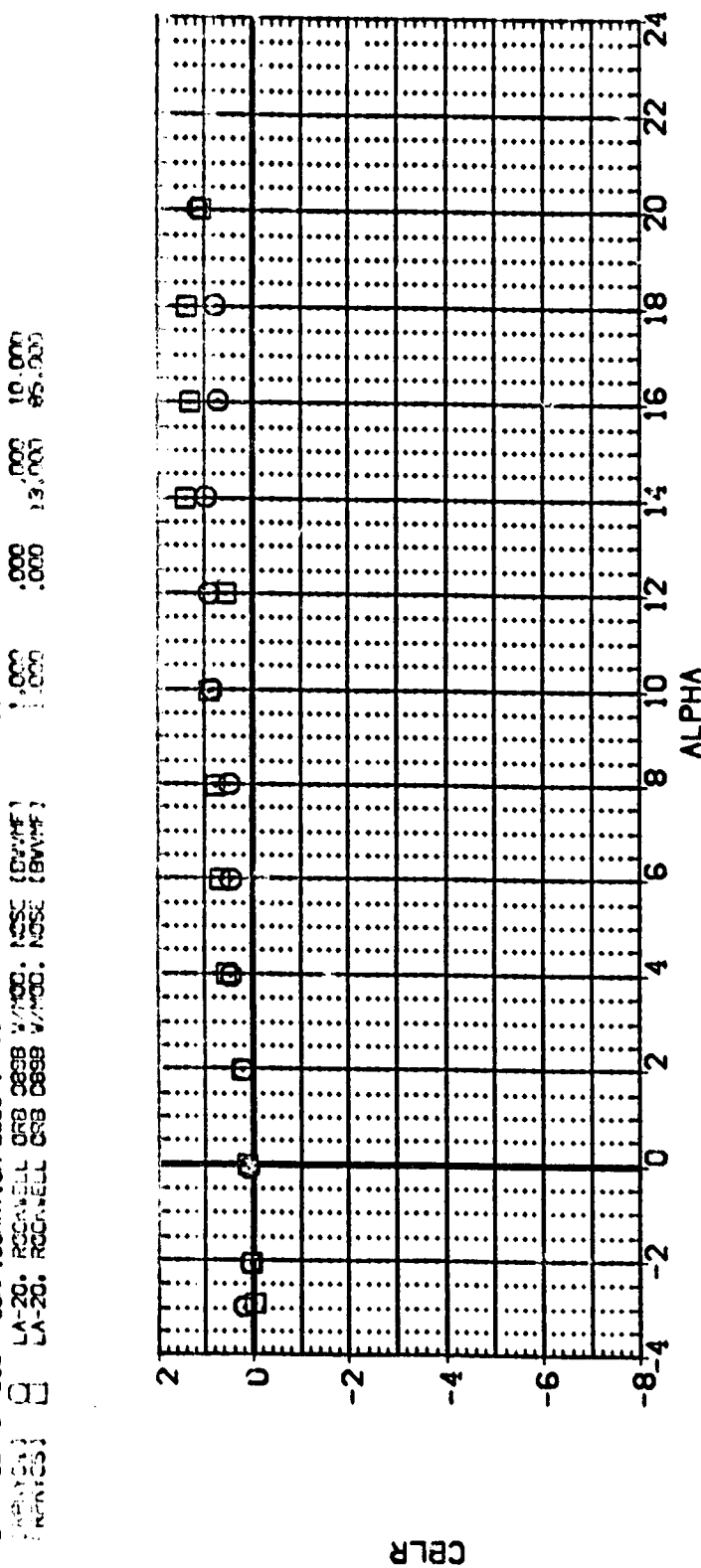


FIGURE 11. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN YAW

(E)MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR
 [RPKRD4] LA-20, ROCKWELL DB99 ORB W/HOO NOSE (BWWVF) 1.000 .000 .000 10.000
 [RPKRD5] LA-20, ROCKWELL DB99 ORB W/HOO NOSE (BWWVF) 1.000 .000 13.000 85.000

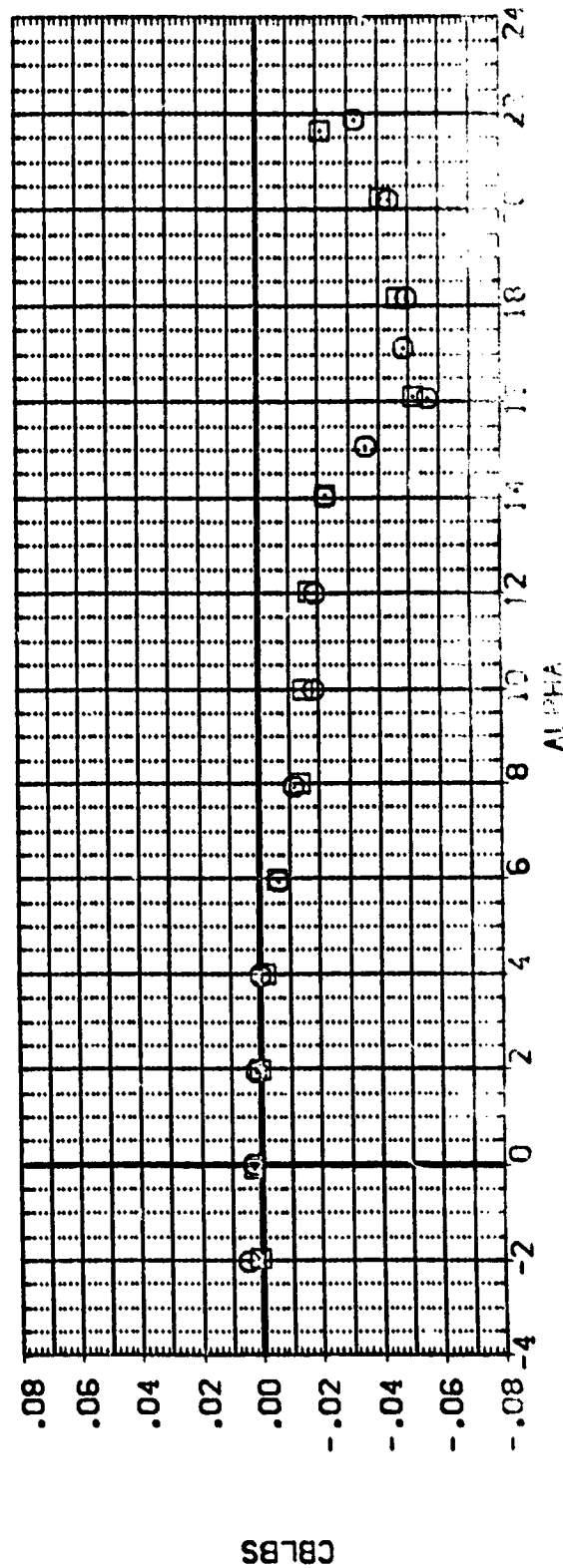
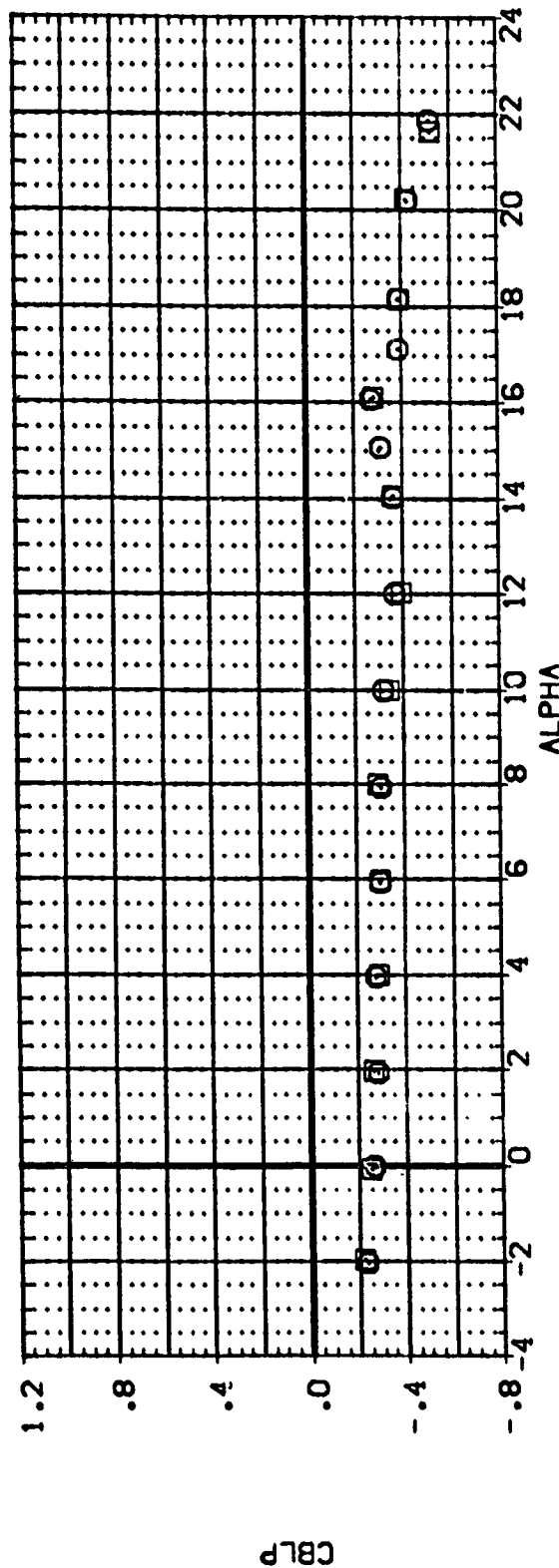


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON LIFT COEFFICIENT PARABOLIC HILL

(A)MACH = .30

DATA SET 5-100L CONFIGURATION DESCRIPTION
 LA-20, ROCKWELL C008 C19 WING NOSE (BNAW)
 LA-20, ROCKWELL C008 C19 WING NOSE (BNAW)
 LA-20, ROCKWELL C008 C19 WING NOSE (BNAW)

CG-100 ELEVTR BODY AP RUPTR
 1.000 .000 1.000
 1.000 1.000 1.000

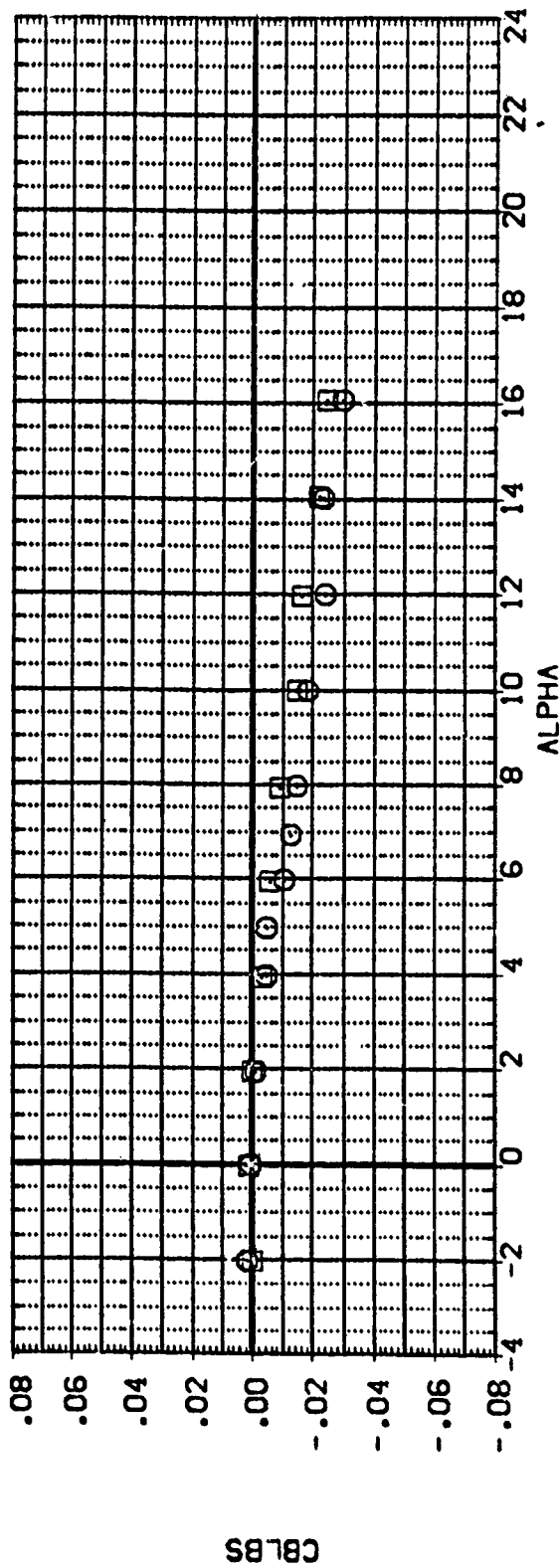
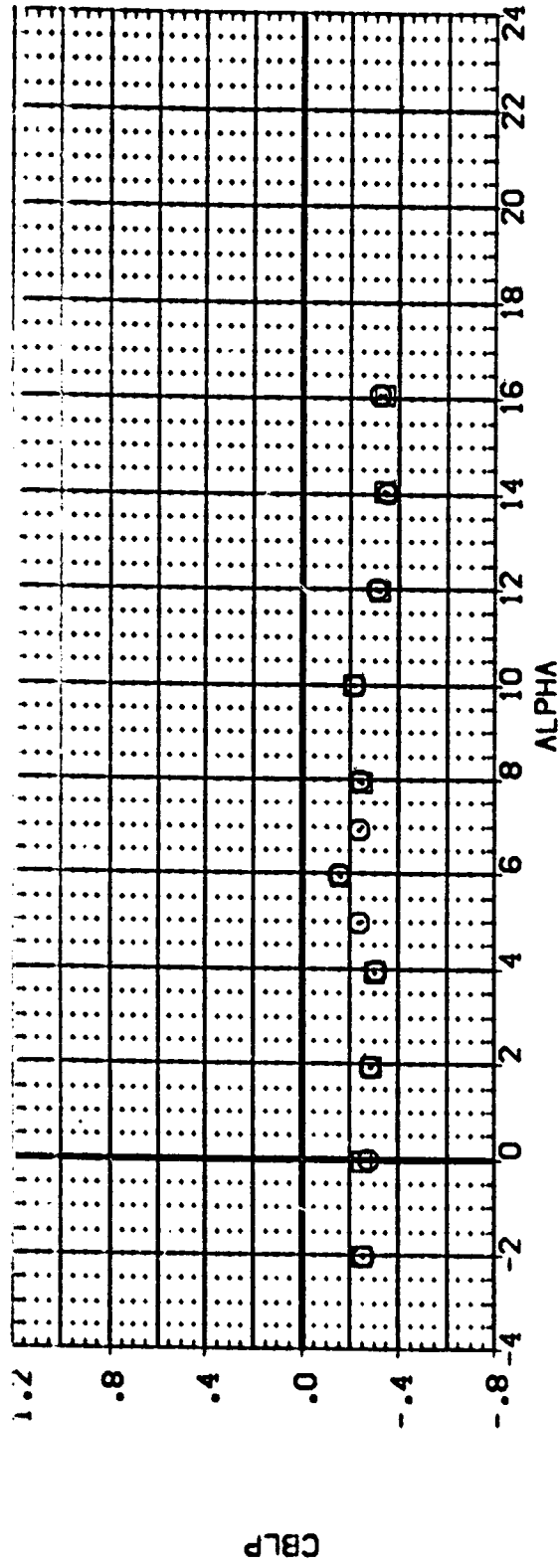


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN ROLL

(B)MACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RPR04) □ LA-20, ROCKWELL 0898 DRB V/HOD NOSE (BNVNF)
 (RPR05) □ LA-20, ROCKWELL 0898 DRB V/HOD NOSE (BNVNF)

CG, IN
 1,000
 1,000

ELEV, IN
 .000
 .000

REF, IN
 .000
 13,000

RUDD, IN
 10,000
 65,000

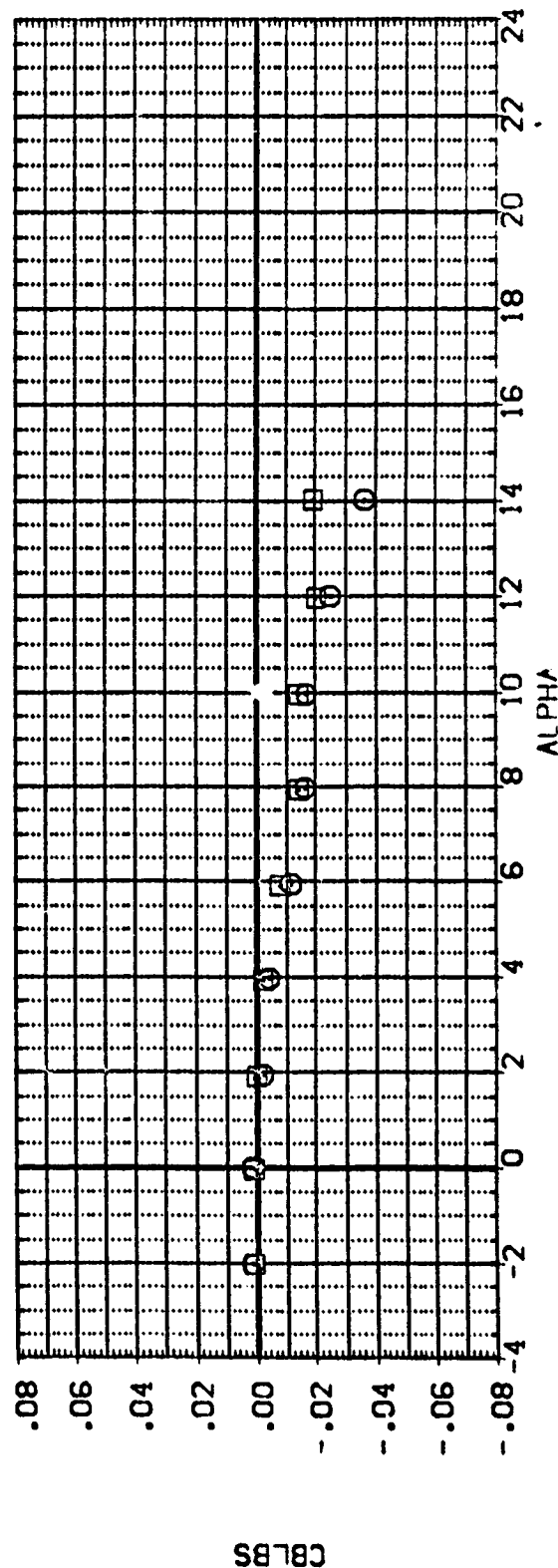
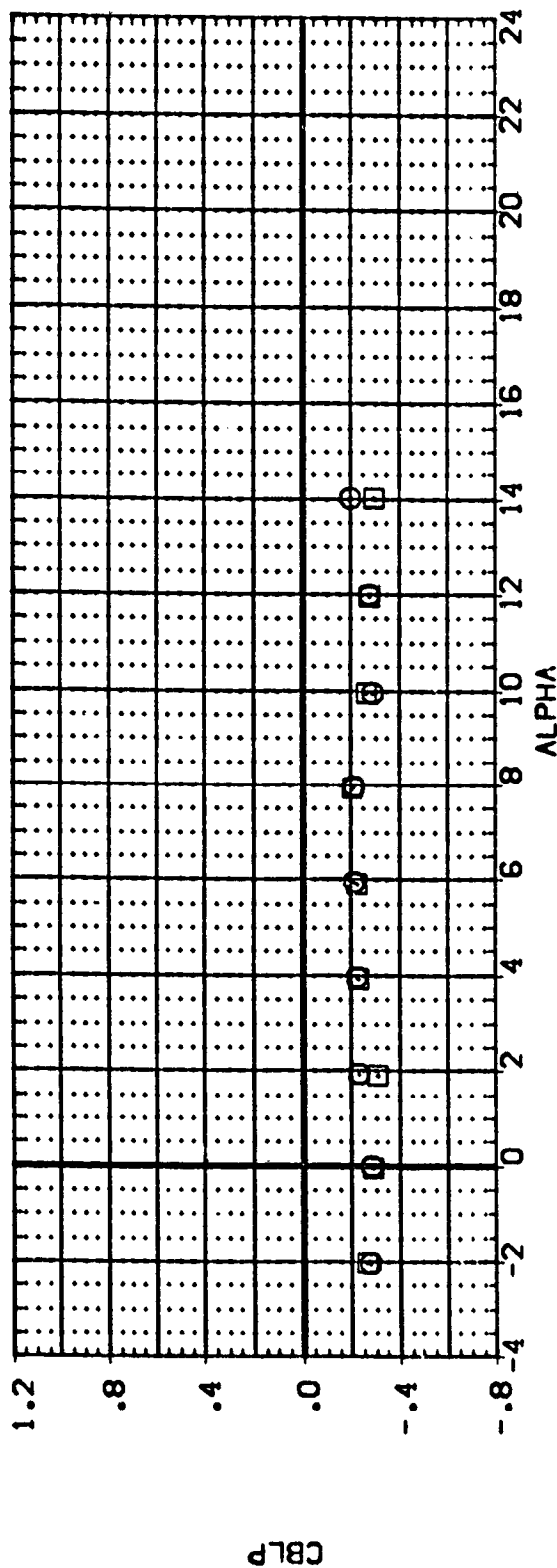


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN ROLL

(C)MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BODY LAP RUDDER

(RPR004) LA-20, ROCKWELL 0898 ORB V/HOD NOSE (BNAVF) 1.000 .000 .000 10.000

(RPR005) LA-20, ROCKWELL 0898 ORB V/HOD NOSE (BNAVF) 1.000 .000 13.000 65.000

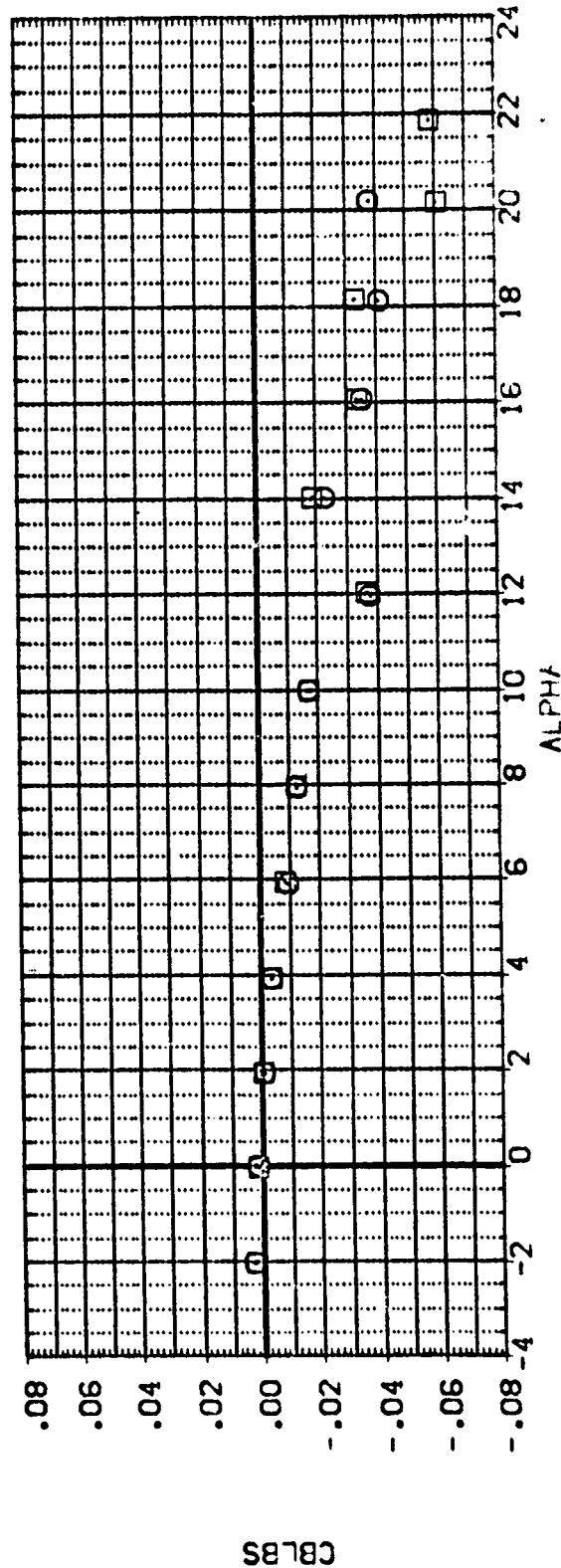
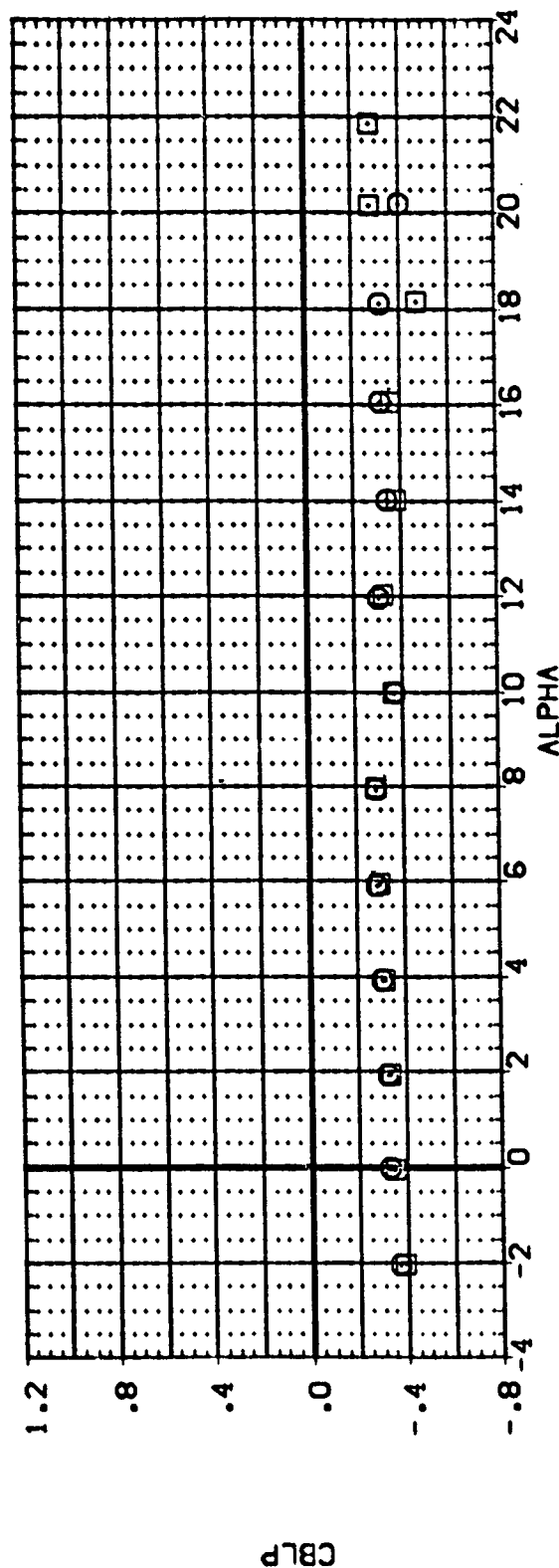


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLAKE ON DYN. STAB. PARAM. IN ROL

(E)MACH = 1.25

Figure 1 is a line graph with a grid. The vertical axis is labeled 'CYNP' and ranges from -0.4 to 0.6 with major ticks every 0.2. The horizontal axis is labeled 'ALPHA' and ranges from -4 to 24 with major ticks every 2 units. There are two data series: one represented by open circles and another by open squares. Both series show a similar trend, with a peak around ALPHA = 12-14 and a dip around ALPHA = 16-18. The square series generally has higher CYNP values than the circle series for ALPHA > 10.

ALPHA	CYNP (Circles)	CYNP (Squares)
0	0.00	-0.05
2	0.05	0.00
4	0.00	-0.05
6	0.05	0.00
8	0.00	-0.05
10	0.05	0.00
12	0.10	0.05
14	0.15	0.10
16	0.10	0.05
18	0.05	0.00
20	0.00	-0.05
22	0.05	0.00


$$[A]_{\text{MACH}} = .30$$

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BD' LAP RUOFLR

[RPKRO4] LA-20. ROCKWELL 089B ORB V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

[RPKROS] LA-20. ROCKWELL 089B ORB V/MOD NOSE (BVMF) 1.000 .000 13.000 65.000

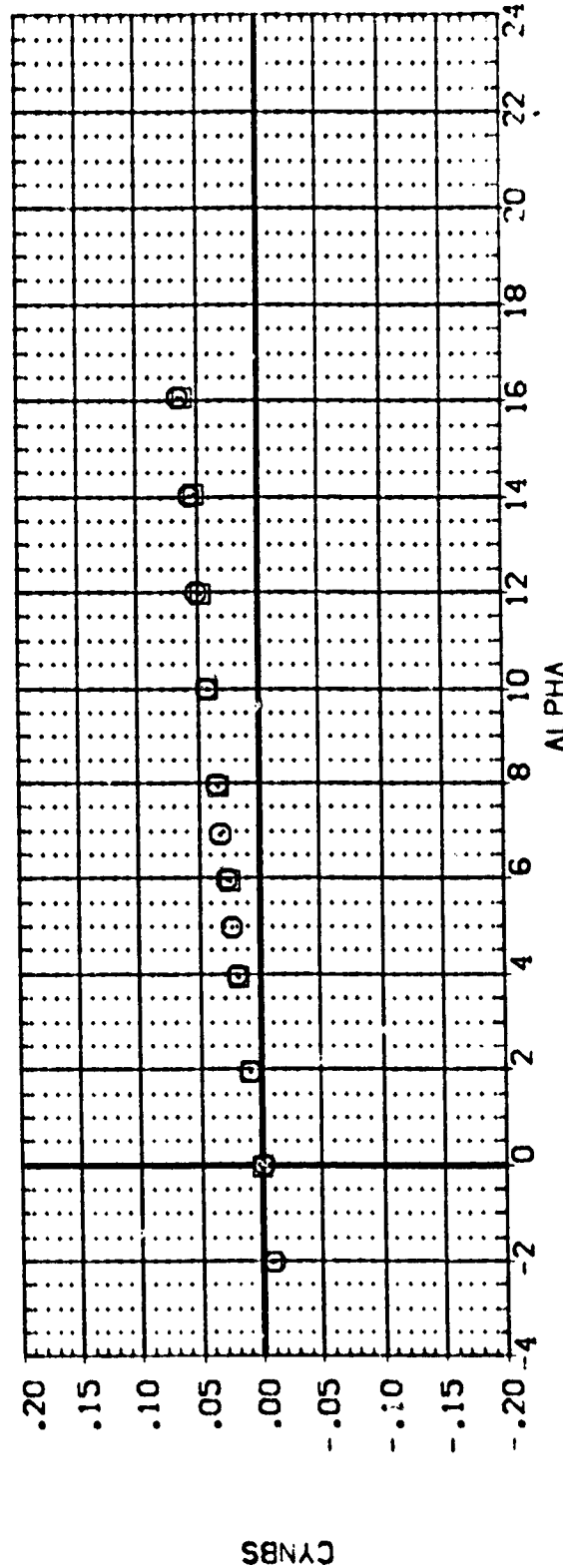
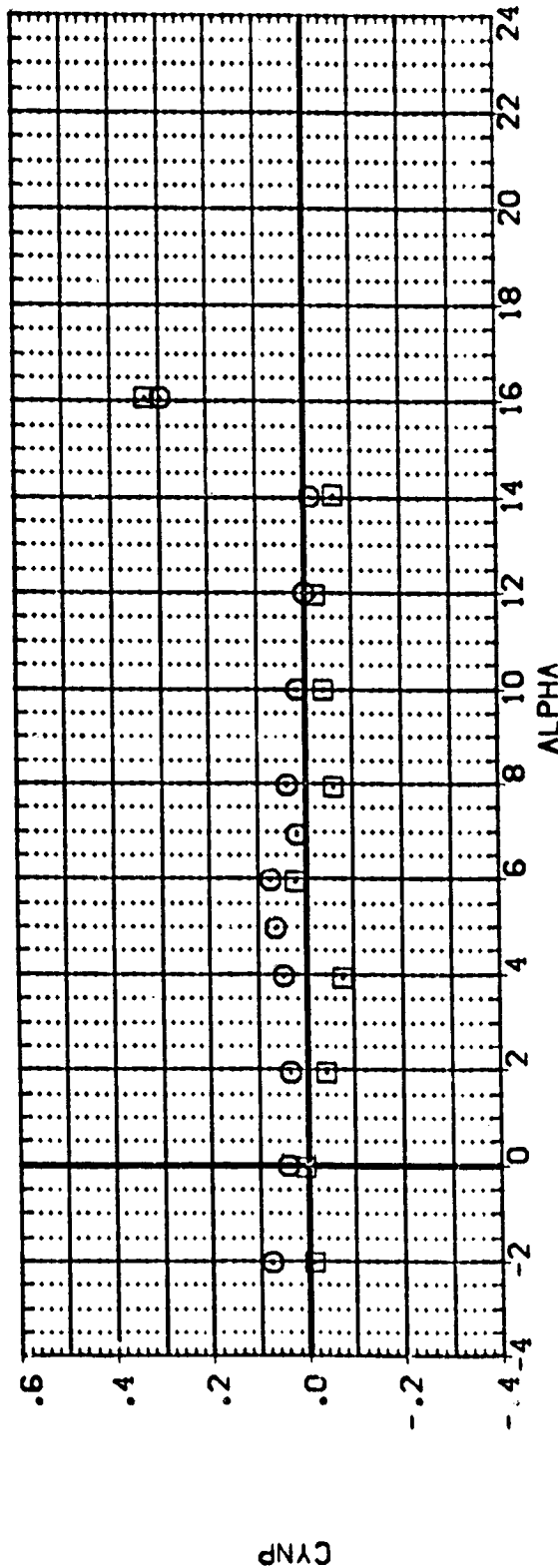


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN ROLL

(B)MACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BODYFLAP RUDDFLR

(P) (R) (A) (S) LA-20. ROCKWELL 0898 ORB V/MOD NOSE (BVVNF) 1.000 .000 .000 10.000

(P) (R) (A) (S) LA-20. ROCKWELL 0898 ORB V/MOD NOSE (BVVNF) 1.000 .000 13.000 85.000

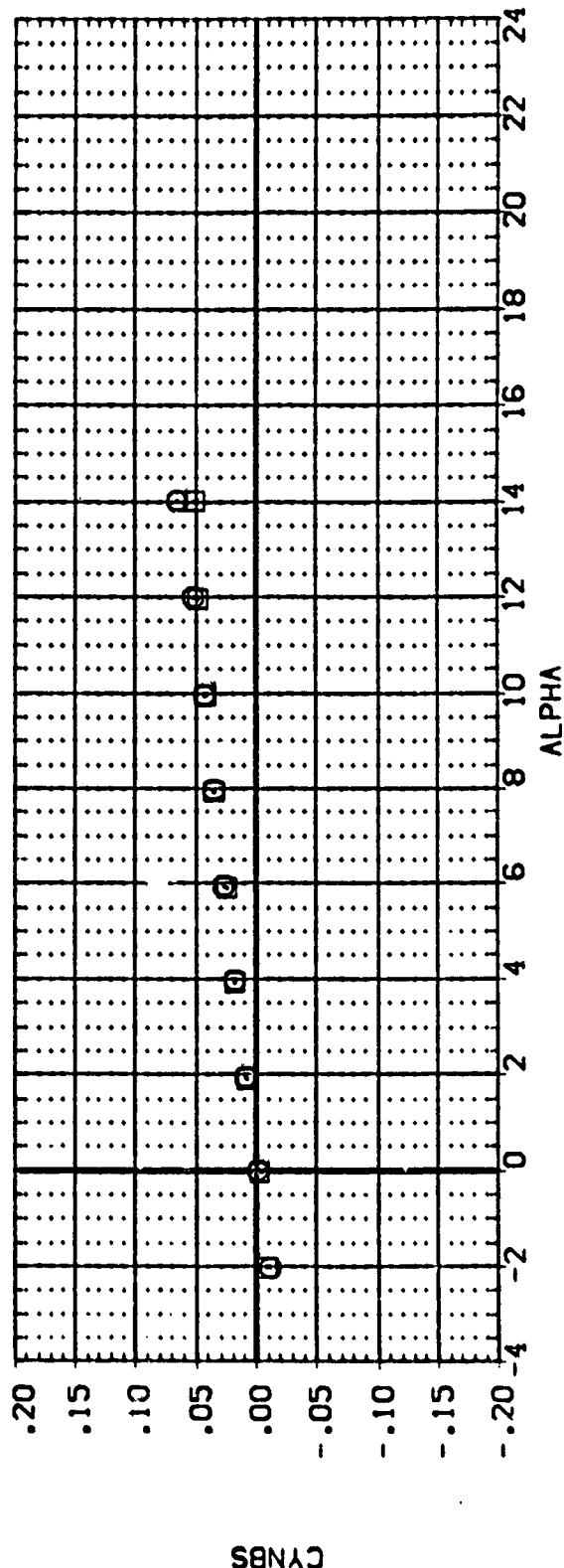
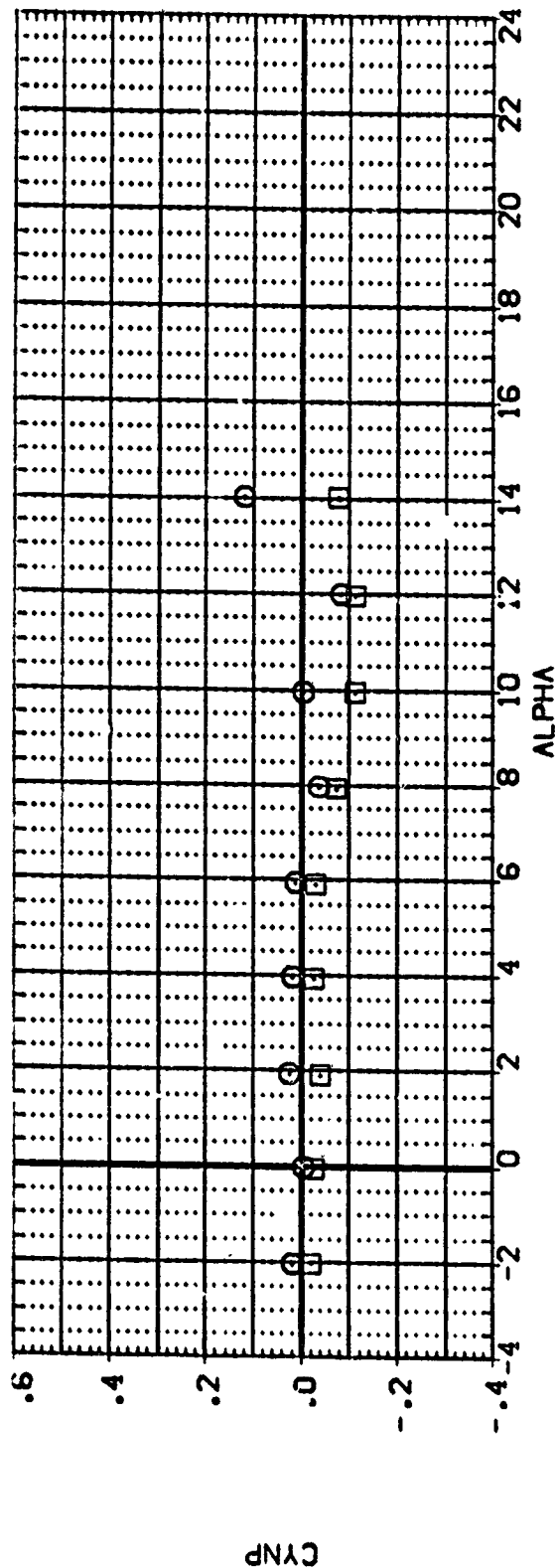


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN ROLL

(C)MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUJFLR

[RPRJ34] LA-20, ROCKWELL 0898 OR8 V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

[RPRJ05] LA-20, ROCKWELL 0898 OR8 V/MOD NOSE (BVMF) 1.000 .000 13.000 85.000

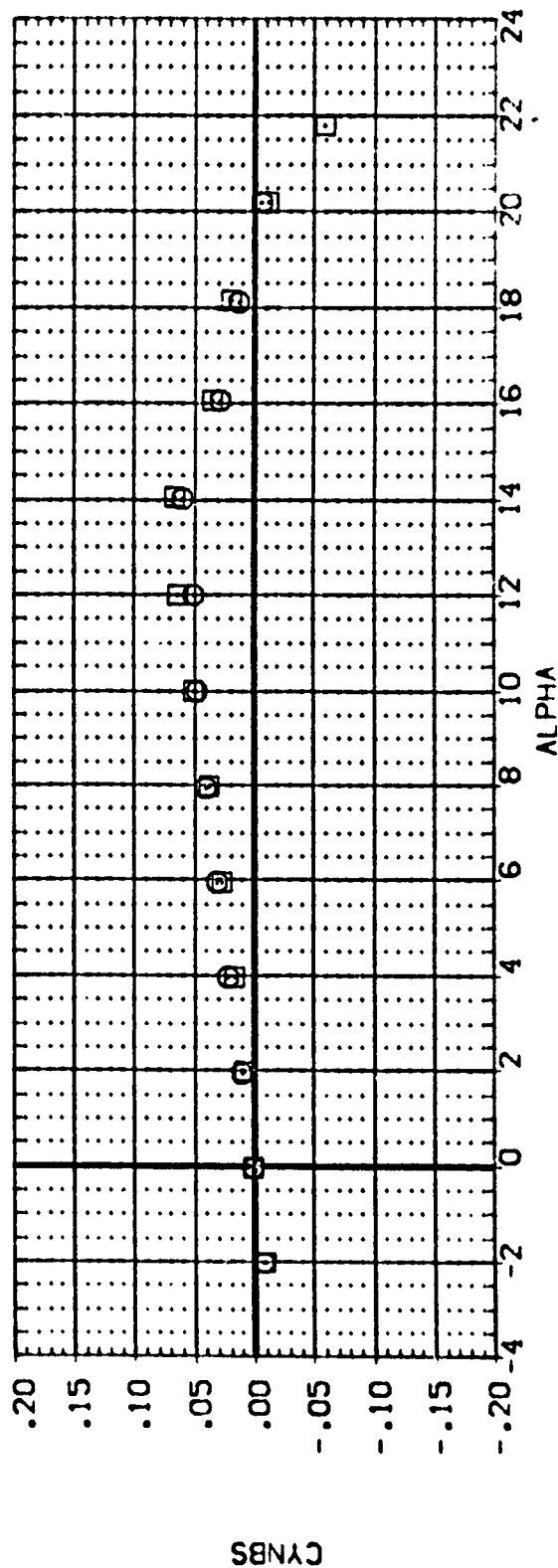
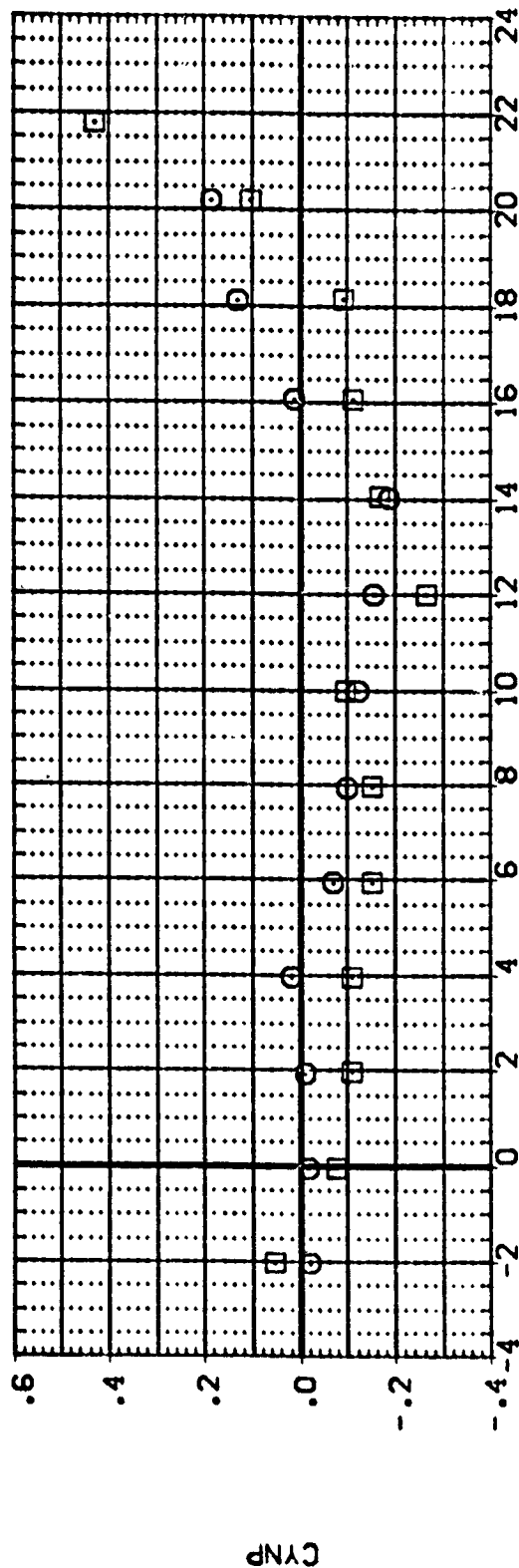


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN POLL

(C)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR
 [RPKRO4] [LA-20, ROCKWELL 0858 OR8 V/MOD NOSE (BVWF)] 1.000 .000 .000 10.000
 [RPKROS] [LA-20, ROCKWELL 0858 OR8 V/MOD NOSE (BVWF)] 1.000 .000 13.000 85.000

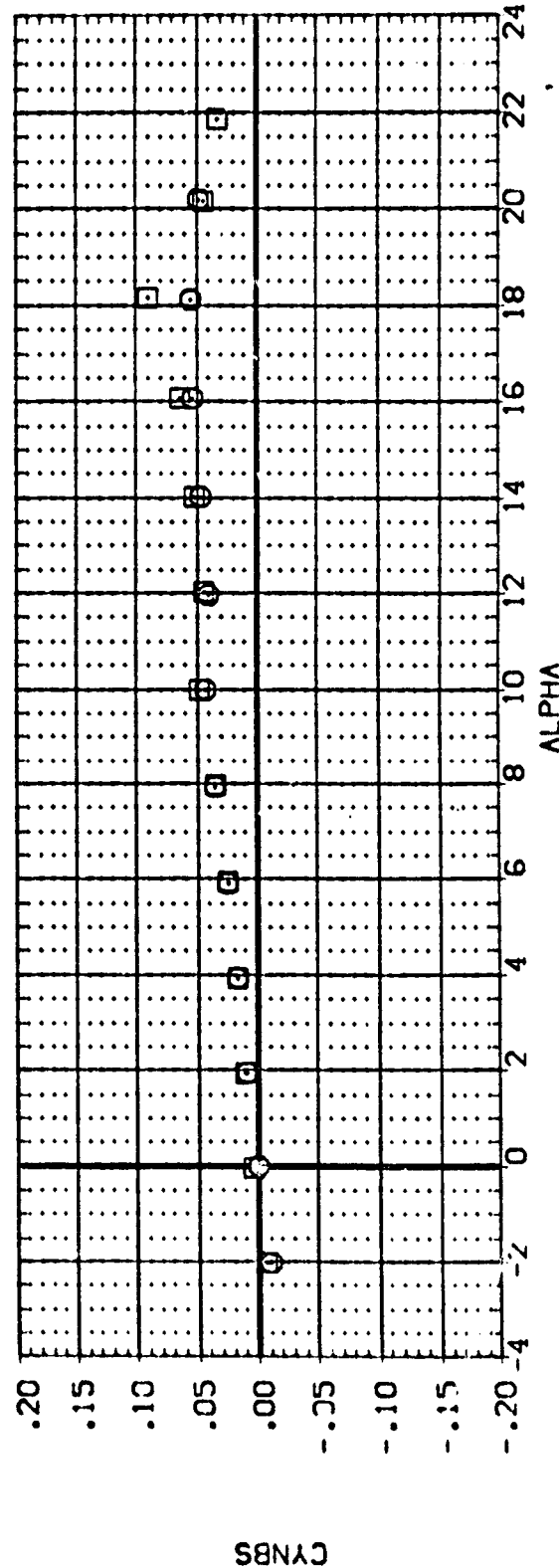
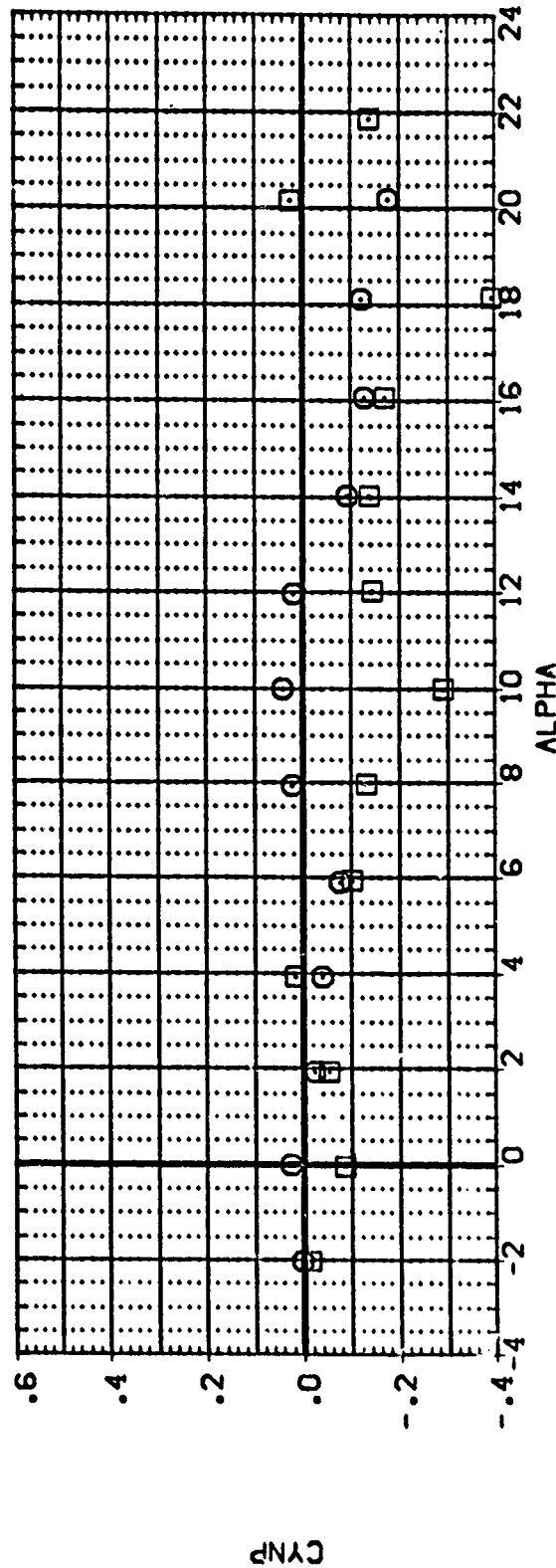


FIGURE 12. EFFECT OF BODY FLAP AND RUDDER FLARE ON DYN. STAB. PARAM. IN ROLL

(CJ)MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RODFLR

[RPKPOO] LA-20, ROCKWELL 089B DRB V/HOD NOSE (BVM) 1.000 .000 .000 10.000

[RPKPO4] LA-20, ROCKWELL 089B DRB V/HOD NOSE (BVM) 1.000 .000 .000 10.000

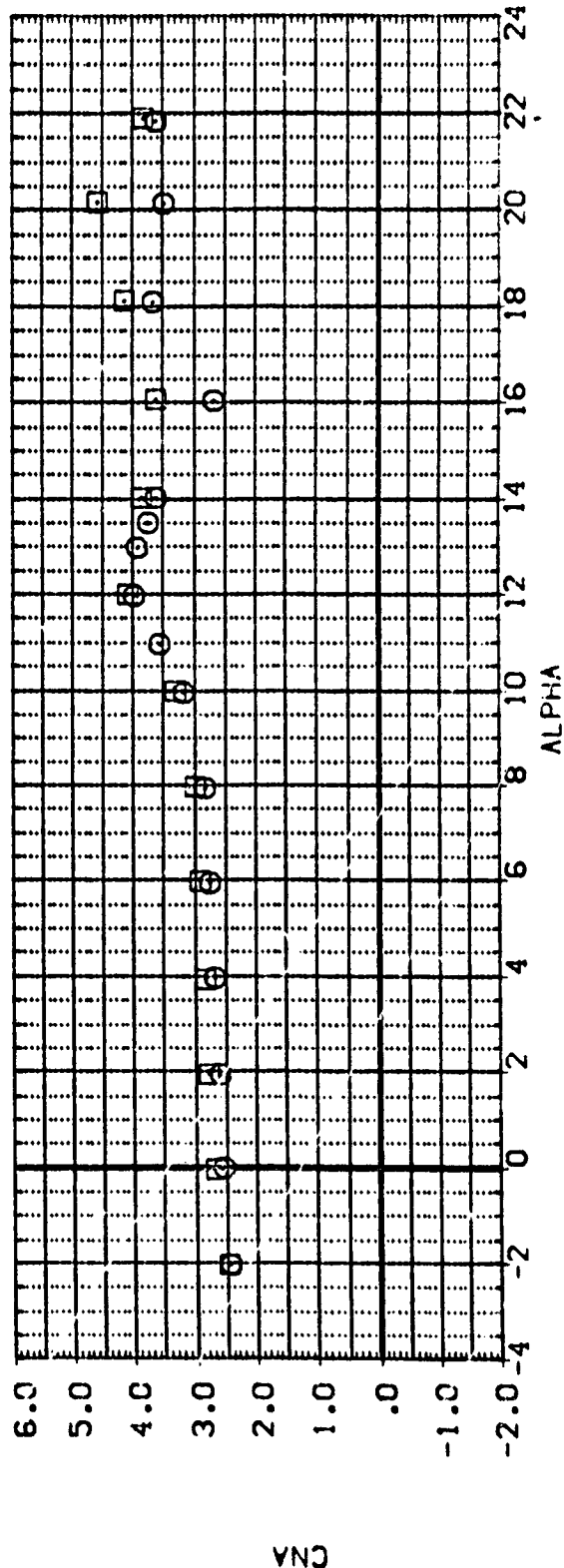
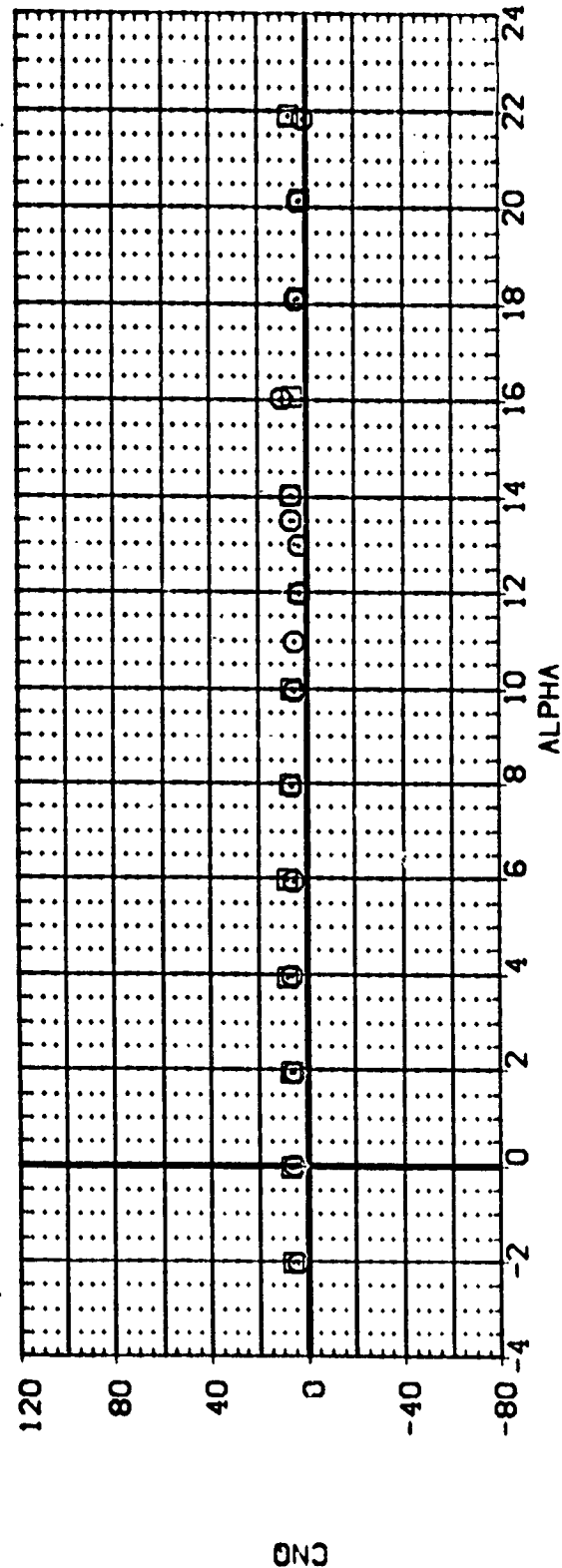


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(A)MACH = .30

DATA SET SYMBOL: (RKP00) (RKP04) CONFIGURATION DESCRIPTION: LA-20, ROCKWELL D858 ORB V/HOD NOSE (BVMH) LA-20, ROCKWELL D858 ORB V/HOD NOSE (BVMH)
 CG-LOC: 1.000 1.000 ELEVTR: .000 .000 BDFLAP: .000 .000 RUDFLR: 10.000 10.000

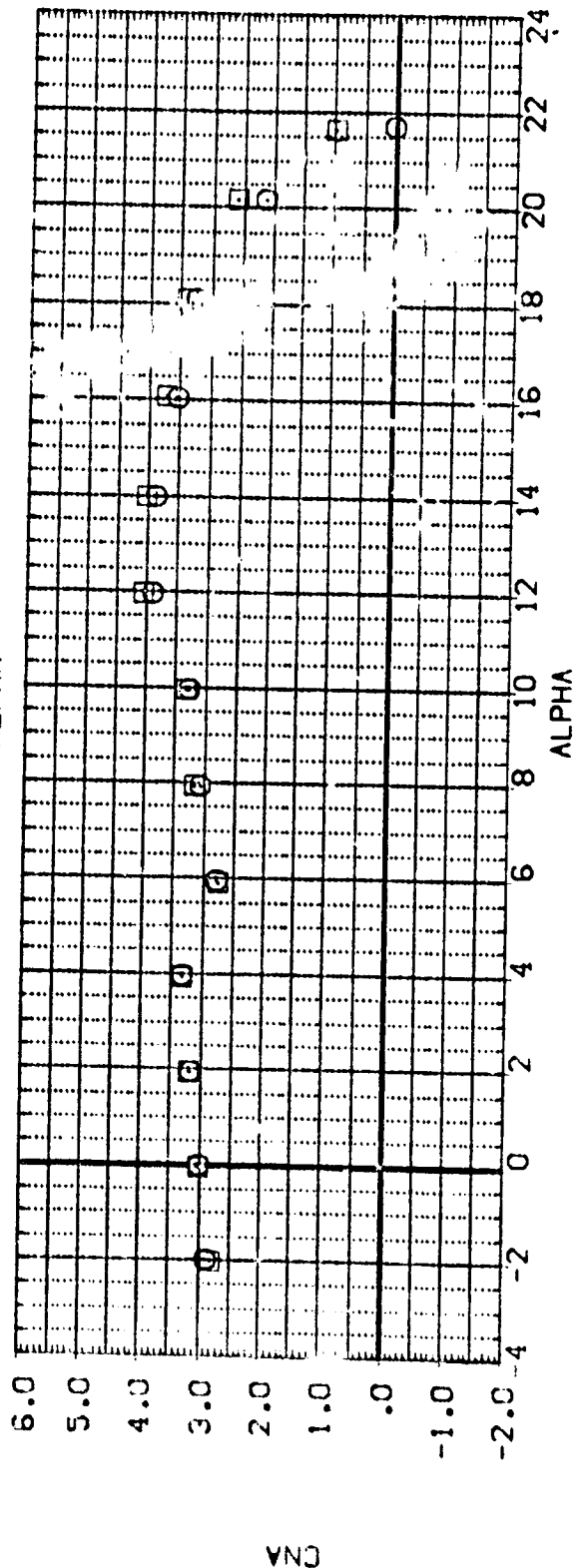
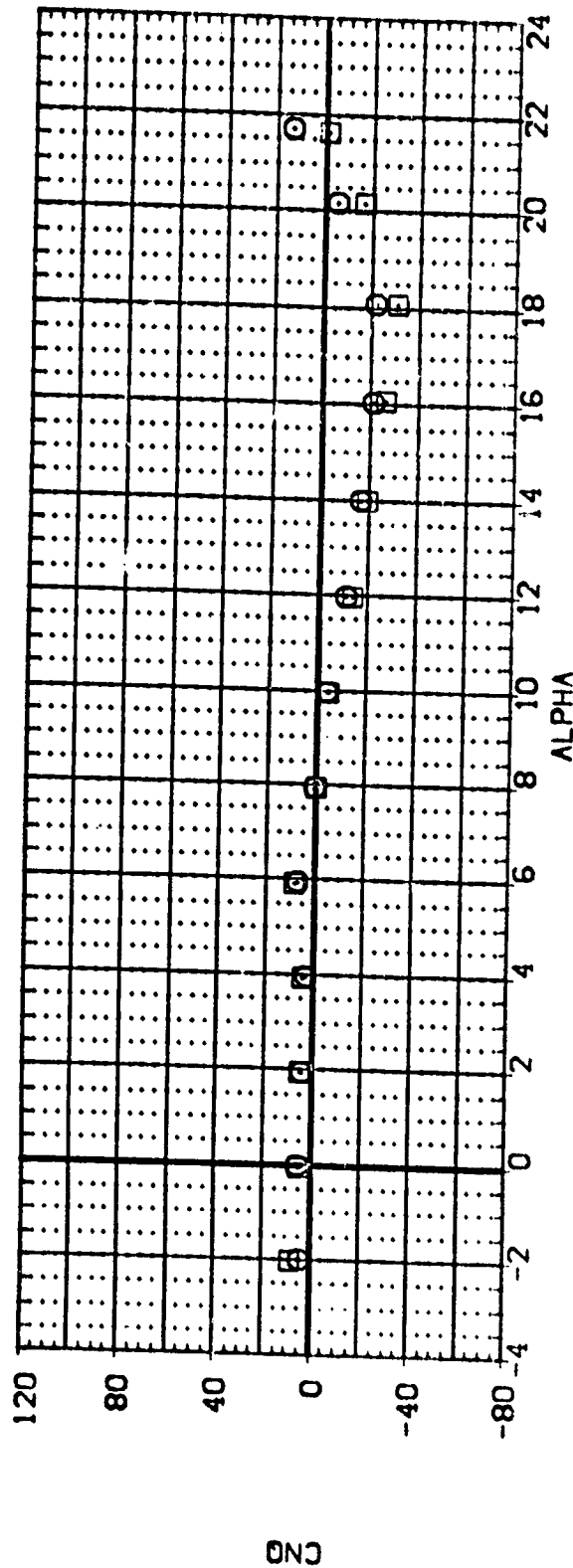


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(B)MACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUDELR
 [RPNP00] LA-20, ROCKWELL D898 ORB V/HOO NOSE (BVMH) 1.000 .000 .000 10.000
 [RPNP04] LA-20, ROCKWELL D898 ORB V/HOO NOSE (BVMH) 1.000 .000 .000 10.000

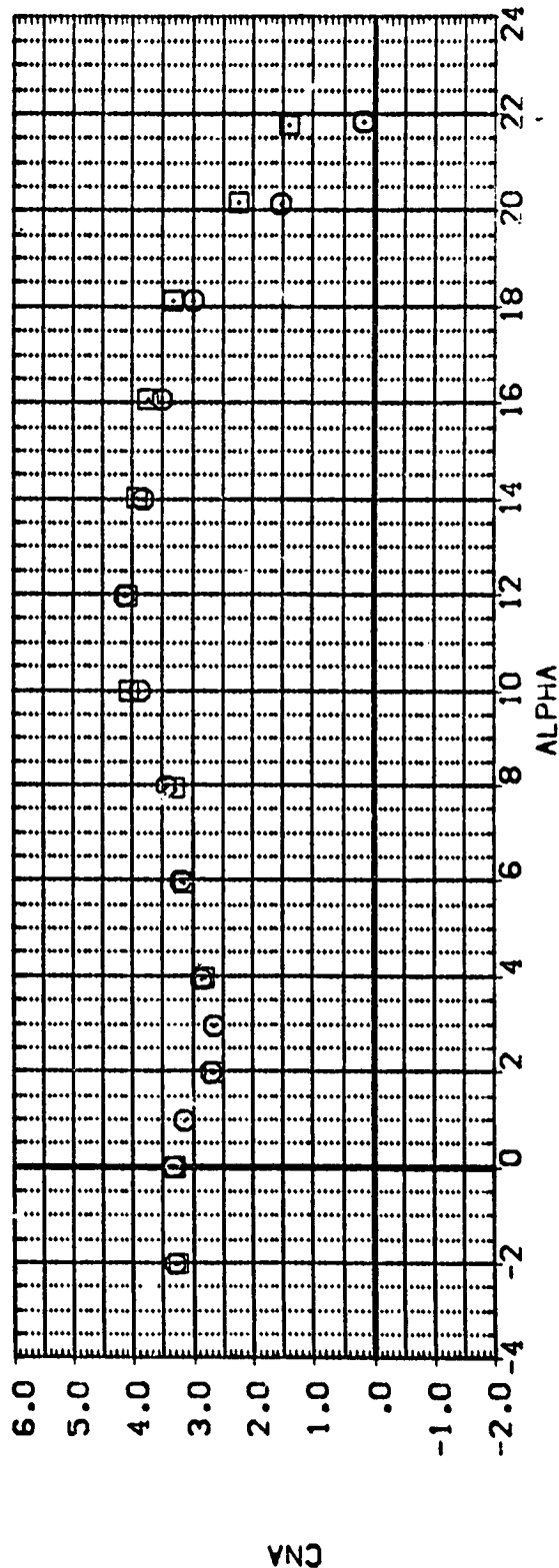
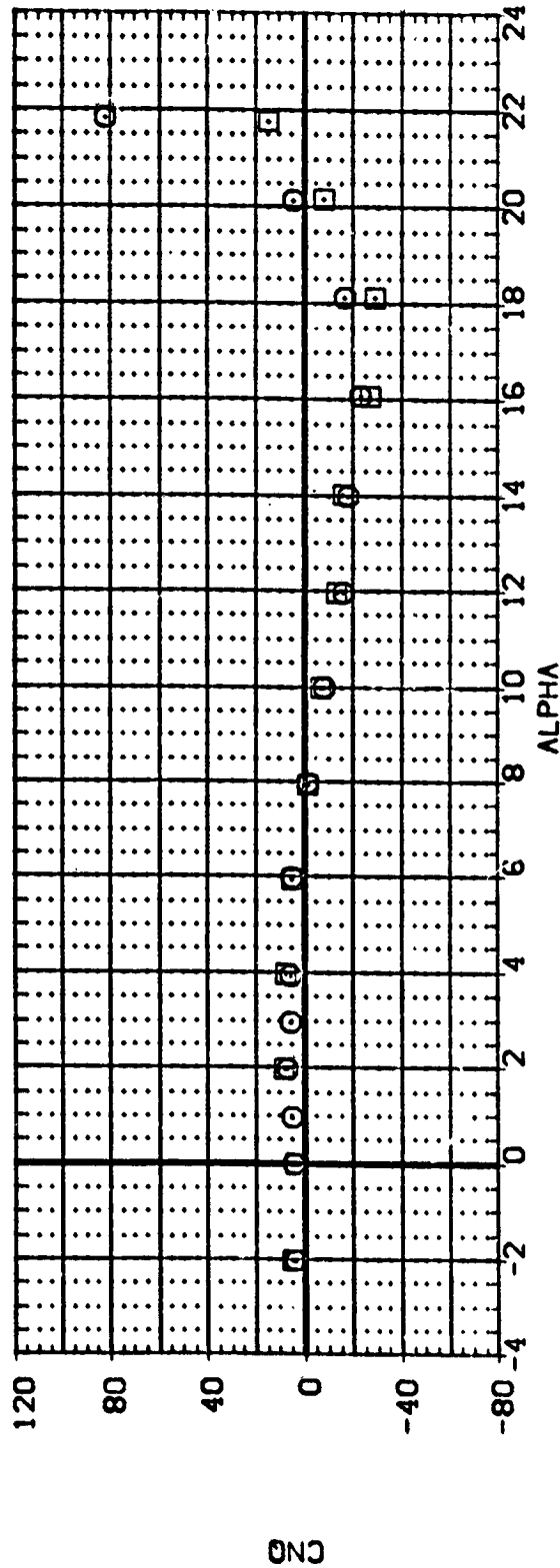


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(C)MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUOFLR
 (RPKPOD) LA-20, ROCKWELL 0898 ORB W/MOD NOSE (BVM) 1.000 .000 10.000
 (RPKPO4) LA-20, ROCKWELL 0898 ORB W/MOD NOSE (BVM) 1.000 .000 10.000

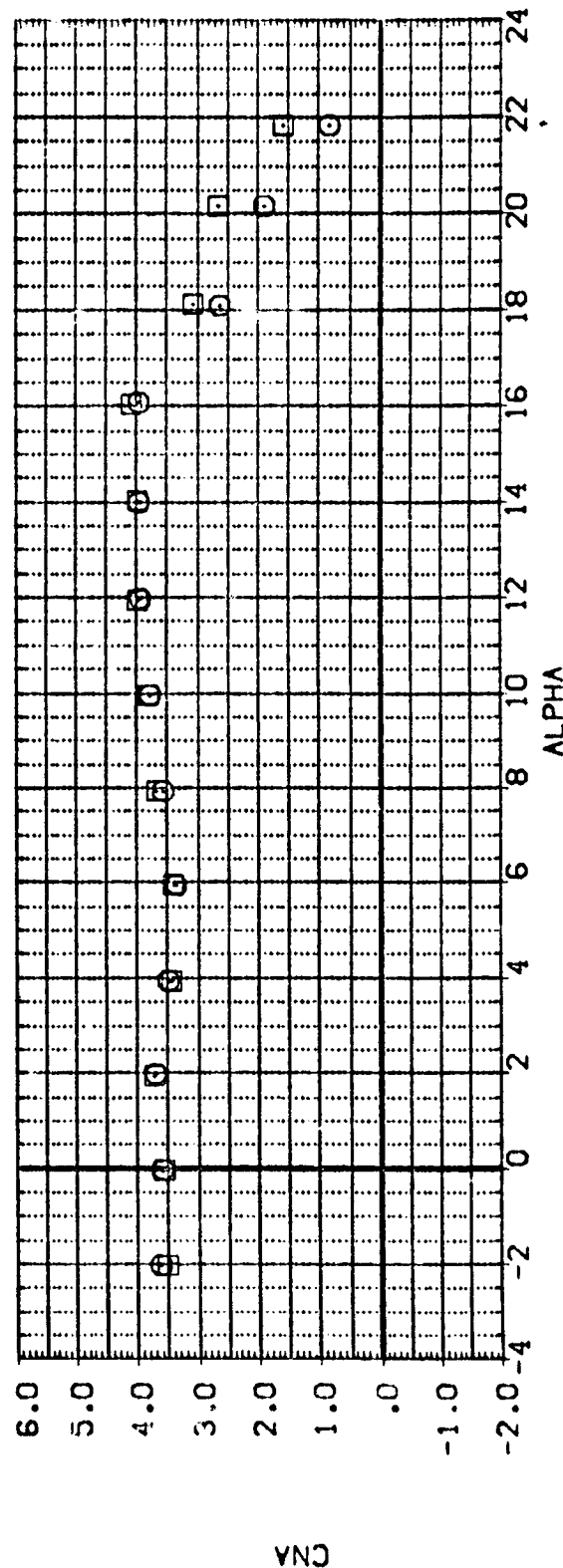
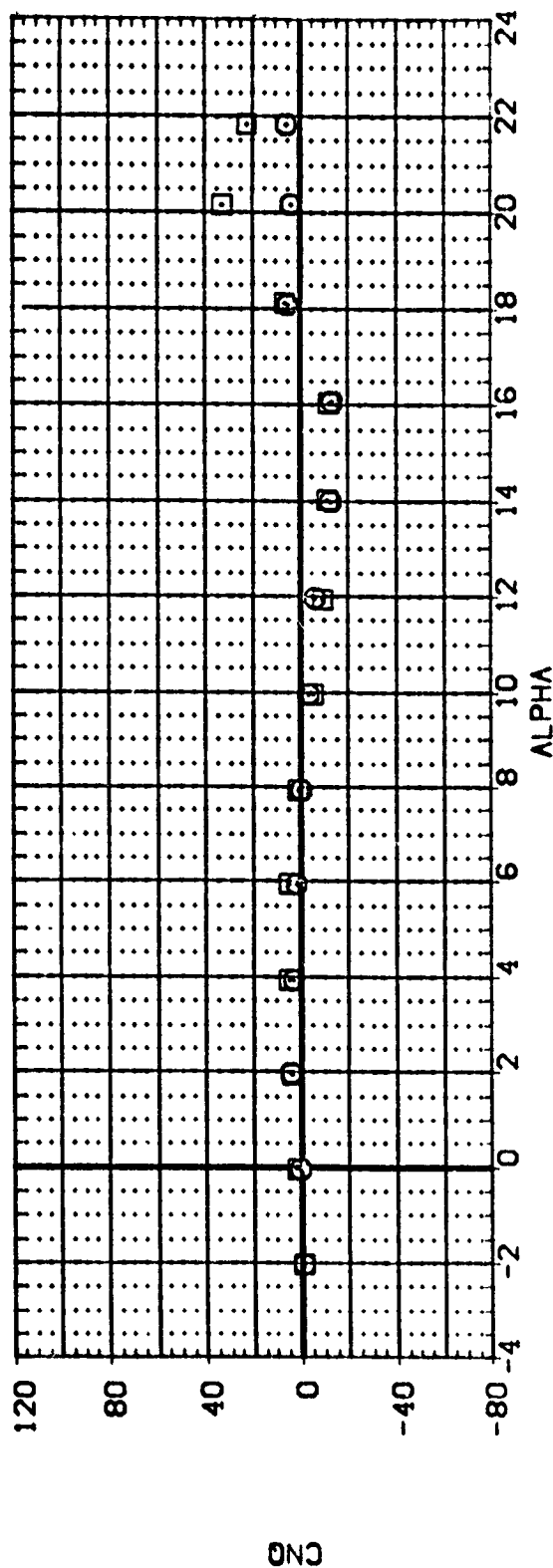


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(D)MACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUFLR

LA-20. ROCKWELL 0898 ORB V/MOD NOSE (BVM) 1.000 .000 10.000

LA-20. ROCKWELL 0898 ORB V/MOD NOSE (BVMF) 1.000 .000 10.000

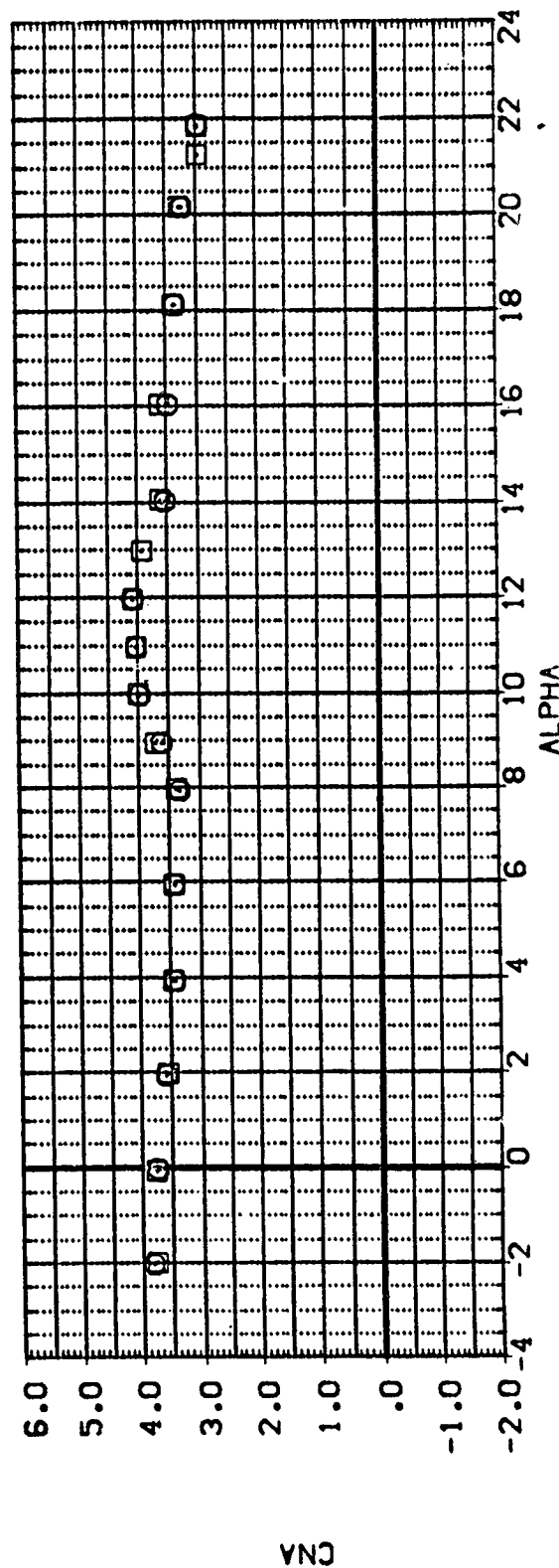
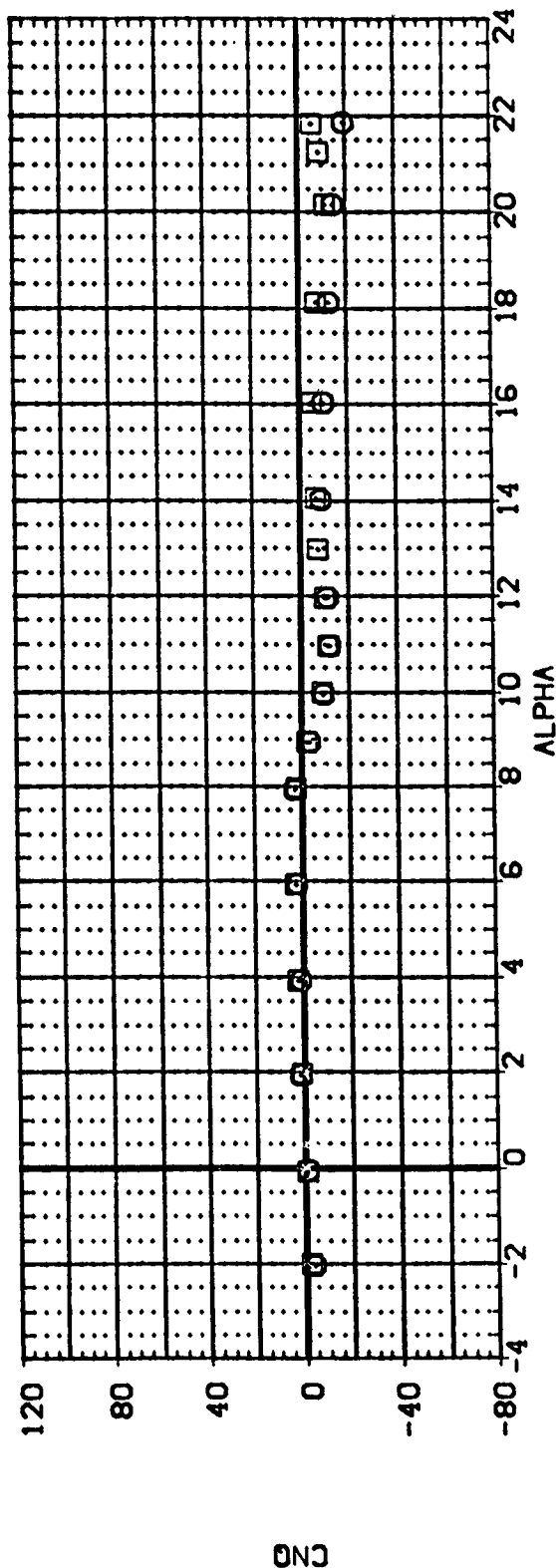


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH ;

(E2MACH = 1.20

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BOFLAP RUDELIR

[RPMPOO] LA-20, ROCKWELL O888 ORB V/MOD NOSE (BVM) 1.000 .000 .000 10.000

[RPMPO4] LA-20, ROCKWELL O888 ORB V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

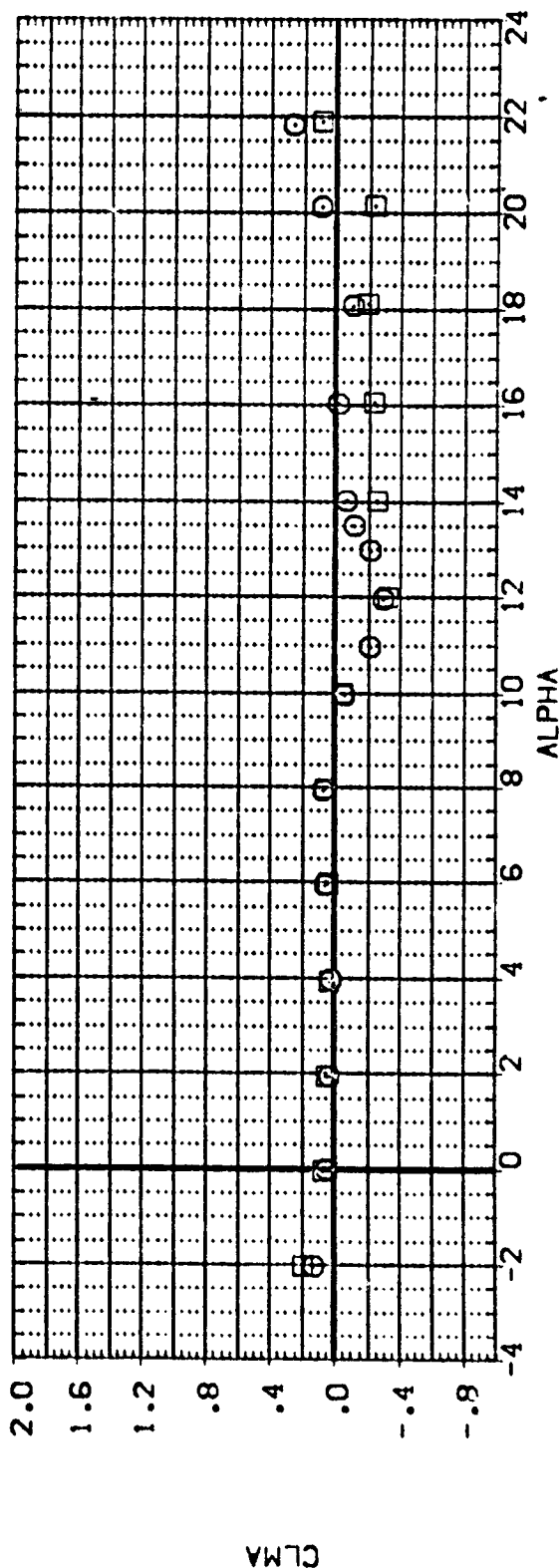
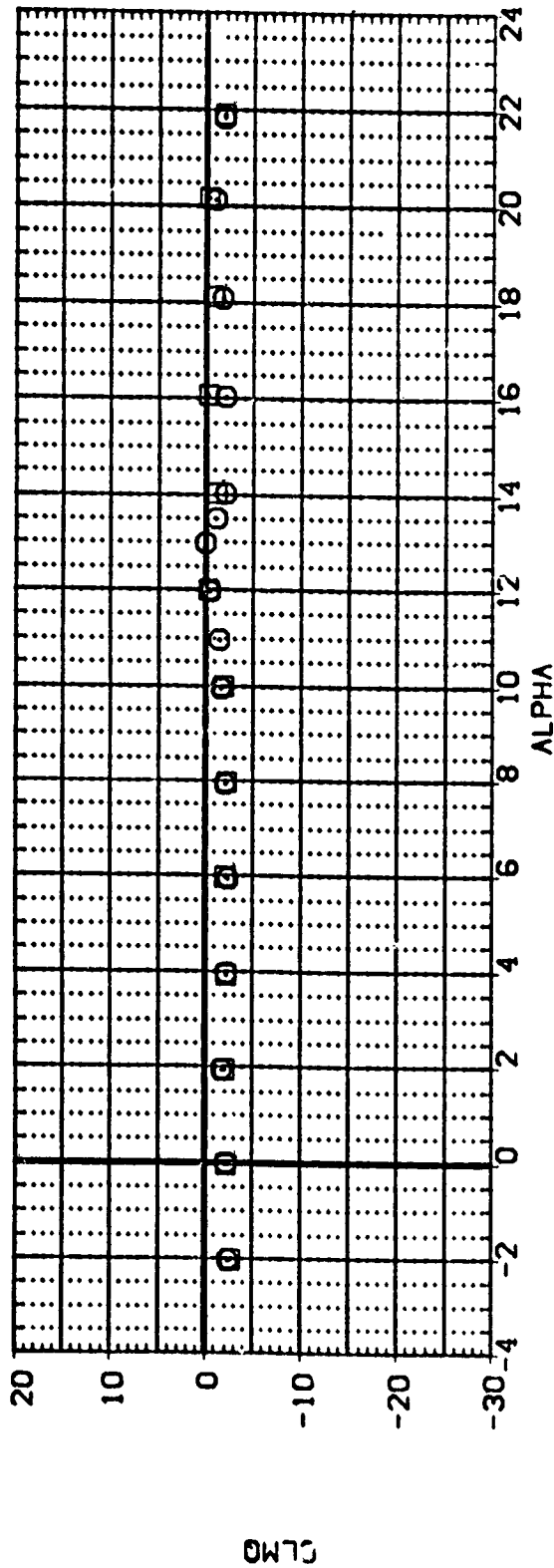


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(AJMACH = .30

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR

(RPKPO0) LA-20, ROCKWELL 0898 OR8 V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

(RPKPO4) LA-20, ROCKWELL 0898 OR8 V/MOD NOSE (BVMF) 1.000 .000 .000 10.000

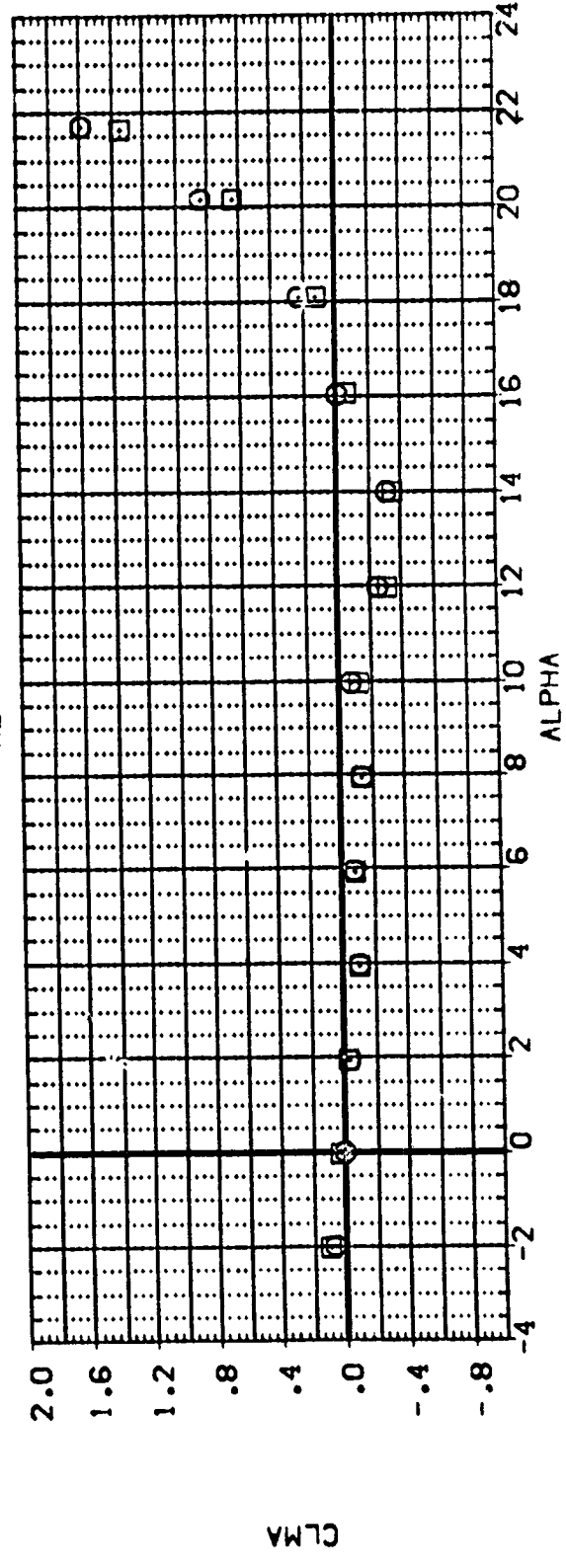
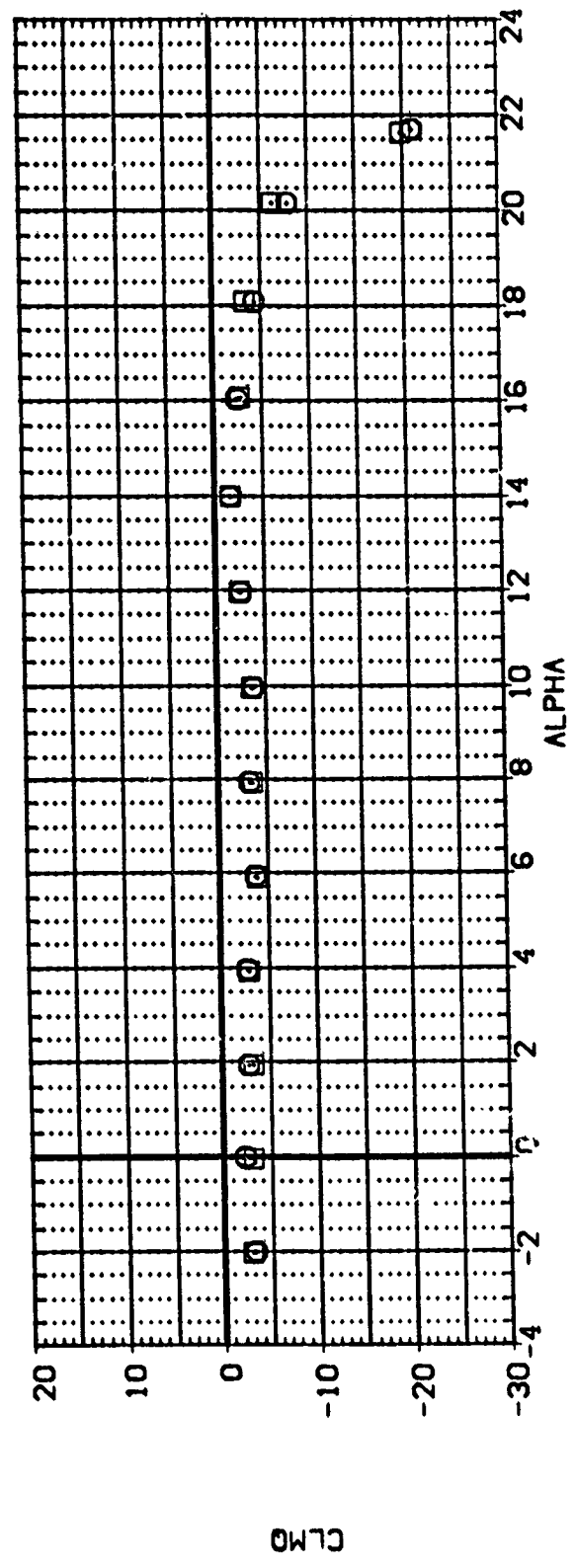


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(B)MACH = .80

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BDFLAP RUOFLR
 (RPN/POD) (RPN/POD) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMH) 1.000 .000 10.000
 (RPN/POD) (RPN/POD) LA-20, ROCKWELL 0898 ORB V/MOD NOSE (BVMH) 1.000 .000 10.000

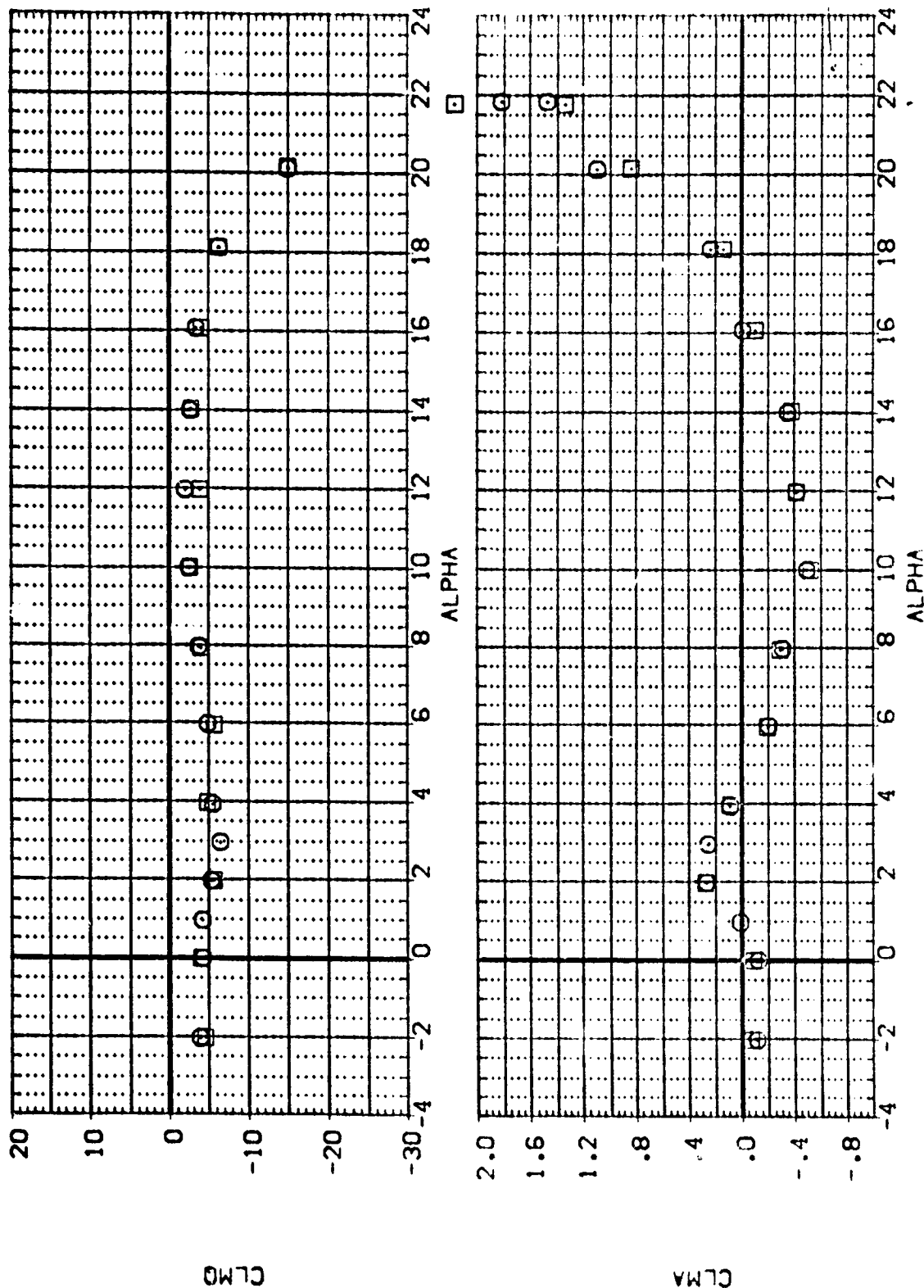


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(C)MACH = .90

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	CG-LOC	ELEVTR	BOFLAP	RUOFLR
(RPKPOO)	LA-20, ROCKWELL 0898 098 V/MOD NOSE (BVMH)	1.200	.000		10.000
(RPKPO4)	LA-20, ROCKWELL 0898 098 V/MOD NOSE (BVMH)	1.000	.000	.000	10.000

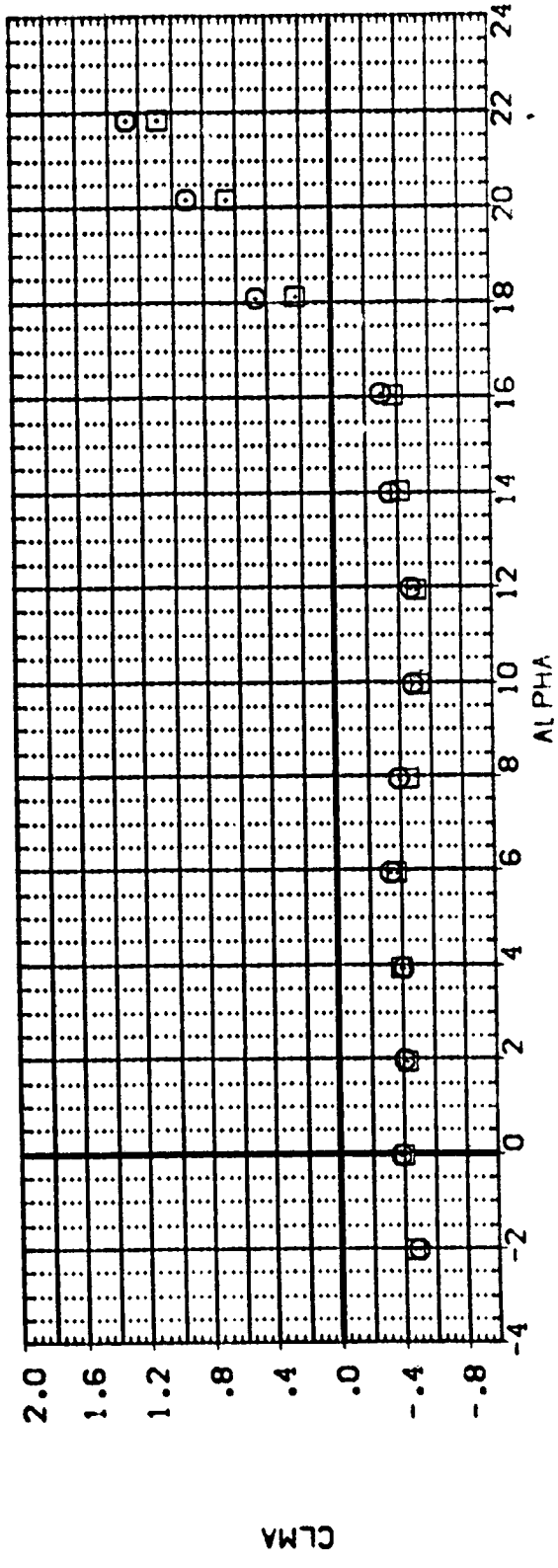
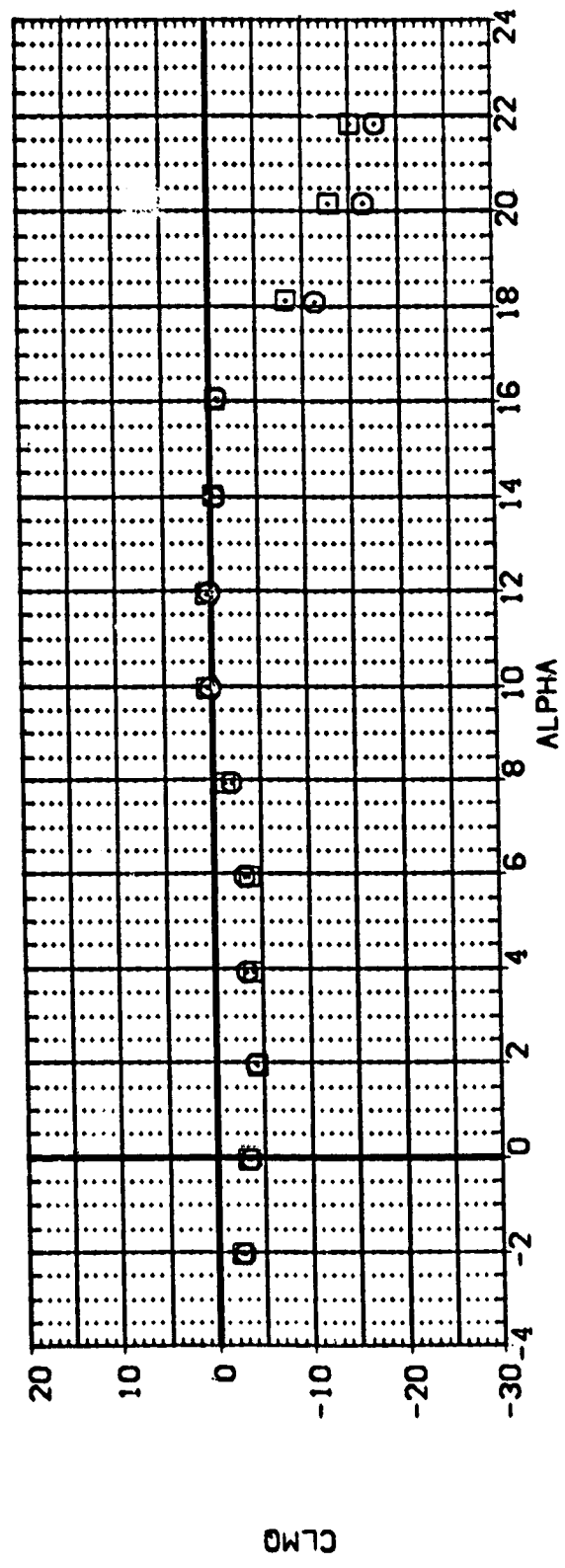


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

COMACH = .98

DATA SET SYMBOL CONFIGURATION DESCRIPTION CG-LOC ELEVTR BODYFLAP RUOFLR

[RPKPOO] LA-20. ROCKWELL 0898 ORB V/MOD NOSE (BVVM) 1.000 .000 .000 10.000

[RPKPO1] LA-20. ROCKWELL 0898 ORB V/MOD NOSE (BVVM) 1.000 .000 .000 10.000

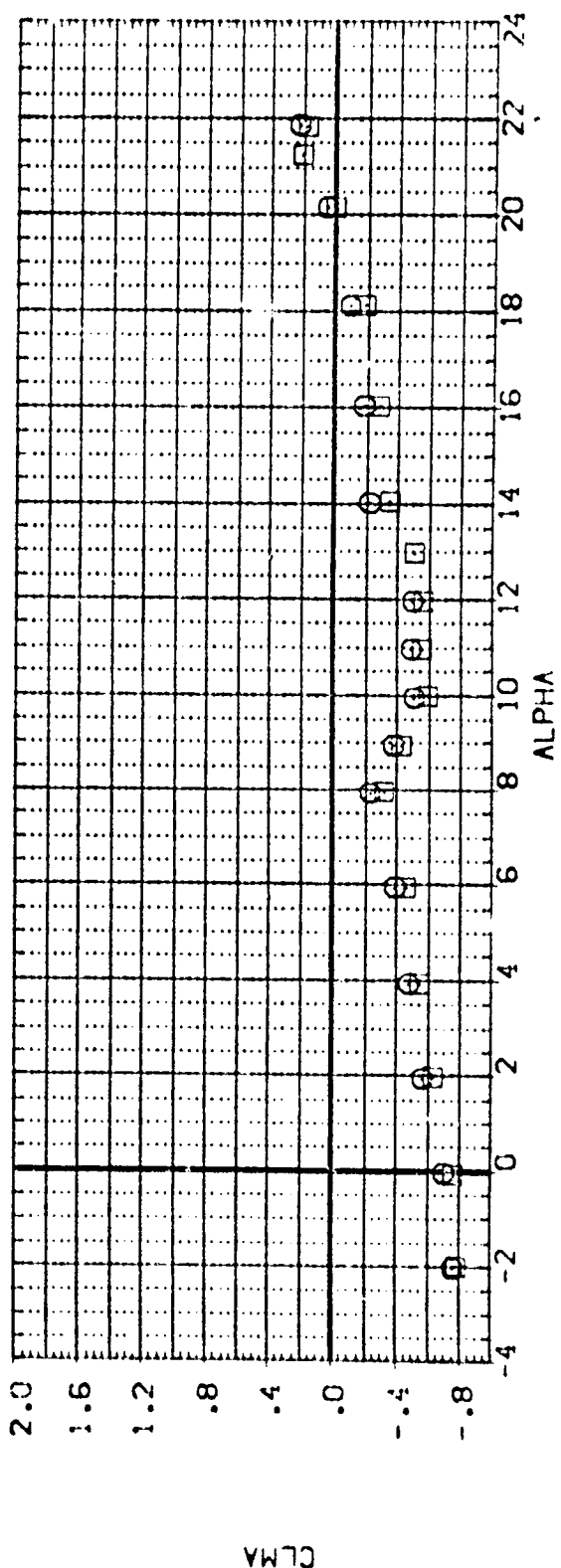
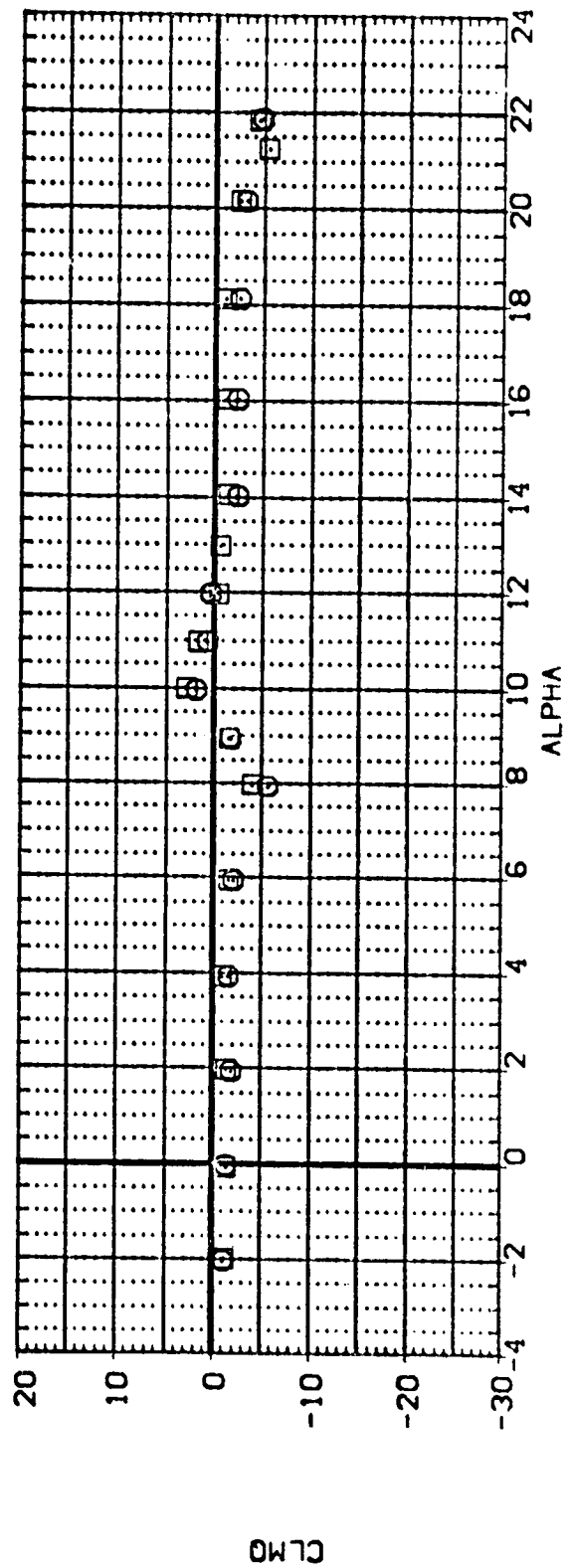


FIGURE 13. EFFECT OF BODY FLAP ON DYNAMIC STABILITY PARAMETERS IN PITCH

(E)MACH = 1.20

APPENDIX
TABULATED SOURCE DATA

Plotted data tabulations are
available from DMS on request.

LA20 TABULATED SOURCE DATA

PAGE 1

LA-20, ROCKWELL ORB 0898 W/MCD. NOSE (RAW F)

(RPRY02)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRCON = .000 CG-LCC = 1.000
 BOFLAP = .000 RUOFUR = 10.000
 RUDDER = .000

RUN NO. 23/ 0

MACH	ALPHA	CYR	CYNBC	CBLR	CBLBC
.300	-2.965	-2.7597	.09800	.07099	-.04469
.300	-2.074	-.29930	.09011	.09636	-.04817
.300	-.050	-.31424	.09203	.09643	-.06410
.300	1.961	-.30762	.09632	.10964	-.07498
.300	3.930	-.29223	.10425	.09783	-.08582
.300	5.986	-.29127	.10707	.16071	-.09369
.300	7.972	-.34134	.10340	.16481	-.11649
.300	9.983	-.34772	.09357	.29969	-.10663
.300	11.987	-.59046	.04619	.34609	-.08724
.300	14.015	-.33710	.09463	.80434	-.11977
.300	16.031	-.75653	.09342	.31680	-.20184
.300	18.013	-.78318	.10718	.60186	-.11182
.300	20.045	-.88683	.09106	.44790	-.18407

RUN NO. 22/ 0

MACH	ALPHA	CYR	CYNBC	CBLR	CBLBC
.800	-2.976	-.25756	.14321	.12967	-.05645
.800	-2.068	-.23199	.14374	.09853	-.06719
.800	-.044	-.22025	.14979	.07893	-.08534
.800	1.978	-.36239	.14935	.23203	-.09953
.800	3.989	-.56909	.14830	.30656	-.13551
.800	5.992	-.63466	.13433	.03219	-.13181
.800	7.989	-.61402	.12608	.22234	-.13870
.800	9.977	-.69349	.11197	.80168	-.10184
.800	11.998	-.71524	.09759	.84909	-.08872
.800	14.015	-.86534	.09244	.86203	-.12269
.800	16.025	-.77088	.04446	1.06333	-.11024
.800	18.042	-.84898	.03431	1.12855	-.10402
.800	20.056	-.97924	-.03317	1.21063	-.10447

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 OF POOR QUALITY

LA20 TABULATED SOURCE DATA

PAGE 2

LA-20, ROCKWELL CRB 0698 W/MCD. NOISE (B/W F)

(RPNY02)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRCON = .000 CG-LCC = 1.000
 BOFLAP = .000 RUOPLR = 10.000
 RUDDER = .000

RUN NO. 21/ 0

MACH	ALPHA	CYR	CYNBC	CBUR	CBLEB
.900	-2.959	-2.8934	.16786	.24188	-.06789
.900	-2.074	-.30700	.16787	.14781	-.07988
.900	-.033	-.29439	.17357	.31723	-.09797
.900	1.955	-.45806	.17077	.39095	-.10461
.900	3.978	-.50279	.16593	.33725	-.10950
.900	5.958	-.59481	.14962	.29878	-.14300
.900	7.972	-.70081	.13395	.30297	-.15475
.900	9.994	-.79754	.10819	.30040	-.12373
.900	11.976	-.83884	.08895	1.39966	-.04255
.900	13.998	-.78000	.06204	.83541	-.14284
.900	16.031	-1.16188	-.01632	1.08194	-.14885
.905	18.059	-1.13462	-.04090	1.21863	-.12399
.900	20.051	-1.23071	-.06309	1.20617	-.13904

RUN NO. 22/ 0

MACH	ALPHA	CYR	CYNBC	CBUR	CBLEB
.980	-2.954	-.35218	.18131	.03829	-.12175
.980	-2.074	-.48536	.17850	.15244	-.12136
.980	-.016	-.48408	.20729	.27088	-.12867
.980	1.961	-.10915	.23535	.48980	-.11726
.980	3.994	-.21631	.23150	.75700	-.10222
.980	5.986	-.37515	.20109	.67748	-.09889
.980	7.989	-.54289	.16227	1.04945	-.11779
.980	9.983	-.67065	.19075	.81058	-.12803
.980	11.998	-1.23035	.17372	.33902	-.17519
.980	14.004	-.75167	.15337	.83174	-.17886
.980	16.025	-.56061	.17359	.88741	-.18563
.980	18.050	.37456	.07020	2.61809	-.16927
.980	20.068	-.42098	.04820	2.04042	-.12344

LA20 TABULATED SOURCE DATA

PAGE 2

LA-20, ROCKWELL CRB DR98 W/MCD. NCSE (BLW F)

(RPMY02)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
AILRON = .000 CG-LCC = 1.000
BDPLAP = .000 RUOPLR = 10.000
RUDDER = .000

RUN NO. 19/ D

MAOH	ALPHA	CYR	CYNBC	CBLR	CBLBC
1.200	-2.841	-.49797	.13687	.02164	-.00887
1.200	-2.074	-.32942	.13359	.08221	-.09834
1.200	-.072	-.50404	.13167	.12609	-.11344
1.200	1.967	-.64080	.12357	.27954	-.11576
1.200	3.983	-.83673	.10206	1.77084	-.04123
1.200	3.975	-.84206	.09654	.66166	-.09453
1.200	7.950	-1.02312	.09161	.98362	-.08676
1.200	9.966	-.44887	.11333	1.97036	-.05372
1.200	11.976	-1.21446	.04305	.82813	-.11551
1.200	14.010	-1.31328	.00680	.99812	-.09933
1.200	15.997	-1.51730	-.01169	1.39897	-.08371
1.200	18.008	-1.33740	-.03474	1.19343	-.12651
1.200	20.016	-1.21321	-.02180	1.43970	-.12644

LA-20, ROCKWELL CRB DR98 W/MCD. NCSE (BLW MF)

(RPMY03)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
AILRON = .000 CG-LCC = 1.000
BDPLAP = .000

RUN NO. 11/ C

MAOH	ALPHA	CYR	CYNBC	CBLR	CBLBC
.300	-2.796	-.07452	-.12518	-.03303	.09697
.300	-2.062	-.12881	-.12490	-.02721	.09210
.300	-.044	-.08174	-.11729	.00409	.07764
.300	1.961	-.11074	-.11064	-.00770	.06437
.300	3.978	-.12208	-.10080	.03792	.04681
.300	5.975	-.15769	-.09625	-.02012	.03582
.300	7.989	-.13519	-.09613	.09074	.01303
.300	9.983	-.15585	-.10581	.14887	.02011
.300	12.004	-.22675	-.14061	.28632	.04599
.300	14.026	-.19698	-.13805	.11693	-.04142
.300	16.025	-.11418	-.09337	.11884	-.07401
.300	18.059	-.31727	-.09070	.04263	-.06315
.300	20.114	-.40046	-.09771	.33101	-.01702

ORIGINAL PAGE IS
OF POOR QUALITY

LA20 CALCULATED SOURCE DATA

PAGE 4

LA-20, ROCKWELL CRB 0898 W/MOD. H2SE (BW MF)

(RPMY03)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRLON = .000 CG-LOC = 1.000
 BDFLAP = .000

RUN NO. 10/ 0

MACH	ALPHA	CYR	CYNBC	CBUR	CBLCB
.800	-2.976	-1.5470	-.12141	-.27593	.11653
.800	-2.031	-1.0410	-.11504	.04587	.10444
.800	-.078	-.08579	-.10754	.03571	.08354
.800	1.967	-.22798	-.09835	.19798	.08889
.800	3.972	-.29195	-.09290	.16578	.05318
.800	5.957	-.35046	-.09538	.04331	.03353
.800	7.978	-.30967	-.09572	.00550	-.00009
.800	9.983	-.40786	-.10843	.57125	.02504
.800	11.987	-.48205	-.11473	.62674	.03825
.800	13.998	-.54026	-.11914	.75073	.00202
.800	16.025	-.43797	-.13346	.42444	-.01762
.800	18.048	-.67849	-.15993	.68298	-.02546
.800	20.056	-.31397	-.18214	.22862	-.07294

RUN NO. 9/ 0

MACH	ALPHA	CYR	CYNBC	CBUR	CBLCB
.900	-2.875	-.10687	-.10167	.35004	.13059
.900	-2.074	-.11906	-.10089	-.12596	.11872
.900	-.067	-.22279	-.08627	.18824	.05042
.900	1.967	-.16774	-.08041	.02914	.05639
.900	3.961	-.20933	-.08317	.14290	.04172
.900	5.958	-.23702	-.07562	.15059	.01418
.900	7.961	-.21749	-.09538	-.12655	.00893
.900	9.983	-.31371	-.11307	.04145	.00292
.900	12.004	-.45419	-.13175	.88843	.06146
.900	14.010	-.49384	-.12779	1.07183	-.01152
.900	16.036	-.86562	-.16665	.95204	-.02912
.900	18.053	-.78115	-.18419	.85527	-.04662
.900	20.051	-1.11915	-.19635	.29520	-.16245

LA20 TABULATED SOURCE DATA

PAGE 5

LA-20, ROCKWELL CRB 0898 WIND. NOISE (24 MF)

(PK103)

PARAMETRIC DATA

BETA = .000 ZLEVTR = .000
 ATLRON = .000 CG-LCC = 1.000
 BDFLAP = .000

RUN NO. 13/ 0

MACH	ALPHA	CYR	CYBC	CBLR	CBLBC
.980	-2.875	-.08788	-.11127	.09682	.08648
.980	-2.074	-.18808	-.10920	.01894	.08154
.980	-.061	-.19921	-.10037	.03085	.04830
.980	1.961	-.26780	-.09005	.18830	.04477
.980	3.961	-.27194	-.07244	.60976	.06620
.980	5.975	-.39904	-.06016	.77120	.07415
.980	7.978	-.64232	-.04632	.63740	.03812
.980	9.983	-.68823	-.04789	.88016	.04589
.980	11.998	-.78968	-.04109	.99737	.00621
.980	14.043	-.77113	-.07771	.72023	-.03282
.980	16.031	-.72154	-.06132	.78093	-.13087
.980	18.042	-.44925	-.05732	1.47049	-.00811
.980	20.056	-1.04354	-.18687	1.35455	-.07367

RUN NO. 12/ 0

MACH	ALPHA	CYR	CYBC	CBLR	CBLBC
1.200	-2.813	-.33597	-.13364	.07276	.07225
1.200	-2.077	-.42942	-.13046	.09109	.06737
1.200	-.030	-.30902	-.11812	-.03439	.05311
1.200	1.961	-.15715	-.11989	.22728	.05654
1.200	4.000	-.29487	-.11149	.28489	.06809
1.200	5.975	-.31398	-.10706	.46395	.05156
1.200	7.983	-.49564	-.10132	.63141	.03661
1.200	9.983	-.56536	-.07830	1.01867	.0262
1.200	12.009	-1.04127	-.11838	.90264	.02423
1.200	14.010	-1.15403	-.13720	.98716	.03896
1.200	16.042	-1.09666	-.13024	.75399	-.01400
1.200	18.036	-1.14689	-.12485	.77043	-.03493
1.200	20.045	-1.13929	-.11718	.98253	-.03083

ORIGINAL PAGE IS
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LA20 TABULATED SOURCE DATA

PAGE 6

LA-20, ROCKWELL CRB 0898 W/WD. NOSE (BMMF)

(RFRY04)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRCON = .000 CG-LCC = 1.000
 BIDFLAP = .000 RUOFLR = 10.000
 RUDDER = .000

RUN NO. 3/ 0

MACH	ALPHA	CYMR	CYNBC	CELR	CELRBC
.300	-2.937	-29311	.09532	.08100	-.02030
.300	-2.037	-31306	.09443	.09951	-.02691
.300	-.027	-30413	.09990	.09041	-.03692
.300	1.967	-39099	.09495	.10035	-.05293
.300	3.961	-27209	.09710	.12749	-.06174
.300	5.975	-31326	.10510	.17227	-.07753
.300	7.978	-26945	.10139	.15540	-.09413
.300	9.977	-28472	.09916	.23213	-.09045
.300	11.996	-26823	.08239	.24284	-.05903
.300	14.021	-22574	.06764	.41931	-.12997
.300	16.036	-42530	.06666	.09812	-.20406
.300	18.042	-42223	.07649	.24282	-.15982
.300	20.074	-42034	.07871	.44232	-.13315

RUN NO. 4/ 0

MACH	ALPHA	CYMR	CYNBC	CELR	CELRBC
.800	-2.835	-32948	.14559	.17034	-.02656
.800	-2.037	-32091	.14614	.03175	-.03531
.800	-.027	-31295	.14399	.08406	-.04961
.800	1.955	-32800	.14743	.12214	-.06380
.800	3.978	-29945	.14996	.07522	-.07991
.800	5.986	-38972	.14129	.06895	-.09628
.800	7.989	-39453	.13206	.03573	-.13311
.800	9.999	-39255	.11752	.30040	-.11963
.800	12.009	-48534	.09310	.41745	-.10397
.800	14.015	-63848	.07624	1.07326	-.11635
.800	16.035	-68933	.04487	.62350	-.13277
.800	18.030	-75297	-.00628	.71469	-.12383
.800	20.053	-75155	-.08394	.40976	-.14104

LA20 TABULATED SOURCE DATA

PAGE 7

LA-20, ROCKWELL CRB 0698 W/MOD. NOISE (BAMMF)

(RPY04)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTURON = .000 CG-LOC = 1.000
 BDFLAP = .000 RUOPLR = 10.000
 RUDDER = .000

RUN NO. 1/0

MACH	ALPHA	CYMR	CYNBC	CLLR	CLBLC
.900	-2.585	-21.594	.19990	.28037	-.00298
.900	-2.057	-2.6310	.19928	.29645	-.01899
.900	-.033	-.28247	.19978	.37470	-.04631
.900	1.961	-.26629	.19727	.09848	-.08845
.900	3.966	-.36864	.18189	.13096	-.10155
.900	5.975	-.43904	.17333	.08999	-.12920
.900	7.978	-.33637	.15882	.17786	-.11887
.900	9.988	-.82373	.12107	1.03260	-.10644
.900	12.015	-.48762	.09732	.76493	-.09532
.900	14.017	-.61275	.08921	.87486	-.12619
.900	16.051	-.78305	-.00930	.88268	-.12452
.900	18.042	-1.08279	-.04781	.88364	-.13029
.900	20.085	-1.37117	-.13875	1.03830	-.19617

RUN NO. 7/0

MACH	ALPHA	CYMR	CYNBC	CLLR	CLBLC
.980	-2.903	-.41342	.19463	.13121	-.09992
.980	-2.085	-.23120	.19000	.04218	-.10729
.980	-.061	-.42784	.16992	.10059	-.11495
.980	1.978	-.41148	.15933	.29882	-.10528
.980	3.978	-.40734	.16211	.63566	-.08297
.980	5.986	-.18901	.20843	.52343	-.08802
.980	7.961	-.36171	.17199	.49331	-.10242
.980	10.005	-.59838	.13982	.51781	-.11203
.980	12.009	-.55707	.12901	.61971	-.13166
.980	14.032	-.69042	.09313	.55653	-.15305
.980	16.042	-.58105	.09972	.68842	-.15988
.980	18.042	-.33428	.11696	1.21731	-.13848
.980	20.056	-1.12886	-.04034	.85793	-.17312

ORIGINAL PAGE IS
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LA20 TABULATED SOURCE DATA

PAGE 8

LA-20, ROCKWELL CRB 0698 WIND. NOISE (BMMF)

(BPKY04)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRCON = .000 CG-LOC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUDDER = .000

RUN NO. 6/0

MACH	ALPHA	CYMR	CYNBC	CLLR	CLLBC
1.200	-2.965	-.42475	.14208	.18997	-.09737
1.200	-2.074	-.43022	.13850	.05410	-.10030
1.200	-.061	-.40072	.13032	.10878	-.10306
1.200	1.972	-.49338	.11827	.21969	-.08954
1.200	3.966	-.54463	.10020	.46154	-.07134
1.200	5.981	-.52026	.08829	.43994	-.08005
1.200	7.983	-.68030	.08733	.50029	-.09018
1.200	9.999	-.55733	.09664	.82950	-.07834
1.200	11.992	-.85913	.08829	.92306	-.08407
1.200	14.021	-1.03235	.07899	.97053	-.06368
1.200	16.036	-1.07816	-.00591	.70642	-.10231
1.200	18.019	-1.03973	-.02363	.77393	-.10877
1.200	20.045	-1.05217	-.03650	1.13392	-.09394

LA-20, ROCKWELL CRB D898 WIND. NOSE (BHWPF)

(RHY03)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRLON = .000 CG-LOC = 1.000
 BDFLAP = 13.000 RUOFLR = 63.000
 RUOGR = .000

RUN NO. 16/ 0

MACH	ALPHA	CYR	CYNBC	CLR	CLBC
.300	-2.892	-.16364	.07468	.12382	.00702
.300	-2.062	-.16129	.07516	.11094	.00324
.300	-.072	-.16773	.08032	.09899	-.00445
.300	1.961	-.16406	.08268	.15967	-.01978
.300	3.935	-.16227	.08390	.15599	-.02889
.300	5.964	-.23892	.08493	.16047	-.03904
.300	7.930	-.25492	.08372	.14660	-.05146
.300	9.972	-.27451	.07258	.16910	-.05500
.300	11.992	-.36136	.04406	.29768	-.02322
.300	13.993	-.43683	.05355	.31329	-.11680
.300	15.966	-.57557	.07901	.42216	-.12119
.300	16.002	-.71429	.10163	.35575	-.10302
.300	19.999	-.78175	.10902	.44682	-.14877

RUN NO. 15/ 0

MACH	ALPHA	CYR	CYNBC	CLR	CLBC
.800	-2.796	-.47796	.09393	.03304	.01564
.800	-2.051	-.50325	.09567	.08767	.00514
.800	-.067	-.50253	.10139	.12233	-.01622
.800	1.978	-.53033	.10977	.11194	-.03481
.800	3.978	-.64330	.11160	.21942	-.04908
.800	5.969	-.75452	.11007	.34287	-.05119
.800	7.978	-.65932	.09857	.27698	-.08497
.800	9.977	-.72767	.08357	.51022	-.07715
.800	11.987	-.82886	.09117	.74364	-.05099
.800	13.987	-.80994	.05373	.66709	-.10687
.800	16.025	-.84887	.02511	.61567	-.09153
.800	16.036	-1.30304	-.02187	-.91092	-.08928
.800	20.062	-.91017	-.08541	-.41193	-.08093

ORIGINAL PAGE IS
 OF POOR QUALITY

LA-20, ROCKWELL CRB 0898 W/MCD. NOSE (BAMF)

(RPMY05)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTROK = .000 CG-LOC = 1.000
 BDFLAP = 13.000 RUOFLR = 63.000
 RUDDER = .000

RUN NO. 14/ 0

MACH	ALPHA	CYR	CYNBC	CLLR	CLLBC
.900	-2.813	-.64599	.09030	.13162	.01938
.900	-2.085	-.67925	.09099	.20061	.00685
.900	-.030	-.73232	.09752	.27841	-.01364
.900	1.961	-.80129	.10394	.32949	-.02664
.900	3.963	-.84324	.10737	.29021	-.03776
.900	5.969	-.92903	.10555	.11167	-.03904
.900	7.972	-.79906	.10110	.12670	-.09199
.900	9.972	-.80828	.09076	.54620	-.08680
.900	11.981	-.78065	.05227	.60748	-.03231
.900	13.987	-.84173	.03036	.88531	-.09624
.900	16.019	-.90274	-.01929	.89803	-.06801
.900	18.008	-1.11790	-.06336	1.37930	-.07635
.900	20.039	-1.08640	-.10794	1.16356	-.11647

RUN NO. 18/ 0

MACH	ALPHA	CYR	CYNBC	CLLR	CLLBC
.980	-2.801	-.44323	.15237	.45096	.04062
.980	-2.079	-.46356	.15647	.36244	.00711
.980	-.030	-.32866	.18029	.53555	-.02002
.980	1.944	-.44851	.17854	.63912	-.03342
.980	3.955	-.53643	.17963	.82433	-.03592
.980	5.975	-.94193	.18129	1.63073	-.05574
.980	7.978	-.71034	.20303	.85124	-.07028
.980	9.994	-.59903	.17331	.56394	-.11647
.980	11.976	-1.20579	.13645	.78038	-.12667
.980	13.998	-.99441	.15032	.74675	-.14956
.980	16.019	-.85192	.16037	.96858	-.13682
.980	18.025	-.09015	.17790	1.68037	-.10618
.980	20.045	-2.06010	-.11925	1.26572	-.10672

LA20 TABULATED SOURCE DATA

PAGE 11

LA-20, ROCKWELL CRB 0898 W/MCD. NOISE (BMMWF)

(BPKY05)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AILRON = .000 CG-LCC = 1.000
 SDPLAP = 13.000 RUOPLR = 85.000
 RUOOR = .000

RUN NO. 17/ 0

MACH	ALPHA	CYR	CYBIC	CBUR	CBIBC
1.200	-2.897	-.21862	.10801	-.02148	-.06192
1.200	-2.062	-.56553	.09428	.03370	-.06779
1.200	-.027	-.45548	.10220	.12157	-.06911
1.200	1.989	-.46834	.09238	.22110	-.07049
1.200	3.989	-.67349	.09916	.56234	-.06187
1.200	5.975	-.83997	.09428	.68718	-.09014
1.200	7.978	-.65773	.10097	.77262	-.08367
1.200	9.966	-.81610	.10637	.92056	-.09497
1.200	11.987	-1.18046	.04980	.56396	-.14799
1.200	13.987	-1.17617	.04216	1.39630	-.07236
1.200	16.008	-1.03670	.03262	1.29038	-.09993
1.200	18.008	-.87084	.05947	1.38042	-.10590
1.200	20.033	-1.39107	-.03295	1.07250	-.13394

ORIGINAL PAGE 1
 OF FOUR QUALITY

LA20 TABULATED SOURCE DATA

PAGE 12

LA-20, ROCKWELL D698 OIB WIND NOSE (BMM)

(BPK100)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTURON = .000 CG-LOC = 1.000
 RUOFLR = 10.000 RUODER = .000

RUN NO. 5/ 0

MAOH	ALPHA	CLMO	CLWA	CNO	CWA
.300	-2.016	-2.3395	.1256	4.40947	2.45436
.300	-.012	-2.20421	.03533	5.94316	2.53207
.300	1.957	-1.75396	.03217	5.73165	2.29401
.300	3.946	-2.25030	.02560	6.17289	2.69871
.300	5.939	-2.33365	.03547	5.90415	2.76399
.300	7.934	-2.06643	.06996	6.59467	2.83549
.300	9.963	-1.56018	-.05636	5.06706	3.17239
.300	10.970	-1.40396	-.20472	4.93927	3.59910
.300	11.972	-.38703	-.29383	2.96692	3.98652
.300	12.982	.06061	-.21145	3.20394	3.93974
.300	13.515	-1.01756	-.11634	6.01948	3.74669
.300	14.013	-2.12799	-.06868	6.39048	3.61509
.300	16.051	-2.07102	-.01837	9.99203	2.67491
.300	18.097	-1.67190	-.11054	4.09462	3.66456
.300	20.156	-.99913	.04663	3.14934	3.47956
.300	21.842	-1.93762	.26702	1.16933	3.60514

RUN NO. 5/ 0

MAOH	ALPHA	CLMO	CLWA	CNO	CWA
.800	-2.000	-3.24293	.07326	4.15580	2.88808
.800	-.024	-2.29028	.00392	5.47922	3.01906
.800	1.945	-2.78943	-.03950	4.59912	3.18439
.800	3.960	-2.73421	-.10323	2.89076	3.32155
.800	5.951	-3.72818	-.07633	7.09270	2.78010
.800	7.946	-3.25723	-.13257	.19958	3.08018
.800	9.963	-3.62201	-.07398	-4.27834	3.27894
.800	11.980	-2.50535	-.23705	-11.04751	3.89906
.800	13.988	-1.55970	-.30913	-16.37823	3.88902
.800	16.051	-2.45240	-.01265	-20.92883	3.54639
.800	18.106	-4.28911	.22894	-21.55204	3.25209
.800	20.164	-7.98457	.83931	-5.08570	2.12061
.800	21.691	-21.04138	1.99186	13.76897	.00739

LA20 TABULATED SOURCE DATA

PAGE 13

LA-20, ROCKWELL D698 CR6 WING NOSE (BMAN)

(BPK000)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ALLCON = .000 CG-LOC = 1.000
 RUOFLR = 10.000 RUDDER = .000

RUN NO. 4/ 0

MACH	ALPHA	CLM0	CLM1	CND	CMA
.900	-2.016	-3.79933	-.11355	3.59099	3.31062
.900	-.024	-4.20214	-.12145	4.09982	3.33826
.900	.962	-4.26448	.00898	5.10369	3.14189
.900	1.969	-5.40080	.28830	7.27493	2.69492
.900	2.948	-6.39941	.23861	5.78045	2.65690
.900	3.940	-5.52473	.08730	5.92591	2.85346
.900	5.963	-4.79958	-.20080	5.19596	3.19306
.900	7.936	-3.77710	-.31125	-1.40339	3.42480
.900	9.975	-2.46321	-.49547	-7.70022	3.86414
.900	11.994	-1.94886	-.41993	-15.58030	4.12892
.900	14.000	-2.50564	-.35267	-17.51272	3.83110
.900	16.063	-3.33092	-.00432	-23.16613	3.51364
.900	18.110	-6.33901	.23331	-16.66875	2.99405
.900	20.130	-14.91232	1.10063	4.20710	1.52318
.900	21.821	-41.83679	1.48093	82.02365	.19879

RUN NO. 13/ 0

MACH	ALPHA	CLM0	CLM1	CND	CMA
.900	-2.034	-2.79957	-.48380	-.61799	3.62885
.900	-.042	-3.53021	-.37680	.28107	3.59981
.900	1.969	-4.14596	-.41302	4.33330	3.68759
.900	3.928	-5.31092	-.41243	2.87026	3.49706
.900	5.939	-3.23049	-.33297	2.01553	3.35404
.900	7.934	-1.83028	-.39302	-.04487	3.58244
.900	9.931	.18463	-.48804	-3.70980	3.76633
.900	11.980	.22507	-.47536	-6.10362	3.93430
.900	13.994	-.49096	-.35132	-12.62012	3.99669
.900	16.099	-.72014	-.29838	-13.54475	3.96510
.900	18.097	-11.36490	.47033	5.15365	2.61620
.900	20.181	-16.34415	.90994	3.53187	1.88246
.900	21.847	-17.68545	1.28385	5.89159	.82062

LA20 TABULATED SOURCE DATA

PAGE 14

LA-20, ROCKWELL 0898 ORB WIND NOISE (BMMH)

(BPKF00)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AILRON = .000 CG-LCC = 1.000
 RUOFLR = 10.000 RUDDER = .000

RUN NO. 14/ 0

MACH	ALPHA	CLM0	CLM1	CND	CW
1.200	-2.028	-1.02962	-76182	-3.41212	3.83216
1.200	-.042	-1.36394	-70026	-.64346	3.72994
1.200	1.939	-1.98599	-56592	1.53339	3.59556
1.200	3.934	-1.63632	-.48129	1.29911	3.44521
1.200	5.939	-2.06314	-.39409	2.64373	3.43201
1.200	7.946	-3.68632	-.23242	3.02262	3.32574
1.200	8.947	-1.80634	-.39435	-3.59422	3.60333
1.200	9.951	1.77044	-.50256	-9.61002	3.97418
1.200	10.959	.68669	-.49101	-12.13745	4.04097
1.200	11.972	.29943	-.49471	-11.31006	4.08780
1.200	14.013	-2.34311	-.22701	-8.57677	3.33745
1.200	16.051	-2.45183	-.18563	-10.23343	3.48859
1.200	18.110	-2.55706	-.09949	-12.89484	3.35089
1.200	20.160	-3.13539	.04992	-14.32102	3.22246
1.200	21.868	-4.85000	.23010	-19.13647	2.96911

LA20 TABULATED SOURCE DATA

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U-20, ROCKWELL 0698 ORB WHOD NOSE (RAW F)

(RPM 002)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTURON = .000 CG-LCC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUOGER = .000

RUN NO. 17/ 0

MACH	ALPHA	CLMD	CLMA	CND	CMA
.300	-2.016	-2.27602	.23491	4.94184	2.22022
.300	-.036	-1.59853	.02810	4.30404	2.36983
.300	1.969	-1.71627	.03064	5.36287	2.62317
.300	3.932	-1.68896	.01870	6.00630	2.67808
.300	5.945	-2.03195	.03043	5.69390	2.83967
.300	7.940	-1.61667	.03766	6.27143	2.98882
.300	9.963	-1.67449	-.00330	5.21795	2.99704
.300	11.980	.61098	-.29686	.47886	4.00329
.300	14.013	-1.90771	-.00066	4.33705	3.34773
.300	16.031	-1.28678	-.16536	7.38893	2.94967
.300	18.123	-1.29197	.00007	3.34447	3.56304
.300	20.175	-1.11363	.10910	1.31335	3.60415
.300	21.881	-2.93248	.34518	4.12742	3.47496

RUN NO. 16/ 0

MACH	ALPHA	CLMD	CLMA	CND	CMA
.600	-2.022	-3.31188	.10198	3.67168	2.88279
.600	-.030	-2.98077	-.00774	4.93112	3.11771
.600	1.945	-3.09234	-.03131	4.97647	3.24897
.600	3.940	-3.35382	-.11404	5.58842	3.18396
.600	5.939	-3.42453	-.14306	6.13223	2.74324
.600	7.934	-3.16770	-.19819	4.01712	3.06440
.600	9.931	-3.21379	-.22907	-.18087	3.63874
.600	11.936	-2.76101	-.22161	-3.00723	3.73469
.600	14.007	-2.09935	-.28963	-6.38404	3.32071
.600	16.031	-3.07076	-.01565	-7.79255	3.40924
.600	18.114	-4.90618	.19880	-3.93798	2.89978
.600	20.156	-12.49727	.83430	9.88878	1.87139
.600	21.849	-38.52198	1.70473	81.15708	-.75032

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LA20 TABULATED SOURCE DATA

PAGE 16

LA-20, ROCKWELL 0898 CRB WIND NOISE (SIW F)

(RPMF02)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTORN = .000 CG-LOC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUDDER = .000

RUN NO. 20/ 0

MACH	ALPHA	CLM0	CLM1	CLM2	CND	CNA
.900	-2.016	-3.62437	-0.5337	3.64146	3.23336	3.23336
.900	-0.048	-3.42799	-0.19048	2.49009	3.48280	3.48280
.900	.978	-3.75731	-0.01653	5.55417	3.13242	3.13242
.900	1.945	-7.99584	.23093	11.93086	2.45845	2.45845
.900	2.948	-5.20792	.11702	10.36935	2.57168	2.57168
.900	3.952	-4.22454	-.08797	7.11315	2.88146	2.88146
.900	5.943	-4.81327	-.20996	7.03112	2.84616	2.84616
.900	7.946	-4.22911	-.37969	3.63130	3.10334	3.10334
.900	9.951	-2.81354	-.41753	-5.83202	3.91224	3.91224
.900	11.992	-2.97217	-.33292	-7.40072	3.93761	3.93761
.900	14.017	-2.43873	-.34954	-6.53945	3.57996	3.57996
.900	16.076	-4.21095	-.11239	-12.83303	3.51113	3.51113
.900	18.118	-5.51164	.17545	-11.37330	2.59486	2.59486
.900	20.181	-27.17055	1.10379	-32.73740	1.94945	1.94945
.900	21.543	-23.77241	1.07223	-22.06596	1.19731	1.19731

RUN NO. 19/ 0

MACH	ALPHA	CLM0	CLM1	CLM2	CND	CNA
.900	-2.028	-1.74553	-.78155	-2.34671	4.13422	4.13422
.900	-.033	-1.74432	-.45922	.62725	3.59087	3.59087
.900	1.957	-.20024	-.25238	-.93412	3.25895	3.25895
.900	2.972	-.75744	-.14784	2.39123	3.01580	3.01580
.900	3.940	-8.82915	.07014	9.62240	2.54359	2.54359
.900	4.933	1.64045	-.34942	-.47917	3.08703	3.08703
.900	5.923	2.16701	-.36299	-.35436	3.04117	3.04117
.900	7.946	1.21760	-.39370	-3.48505	3.49793	3.49793
.900	9.963	2.90337	-.55622	-6.64439	3.72441	3.72441
.900	11.992	3.33210	-.56657	-11.83771	3.85199	3.85199
.900	14.025	3.60017	-.54567	-13.08272	4.01729	4.01729
.900	16.071	-2.07368	.04379	-4.78681	3.27491	3.27491
.900	18.118	-10.77123	.51051	19.36566	1.93607	1.93607
.900	20.194	-12.74419	.80292	33.81459	1.03312	1.03312

LA20 TABULATED SOURCE DATA

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LA-20, ROCKWELL 0898 ORS WIND NOISE (SNW F)

(RPM F02)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRSON = .000 CG-LOC = 1.000
 BDFLAP = .000 RUOPLR = 10.000
 RUOOR = .000

RUN NO. 18/ 0

MACH	ALPHA	CLM0	CLM1	CM2	CM3
1.200	-2.004	-8.7177	-7.7370	-8.5195	3.64193
1.200	-.030	-1.29092	-.29453	.97779	3.44932
1.200	1.925	-1.06755	-.57857	.16030	3.43351
1.200	3.916	-4.35721	-.32063	4.66062	3.09714
1.200	5.939	-2.20077	-.33290	1.95791	3.22530
1.200	6.948	-3.45720	-.37993	10.19040	3.19079
1.200	7.433	-13.40854	-.19000	19.29149	2.87013
1.200	7.938	-17.20359	-.10646	19.31470	2.81134
1.200	8.967	-4.20641	-.27770	.83334	3.34266
1.200	9.943	-.80394	-.39372	-3.67283	3.62913
1.200	11.960	.91320	-.52536	-8.06101	3.97394
1.200	14.021	-1.91571	-.28619	-5.50450	3.45002
1.200	16.031	-2.77245	-.14520	-4.72157	3.24098
1.200	18.110	-2.26659	-.07285	-6.59768	3.14009
1.200	20.181	-3.79973	.09778	-5.25655	2.97755
1.200	21.834	-2.85646	.04266	-8.09639	3.02542

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LA20 TABULATED SOURCE DATA

PAGE 18

LA-20, ROCKWELL 0898 ORB WIND NOISE (SIAMF)

SRK(RD4)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRCN = .000 CG-LCC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUDDER = .000

RUN NO. 12/ 0

MACH	ALPHA	CLM0	CLM1	CND	CN1
.300	-2.016	-2.7238	.1968	6.30236	2.47961
.300	-.032	-2.17986	.07167	6.80992	2.66882
.300	1.937	-2.12049	.04801	6.97793	2.78973
.300	3.922	-2.19415	.02897	8.03724	2.82831
.300	5.981	-2.07354	.03907	8.07220	2.91891
.300	7.976	-2.20974	.06368	7.28930	2.97846
.300	9.993	-1.89616	-.05090	6.41932	3.31412
.300	12.011	-.33593	-.31681	3.04068	4.07836
.300	14.037	-1.02909	-.23244	6.30691	3.85377
.300	16.082	-.23985	-.23624	5.88392	3.61106
.300	18.110	-.77935	-.19553	3.19844	4.13976
.300	20.168	-.33048	-.23993	2.13111	4.57636
.300	21.887	-1.93928	.09203	6.67373	3.83089

RUN NO. 6/ 0

MACH	ALPHA	CLM0	CLM1	CND	CN1
.800	-2.028	-2.82466	.09893	8.33104	2.79904
.820	-.048	-3.08616	.03023	5.12477	3.02585
.800	1.937	-3.14363	-.02892	5.32615	3.19309
.800	3.928	-2.88961	-.10725	4.38280	3.33345
.800	5.927	-3.83053	-.04960	8.11294	2.76102
.800	7.946	-3.53386	-.13710	-.97058	3.16106
.800	9.931	-3.47668	-.12074	-5.02673	3.32977
.800	11.968	-2.42366	-.31328	-13.92196	4.05875
.800	14.004	-1.55789	-.33478	-19.06624	4.02368
.800	16.067	-2.79503	-.06620	-26.52833	3.72755
.800	18.097	-3.40524	.12332	-30.67306	3.40224
.800	20.181	-6.27668	.63242	-16.19448	2.59271
.800	21.637	-19.89398	1.34807	-1.36884	.97331

LA20 TABULATED SOURCE DATA

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LA-20, ROCKWELL 0898 CRB WING NOSE (5WMPF)

(BPM04)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTORN = .000 CG-LOC = 1.000
 BCLAP = .000 RUOFLR = 10.000
 RUOGR = .000

RUN NO. 7/ 0

MACH	ALPHA	CLM0	CLM1	CND	CW
.900	-2.016	-4.4683	-0.0665	4.94218	3.25546
.900	-0.024	-3.95758	-0.0688	4.00270	3.31293
.900	1.961	-3.68019	-2.7905	8.39738	2.66129
.900	3.946	-4.84906	.08491	7.83238	2.81190
.900	5.937	-5.61473	-1.9784	4.74367	3.13946
.900	7.934	-4.08366	-.29594	-1.64818	3.31221
.900	9.975	-2.58949	-.32392	-6.58731	4.07164
.900	11.940	-3.89664	-.42096	-13.53784	4.09017
.900	14.013	-2.68232	-.39070	-16.10634	3.94128
.900	16.057	-3.99199	-.10810	-26.62727	3.74199
.900	18.123	-6.34126	.13643	-29.02756	3.33929
.900	20.181	-14.86912	.83929	-8.21320	2.24509
.900	21.778	-35.95168	1.33875	14.62312	1.39458

RUN NO. 11/ 0

MACH	ALPHA	CLM0	CLM1	CND	CW
.900	-2.028	-2.44266	-.48080	-.89545	3.31959
.900	-0.040	-3.21196	-.40297	2.07767	3.35803
.900	1.945	-4.31842	-.42381	3.94911	3.70886
.900	3.940	-4.06545	-.34608	5.19165	3.43992
.900	5.943	-3.74886	-.36870	4.75564	3.41040
.900	7.946	-1.26117	-.45871	1.25830	3.68136
.900	9.963	.48040	-.53732	-5.63314	3.80499
.900	11.948	.54723	-.51919	-9.23801	3.98070
.900	14.013	-.32808	-.41727	-11.44978	3.99000
.900	16.051	-.58074	-.38218	-12.04921	4.09155
.900	18.110	-8.18361	.23078	6.43680	3.05909
.900	20.185	-12.83326	.65633	32.71966	2.66277
.900	21.825	-15.13201	1.08502	22.47894	1.57339

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LA-20, ROCKWELL D898 CRB W/NOO NOSE (BMMF)

(RPR04)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRCON = .000 CG-LCC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUOBER = .000

RUN NO. 10/ 0

MACH	ALPHA	CLM0	CLM1	CMA	CMB	CNA
1.200	-2.004	-1.22047	-1.78110	-2.08571	3.78367	
1.200	-0.089	-1.36065	-1.76164	-1.70430	3.76971	
1.200	1.965	-1.43186	-1.62917	1.39513	3.55697	
1.200	3.944	-1.92276	-1.54751	2.26277	3.45516	
1.200	5.931	-1.55267	-1.46056	3.02142	3.42187	
1.200	7.970	-4.03167	-1.32497	2.45097	3.39299	
1.200	8.947	-1.63756	-1.42393	-2.26789	3.70729	
1.200	9.987	2.82800	-1.59740	-8.75175	4.01612	
1.200	10.964	1.29856	-1.55713	-11.30557	4.02793	
1.200	11.980	-3.6059	-1.56302	-10.07721	4.07440	
1.200	12.994	-1.60392	-1.53873	-7.35027	3.93988	
1.200	14.050	-1.37671	-1.35286	-6.95201	3.61914	
1.200	16.051	-1.25094	-1.28030	-5.53297	3.60430	
1.200	18.135	-1.09778	-1.18453	-7.52646	3.37499	
1.200	20.160	-2.62616	.00413	-11.26622	3.27930	
1.200	21.240	-5.54023	.20325	-8.90635	2.96932	
1.200	21.834	-4.49222	.17905	-5.81699	2.95766	

LA20 TABULATED SOURCE DATA

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LA-20, ROCKWELL 0898 ORB W/MCD NOSE (BMMF)

(RPMF05)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRLON = .000 CG-LOC = 1.000
 BOFLAP = 13.000 RUOFLR = 85.000
 RUODR = .000

RUN NO. 23/ 0

MACH	ALPHA	CLM0	CLM1	CND	CNA
.300	-2.010	-2.72180	.22286	3.84400	2.26670
.300	-0.72	-2.27081	-.03463	4.90878	2.70639
.300	1.963	-2.33923	-.04482	2.81567	2.73937
.300	3.934	-2.42482	-.06498	6.02317	2.78348
.300	5.933	-2.33750	-.03464	5.85959	2.91561
.300	7.934	-2.84372	-.01943	4.16404	3.03793
.300	9.931	-2.44275	-.04639	3.88583	3.12006
.300	11.940	-1.89442	-.27343	1.28269	3.94363
.300	14.013	-2.47489	-.08465	4.86331	3.58282
.300	16.057	-3.13391	-.12538	7.79502	2.93716
.300	18.116	-2.53577	-.11086	3.61669	3.91375
.300	20.167	-2.79966	.04393	3.31022	3.76719
.300	21.842	-2.46693	.16126	1.91819	3.81047

RUN NO. 24/ 0

MACH	ALPHA	CLM0	CLM1	CND	CNA
.800	-2.040	-4.47922	.02001	5.37101	2.98879
.800	-1.048	-4.34000	-.03463	5.93653	3.13088
.800	1.937	-4.92431	-.08759	5.59189	3.34092
.800	3.928	-4.08293	-.17430	4.09271	3.53131
.800	5.927	-3.10309	-.17674	10.69591	2.88658
.800	7.922	-4.77059	-.19987	2.39512	3.50998
.800	9.951	-4.96010	-.14658	-.43864	3.36303
.800	11.968	-4.25408	-.26463	-7.04135	3.96577
.800	14.004	-1.47761	-.43344	-13.36017	4.29779
.800	16.051	-4.30494	-.01225	-9.81134	3.50770
.800	18.104	-5.37755	.09820	-11.19687	3.19535
.800	20.168	-6.67387	.50134	.28455	2.29286
.800	21.855	-11.89982	1.07208	6.19558	1.29668

ORIGINAL PAGE 1
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LA-20, ROCKWELL 0899 CRB WIND NOISE (BMMF)

(BPKF05)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AILTRON = .000 CG-LOC = 1.000
 BDFLAP = 13.000 RUOFLR = 85.000
 RUDDER = .000

RUN NO. 23/ 0

MACH	ALPHA	CLM0	CLM1	CLM2	CND	CWA
.900	-2.026	-5.44374	-.04620	3.55645	3.26720	
.900	-.048	-4.98797	-.15639	4.43960	3.41551	
.900	1.945	-5.46722	-.14768	7.73445	3.42775	
.900	3.916	-6.53699	-.09930	9.51318	2.99375	
.900	5.927	-5.68397	-.36384	6.34780	3.29480	
.900	7.922	-5.53049	-.36732	2.01730	3.58698	
.900	9.939	-4.89263	-.49617	-2.37568	3.99426	
.900	11.968	-3.01090	-.39032	-6.71019	3.94105	
.900	14.000	-3.85667	-.39465	-9.09975	3.68219	
.900	16.045	-5.16336	-.10215	-15.59998	3.99293	
.900	18.110	-6.70550	.06267	-16.10959	3.25574	
.900	20.181	-8.09148	.47576	.83312	2.28426	
.900	21.730	-13.98265	.68396	1.12639	1.87447	

RUN NO. 22/ 0

MACH	ALPHA	CLM0	CLM1	CLM2	CND	CWA
.900	-2.040	-5.58033	-.46532	2.37086	3.67122	
.900	-1.020	-4.38825	-.80121	1.53924	3.84165	
.900	-.060	-6.14309	-.23226	3.93484	3.33966	
.900	.942	-6.64217	-.19959	5.15633	3.31085	
.900	1.933	-14.62167	.19773	9.62752	2.86537	
.900	2.948	-9.49879	.21162	12.81062	2.59028	
.900	3.928	-6.32247	-.13122	3.31209	3.11375	
.900	5.927	-2.81149	-.42936	.95514	3.26281	
.900	7.934	.82731	-.62961	-4.77524	3.75343	
.900	9.951	2.03895	-.69606	-6.64111	3.91383	
.900	11.968	3.01910	-.80389	-8.63372	3.89188	
.900	13.007	3.19303	-.63245	-9.53051	3.94411	
.900	14.019	4.01936	-.68172	-10.86119	4.07328	
.900	15.035	1.99042	-.52453	-9.88856	4.00285	
.900	16.038	-1.02181	-.30432	-10.08225	3.67232	
.900	18.110	-4.74721	.30432	-1.28648	2.57646	
.900	20.181	-7.87409	.41675	-10.36051	2.55102	
.900	21.855	-9.47386	.74398	-9.40561	1.86888	

LA20 TABULATED SOURCE DATA

PAGE 23

LA-20, ROCKWELL D898 CRB W/NOSE (BAMMF)

(RPM05)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AILRON = .000 CG-LCC = 1.000
 BIDFLAP = 13.000 RUOFLR = 85.000
 RUODER = .000

RUN NO. 21/ 0

MACH	ALPHA	CLMD	CLMA	CND	CNA
1.200	-2.028	-1.91748	-90638	-2.22967	4.00873
1.200	-0.00	-2.09612	-77738	-.69963	3.79034
1.200	1.945	-2.38832	-70062	.09802	3.61640
1.200	3.928	-2.36273	-.57751	-.06598	3.42344
1.200	5.927	-3.01898	-.47020	2.02031	3.33885
1.200	6.473	-9.40492	-.22087	10.63274	2.92482
1.200	6.934	-11.31281	-.14512	11.74990	2.84264
1.200	7.928	-7.13353	-.28724	1.64018	3.29931
1.200	8.929	-3.43781	-.43559	-.84742	3.45739
1.200	9.933	-1.73766	-.52354	-4.30687	3.64918
1.200	11.980	-.92545	-.61257	-8.97905	3.95663
1.200	14.013	-4.32158	-.20848	-5.31844	3.26268
1.200	15.031	-4.39209	-.12171	-9.48684	3.26320
1.200	18.110	-5.32468	.07599	-10.30341	2.98735
1.200	20.211	-8.83153	.31569	-9.13180	2.60755
1.200	21.778	-3.43145	.06487	-10.27359	2.82815

LARO TABULATED SOURCE DATA

PAGE 24

LA-20, ROCKWELL 0896 ORB WIND NOISE (BMMWF)

(RPR06)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRLON = .000 CG-LCC = 2.000
 BOFLAP = .000 RUOFLK = 10.000
 RUDDER = .000

RUN NO. 26/ 0

MACH	ALPHA	CLM0	CLMA	CNO	CNA
.300	-2.010	-2.12436	.34533	5.36727	2.24546
.300	-.036	-1.98130	.20059	4.39736	2.64796
.300	1.957	-1.97264	.20336	4.93196	2.69613
.300	3.952	-1.97410	.18995	4.99620	2.73797
.300	5.951	-1.79271	.18362	5.64134	2.89537
.300	7.958	-1.91221	.18443	4.46102	2.98009
.300	9.963	-1.52044	.19945	4.77647	3.07422
.300	11.974	-.69119	-.07461	1.22837	3.98753
.300	14.025	-1.24461	.06292	4.63903	3.99409
.300	16.051	-.30621	-.04500	6.50440	3.00377
.300	18.110	-1.19974	.09950	2.13209	3.84058
.300	20.181	-1.61594	.24329	1.08629	3.73459
.300	21.842	-2.25252	.41993	2.63292	3.72304

RUN NO. 28/ 0

MACH	ALPHA	CLM0	CLMA	CNO	CNA
.800	-2.036	-2.73265	.24553	7.30309	2.86746
.800	-.060	-3.71792	.19183	3.47825	3.01732
.800	1.945	-2.71519	.12761	4.36893	3.20359
.800	3.940	-2.78147	.09997	3.47121	3.29942
.800	5.951	-3.20642	.02054	4.76649	2.87991
.800	7.946	-2.80488	.02237	-.41430	3.23001
.800	9.943	-3.29498	.04666	-3.45500	3.17355
.800	10.978	-2.21115	-.12693	-8.98443	3.79054
.800	11.990	-1.81884	-.10021	-13.75349	3.92976
.800	14.021	-1.00632	-.18039	-19.83149	4.06653
.800	16.063	-2.77745	.10859	-21.27743	3.33946
.800	18.110	-3.10320	.27786	-24.61157	2.83696
.800	20.181	-6.90474	.74185	-24.73505	2.83140
.800	21.855	-16.82834	1.38304	4.99977	-.31789

LA20 TABULATED SOURCE DATA

LA-20, ROCKWELL 0698 CRB W/NOSE (BAMF)

(RPM06)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATTORN = .000 CG-LCC = 2.000
 BOFLAP = .000 RUOFLR = 10.000
 RUOER = .000

RUN NO. 27/ 0

MACH	ALPHA	CLND	CLMA	CND	CMA
.900	-2.034	-4.09009	.09690	3.34133	3.23460
.900	-.060	-3.69566	.07569	3.41179	3.27648
.900	1.945	-5.32786	.49825	6.78340	2.63359
.900	2.936	-5.99339	.44153	5.12201	2.58022
.900	3.928	-4.52924	.18543	6.93268	2.86348
.900	5.923	-3.70905	-.06987	2.80423	3.14992
.900	7.930	-3.44269	-.15786	.51972	3.34999
.900	8.931	-3.47291	-.22343	-4.46206	3.63499
.900	9.939	-2.44397	-.30201	-4.91775	3.94969
.900	10.939	-2.09567	-.32548	-7.00539	4.12078
.900	11.976	-3.02920	-.19886	-8.32429	3.90540
.900	14.013	-2.34430	-.19172	-7.33717	3.68232
.900	15.011	-4.39947	.04048	-11.43969	3.29018
.900	16.039	-4.17648	.07998	-16.76356	3.29676
.900	18.110	-5.14348	.26153	-27.80520	2.82727
.900	20.168	-10.27539	.90624	-23.34993	1.58614
.900	21.855	-29.94590	1.35145	72.15391	.16665

RUN NO. 29/ 0

MACH	ALPHA	CLND	CLMA	CND	CMA
.900	-2.016	-2.18018	-.30370	-.07523	3.56369
.900	-.060	-2.51708	-.23888	1.39342	3.59126
.900	1.945	-2.73294	-.25785	1.09284	3.69234
.900	3.932	-3.98085	-.15149	4.58039	3.55639
.900	5.931	-4.46185	-.08428	3.42615	3.32169
.900	7.946	-1.43673	-.27182	.31690	3.52710
.900	9.963	-.33269	-.34809	-6.35534	3.78625
.900	11.969	.51726	-.30015	-12.61969	3.73434
.900	14.025	.28136	-.21440	-19.09429	3.80491
.900	16.071	-.33727	-.18686	-21.87659	3.81035
.900	18.123	-8.12741	.42733	-10.87781	2.59211
.900	20.189	-10.45516	.66729	-11.28347	1.95292
.900	21.778	-19.47595	1.37914	15.69051	.00461

LA20 TABULATED SOURCE DATA

PAGE 26

LA-20, ROCKWELL 0698 CRB W/MCD NOISE (BAMMF)

(RKF06)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATLRCN = .000 CG-LOC = 2.000
 BDFLAP = .000 RUOFLR = 10.000
 RUDDER = .000

RUN NO. 30/ 0

MACH	ALPHA	CLM2	CLM4	CND	CNA
1.200	-2.028	-1.00288	-60297	-3.18846	3.83279
1.200	-0.40	-1.29934	-54397	-1.14168	3.75816
1.200	1.945	-1.00945	-48417	-.03998	3.63961
1.200	3.940	-1.31036	-33362	1.19933	3.49420
1.200	5.939	-3.17975	-23055	5.47132	3.29903
1.200	7.946	-3.31996	-11819	2.43546	3.29292
1.200	8.959	-1.63731	-25630	-5.08399	3.57523
1.200	9.975	.53531	-34987	-8.20717	3.89951
1.200	10.974	.76727	-38342	-11.10576	3.99346
1.200	11.988	-.23075	-40610	-11.44203	4.04474
1.200	14.013	-2.02862	-13997	-9.71057	3.42665
1.200	16.076	-2.28479	-.09303	-12.97003	3.29376
1.200	18.123	-2.29761	-.00740	-20.81935	3.18189
1.200	20.168	-4.76877	.23308	-32.20947	2.94104
1.200	21.547	-3.17455	.25396	-42.09606	2.84759

LA20 TABULATED SOURCE DATA

PAGE 27

LA-20, ROCKWELL 0852 CRB WHOD NOSE (BW F)

(RFXR01)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATLROH = .000 CG-LCC = 1.000
 BDFLAP = .000

RUN NO. 81/ 0

MACH	ALPHA	CLP	CLBS
.300	-2.032	-.22036	-.00146
.300	-.060	-.23849	.00342
.300	1.949	-.26136	.00435
.300	3.932	-.26682	.00376
.300	5.939	-.27342	.00122
.300	7.946	-.26535	-.00125
.300	9.963	-.31366	-.00165
.300	12.017	-.35285	.00076
.300	14.045	-.39235	.00357
.300	16.105	-.31363	-.02915
.300	18.148	-.38846	-.01757
.300	20.223	-.45448	-.00929
.300	21.568	-.60164	.00909

RUN NO. 82/ 0

MACH	ALPHA	CLP	CLBS
.600	-2.040	-.20543	-.00284
.600	-.048	-.23039	.00161
.600	1.957	-.25140	.00267
.600	3.948	-.20746	.00308
.600	5.955	-.16427	.00248
.600	7.958	-.15545	.00161
.600	9.987	-.27935	.01145
.600	11.988	-.21740	.00301
.600	14.030	-.33598	.00473
.600	16.080	-.31985	-.00109

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 OF POOR QUALITY

LA20 TABULATED SOURCE DATA

PAGE 28

LA-20, ROCKWELL D898 CRB W/ WIND NOISE (BAY F)

(RPM002)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRLON = .000 CG-LCC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUOGR = .000

RUN NO. 77/ 0

MACH	ALPHA	CBLP	CBLS	CYNP	CYNBS
.300	-2.060	-2.4503	.00456	.09999	-.00293
.300	-.060	-.27102	.00244	.05939	.00279
.300	1.969	-2.7135	-.00164	-.11689	.01604
.300	3.940	-.27366	-.00636	-.15623	.02181
.300	5.947	-.25552	-.01421	-.21664	.03154
.300	7.970	-.24874	-.02512	-.19390	.04329
.300	9.999	-.30390	-.02992	-.15109	.05066
.300	10.982	-.42991	-.01874	-.04530	.04904
.300	12.013	-.32631	-.02922	-.07444	.05994
.300	13.027	-.39973	-.01533	.00097	.06759
.300	14.050	-.39315	-.03079	-.08724	.07252
.300	15.068	-.34419	-.04270	.03415	.07341
.300	16.096	-.24576	-.07018	-.09194	.08577
.300	18.143	-.39101	-.05993	-.06172	.09946
.300	20.232	-.43074	-.08046	-.03186	.10297
.300	21.919	-.52069	-.03915	-.14318	.10570

RUN NO. 76/ 0

MACH	ALPHA	CBLP	CBLS	CYNP	CYNBS
.800	-2.056	-.23904	.00195	-.04948	-.00595
.800	-.064	-.27263	.00112	-.06777	.00296
.800	1.945	-.29393	-.00203	-.01532	.01270
.800	3.940	-.29396	-.00698	.02473	.02348
.800	5.943	-.16167	-.01343	-.02605	.03246
.800	7.934	-.19672	-.01807	-.05721	.04336
.800	9.975	-.25188	-.01635	-.04719	.04952
.800	12.013	-.28717	-.02403	-.03386	.05989
.800	14.033	-.34924	-.02232	-.04491	.06032
.800	16.098	-.35540	-.03459	.39973	.06768

LA20 TABULATED SOURCE DATA

PAGE 29

LA-20, ROCKWELL 0696 ORB W/WD NOSE (B/W F)

(IPR002)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 AIRLON = .000 CG-LOC = 1.000
 BOFLAP = .000 RUOPLR = 10.000
 RUOBER = .000

RUN NO. 75/ 0

MACH	ALPHA	CLP	CLBS	CNP	CYBS
.900	-2.052	-.27154	.00263	.03748	-.00866
.900	-.056	-.28645	.00099	.01027	.00175
.900	1.937	-.22644	-.00247	.06678	.01062
.900	3.928	-.20784	-.00704	-.00705	.02242
.900	5.939	-.17742	-.01494	-.04525	.03100
.900	7.942	-.24207	-.01133	-.12645	.04026
.900	9.967	-.22785	-.02199	-.11125	.04778
.900	11.997	-.24140	-.02511	-.10166	.05134
.900	14.037	-.24226	-.03359	-.08963	.05315

RUN NO. 75/ 0

MACH	ALPHA	CLP	CLBS	CNP	CYBS
.900	-2.028	-.32618	.00304	-.05462	-.00565
.900	-.036	-.24799	.00028	-.04375	.00593
.900	1.953	-.20110	-.00346	-.07346	.01493
.900	3.956	-.19790	-.00936	-.05643	.02250
.900	5.963	-.25894	-.01267	-.07360	.03142
.900	7.934	-.24322	-.01425	-.04307	.04014
.900	9.987	-.25146	-.02415	-.13266	.04787
.900	11.992	-.29562	-.03391	-.19223	.05952
.900	14.050	-.30190	-.04368	-.43135	.07493
.900	16.096	-.26804	-.05117	-.17953	.06394
.900	18.160	-.26358	-.05901	-.15760	.03805
.900	20.228	.56348	-.06101	.40922	.00541

LA20 TABULATED SOURCE DATA

PAGE 30

LA-20, ROCKWELL D898 CRB W/MCD N2SE (BW F)

(RPR002)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATLORN = .000 CG-LCC = 1.000
 BDFLAP = .000 RUOFLR = 10.000
 RUOBER = .000

RUN NO. 76/ 0

MACH	ALPHA	CBLP	CBLBS	CYNP	CYNBS
1.200	-2.044	-3.6404	.00282	-.06192	-.00263
1.200	-.036	-.32765	.00094	-.11261	.00730
1.200	1.933	-.33316	-.00236	-.09270	.01442
1.200	3.932	-.37745	-.00096	-.06972	.02211
1.200	5.963	-.26094	-.01042	-.10546	.03040
1.200	7.970	-.28864	-.01303	-.06902	.04115
1.200	10.003	-.32000	-.01753	-.09114	.05121
1.200	12.017	-.29847	-.04374	-.12725	.03234
1.200	14.037	-.31963	-.03246	-.14179	.05390
1.200	16.105	-.44116	-.02329	-.29144	.09332
1.200	18.180	-.36141	-.04394	-.39239	.07607
1.200	20.240	-.28751	-.05325	-.39642	.07051
1.200	21.861	-.22756	-.06947	-.03747	.05735

LA-20, ROCKWELL D898 CRB W/MCD N2SE (BW MF)

(RPR003)

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATLORN = .000 CG-LCC = 1.000
 BDFLAP = .000

RUN NO. 67/ 0

MACH	ALPHA	CBLP	CBLBS
.300	-2.028	-.19423	-.00206
.300	-.054	-.22598	.00068
.300	1.945	-.23513	.00287
.300	3.940	-.25159	.00395
.300	5.939	-.24302	.00253
.300	7.958	-.24404	-.00192
.300	9.947	-.30133	-.00116
.300	12.017	-.35373	.00149
.300	14.056	-.34327	.00140
.300	16.071	-.25522	-.03049
.300	18.173	-.37325	-.02037
.300	20.226	-.40744	-.01012
.300	21.624	-.44745	.00378